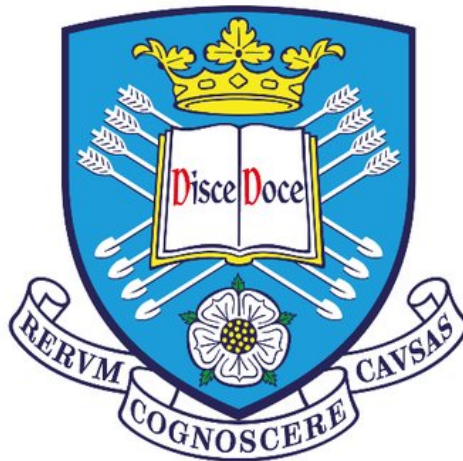


1 **Characterisation of Blast Loading from**
2 **Ideal and Non-Ideal Explosives**

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4

5 The thesis presented is for consideration towards the degree of
6 Doctor of Philosophy in

7 THE DEPARTMENT OF CIVIL AND STRUCTURAL ENGINEERING,
8 UNIVERSITY OF SHEFFIELD

9

Monday 11th December, 2023

10 **Declaration**

11 I, Dain George Farrimond, certify that all the material contained within this document is
12 my own work, except where it is clearly reference to others.

13

14 Date:

15

16 Signed:.....

17

18 Abstract

19 Explosive detonation in its simplest form can be characterised by an instantaneous release
20 of energy at an infinitely small point in space as a solid explosive material. This is a result
21 of chemical decomposition of an explosive which reforms as high pressure and temperature
22 gases which expand radially. This supersonic expansion of detonation products compresses
23 the surrounding medium resulting in a shock wave discontinuity which propagates away
24 from an explosive epicentre at high speeds. This has the potential of significant damage
25 to anything the shock wave interacts with.

26

27 Shock wave quantification work conducted in 1940's through to the 1980's was done so
28 to understand the effects of large scale explosive detonation which was an immediate
29 threat due to the discovery of the nuclear bomb. Highly skilled experimental and theo-
30 retical scientists were assigned the task of capturing the effects of large scale detonations
31 through innovative solutions and development of pressure gauges. The in-depth funda-
32 mental understanding of physics, combustion and fluid dynamics the researchers utilised
33 resulted in the well-favoured semi-empirical blast predictions for simplistic free-field spher-
34 ical/hemispherical blasts.

35

36 A broad amount of literature has been published on free-air characterisation of spher-
37 ical/hemispherical explosives, with the detonation process and subsequent shock wave
38 formation mechanics being well understood. However, there is yet to be a definitive and
39 robust understanding of how deterministic a shock waves spatial and temporal parameters
40 are for simplistic scenarios. This goes as far as some studies suggesting that semi-empirical
41 tools are not as effective as previously assumed. Often the use of numerical simulations
42 provide reasonable insights to blast loading conditions imparted on structures and sce-
43 narios with higher complexities. However, when the validation data used is assumed to
44 exhibit erroneousess, the schemes are no longer characteristically high in fidelity. The

45 lack of quantified variability and confidence in the data which is published, are significant
46 issues for engineers when designing infrastructure that is both robust enough to withstand
47 extreme loading, and not overly conservative that there are cost and material waste impli-
48 cations. This issue is investigated thoroughly within this thesis, highlighting the sensitivity
49 of blast parameters across the scaled distance ranges, and determining their predictability
50 with both numerical simulation and semi-empirical tools.

51

52 The vast majority of free-field characterisation has been conducted using military grade
53 explosive which exhibit ideal detonation behaviours; meaning the detonation reaction is
54 effectively instantaneous. Ideal explosives, by the theoretical definition, can be categorised
55 by a simplistic instantaneous energy release. In far-field regimes, any explosive with ideal-
56 like compositions and behaviours should be scalable with mass. This assumption is not
57 valid for homemade explosives (HME), such as ANFO (Ammonium Nitrate + Fuel Oil),
58 whose compositions are usually homogenous, resulting in a finite reaction zone length.
59 These can be long enough to cause failures in detonations and exhibit a variety of dif-
60 ferent energy releases depending on the mass of the charge resulting in HME's having
61 different TNT equivalence values depending on their scale.

62

63 Early works of ANFO characterisation was done so in the desire to replace TNT, to assess
64 its capability of producing similar yields for a fraction of the manufacturing costs. This
65 meant the hemispherical detonations of ANFO which have led to its overall classification,
66 were done using charges of over 100kg and therefore non-ideal reaction zone effects become
67 negligible in comparison to the overall charge size. Yields presented in this region were
68 consistently measured at around 80% of a similar TNT detonation and has therefore been
69 incorrectly assumed a rule for ANFO across all mass ranges within published literature.

70

71 There is a distinct lack of characterisation of non-ideal explosives throughout the mass
72 scales, posing a significant implication for designing structures to withstand the threat
73 of HMEs. With the knowledge that energy is released at a much slower rate when deto-
74 nating these compositions, the assumption that large scale trials accurately capturing the
75 behaviour of a small charge masses, when scaled down, is not verified. Most HMEs will
76 be hand held devices or, at the very least, backpack size, meaning the threat currently
77 is not predictive with confidence through validated data conducted under well-controlled
78 conditions. Small scale ANFO trials have demonstrated this to be the case within this

79 thesis, with theoretical mechanisms proposed which offering a prediction method of the
80 behaviour of non-ideal detonation across all mass scales.

81

82 Findings in this PhD thesis will offer a conclusion on whether shock waves in free-field
83 scenarios are deterministic for both ideal and no-ideal explosives, with a particular em-
84 phasis on the far-field range. The results presented are developments in the accurate
85 quantification of shock wave loading conditions a structure is subjected to through ex-
86 plosive detonation and should be used by engineers to establish robust, probabilistic but
87 accurate designs.

88

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535 for this body of research and article 128

536 Nomenclature

537	A	Area
538	D_{CJ}	Chapman-Jouguet detonation velocity
539	D_{∞}	Infinite charge diameter
540	E	Detonation energy per unit mass
541	F	Original image
542	i	Specific impulse
543	i_r	Reflected positive phase specific impulse
544	$i_{r,min}$	Reflected negative phase specific impulse
545	i_{so}	Incident positive phase specific impulse
546	$i_{so,min}$	Incident negative phase specific impulse
547	K	Hopkinson-Cranz scaling factor
548	N	Resulting subtracted image
549	P	Pressure
550	P_{CJ}	Chapman-Jourguet pressure
551	P_{max}	Max Overpressure
552	P_o	Ambient air pressure
553	P_r	Reflected pressure
554	$P_{r,neg}$	Negative reflected pressure
555	P_{so}	Incident pressure
556	$P_{so,neg}$	Negative Incident pressure
557	P_{SS}	Secondary shock pressure
558	R	Distance from charge centre (stand-off)
559	R^2	Coefficient of determination
560	ρ	Charge density
561	t	Time
562	t_a	Time of arrival

563	t_d	Positive phase time duration
564	t_{d-}	Negative phase time duration
565	td_{ss}	Secondary Shock positive phase time duration
566	τ_{ss}	Secondary shock delay parameter
567	$\delta\tau_{ss}$	Time difference between arrival of primary and secondary shock
568	V_{od}	Velocity of detonation
569	W	Charge mass
570	Z	Scaled distance
571		

572 Chapter 1

573 Introduction

574 The overall aim of this PhD thesis is to comprehensively characterise the effects various
575 explosive compositions and sizes have when detonated across a full range of distances from
576 a target. This will focus primarily on the far-field range of distances, the time at which
577 the shock wave has propagated to a significant distance away from the explosive centre as
578 to not be affected by late-time chemical reactions within the fireball. The context of this
579 work looks at providing key analytical procedures which are validated for the most simple
580 and ideal explosive compositions and assessing their application to non-ideal compositions.

581

582 This chapter will provide an outline of the structure of the thesis as well as detailing the
583 motivation behind the current research.

584

585 1.1 Background and Motivation

586 The discovery and use of the nuclear bomb in 1945, initiated a critical need across the
587 globe to better understand the effects of explosions with extremely large yields. This
588 was essential to ensure procedures and strategies could be implemented for civilian safety
589 against potential large scale threats. Between 1945-1963, a large number of nuclear tests
590 were undertaken by the United States in the attempt to quantify the disastrous conse-
591 quences these events would have on civilian infrastructure. The severe environmental
592 damage caused by these tests induced the first instance of global cooperation to eliminate
593 the testing of nuclear weapons. There was however still a need for quantifying the effects

594 of said events to understand the fundamentals of explosions and how this knowledge could
595 be used to improve design practices of structures to withstand them. Vast investment
596 at this time into testing of high explosives, with yields within similar margins to nuclear
597 detonation, produced some of the more widely regarded quantification data sets^[87]. This
598 was then utilised to develop a semi-empirical prediction tool for free-air blast scenarios^[90].
599 Blast resilient design specifications to this day still make use of these tools for prescribing
600 loading conditions a given target will be subjected to for a given mass of Trinitrotoluene
601 (TNT) - the baseline to which all comparisons are made in blast engineering.

602

603 This tool has however had its accuracy questioned over more recent years with researchers
604 debating whether it provides reasonable enough representations of an explosive event
605 across a full-field of distances between charge and target^[13]. Logic would suggest that
606 with more modern and precise experimental procedures, the results should tend towards
607 those predicted, with a reduction in the uncertainty. However, the problem has been that
608 many modern researchers have reported a significant lack of repeatability in experimental
609 measurements when comparing directly to the tool's predicted parameters, and the much
610 older data sets used to produce them^[56]. There is however a clear divide in the blast
611 protection community regarding the fundamental understanding of explosive characterisa-
612 tion; those who exhibit large variability in experimental blast measurements and interpret
613 this to be inherent of explosive detonation, and those which undertaking similar testing,
614 recorded consistent and deterministic data.

615

616 Modern day architecture has moved towards more material and energy efficient designs
617 whilst appealing aesthetically through atrium-like open spaces, glazed cladding for addi-
618 tional lighting and unique structural forms for visual effect. These three additions, among
619 others, to civilian infrastructure have one key disadvantage: the catastrophic effects, both
620 structurally and to civilians, they can have as a result of being subjected to explosive
621 loading conditions. This in conjunction with the rapid expansion of the geographic extent
622 of urban areas has resulted in a rise in spatially complex civilian environments. Blast
623 resilient design of infrastructure has become essential as a results of an increasing number
624 of explosions occurring in civilian areas. In 2021 alone there were a total of 5,226 recorded
625 terrorist attacks on civilian structures, with the 18 worst events resulting in 1241 deaths
626 alone^[79]. The frequency and variety in modern day explosive events and their interac-
627 tion with structures provides enough of a challenge for engineers to combat without the

628 predicted methods used for design being questioned. Whilst ever there is uncertainty in
629 the predictive tools used to design civilian infrastructure, there is also an uncertainty in
630 safety assurances of structures when subjected to explosive loading. There is, therefore, a
631 need to establish whether the prediction tools developed by Kingery and Bulmash [87] are
632 accurate for a full range of distances between charge and target across various explosive
633 types.

634

635 This thesis presents research which will contribute to the existing knowledge of quantifying
636 reflected blast loading on targets by characterising the effects across a range of charge-to-
637 target distances for a variety of explosives. The research aims to provide evidence which
638 supports the notion of explosive variability through re-examination of both historic and
639 newly recorded data, whilst developing new and robust experimental data processing tech-
640 niques. The investigation into different types of explosives will provide improvements to
641 the fundamental understanding of the mechanisms and characteristics of their detonation,
642 and the resulting yields.

643

644 1.2 Scope and Objectives of the Thesis

645 When designing infrastructure to withstand explosive loading conditions, the engineer is
646 faced with the task of assigning a given charge composition, shape, mass and location
647 from a target. These variables often will not be known, and therefore a probabilistic anal-
648 ysis is required. The ability to determine an accurate prediction for a prescribed event
649 is essential to introduce confidence in probabilistic studies. Whilst all of these variables
650 can significantly alter the pressures associated with the resulting blast wave, there still is
651 the issue of whether current empirical and numerical predictive methods are reasonable
652 estimations to a real event.

653

654 Whilst there have been significant developments in numerical analysis and their potential
655 to output accurate predictions, it is important to revisit experimental trials to somewhat
656 re-establish the benchmark fundamental explosive parameters from simple scenarios. Only
657 when the discussion over inherent explosive variability is finalised, can reliable and accurate
658 numerical models be developed. This thesis, has several objectives which are directly
659 related to issues outlines above:

- 660 1. To identify a clear benchmark of free-field explosive characterisation of ideal ex-
661 plosives, and its associated variability with distance, which can be used in yield
662 comparison studies of various composition types.
- 663 2. To develop validated experimental methodologies and tools for analysis to establish
664 reliable data to inform the understanding of fundamental explosive physics.
- 665 3. To establish scalability of explosives which exhibit non-ideal detonation behaviours,
666 and evaluate whether TNT equivalence theories are valid.

667 1.3 Thesis Outline

668 The remainder of this thesis is organised into the following chapters:

- 669 • **Chapter 2 - Literature Review and Theoretical Background**

670 This chapter will provide a detailed overview of previously published literature re-
671 garding explosive blast loading gained through both experimental and numerical
672 analysis across a number of charge compositions, sizes, shapes and location from a
673 target. The aim of this chapter is to scrutinise existing hypotheses and conclusions
674 authors have made regarding the fundamental principles of blast wave development
675 and loading alongside the quantification of variability a given explosive type exhibits
676 in line with Objective 1.

- 678 • **Chapter 3 - Standardising Experimental Methodologies and Analysis**

679 This chapter establishes both experimental methodologies and methods of analysis
680 through extensive validation against recorded data and numerical modelling with
681 Objectives 2 and 3 in mind. The effects of generally assumed insignificant experi-
682 mental features are studied and highlight the need for strict control measures during
683 explosive testing and modelling.

- 685 • **Chapter 4 - Ideal Explosive Blast Characterisation**

686 The validated techniques developed in Chapter 3 were used across a range of differ-
687 ent ideal high explosive compositions in the far-field ranges to determine whether
688 they are applicable across simplistic free-air scenarios to achieve Objective 4. These

689 methods are exercised against current standards of readily available prediction tools
690 to access their accuracy and consistency.

691

692 • **Chapter 5 - Non-Ideal Explosive Blast Characterisation**

693 This chapter aims to complete Objective 5 by providing a detailed characterisation
694 of non-ideal explosives not yet fully discussed within the available literature. The
695 content attempts to assess the scalability of non-ideal explosives and compares the
696 resulting yields to that of ideal explosives and the respective prediction tools.

697

698 • **Chapter 6 - Summary, Conclusions and Future Work**

699 This chapter summarises the research presented within this thesis and makes sug-
700 gestions for future developments towards improving the current understanding of
701 explosive loading conditions within a full-scale range of standoff distances in line
702 with Objective 6.

703

704 **1.4 Published Articles**

705 It is important to note that all experimental work alongside the analysis was undertaken
706 by the author of this thesis with technical assistance from Blastech Ltd. All numerical
707 simulations were undertaken by collaborators at DSTL whom had no further input into
708 the narrative of this research. The work contained within this thesis has been used to
709 publish peer-reviewed journal papers which can be quickly accessed using these QR codes:

710



(a) *Farrimond et al.*^[49]



(b) *Farrimond et al.*^[50]



(c) *Hobbs et al.*^[72]



(d) *Barr et al.*^[10]

Figure 1.1: QR code links to published articles

711 Chapter 2

712 Literature Review and Theoretical 713 Background

714 2.1 Introduction

715 This chapter discusses the fundamental understanding of explosive detonation and the
716 subsequent blast wave propagation and interaction with structures. Throughout this dis-
717 cussion, a review of published literature will be undertaken relevant to the subsequent
718 subject areas investigated within the content of this thesis.

719

720 2.2 The Detonation of Explosives

721 During the detonation process of an explosive, enough energy is provided to the physi-
722 cal solid mass which propagates through the composition in the form of a reaction wave.
723 This results in an exothermic decomposition over a very short duration of time producing
724 hot dense gases^[1]. The reaction process occurs over a given amount of time (varying
725 dependant on charge composition) and is confined within the explosive mass itself until
726 the wave reaches the extents of the explosive. This means that the gaseous detonation
727 products occupy the same volume as the solid mass previously did and therefore exhibit
728 extremely high temperatures and pressures^[101] of around 10-30GPa and 3000-4000°C re-
729 spectively^[32]. The detonation wave is continually reinforced by the energy released from
730 the material detonating immediately ahead of it until the explosives extents are reached.

731

732 Developing the knowledge of the detonation process is fundamental to investigating and
733 characterising the behaviours of detonating explosive compositions. This will be consid-
734 ered regardless of reaction chemistry and detonation dynamics, across a range of masses
735 and distances from a target. A clear review of the different detonation processes are pre-
736 sented with a discussion around the effect these can have on free air shock loading.

737

738 Taylor^[167–169] proposed a very simplistic approach to explosive detonation which hypoth-
739 esised an instantaneous release of energy from a infinity small point in space resulting in
740 a radially expanding shock front. In actual fact, detonation occurs over incredibly small
741 time scales, with the ongoing reactions exhibiting high complexity chemical decomposition
742 and oxidation processes. Zukas and Walters^[182] proposed assumptions for the detonation
743 process which enabled a more robust approach to be adopted to understand an explosives
744 detonation behaviour:

- 745 • The implementation of a one dimensional analysis which considers detonation prod-
746 ucts only propagating behind the shock front with no lateral losses i.e. and infinite
747 charge diameter.
 - 748 – In reality, all charges have a finite boundaries which means losses in energy de-
749 velopment will incur due to product expansion prior to full potential detonation
750 reactions occur
 - 751 • A perfectly planar detonation wave propagates through the explosive to avoid shock
752 front curvatures.
 - 753 – When the detonation front reaches the finite boundaries and energy is lost,
754 the front slows at its peripheries and propagates at higher speeds towards the
755 centre of the charge, therefore exhibiting a curvature.
 - 756 • The reaction is perfectly stoichiometric and occurs instantaneously, with detonation
757 products being in thermodynamic equilibrium.
 - 758 – Detonation by nature is not in thermodynamic equilibrium, the reaction process
759 occurs over some finite amount of time.
 - 760 • The whole process occurs at constantly velocity which in itself assumed a steady
761 state reaction zone.
-

762 – Detonation velocity is directly proportional to the reaction process and therefore
763 is not a constantly velocity in line with the point above.

764 With these assumptions in mind, the accepted view of a high explosives detonation struc-
765 ture consists of four distinct reaction regions, quoted by Paterson^[111] and displayed in
766 Figure 2.1:

- 767 1. The pre-detonation region which is yet to be subjected to the propagating shock
768 wave and therefore the composition is unaffected.
 - 769 2. The leading shock front, travelling at supersonic velocities, compresses the explosive
770 and initiates the detonation through exothermic decomposition of molecular bonds.
771 In the simplest form, detonation theories presented by Chapman^[26] and Jouguet^[84]
772 (C-J) assumed the reaction to occur instantaneously with the flow becoming sonic
773 at the C-J point. At this point, the energy released behind the sonic front is able to
774 support the detonation front at constant velocity and pressure^[55]. von Neumann^[177]
775 however argued that detonation was to occur over a finite time and reaction zone
776 length, resulting in a significant difference in pressure and temperature either side
777 of this region. This was further developed to consider the existence of an induction
778 period of time related to reaction initiation time lag once compressed by the leading
779 detonation wave^[166].
 - 780 3. The reaction zone, in which an increasing amount of energy is released with a function
781 of distance travelled along a given explosive composition, results in an acceleration
782 of the products away from the shock front and therefore propelling the shock front
783 forward^[11]. The reaction zone is a function of characteristic time taken for the
784 chemical decomposition of the explosive and therefore the rate at which energy is
785 released.
 - 786 4. Upon reaction completion, the detonation products begin to expand with the loss
787 of inertial confinement of the solid state with a rarefaction wave travelling in the
788 opposite direction the detonation wave^[31].
-

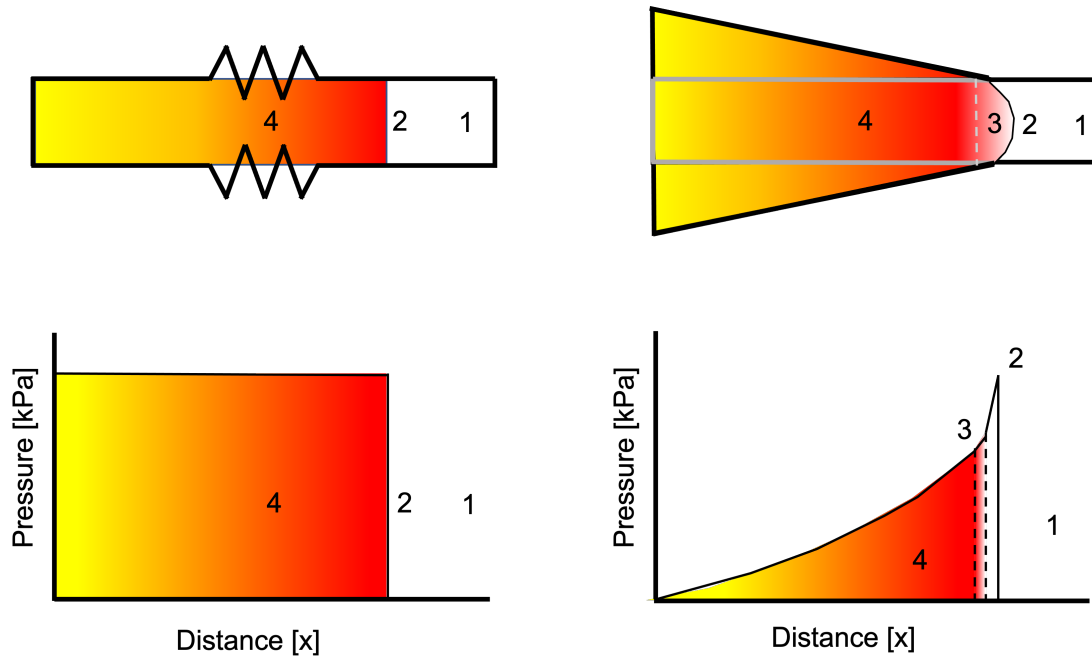


Figure 2.1: Schematic representations of detonation theories: C-J theory (left) displaying the planar like pressure wave with distance and no lateral losses, and the theory proposed by von Neumann^[177] (right) which details finite reaction zones and detonation product expansions. Note the numbering system aligns with detonation process discussed above.

789 The detonation theories outlined categorise explosives in to two groups: Ideal and non-
 790 ideal explosives. Ideal explosives, when detonated, exhibit behaviours which are captured
 791 by C-J theory with high levels of agreement through the use of thermochemical com-
 792 puter codes^[48], such as Cheetah^[58] and EXPLO5^[161]. These tools have integrated the
 793 C-J theory into their methodology, to predict the behaviour of an explosive based on its
 794 composition, heat of formation and density^[179]. In reality, most commercial available
 795 explosives, and certainly all home-made explosives (HME), often result in detonation ve-
 796 locities and pressures which do not follow C-J theory, labelled non-ideal^[159]. This group of
 797 explosives are generally heterogeneous fuel-oxidiser mixtures which when detonated result
 798 in non-monotonic relationships between detonation and density which is not captured by
 799 thermochemical calculations^[114].

800

801 The C-J theory makes the assumption that for a planar detonation in an infinite diam-
 802 eter charge, D_∞ , the reaction zone structure would not alter the maximum velocity of
 803 detonation to which is calculated^[75]. The reality is that charges have a finite diameter
 804 and therefore are subjected to lateral energy losses through detonation product expansion,
 805 thus never exhibiting their maximum potential velocity. Fundamentally, as the diameter

806 of an explosive composition reduces, the amount of energy lost through product expansion
807 increases, resulting in detonation behaviours unclassifiable by standard C-J theory. This
808 reduction in energy continues up to the diameter at which the reaction zone no longer
809 releases enough energy to sustain the detonation process^[142], called the critical diameter
810 of an explosive composition, d_{cr} .

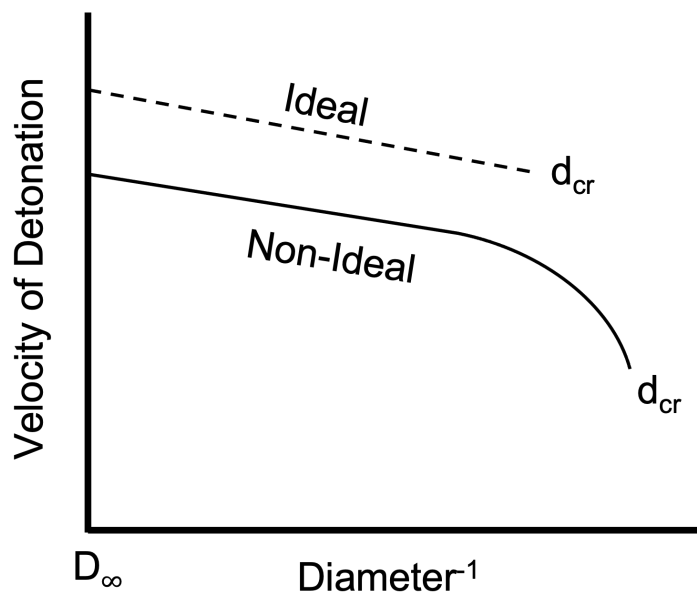


Figure 2.2: Schematic representations of detonation velocities and how they vary with charge diameter for ideal (dashed) and non ideal (solid) explosives adapted from work presented by Scott^[142].

811 The reason C-J theory is unable to capture the behaviour of non-ideal explosives is due
812 to the assumption that reactions occur instantaneously. An increase in non-ideality of
813 an explosive is a direct result of a larger reaction zone and characteristic time for full
814 decomposition of the explosive to occur. For military grade explosives, which exhibit ideal
815 like behaviour, the reaction zone length are small ($\sim 1\text{mm}$) and therefore the d_{cr} can be
816 within a few millimetres. For the non-ideal explosives, the reaction zone is significant
817 larger ($\sim 10\text{mm}$) meaning the d_{cr} is orders of magnitude bigger^[94]. This behaviour is rep-
818 resented by Figure 2.2.

819

820 2.3 Air Shock Formation

821 Once the detonation process reaches the extents of an explosive composition there is a dis-
 822 continuity between it and the surrounding medium. This discontinuity results in both the
 823 rapid expansion of detonation products into the surrounding medium and a reflective wave
 824 propagating back towards the detonation epicentre. The expanding detonation products
 825 displace the surrounding medium at speeds dependent on the magnitude of energy con-
 826 tained within the wave itself. Air is the medium of choice within this thesis it is important
 827 to note that it is a compressible fluid. This means that rapid expansions and disturbances
 828 to the surrounding air results in higher pressure components of the expansion travelling
 829 quicker than the lower pressure components over small periods of time. This transitions
 830 the expansion into an effectively instantaneous increase in pressure and density, propa-
 831 gating away from the explosive in the form of a shock (or blast) wave^[7], represented by
 832 Figure 2.3. It is important to note that Figure 2.3 represents the fireball and shock wave
 833 breakout of a hemispherical charge. Shapes which exhibit geometrical variations would
 834 result in different breakout procedures and thus loading conditions in near-field regimes
 835 which self-heals far-field.

836

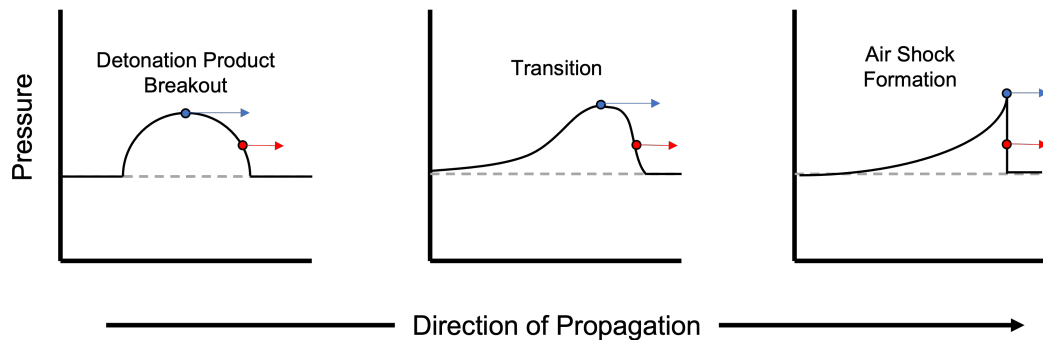


Figure 2.3: Schematic representation of the formation of air shock from hemispherical explosive detonation: Depicted by Initial pressure pulse related to the detonation product breakout (left), transitioning into a front loaded pulse due to a range of particle velocities (middle) to a point where a near-discontinuous shock wave forms (right), adapted from Kinney and Graham^[90].

837 The detonation of an explosive and the subsequent expansion of the gaseous fireball prod-
 838 ucts displace the surrounding air and compresses it into a high pressure discontinuity with
 839 undisturbed air ahead of it. As the expansion propagates, a temporal decay in pressure
 840 and velocity back to ambient air pressure is experienced. For spherical/hemispherical ex-

841 plosive charges, the detonation product cloud halts its expansions at distances some 20
 842 charge diameters away from the epicentre, which up til this point would be described as
 843 the near-field regime^[173]. The shock wave preceding the detonation products continues to
 844 propagate radially until it either interacts with a target (reflected) or decays away (inci-
 845 dent), defined as the far-field regime. Figure 2.4 provides a visual representation of this
 846 behaviour.

847

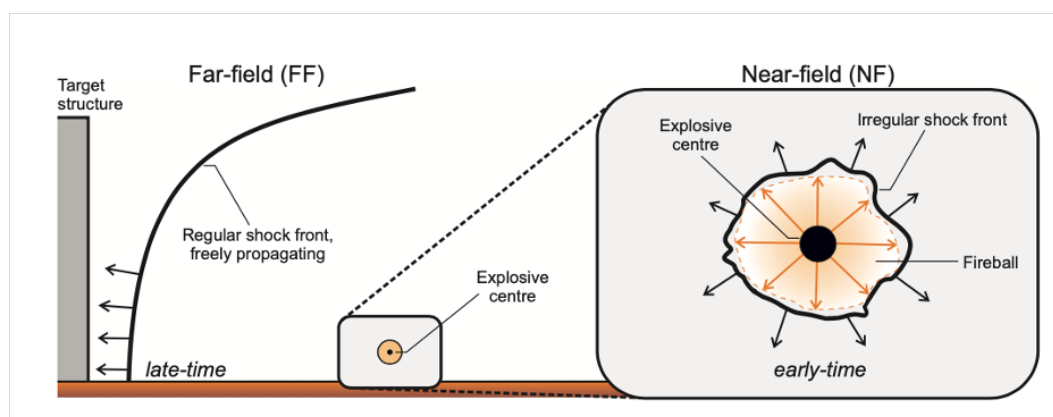


Figure 2.4: Schematic representation of the the different scaled distance regimes discussed by Tyas^[173]

848 It is important to note that common blast engineering practices consider far-field regimes,
 849 in which an air shock is fully formed and has propagated far enough away from the detona-
 850 tion epicentre to not be affected by the fireball. In close proximity to the detonation, where
 851 the air shock has not detached from the fireball, the current knowledge of blast loading
 852 is significantly undefined. This is due to the difficulty in measuring such high pressures
 853 and impulses without damaging equipment. In conjunction to this, any interaction in this
 854 regime is a result of both fireball products and potentially a slower loading profile, as seen
 855 in Figure 2.3.

856

857 Naturally, there are a number of variables which dramatically affect the characteristics of
 858 these shock waves including: the distance the explosive is away from a target, known as
 859 the standoff distance; the shape, size and type of the explosive used; and both the detona-
 860 tion and interaction mechanisms occurring which alter the intrinsic characteristics of the
 861 waves. For the contents of this thesis, hemispherical charges are the only shape considered
 862 to make direct comparisons with the broad range of historical data available. Having an
 863 understanding of how each of these parameters effects the characteristics of a shock wave

864 is important for probabilistic-based analysis of loading conditions on a structure.

865

866 Figure 2.5 presents an idealised shock wave pressure-time history defined as the modified
 867 Freidlander waveform which can be approximated by Equation (2.1) [57]. Considering a
 868 point far enough away from the detonation epicentre to be characterised by a fully formed
 869 air shock, the behaviour is characterised by a near-instantaneous rise in overpressure, P_{so}
 870 and P_r for incident and reflected results respectively. This has a characteristic arrival time,
 871 t_a , and a temporal decay back to ambient conditions, P_o , which occurs over a duration
 872 of time, t_d . The waveform parameter, b , controls the decay of the pressure-time curve.
 873 Through integrating the pressure-time history with respect to time, a impulse-time history
 874 can be plotted. This can be attributed to the change in momentum, or specific impulse
 875 i , a given area experiences from the different loading conditions, denoted by the coloured
 876 regions in Figure 2.5. This period of a shock waves behaviour is defined as the positive
 877 phase which is considered the most destructive part of blast loading [182].

878

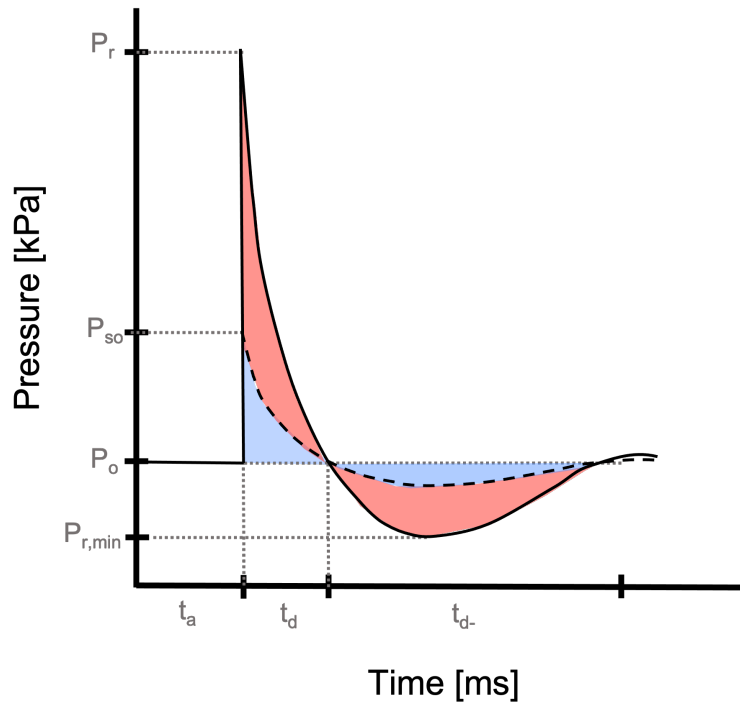


Figure 2.5: Indicative Freidlander shock wave pressure-time history with reference to both reflected (red) and incident (blue) forms with respect to each other.

$$P(t) = P_{max} \left(1 - \frac{t}{t_d}\right) e^{-b \frac{t}{t_d}} \quad (2.1)$$

879 The rapid overexpansion of the medium results in a lower density partial vacuum be-
 880 hind the shock front which forces air back towards the detonation epicentre to maintain
 881 equilibrium conditions, defined as the negative phase of a shock wave^[107]. Similarly to
 882 the positive phase, the time taken for conditions to return to ambient is defined as the
 883 negative phase duration, t_{d-} , with the maximum underpressure defined by $P_{so,min}$ and
 884 $P_{r,min}$. Figure 2.6 presents a schematic of the negative phase mechanism which details
 885 the aforementioned processes. It is important to note that for hemispherical charges, the
 886 negative phase measured at the ground surface is found to be greater due to the rising
 887 fireball. This creates a convectional lower pressure regions, thus resulting in a greater
 888 underpressure^[133]. Further to this, at some time after the arrival of the positive phase,
 889 secondary and tertiary waves arrive at the same recording position. This phenomena is
 890 a result of rarefaction waves from detonation product expansion, coalescing at the point
 891 of detonation, and reflecting radial outwards. This is similar to a pulsing-like effect with
 892 subsequently lower yields associated with each iteration of shock.

893

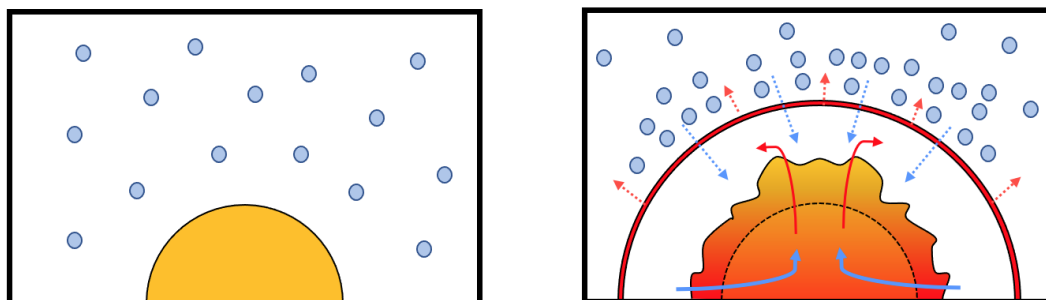


Figure 2.6: Indicative behaviour of negative phase process: Pre-detonation of charge with reference air particles (left), and Post-detonation with reference to shock wave condensed air particles, leaving a vacuum with an under pressure.

894 2.4 Secondary Shock

895 Needham et al.^[106] discussed the phenomena of secondary shocks occurring sometime af-
 896 ter the formation and propagation of a primary shock, resulting in an imploding moving
 897 shock which coalesces at the detonation epicentre and reflects back outwards. The results
 898 of this process is presented in Figure 2.7, where P_{SS} is the associate shock pressure and
 899 td_{ss} is the phase duration of the secondary shock. Whilst the feature is documented across

900 a number of articles^[7], historically there seems to have been little effort made to quantify
 901 or characterise its effects. This is mainly due to exhibiting low yields and therefore corre-
 902 sponding to low impact on structures or civilians.

903

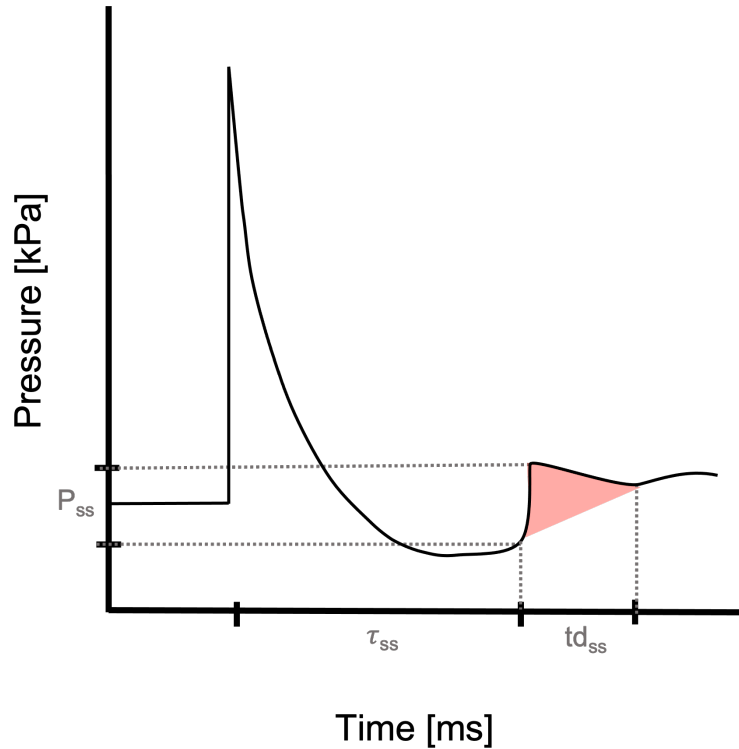


Figure 2.7: Indicative Freidlander shock wave pressure-time history with reference to the secondary shock phenomena and the blast parameters associated.

904 Work presented by Gitterman and Hofstetter^[62] began to compile a considerably large
 905 data set of the secondary shock arrival time, and its subsequent delay after the arrival of
 906 the positive phase, across a range of explosive compositions, sizes and scaled distances.
 907 The reason for this analysis was to help improve the predictions of negative phase pa-
 908 rameters as currently numerical models struggle to capture them with definitive accuracy.
 909 With near-field blast parameters within the fireball holding difficulty to prescribe with
 910 standard methods, it is believed that the behaviour of a secondary shock could be used as
 911 an indicator for post-detonation products within the fireball.

912

913 With the aforementioned in mind, the secondary shock delay parameter, τ_{ss} , was intro-
 914 duced by Rigby and Gitterman^[129] who attempted to normalise each explosive by scaling
 915 the delay parameter by standard detonation velocities and the cube-root of the packing

916 density, defined by Equation 2.2. The thinking behind this methodology was to correlate
 917 the parameter with the characteristic time for complete detonation of a finite hemispheri-
 918 cal charge, where ρ is charge density, V_{od} is velocity of detonation, W is charge mass and
 919 δt_{ss} is the time difference in arrival times between primary and secondary shocks. The
 920 data presented in Figure 2.8 was obtained from the authors of the aforementioned articles
 921 and improved by including standard TNTe values for each explosive, which seemed to
 922 provide increased linearity of the data when compared to the least-squares fit defined by
 923 Equation 2.3.

$$\tau_{ss} = \delta t_{ss} V_{od} / (\rho W)^{1/3} \quad (2.2)$$

$$\tau_{ss} = 2.45 \log_{10}(Z) + 1.36 \quad (2.3)$$

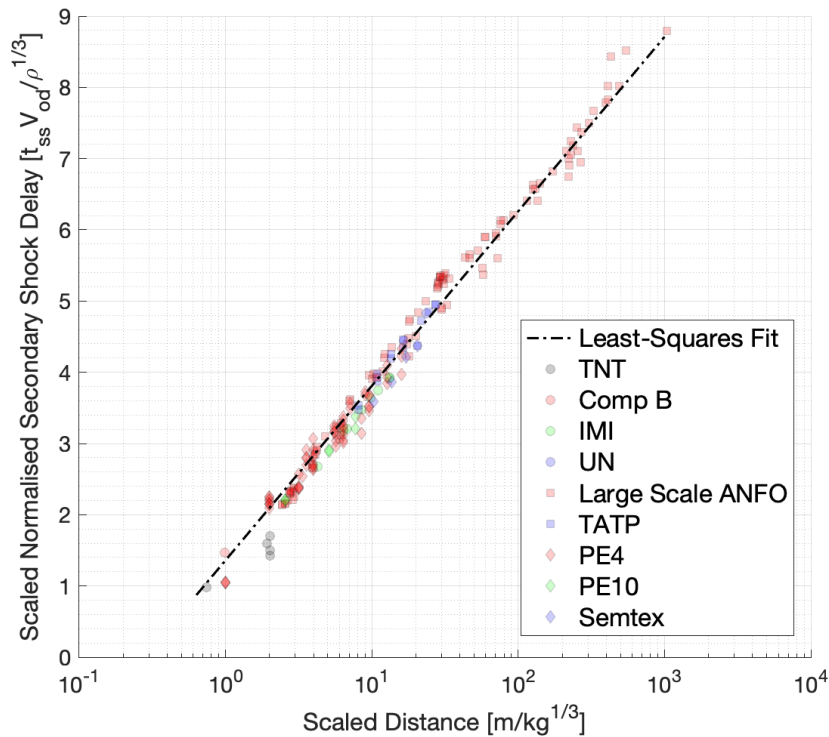


Figure 2.8: Secondary shock delay parameter evaluated for all explosives scaled accordingly compared with a least-squares fit to data, extracted from Rigby and Gitterman^[129]

924 The least-square fit, detailed by Equation 2.3, was tested against experimental data and
 925 theoretical examples presented by Rigby and Gitterman^[129] and was found to provide

926 remarkably accurate results. The issue with this methodology is that the secondary shock
 927 parameter defined by this analysis has units of $\text{m}^2/\text{kg}^{2/3}$ which holds no physical mean-
 928 ing and therefore a different consideration is required to assign a real mechanism to the
 929 behaviour.

930

931 Despite analysis undertaken of the large data for the delay in arrival time^[62,129,133], there
 932 is still a gap in characterising the other secondary shock blast parameters. There is a need
 933 for a theoretical mechanism which captures the behaviour of secondary shocks across a
 934 variety of explosives and scaled distances which can be implemented into predictive tools
 935 to provide more realistic and valid predictions.

936

937 2.5 Scaling Laws

938 2.5.1 Mass and Energy Proportionality

939 A fundamental finding proposed separately by Hopkinson^[74] and Crazz^[33], suggested
 940 that through cube-root scaling of the distance a charge is away from a target, results
 941 in consistencies in the yields from two different explosive charge. This can be achieved
 942 provided the same shape of charge and direct equivalent masses can be established. This
 943 means that the blast parameters at a given distance away from the centre of the explosive,
 944 R , with mass, W , will be comparable to that at a distance, KR from a mass of K^3R .
 945 As a blast wave spherically propagates, energy dissipates into the surrounding medium in
 946 all three dimensions, hence the cube-root scaling, which led the aforementioned authors
 947 to develop Equation 2.4, where E , is the energy released per unit mass of the explosive
 948 material.

$$\frac{R}{E^{1/3}} = \frac{KR}{(K^3E)^{1/3}} \quad (2.4)$$

949 The amount of resulting energy produced upon detonation of an explosive, is inherently
 950 dependent on the equivalent mass of the explosive, W . This leads to Equation 2.5, which
 951 introduced the notion of scaled distance, Z , measured in units of $\text{m}/\text{kg}^{1/3}$.

$$Z = \frac{R}{W^{1/3}} = \frac{KR}{(K^3W)^{1/3}} \quad (2.5)$$

952 The Hopkinson^[74] and Cranz^[33] cube-root scaling, assumes the pressure associated with
 953 a blast is directly proportional to the energy released so does not require scaling. It is
 954 only time-related parameters which require scaling by the cube-root of the explosive mass
 955 Kinney and Graham^[90].

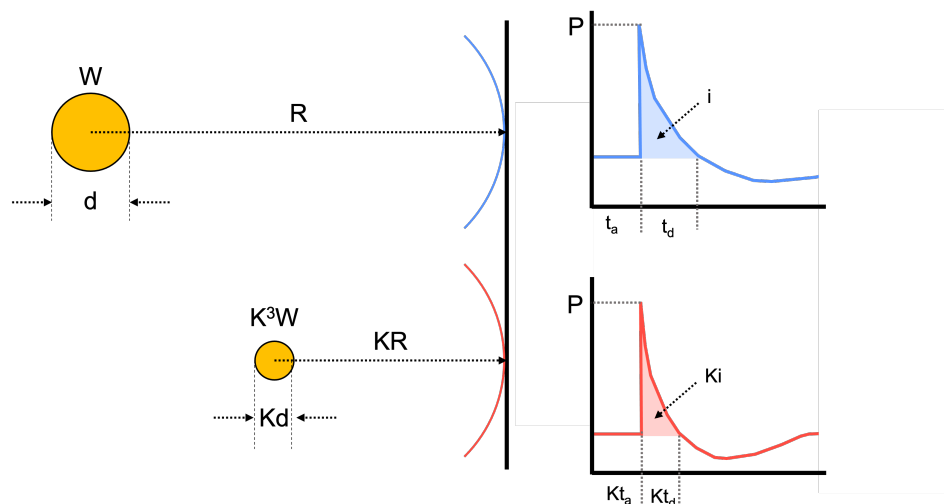


Figure 2.9: Representative schematic of Hopkinson^[74]-Cranz^[33] scaling, adapted from Baker^[7]

956 2.5.2 Charge Shape

957 The effect of charge shape has been investigated throughout literature to assess and char-
 958 acterise the differences in the blast parameters induced. Both Wu et al.^[180] and Sherkar
 959 et al.^[146] undertook the numerical simulation of explosive detonation for spherical and
 960 cylindrical across a range of scaled distances, charge orientations and detonation positions
 961 on the charge. Presented were the distributions in blast parameters which concluded that
 962 charge shape has a significant effect. When standoff distance from the detonation point is
 963 greater than 35 charge diameters $Z > 3m/kg^{1/3}$, the effect of charge shape can be ignored.
 964 Similar ranges were presented by Xiao et al.^[181], but all came to the agreement that the
 965 information presented was not sufficient enough to identify the precise scaled distance at
 966 which charge shape effects could be ignored definitely. Shi et al.^[147] countered this finding
 967 when a variety of 1kg TNT explosive trials consisting of four different charge shapes in
 968 near-field scenarios were conducted and agreed that the blast parameters recorded are
 969 far more sensitive in the near-field when non-spatial symmetry of an explosive creates
 970 non-uniform detonation processes. This was displayed by high speed video frames which
 971 showed how variability in breakout shape decreased with scaled distance but only becom-
 972 ing negligible at $Z > 5m/kg^{1/3}$.

973

974 Spherical and hemispherical charges exhibit far more similarities through radial detona-
975 tion mechanics and subsequent shock wave breakout. Early works presented by Brode^[19]
976 undertook numerical simulations of spherical blast waves which have been used in conjunc-
977 tion with experimental trials using large scale TNT hemispheres^[87] to develop fast-running
978 engineering computer codes, such as ConWep^[76]. These methods were all based on the
979 assumption of spherical charges detonation in free air, or a hemispherical charge detona-
980 tion on a flat surface.

981

982 Assuming a perfect reflecting plane, a hemispherical charge of mass, W , in theory a spher-
983 ical charge of mass, $2W$, would produce the equivalent blast parameters. The issue with
984 this theory is that in reality, hemispherical charges detonated on a ground surface would
985 result some energy be lost through cratering. A shape factor of 2 in this instance would
986 be conservative, however the aforementioned work presented a realistic factor of 1.8. This
987 has been integrated within design code and are validated for spherical and hemispherical
988 charge^[135]. These simplistic scenarios are fundamental to continue the understanding into
989 both detonation and shock wave mechanics. Although uncommon within real-world blast
990 application, without a robust knowledge on the behaviour of blast loading in simplistic
991 scenarios, confidence in the understanding of non-spherical explosive detonation will never
992 be established.

993

994 **2.6 Semi-Empirical Prediction Methods**

995 **2.6.1 Positive Phase Parameters**

996 As mentioned previously, Taylor^[167], Brode^[19] and Granstrom^[66] were the pioneers of
997 explosive shock wave prediction. Their developing theories which assumed instantaneous
998 energy release at a infinitesimal small point source captured the behaviour of simple ex-
999 plosive scenarios. These however were not perfectly applicable for use when considering
1000 the effects loading as on infrastructure^[126].

1001

1002 There has been significant research since, expanding the fundamental understanding of ex-
1003 plosive events alongside characterising resulting blast wave features through experimental

1004 analysis^[85]. Baker^[7], Remennikov^[123] and Esparza^[45] all provide summaries of contem-
 1005 porary and historical experimental quantification of blast parameters.

1006

1007 Discussed in more detail within Section 2.9.2, is this history around the quantification
 1008 of blast loading and the effects imposed on structures. The broad range of works in-
 1009 vestigating high explosive detonations was collated by Kingery and Bulmash^[87], whom
 1010 presented high-order polynomial relationships which capture the explosive behaviour of
 1011 spherical and hemispherical TNT detonations across scaled distances spanning from the
 1012 edge of a explosive charge, out to around $Z \simeq 40\text{m}/\text{kg}^{1/3}$. It is notable that the data
 1013 collected was a combination of both direct measurements, in the far-field region, and in-
 1014 ferred measurements from tracking smoke trials^[37]. This was supported with rudimentary
 1015 numerical analysis, in the near-field^[19], therefore holding this region in doubt of its fidelity.

1016

1017 The high-order polynomial relationships only exhibit the behaviour of TNT detonations
 1018 across a scaled distance range and therefore should not be mistaken for a representation
 1019 of a physical mechanism directly. With this being said, the relationships are usually sum-
 1020 marised and used as a set of well defined curves, as presented in Figures 2.10a-b.

1021

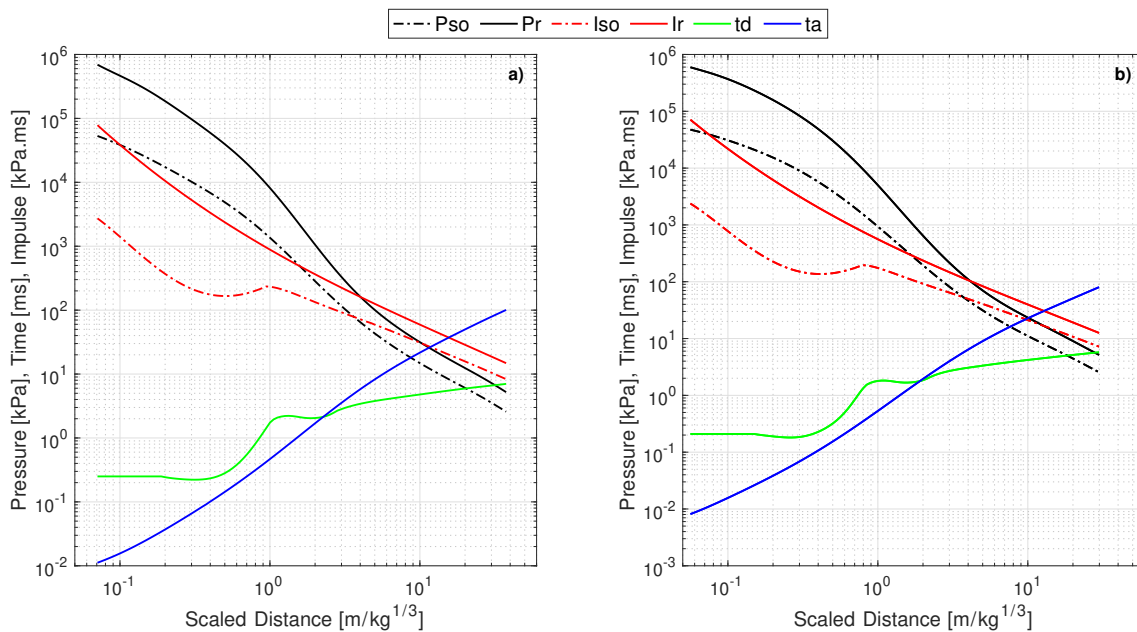


Figure 2.10: The semi-empirical curves for incident and reflected positive phase blast wave parameters for a 1kg TNT explosive detonation with respect to scaled distance, established by Kingery and Bulmash^[87]: a) Hemispherical charge of TNT detonated of a ground surface, b) Spherical charge of TNT in free air.

1022 2.6.2 Negative Phase Parameters

1023 The positive phase of a blast event is reasonably understood in terms of the underly-
 1024 ing fundamental physics. Vast quantities of experimental and numerical work have been
 1025 conducted to characterise and quantify the associated blast parameters and to validate
 1026 empirical predictions, as discussed in Section 2.9. Whilst negative phase parameters are
 1027 available in literature and blast related design manuals^[176], Bogosian et al.^[13] presented
 1028 scepticism of the associated empirical curves due to the original data source being unclear.
 1029 Rigby^[126] undertook a thorough review of available historical literature and highlighted
 1030 the work by Granstrom^[66] to be the likely source of data used to develop the empirical
 1031 curves, seen in Figure 2.11. This is defined as the ‘Cubic Negative’ curve fit approximation
 1032 (detailed in the second case of Equation 2.6).

1033

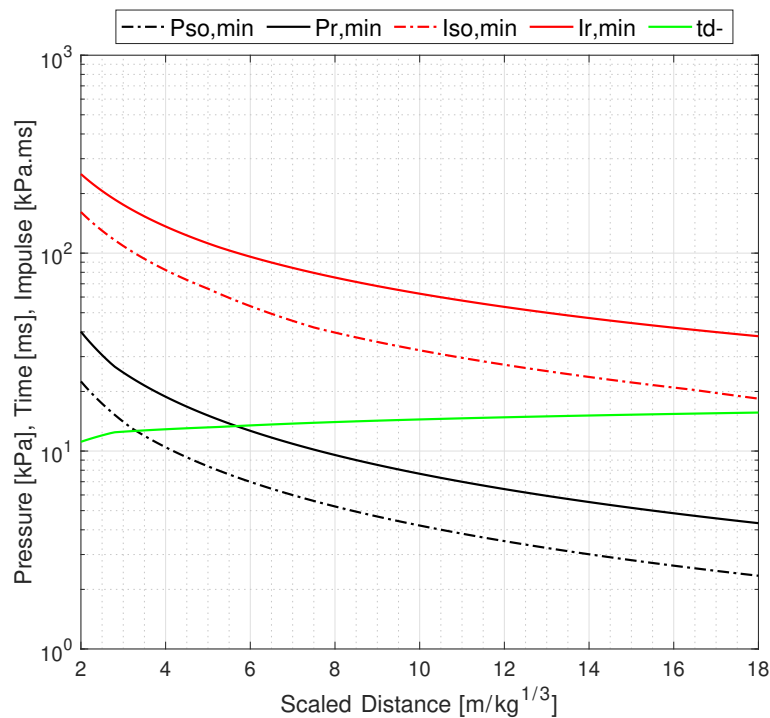


Figure 2.11: The semi-empirical curves for incident and reflected negative phase blast wave parameters for a 1kg TNT explosive detonation with respect to scaled distance, established by Granstrom^[66]: a) Hemispherical charge of TNT detonated of a ground surface, b) Spherical charge of TNT in free air.

1034 There is however, limited validation of the negative phase on the whole with many re-
 1035 searchers historically omitting their effects due to the associated parameters being orders
 1036 of magnitude lower than the positive phase. The physical mechanisms ongoing during this

1037 time of lower pressure in reality can still pose some serious threats to infrastructure and
 1038 civilians.

1039

$$P_r(t) = \begin{cases} P_{r,max}(1 - \frac{t}{t_d})e^{-b\frac{t}{t_d}}, & t \leq t_d. \\ -P_{r,min}(\frac{6.75(t-t_d)}{t_d})(1 - \frac{(t-t_d)}{t_d})^2, & t_d < t \leq t_d + t_{d-}. \end{cases} \quad (2.6)$$

1040

1041

1042 2.7 TNT Equivalence

1043 The mass term, W , referred to throughout this section is actually the equivalent charge
 1044 mass in kilograms of trinitrotoluene (TNT), which is widely accepted as the benchmark
 1045 in explosive characterisation. Discussed in more detail with Section 2.9.2, the quantity
 1046 of TNT free-air testing which has been conducted over a multiple decades. These were
 1047 collated by Kingery and Bulmash^[87] and used to develop semi-empirical predictive curves
 1048 for 1kg TNT explosives, shown in Figures 2.10 and 2.11. Using these, the results of det-
 1049 onating explosives of differing compositions have been collated, with their masses scaled
 1050 until predictions of 1kg of TNT are able to accurately capture the behaviour of an equiv-
 1051 alent mass of the given explosive. This scaling factor is defined as the TNT equivalence,
 1052 TNT_e .

1053

1054 The problem with TNT equivalence is mainly the lack of significant rigour associated
 1055 with the testing methodologies adopted for establishing the parameter. Cooper^[30] out-
 1056 lined in the overview on TNT equivalence that historically there have been various and
 1057 widely different testing methodologies to estimate the factor, all of which exhibit their
 1058 own uncertainties. This is a direct cause of the published spreads in TNT_e for nominally
 1059 identical charge masses and compositions tested, and therefore the same energy release.
 1060 It is noted, that in the extreme near-field, where the recorded blast parameters are highly
 1061 proportional to the fireball chemistry, different explosive compositions will exhibit a vary-
 1062 ing TNT_e as a result different chemical de-combustion^[81,151]; a finding which was verified
 1063 by Shin et al.^[149].

1064

1065 Further justifying the sensitivity of detonation characteristics in the near-field was works

1066 presented by Simoens and Lefebvre^[152]. A variety of experimental trials highlighted how
1067 the detonator position, and thus the associated detonation process, caused variations in
1068 pressure recordings but impulse effectively stays constant. This is due to impulse being
1069 directly related to the overall energy released during detonation. If using impulse as the
1070 only measuring parameter for TNT equivalence, the theory proposed by Taylor^[167, 168, 169]
1071 on explosives being theorised as infinitesimally small points of energy release would hold
1072 credibility.

1073

1074 Far-field scenarios, the region in which the shock wave detaches from the fireball and
1075 therefore no longer influenced by the ongoing thermochemical combustion reactions, there
1076 is presently no physical explanation for TNT equivalence to vary. Dewey^[38] used high
1077 speed video techniques to track a propagating shock wave resulting from a hemispheri-
1078 cal propane-oxygen explosion and converting its velocity into the shock waves pressure
1079 through Rankine-Hugoniot relationships^[7]. A constant TNT equivalence value within the
1080 far-field range, $Z > 3\text{m}/\text{kg}^{1/3}$, was evaluated. Similar findings have been presented when
1081 conducting both well-controlled PE4 explosive trials in far-field regimes and comparing
1082 them to high fidelity numerical simulations, both converge on a $\text{TNT}_e \simeq 1.2$ ^[132, 135]. How-
1083 ever despite a broad range of published literatures attempting to characterise this factor for
1084 different explosives, there is no widely accepted methodology of establishing TNT_e ^[32].
1085 Some authors exhibiting large variation in TNT equivalence across an entire scaled dis-
1086 tance rather than variability being confined to the near-field domain^[56].

1087

1088 The TNT equivalency factor of an explosive composition is highly related to the variability
1089 associated in experimental data captured. It therefore requires further investigation to
1090 establish whether all explosives can be defined by an equivalent mass of TNT, or whether
1091 variations are inherent to explosions.

1092

1093 **2.8 Blast Variability - Real or Systematic?**

1094 Explosive testing within the available literature provides a wide variety of methodologies
1095 for undertaking experimental blast work and the analysis of the resulting data. However,
1096 varying conclusions regarding the output yield of a given explosive have when detonated at
1097 certain distances from a target, alongside the associated variability it exhibits between

1098 nominally identical tests, have been made. There is therefore a necessity to further im-
1099 prove current practices and develop new techniques for experimental testing and data
1100 processing to ensure the fundamental understanding of shock wave propagation is fully
1101 understood. Clearly, if blast experiments are viewed as naturally varying, approximate,
1102 and even first-order in nature, then our ability to rigorously validate numerical modelling
1103 approaches is hampered. This clearly has negative connotations for the use of modelling
1104 tools for design. It is important to fully understand the nature of blast parameter vari-
1105 ability, and its dependence on extrinsic features such as experimental set up, control, and
1106 interpretation of data. The contention around this subject matter however poses a clear
1107 need for defined experimental procedure which adopt precision. These require validation
1108 to encouraging a movement towards an acceptable experimental and analytical standard
1109 which provides a consistency in resulting blast parameters from explosive testing.

1110

1111 Whilst any published article will defend and justify its findings, multiple contrasting con-
1112 clusions regarding explosive variability from similar testing regimes, cannot all be scientif-
1113 ically correct. The philosophy around research in modern times, for blast engineering in
1114 particular, has seemingly moved away from critically analysing why certain data appears
1115 to exhibit particular trends, making links through tangible justification of fundamental
1116 physical mechanisms and principles. Instead, a more literal acceptance of recorded data
1117 is commonly taken. Inductively creating theories justifying the accuracy of the data re-
1118 gardless of the underlying physics required for the data to be valid.

1119

1120 The Falsification Principle, proposed by Popper^[113], suggested that scientific theory should
1121 make predictions which can be tested, and the theory rejected if these predictions are
1122 shown to be incorrect. Adopting critical rationalism is something which enables science
1123 to hone in on realistic physical theories which is void of highly-controlled experimental
1124 data to falsify them. Take the paradoxical opinions in the blast community regarding con-
1125 sistency in blast parameters; what are the underlying reasons for why different research
1126 groups are experiences varying levels of consistency?

1127

1128 It is of great importance to establish the route causes for these conflicting conclusions
1129 and to develop best practices based on the findings. This chapter will access the features
1130 of a blast waves development as it propagates away from the source of the detonation.
1131 Through careful analytics, experimental procedures and improved methods of analysis are

1132 proposed for use across a range of idealised high explosive types.

1133 2.9 Historical Review of ‘Best’ Practices

1134 The measurement of shock wave data from explosive events has been of interest since the
1135 first explosion occurred, most probably around similar times as the invention and subse-
1136 quent detonation of gun powder. The ability to release significantly large quantities of
1137 energy over small periods of time through detonation, created an interest in developing
1138 weapons to release shock waves of catastrophic capabilities^[63].

1139

1140 With increasing interest into powerful energetics came extensive testing regimes which
1141 looked at detonating large quantities of various types of explosives. This required instru-
1142 mentation to record characteristics of these events to provide a fundamental understanding
1143 of shock wave phenomena. Due to the threat severity of nuclear war at the time, instru-
1144 mentation engineers and scientists were faced with an incredibly difficult task of recording
1145 the effects of equivalent blasts with limited resources and time constraints.

1146

1147 The development of best practice for blast damage and shock data capture is not some-
1148 thing which came easy, and has taken many decades to establish reliable instrumentation
1149 in which these events can be recorded with the highest levels of precision. Shelton^[145]
1150 documented that during the first nuclear tests conducted by the US, ‘blast damage and
1151 shock data were obtained by two methods: those that were improvised by inquisitive sci-
1152 entists; and those that were measured by sophisticated instruments’. The creativity and
1153 innovation from the scientists working on these large scale trials are to be commended
1154 for having a well-rounded understanding of fundamental explosion physics enabling their
1155 efforts in effective instrumentation development. The iterative procedure of testing and
1156 modifying instrumentation over many years resulted in the technology available for us
1157 today. It is however important to establish the historical context of how these instruments
1158 came to be developed. This will help understand why the current methodologies adopted
1159 for testing results in sporadic data sets between nominally identical trials.

1160

1161 Reisler et al.^[118] meticulously reviewed all available literature, both within the public
1162 domain and that which is classified. The article detailing the testing and development
1163 of systems which were capable of not only withstanding, but recording disastrous effects

1164 from nuclear explosions. The vast majority of devices deployed during these early tests
1165 were passive, inexpensive to produce and required minimal effort to deploy. These how-
1166 ever, reported spreads in accuracy of shock wave parameters between 10-20% of the values
1167 numerically predicted. As the development of electronic transducers began, data could be
1168 recorded with respect to a time base, thus establishing a detailed understanding of the
1169 development of explosive shock wave phenomena from the instance of detonation. These
1170 transducer measurements were found to obtain accuracies of between 3-5% which showed
1171 vast improvements in the quality of data recorded despite their need to be coupled with
1172 electronic recording systems such as oscilloscopes.

1173

1174 This brings the thinking back to that discussed by Shelton^[145], in that there was a skill,
1175 and in some scenarios improvisation, required to use these systems to record highly accu-
1176 rate data. The capabilities of these scientists to understand and predict what the effects
1177 of a given explosive should be, and testing their predictions against constantly evolving
1178 instrumentation provides definitive confidence in their research. It is safe to conclude that
1179 the instrumentation developed during the nuclear period was successful at capturing shock
1180 parameters with accuracy for long duration blast waves. Despite the era these instruments
1181 were developed, the quoted accuracy of between 3-5% when comparing recorded data to
1182 numerical models are within the similar regions to modern research programmes looking
1183 into high explosive detonations^[135].

1184

1185 The use of high explosives for large scale blast phenomena analysis closely followed the
1186 initial ban on atmospheric testing of nuclear explosives in 1958. The reason for this was to
1187 provide effective simulations of a nuclear weapon's blast characteristics without the ma-
1188 jor environmental concerns of nuclear detonation. The majority of the instrumentation
1189 adopted for testing at the start of this era was that which were established during the
1190 nuclear testing era. This movement meant that measuring devices no longer required the
1191 need to be radiation-resistant but did require the response and recording rates of gauges
1192 to be considerably faster relating to much shorter shock wave durations associated with
1193 high explosives^[8]. Reisler et al.^[119] again provided a comprehensive overview of the de-
1194 velopment of instrumentation during this period. Detailed was how the state of the art
1195 recording systems for electronic gauges progressed from the 20-40 kHz frequency of record-
1196 ings range to 500 kHz through the use of digital recording systems which are still used
1197 today. This still achieved the high accuracy recorded data when compared to numerical

1198 models.

1199

1200 Blast related research placed greater demands on data capture systems to provide increased
1201 frequency response whilst upholding precision in the data. However, greater temporal ac-
1202 curacy of recordings does not necessarily result in consistency and reliability of recorded
1203 data from repeat trials. The knowledge of what an experimental trial is aiming to accom-
1204 plish and an understanding of the physical processes any given instrumentation uses to
1205 capture the data is paramount for achieving reliable and accurate data. When reviewing
1206 the quality of published works between the start of shock wave quantification in the 1940's
1207 all the way through to modern times, there is a clear discrepancy in the results published
1208 which requires investigating.

1209

1210 **2.9.1 Data Capture between 1950 - 1980**

1211 Corresponding to the era in which instrumentation was being constantly tested and im-
1212 proved, the research undertaken into blast phenomena produced some of the more widely-
1213 regarded data sets which helped develop semi-empirical prediction tools for free-air blast
1214 scenarios^[87]. This well-known method utilises poly-logarithmic curves fitted to the com-
1215 pilation of both rudimentary numerical analysis results and experimental measurements
1216 ranging from small to large scale events ($< 1\text{kg}$ to $>400,000\text{kg}$). These semi-empirical
1217 curves are generally accepted standard practice for predicting blast loads from a given
1218 explosive mass at a given distance from the target. As a result, the KB method has been
1219 implemented into the UFC-3-340-02 design manual^[176], the predictive computer code Con-
1220 wep^[76,77], and commercial finite element code LS-DYNA^[116].

1221

1222 Figures 2.12a-d present experimental recordings from TNT charges with masses between
1223 $0.45 - 450,000\text{kg}$, which were spherical and hemispherical in shape, recovered from 22
1224 individual references dated between 1940 and 1980 (of which were discovered in Reisler
1225 et al.^[119] and documented in Table 2.1, The data points have been scaled to an equivalent
1226 hemispherical shape, if required, using a shape factor of 1.8^[76,77], and 1kg in mass to
1227 compare accordingly with KB predictions. Despite a low amount of anomalous results,
1228 the four positive phase parameters compare well with the KB curves and are fairly con-
1229 sistent across the entire tested scaled distance range. It is worth noting that Kingery and
1230 Bulmash^[87] made use of some of the presented data in order to produce the high order

1231 polynomial predictive curves, so whilst this comparison is self-referential, it does highlight
 1232 the general consistency of the measured blast parameters during this era across a full range
 1233 of charge masses and distances.

1234

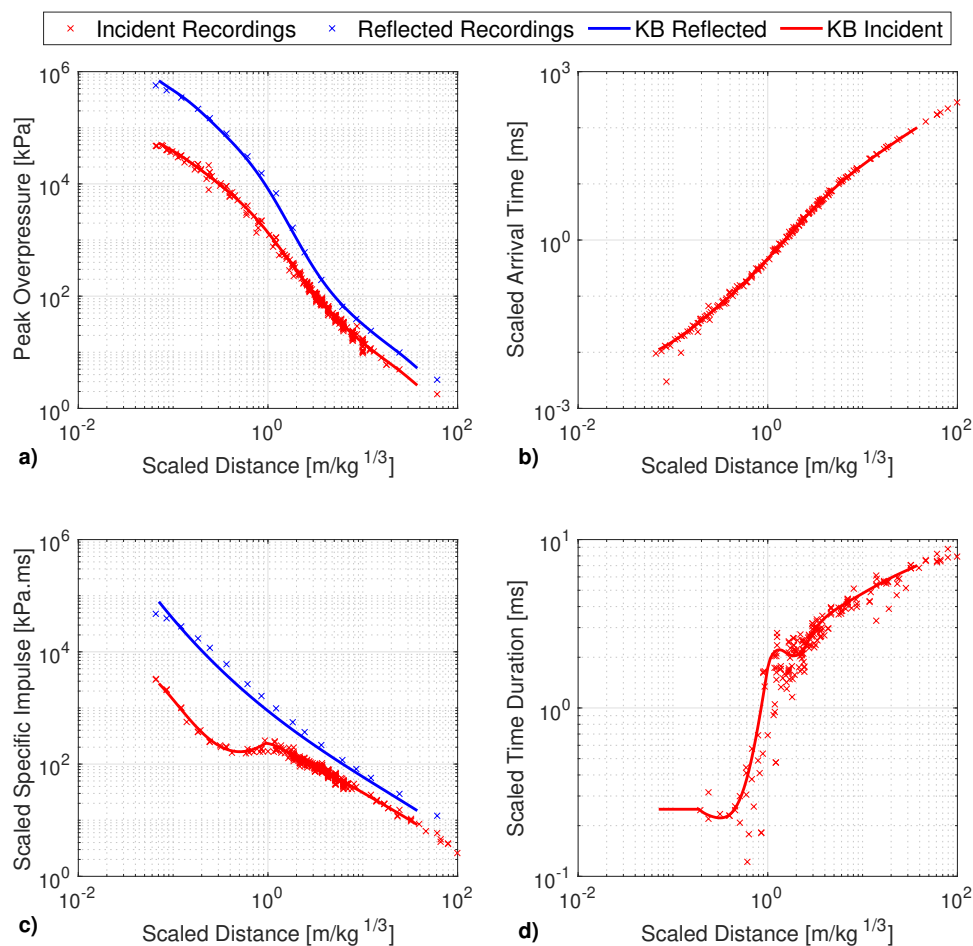


Figure 2.12: Compiled blast parameters from TNT explosive trials with varying mass as a function of scaled distance, which has been scaled to a 1kg equivalent hemispherical charge, using a shape scaling factor of 1.8 and compared with KB predictions: a) Peak overpressure, b) Scaled peak specific impulse, c) Scaled arrival time, d) Scaled positive phase duration

1235 Figures 2.13a-d again present experimentally recorded blast parameters when detonating
 1236 Pentolite - another commonly-used high explosive at the time. A variety of smaller scale
 1237 masses of between 0.1-3.8kg, of which a collection of spherical and hemispherical charges
 1238 were used. The raw data was scaled accordingly to be equivalent in shape and size to a
 1239 surface burst 1kg hemisphere of TNT. The data presented was extracted from 6 individual
 1240 references, detailed in Tables 2.1 scaled to an equivalent TNT explosive using a TNTe =

1241 1.2. This value is in line with averages for both pressure and specific impulse in far-field
 1242 scenarios evaluated by Shin et al.^[149].

1243

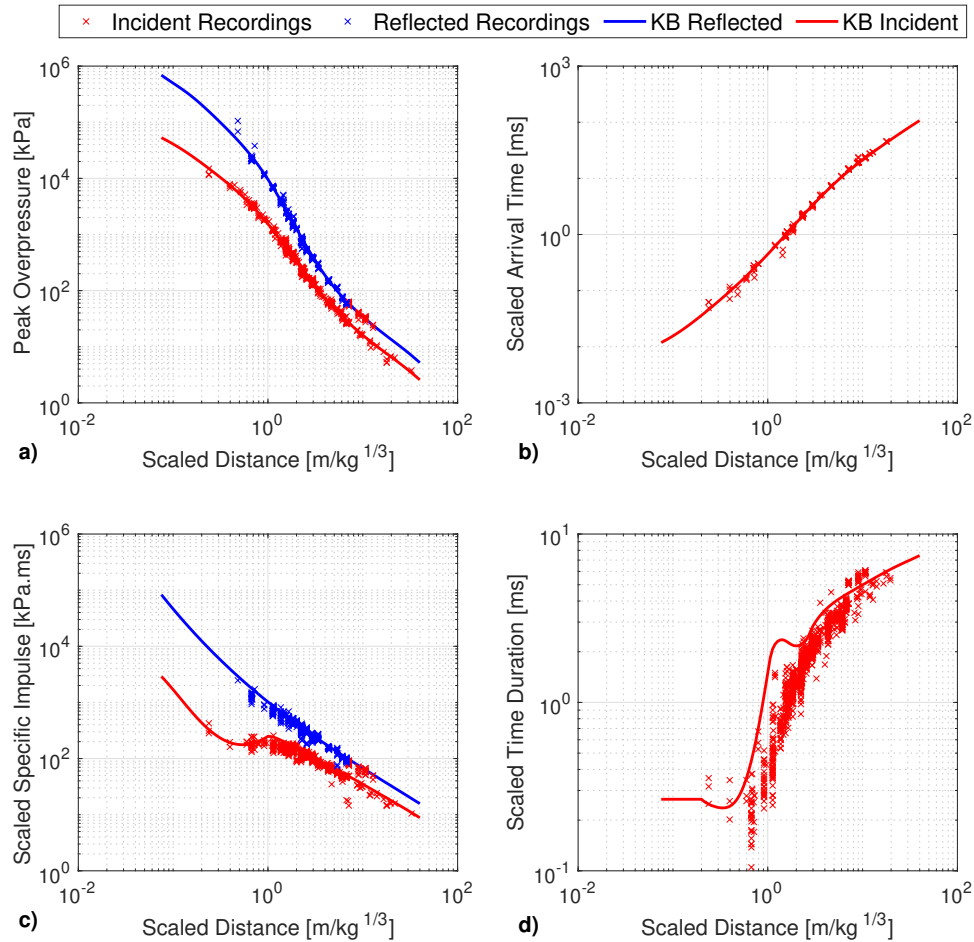


Figure 2.13: Compiled blast parameters from Pentolite explosive trials with varying mass as a function of scaled distance, which has been scaled to a 1kg equivalent TNT hemispherical charge using a shape scaling factor of 1.8, and a $TNTe = 1.2$ (Shin et al.^[149]), comparing to KB predictions: a) Peak overpressure, b) Scaled peak specific impulse, c) Scaled arrival time, d) Scaled positive phase duration

1244 The consistency in this data is much the same as that presented in Figures 2.12a-d and
 1245 holds good agreement with the KB predictions. These two extensive data sets provide rea-
 1246 sonable evidence to support the hypothesis that explosive positive phase parameters are
 1247 predictable for ideal high explosives across a range of charge sizes and scaled distances. It is
 1248 also important to note that, although not perfect for every single data point, a single TNTe
 1249 value can be used to establish a generalised equivalent behaviour of ideal explosives. This
 1250 does however begin to exhibit higher variations at small scaled distances, $Z < 1\text{m/kg}^{1/3}$.

1251 It is a reasonable conclusion to make that for far-field scenarios, $Z > 3\text{m}/\text{kg}^{1/3}$, where the
 1252 propagating shock wave has detached and is no longer effected by expanding fireball fea-
 1253 tures, the positive phase blast parameters are consistent and can be defined by a single
 1254 TNTe value.

1255

Reference	Explosive Type
Granstrom ^[66]	TNT
Ammann and Whitney ^[2]	TNT
Dobbs et al. ^[40]	TNT
Fisher and Pittman ^[54]	TNT
Shear and Wright ^[144]	TNT
Kingery and Pannill ^[89]	TNT
Groves ^[69]	TNT
Smale and Sigs ^[155]	TNT
Chabia et al. ^[25]	TNT
Shear and Day ^[143]	TNT
Kingery ^[86]	TNT
Reisler et al. ^[121]	TNT
Reisler et al. ^[122]	TNT
Davis et al. ^[35]	TNT
Swisdak ^[163]	TNT
Esparza ^[44]	TNT
Kingery and Coulter ^[88]	TNT
Rudlin ^[137]	TNT
Fisher ^[53]	TNT
Weibull ^[178]	TNT
Lutzky ^[95]	TNT
Reisler et al. ^[120]	TNT
Esparza and Moroney ^[46]	Pentolite
Johnson et al. ^[83]	Pentolite
Kingery and Coulter ^[88]	Pentolite
Sultanoff and McVey ^[162]	Pentolite
Goodman ^[64]	Pentolite
Goodman and Giglio-Tos ^[65]	Pentolite
Hoffman and Mills ^[73]	Pentolite

Table 2.1: Reference list for historic data extracted from within this era for both spherical and hemispherical TNT and Pentolite detonations.

1256 2.9.2 Data Capture between 1980 - Today

1257 The accuracy of the KB method has been questioned over more recent years. Paradox-
1258 ically, it would seem that with more modern and precise experimental procedures, the
1259 results should exhibit a reduction in uncertainty and an improvement on the precision.
1260 Many contemporary researchers have however reported a significant lack of repeatability
1261 in experimental measurements when comparing directly to KB parameters, and the much
1262 older data sets used to produce them.

1263

1264 Bogosian et al.^[13] utilised an extensive experimental database of recorded explosive pa-
1265 rameters and compared it with KB predictions of similar explosives. Typical uncertainties
1266 of between 70–150% and 50–130% for peak pressure and specific impulse respectively
1267 from nominally identical far-field ($1 < Z < 40\text{m/kg}^{1/3}$) tests were presented. Similar levels
1268 of uncertainty have been found in related studies, with the general observation being a
1269 reduction in uncertainty as distance from the charge increases^[16]. Formby and Whar-
1270 ton^[56] explored the TNT equivalence of a variety of hemispherical explosive compositions
1271 with the results demonstrating relatively high levels of variability: $\pm 30\%$ and $\pm 15\%$ for
1272 pressure and specific impulse respectively. These recordings were at large scaled distances,
1273 leading to a general impression that there will always be some degree of randomness in
1274 blast pressure measurements^[156].

1275

1276 In Figure 1 of the article presented by Formby and Wharton^[56], a site layout includ-
1277 ing pressure gauge position and height, in relation to explosive centre, was presented to
1278 highlight terrain features and large changes in surface height. The authors note that the
1279 presence of these features could influence the recorded data through inducing both wave
1280 reflection and/or shielding yet continue to suggest the yield of an explosive varies with
1281 scaled distance in a free-air scenario. Borenstein^[18] performed a sensitivity study of blast
1282 parameters taken from the 303 individual measurements discussed in Bogosian et al.^[13]
1283 and highlighted that these comprised of different explosive shapes, sizes and composition
1284 and were scaled directly to TNT as a collective. This resulted in quantification of a more
1285 extrinsic representation of blast parameter variability, which the authors seemingly incor-
1286 rectly attributed to the inherent randomness of explosives.

1287

1288 Stoner and Bleakney^[160] reported data recorded from free-air tests, using a variety of

1289 charge shapes, sizes and compositions. When separated into nominally identical test
1290 groups and analysed, the results presented much lower levels of variability in pressure,
1291 between ± 0.6 – 6.5% , again with the observation that variability reduced with an increase
1292 in scaled distance. Esparza^[44, 45] presented blast parameter results from a variety of high
1293 explosive compositions. Despite not quantifying variability in the recordings, visually the
1294 data holds agreement with itself for nominally identical tests.

1295

1296 Tang et al.^[164] also undertook a variety of incident and reflected measurements from a
1297 range of PE4 masses, formed into spherical and hemispherical charges, resulting in similar
1298 magnitudes of variability quoted by Stoner and Bleakney^[160]. This experimental data was
1299 compared against both KB predictions and hydrocode numerical simulator, Air3D, which
1300 presented reasonable levels of agreement for medium to large scale charges^[165]. Reflected
1301 data agreed significantly better to KB predictions than incident pressures justified to be a
1302 result of the difficulty in recording purely incident shock waves. Rickman and Murrell^[125]
1303 and Tyas et al.^[175] conducted well controlled small-scale experimental high explosive test-
1304 ing using pressure transducers which provided comparable results to KB predictions for
1305 normally reflected conditions at far-field scaled distances.

1306

1307 Ohashi et al.^[108] made use of optical methods to record shock wave propagation of small-
1308 scale explosive tests with varying masses. The article presented a method of converting
1309 radius-time data into incident pressure of a given shock wave using Rankine-Hugoniot
1310 jump conditions. The results of this analysis, when scaled, provided remarkably low vari-
1311 ability between tests and also compare well with KB predictions.

1312

1313 The articles mentioned show the other contrasting opinion of the blast industry in that
1314 KB predictions are incredibly accurate, and that variability between nominally identical
1315 tests should be minimal. Rigby et al.^[135] gave an in-depth literature review of articles
1316 which discuss experimental variability of far-field blast parameters and how they compare
1317 with the KB predictions. Systematic experimental and analytical errors are suggested to
1318 be the reason as to why such high levels of variability have been documented, rather than
1319 inherent randomness of explosive events. In an attempt to tackle systematic analytical
1320 variability, the authors used an exponential ‘Friedlander’ curve fitting method to deter-
1321 mine blast parameters from each experimental trial in an unbiased manner.

1322

1323 The results showed very good test-to-test consistency across the measured blast param-
1324 eters (arrival time, reflected peak overpressure, and reflected specific impulse) typically
1325 within a range of $\pm 2.5\%$ (arrival time) and $\pm 6\text{--}8\%$ (pressure and impulse) of the mean
1326 values for each set of repeat tests. The recorded positive duration was the only parameter
1327 to exhibit higher levels of variability with all but one value achieving a $\pm 9\%$ range of
1328 the mean value. Positive phase duration typically exhibits higher levels of experimental
1329 spread due to the difficulty in determining when overpressure returns to zero when signal
1330 noise is present^[96]. No signal will ever be perfectly noiseless, and therefore the positive
1331 phase duration will always carry an enhanced level of uncertainty. Errors or uncertainties
1332 in this parameter contribute very little to the overall loading, since a curve fit can always
1333 be tailored with a different decay parameter in order to match a prescribed specific im-
1334 pulse value. Hence, studying sources of uncertainty in positive phase duration are of lesser
1335 importance and will not be considered further.

1336

1337 Chiquito et al.^[27] and Bogosian et al.^[14] both used similar methods of functional fitting
1338 curves to experimental data, and both presented results exhibiting enhanced consistency
1339 and reduced subjectivity of analysed blast data. The analytical techniques proposed ac-
1340 count for errors during the post processing section of the testing but not account for any
1341 systematic errors which may have been introduced into the experiments during test setup.

1342

1343 The aforementioned articles and resulting data presents a clear divide in the blast com-
1344 munity in whether it is reasonable or not to accept free air far-field blast parameters as
1345 inherently variable. As stated by Borenstein^[18], there are clear reasons as to why extrinsic
1346 and intrinsic sources of variability should be considered:

- 1347 1. The generalised real-world application of predicting explosive parameters which in-
1348 clude safety factors and variability margins accounting for the unpredictability of
1349 explosive size, shape, composition and separation distance from a target (extrinsic).
- 1350 2. The specific, scientific approach of assigning precise loading characteristics for a
1351 particular charge shape, size and composition and looking at how removing those as
1352 independent variables results in increased consistency (intrinsic).

1353

1354 This provides an important, but seemingly overlooked steer to the blast research commu-
1355 nity: in order to produce robust and reliable models that account for *extrinsic* variations in

1356 blast properties, it is imperative to understand and be able to quantify *intrinsic* sources of
1357 variability. This is where well-controlled scientific testing and processing can make signif-
1358 icant contributions to our understanding of, and ability to predict, blast load parameters
1359 from *known* explosive sources.

1360

1361 2.10 Non-Ideal Explosive Characterisation

1362 When reviewing the last 50 years of both accidental and intentional terror related large
1363 scale explosive events, presented in Table 2.2, there has been an large number of lives
1364 lost, people exhibiting significant injury and money spent to remediate the destructive
1365 effects. Whilst the information presented in Table 2.2 does not cover every instance of
1366 explosive event across the world, it provides a range of masses and types of explosives
1367 causing catastrophic effects.

1368

1369 Non-ideal explosives are common across the vast majority of events listed. Home-made
1370 compositions, such as ANFO, are the weapon of choice for most terror related threats
1371 due to the relative ease and low costs associated with acquiring their constituents. The
1372 accidental explosions noted in Table 2.2 were a direct result of stockpiling large quanti-
1373 ties of material which does foster the ability to detonate under extreme circumstances of
1374 pressure and temperature. Take the most recent example which discusses the explosion
1375 of approximately 2750 tonnes of AN following a fire in the warehouse where it was being
1376 stored^[131].

1377

1378 In 1963, the Nuclear Test Ban Treaty was enacted, which prohibited the detonation of nu-
1379 clear weapons for any form of research. This caused problems for departments of defence
1380 and security across the globe due to an essential requirement to better understand the
1381 effects of devastating explosions on both military and civilian infrastructure. Large scale
1382 trials were conducted using TNT-stacked charges and provided data which is held in high
1383 regard for its accuracy (see Chapter 4), and was used to develop fast-running empirical
1384 models still used today^[87]. This data was the pinnacle of blast data capture for its time
1385 and has provided insights into the fundamental mechanisms and characteristics of explo-
1386 sive shock waves. Through vast quantities of research making efforts towards quantifying
1387 and characterising other ideal high explosives (such as Pentolite, PE4, PE8, PE10 etc)

1388 across an entire range of scaled distances, numerical simulations have been extensively
1389 validated.

1390

1391 The problem with testing regimes using ideal explosives is the cost associated with the
1392 large-scale trials and so an inexpensive and readily-available alternative in ANFO, was
1393 investigated to compare directly with tests conducted using TNT, and nuclear weapons
1394 prior^[112]. Large scale trials were conducted from the 1970s onwards, making use of hemi-
1395 spherical charges with masses ranging between 18,000-572,000kg, boosted with various
1396 compositions of explosives at such small percentages of the overall mass they were con-
1397 sidered negligible^[61,62,112,138,139]. The findings from these tests began to characterise the
1398 general behaviour of ANFO as an explosive used alternatively and led to an empirical
1399 $TNTe=0.82$ ^[112].

1400

1401 As previously mentioned, an ideal detonation is one in which its temporal characteristics
1402 of velocity, time and pressure match those of which can be predicted through theoretical
1403 evaluation. Assuming a planar detonation in an infinite diameter charge, the detonation
1404 process would be independent of reaction zone size, in which case the detonation wave
1405 would propagate at a maximum theoretical value within the charge Horie^[75]. In reality,
1406 charges with a finite diameter are dependent on reaction zone, of which large ones result in
1407 larger losses of energy and thus detonation performance. For every explosive composition
1408 there is a thermodynamic break point at which the diameter is so small detonation cannot
1409 be sustained, denoted by the critical diameter C_d . Ideal explosives are described to exhibit
1410 small reaction zones and thus compare well to theoretical predictions of temporal char-
1411 acteristics. Non-ideal explosives, such as ANFO, are known to have much larger reaction
1412 zones - in the order of 100mm^[29]. Scott^[142] suggested that explosive compositions with
1413 large critical diameters as a result of large reaction zones, require much greater masses to
1414 achieve steady detonation which compare to theoretical values. Despite this knowledge,
1415 the findings from the large scale testing of ANFO when collated by Petes et al.^[112] in-
1416 fluenced the idea that the non-ideality of ANFO was irrelevant as a seemingly constant
1417 $TNTe$ was assigned to the composition.

1418

1419 Figure 2.14 contains a compilation of data from 7 different testing regimes (see Table 2.3),
1420 6 of which are large scale hemispherical and spherical ANFO trials in free field scenar-
1421 ios and the2 containing smaller mass regimes. It is important to note that whilst the

Year	Event	Explosive	Estimated Mass [kg]	Effects
1970	Sterling Hall Bombing (Madison, USA)	ANFO	~100	57 I, \$13.6million
1983	Beirut Barracks Bombing (Beirut, Lebanon)	PETN	~9000	307 D, 150 I
1983	US Embassy Bombings (Beirut, Lebanon)	ANFO	~900	63 D, 120 I
1992	St. Mary Axe Bombing (London, United Kingdom)	Sentex	~45	3 D, 91 I
1993	World Trade Center Bombing (New York, USA)	Urea Nitrate	~600	6 D, 1042 I, \$65 million
1993	Bishopsgate Bombing (London, United Kingdom)	AN/Sugar	1000-2000	1 D, 44 I
1995	Oklahoma City Bombing (Oklahoma City, USA)	AN/NM	~2500	168 D, 680 I, \$652 million
1996	Manchester Shopping Mall (Manchester, United Kingdom)	AN/Sugar	~1500	212 I
1996	South Quay bombing (London, United Kingdom)	AN/Sugar	~1500	1 D, 44 I
1996	Khobar Towers Bombing (Khobar, Saudi Arabia)	C4	~2300	19 D, 498 I
1998	US Embassy Bombings (Tanzania, Kenya)	Equivalent TNT	~900	224 D, Over 4000 I
2000	USS Cole Bombing (Aden, Yemen)	Equivalent TNT	~450	17 D, 40 I
2000	Enschede fireworks disaster (Enschede, Netherlands)	Equivalent TNT	~4000	23 D, 1000 I
2002	Bali Nightclub Bombing (Bali, Indonesia)	KClO ₃ /S/Al	~1000	204 D, 209 I
2003	Marriott Hotel Jakarta Bombing (Jakarta, Indonesia)	KClO ₃ /S/Al	~50	12 D, 150 I
2003	British Consulate Bombing (Istanbul, Turkey)	AN/Al	500-1000	59 D, 750 I
2003	Casablanca Bombings (Casablanca, Morocco)	TATP/AN	5-10	45 D, 100 I
2004	Australian Embassy Attack (Jakarta, Indonesia)	KClO ₃ /S/Al	500-1000	9 D, 150 I
2004	US Embassy Attack (Tashkent, Uzbekistan)	AN/Al	5-10	2 D, 9 I
2004	Madrid Train Bombings (Madrid, Spain)	Dynamite	5-10	193 D, 2050 I
2005	7/7 Underground Bombing (London, United Kingdom)	HP/BP	5-10	52 D, 770 I
2008	US Embassy Attack (Sana'a, Yemen)	Equivalent TNT	~50	18 D, 16 I
2011	Oslo Bombing (Oslo, Norway)	ANFO/Al	~1000	77 D, 319 I
2013	Boston Marathon Bombings (Boston, USA)	Equivalent TNT	5-10	3 D, 280 I
2013	West Fertilizer Company explosion (Texas, USA)	AN	7500-10,000	15 D, 160 I
2015	Paris Attacks (Paris, France)	TATP	5-10	137 deaths, 416 injured
2015	Tianjin explosions (Tianjin, China)	AN	~800,000	181 D
2016	Brussels Attacks (Brussels, Belgium)	TATP	15-20	35 D, 340 I
2017	Concert Bombing (Manchester, United Kingdom)	TATP	Unspecified	23 D, 107 D, £13 million
2020	Beirut explosion (Beirut, Lebanon)	AN	~2,750,00	Over 200 D, Over 6500 I, \$10-15 billion

Table 2.2: Historical explosive events across the world noting the explosives which detonated and the overall effects associated adapted from National Academies of Sciences and Medicine [105]. The events noted in bold are accidental whilst the rest are terror events with D and I corresponding the Deaths and Injuries respectively.

1422 data extracted from Carton^[22] is presented, the charge shape used was flat and square,
 1423 detonated in one corner rather than centrally which introduces complexity. The lower
 1424 values of TNTe extracted have been directly related to irregularities in the detonation
 1425 mechanics. This occurred due to the variety of geometrical differences this methodology
 1426 included when comparing to results from a centrally detonated spherical or hemispherical
 1427 charge. Sherkar et al.^[146] explained that charge shape needs to be taken into consideration
 1428 when regarding resulting blast parameters, provided standoff distance from the detonation
 1429 point is less than 35 charge diameters. Past this standoff the effect of charge shape can
 1430 be ignored. The lower two values of TNTe extracted from Carton^[22] were measured at
 1431 much closer standoff distances than the quoted 35 charge diameter. They were therefore
 1432 omitted from any further analysis to avoid any reasonable doubt relating to charge shape
 1433 and detonator position.

1434

Reference	Explosive Type
Sadwin and Swisdak ^[139]	ANFO
Carton ^[22]	ANFO
Gitterman and Hofstetter ^[62]	ANFO
Giglio-Tos and Reisler ^[61]	ANFO
Petes et al. ^[112]	ANFO
Figuli et al. ^[52]	ANFO
Edwards ^[42]	ANFO

Table 2.3: Reference list for historic data extracted for both spherical and hemispherical ANFO detonations.

1435 Clearly shown from the Petes et al.^[112] data alone is that once above a mass of around
 1436 120kg, ANFO behaves with a seemingly constant TNTe=0.82. This finding was related to
 1437 a stable detonation and having an average detonation velocity of around 4200m/s (~80%
 1438 of that of TNT). For charges smaller than 120kg, the results suggested that the charge
 1439 diameter was not large enough to permit steady state detonation conditions, thus re-
 1440 sulting in lower equivalence values. This information seems to have been lost in time,
 1441 with more modern approaches towards quantifying the effects of smaller-scale ANFO both
 1442 experimentally^[22,27,52] and numerically^[67,68,157] making reference to equivalency features
 1443 relating to larger scaled charges. Instead an approach which determines a non-steady det-
 1444 onation state of ANFO and the resulting blast parameters for smaller charges is necessary.

1445

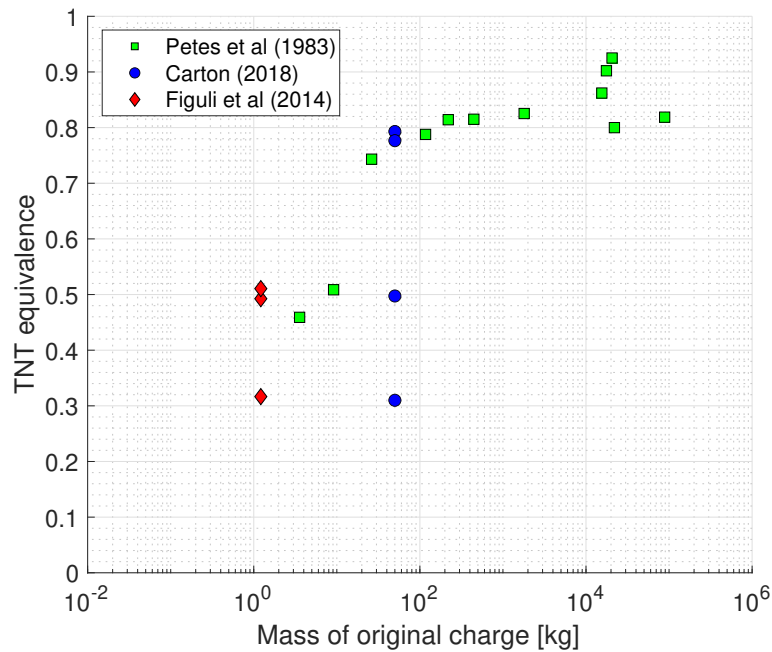


Figure 2.14: Pressure evaluated TNT equivalence ratio for unconfined ANFO with respect to the mass of the charge extracted from Petes et al.^[112], Figuli et al.^[52] and Carton^[22]

1446 The data from the aforementioned reports of large scale trials has been compiled for the
 1447 purpose of validating the assumption that constant TNTe across all scaled distances and
 1448 masses of charge was made shown in Figures 2.15a-d. The raw data points have been
 1449 scaled to an equivalent hemispherical shape, if required, using a shape factor of 1.8^[76,77],
 1450 and 1kg in mass of TNT using a TNTe=0.82, as defined by Petes et al.^[112], to compare
 1451 accordingly with KB predictions. It is clear to see the data presented across all positive
 1452 phase parameters agree remarkably well with the aforementioned KB predictions when
 1453 scaled appropriately. This finding suggests that whilst non-ideal explosives are considered
 1454 to have different time scales associated with their detonation chemistry, that once a certain
 1455 scale is reached they behave like an ideal explosive with consistent comparisons to TNT.
 1456 This provides validation to^[167] who suggest the theory that all explosive detonations can
 1457 be considered as point source energy releases, with localised variations, reaching an equi-
 1458 librium at a given distance away from the explosive centre.

1459

1460 There is a reasonable increase in the variability of data at smaller scaled distances, simi-
 1461 larly to the ideal explosives in Figures 2.12a-d and Figures 2.13a-d. One safe assumption
 1462 for this would be relating this directly to the variations in booster composition and size
 1463 creating highly variable chemistry within the regions in which the fireball still has an effect

1464 on the resulting shock wave properties.

1465

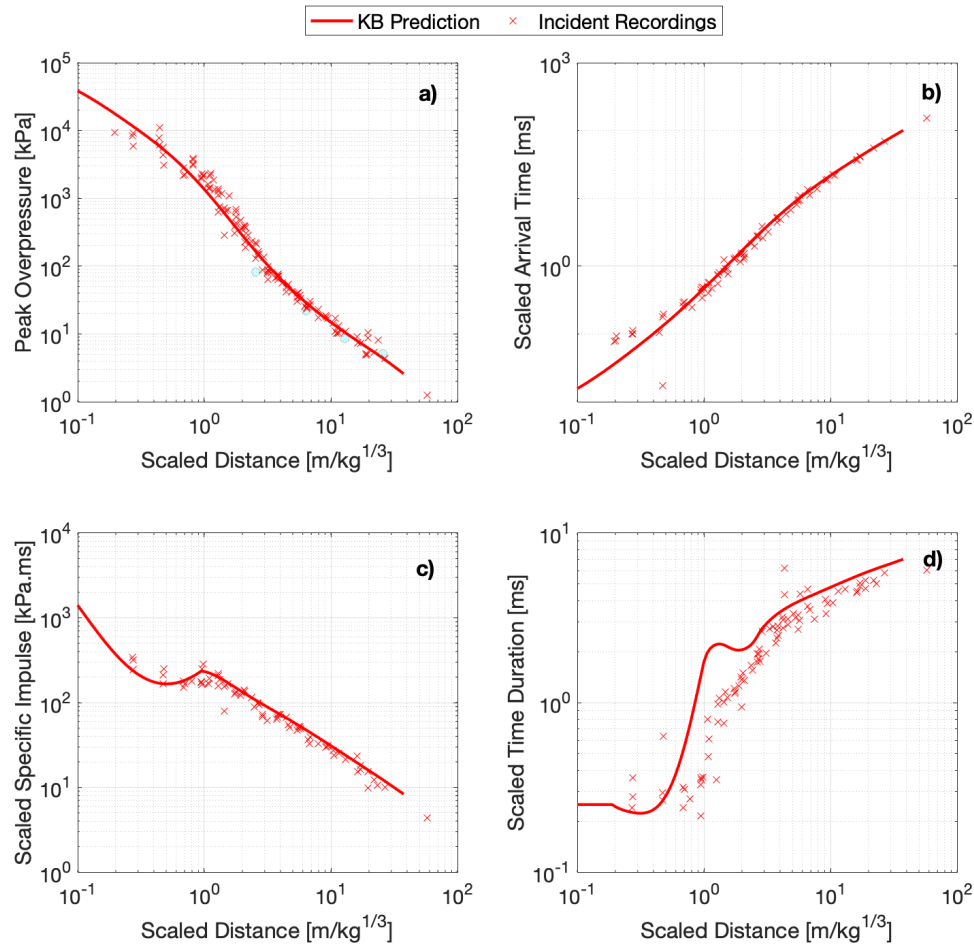


Figure 2.15: Compiled blast parameters from ANFO explosive trials with varying mass as a function of scaled distance, which has been scaled to a 1kg equivalent hemispherical charge, using a shape scaling factor of 1.8, a $TNT_e=0.82$ ^[112], and compared with KB predictions: a) Peak overpressure, b) Scaled arrival time, c) Scaled peak specific impulse, d) Scaled positive phase duration

1466 Another justification could be related to the inconsistencies in ANFO used for the trials,
 1467 again referring back to the discussion of introducing systematic errors into a testing regime
 1468 in Chapter 3. ANFO comes in prill form, with a variety of sizes and fuel oil absorption
 1469 variations. Without careful consideration of these parameters, explosive yields are specu-
 1470 lated to vary. Considering the scale of the historically tested charges, the prill volume is
 1471 assumed to be generally homogeneous exhibiting consistent characteristics, however when
 1472 taking small samples, the composition would be heterogeneous.

1473

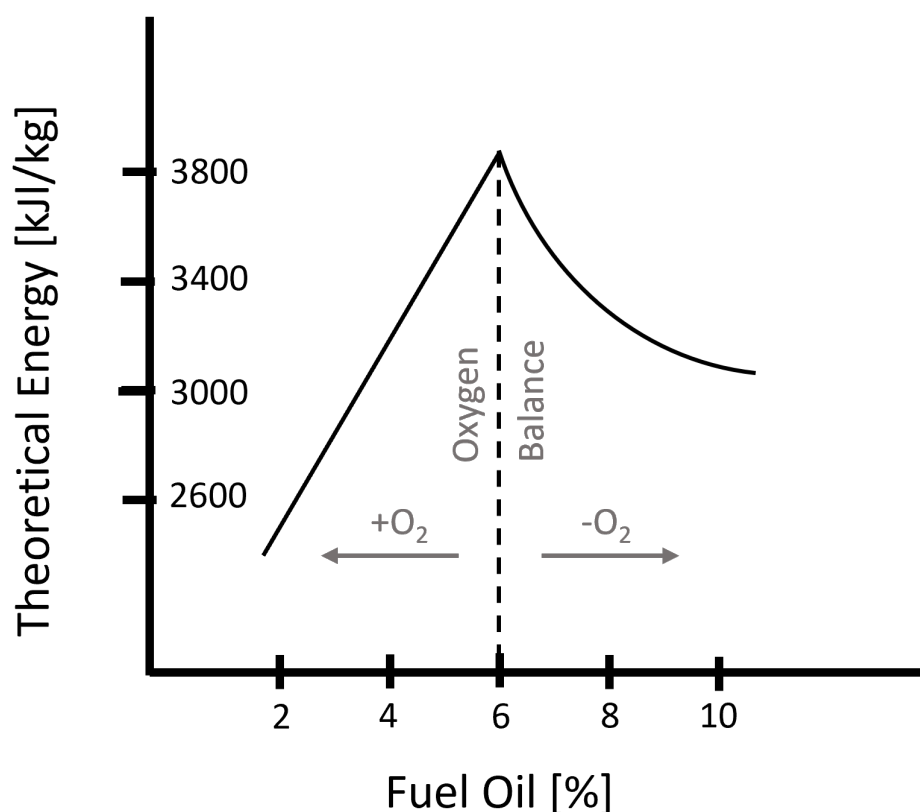


Figure 2.16: The theoretical energy of ANFO as a function of fuel oil content, adapted from Petes et al.^[112]

1474 Figure 2.16 represents the theoretical energy output and sensitivity of ANFO as a func-
1475 tion of the fuel oil content. As detailed by Petes et al.^[112], at 6% FO the composition
1476 is stoichiometric, meaning full potential energy release from both the AN and the FO is
1477 attained during the detonation process. From a thermodynamic standpoint, both oxy-
1478 gen rich and deficient compositions result in a reduction in the total detonation energy
1479 available, and therefore the percentage of FO has a major influence on the overall energy
1480 released. Whilst strict regimes of quality control may have been implemented, ensuring an
1481 accurate percentage of FO throughout the entire composition of the charges across each
1482 test would be difficult. With large scale charges, self-compaction and void collapse could
1483 result in displacement of FO within regions of the charge and therefore variations could
1484 be experienced.

1485

1486 Similar can be said for the prill size distribution. Dobrilovic et al.^[41], McKay et al.^[99], Pe-
1487 tes et al.^[112], Salyer et al.^[140] all investigated varying the densities of ANFO charges to

1488 quantify how it affected velocity of detonation and other blast parameters. The findings
1489 suggested that generally larger prills are harder to detonate than smaller prills, and that
1490 a steady state detonation is only consistently achieved at bulk density of between 0.80-
1491 $0.85g/cm^3$). This means densities either side of this region would result in variations in
1492 detonation affectivity and thus the resulting blast parameters. In the larger scale charges,
1493 ensuring a constant bulk density throughout a charge would be incredibly difficult due to
1494 hydrostatic compaction of the lower layers of ANFO.

1495

1496 A further difference between ANFO and ideal explosives is the lack of binder, and there-
1497 fore the overall density, and consequential air voids in a ANFO charge, are thought to be
1498 an influence on yields through meso-structural detonation mechanics. With large scale
1499 charges, self compaction of the charge would result in a more homogenous material and
1500 therefore a more efficient detonation process, comparing closely to ideal explosives. Smaller
1501 scale charges could begin to provide evidence to help the fundamental understanding of
1502 increased variability in ANFO compared to ideal explosives.

1503

1504 Despite ANFO being used both commercially in mining/quarrying processes, and a com-
1505 mon explosive for terror threats, there seems to be little available published data char-
1506 acterising the explosive across a range of scaled distances for small charge sizes. Figuli
1507 et al.^[52] undertook free-field ANFO explosive testing using 1kg spherical charges, which
1508 showed yields equating to a TNTe 0.32-0.51, different to figures used within numerical
1509 modelling strategies suggested by Sochet et al.^[157] and Grisaro and Edri^[67]. The lack of
1510 data results in uncertainties on how best to model and predict the blast wave properties
1511 emerging from detonations of small ANFO charges. Specifically, it cannot be definitively
1512 concluded if the TNTe value for ANFO is consistent across scales, or whether the reaction
1513 mechanics are affected by the physical length scale over which the detonation wave transits.

1514

1515 **2.11 Summary**

1516 Whilst explosive characterisation has been investigated heavily over the last few decades,
1517 there is still no definitive conclusion on whether or not the associated parameters are de-
1518 terministic. There requires a fundamental and somewhat crude review of experimental
1519 data capture and analysis methods to understand why there are discrepancies between

1520 researchers and to conclude on inherent variability of a given explosive. Standardised
1521 techniques are required to investigate the effects of different compositions with confidence.
1522 Whilst ever there are differences in experimental control, there is an expectation of variable
1523 data. This thesis aims at developing validated methods of characterisation for simplistic
1524 blast scenarios as a result of any given composition detonation. A comparison will be
1525 made between ideal and non-ideal explosives, which exhibit detonation behaviours on sig-
1526 nificantly different time scales.

1527

1528 Chapter 3

1529 Standardising Experimental 1530 Methodologies and Analysis

1531 3.1 Development of Experimental and Analytical Procedures

1532 This chapter will detail how the experimental methodology used by the University of
1533 Sheffield has been modified and improved through the careful consideration of the control
1534 measures in place for each test and the results gathered from them. The improvements to
1535 the testing regime have been compared to historic data and numerical simulations to both
1536 co-validate the precision of experimental testing and the value of established numerical
1537 models. The aim of this chapter is to confirm the underpinning consistency of shock wave
1538 parameters through developing experimental and analytical techniques of data capture
1539 and processing respectively.

1540

1541 3.1.1 Preliminary Test Methodology

1542 The far-field arena-style tests undertaken at the University of Sheffield (UoS) Blast and
1543 Impact Laboratory in Buxton, UK, has been a long established capability resulting in a
1544 number of blast quantification research articles^[49,128,129,132–134,175]. Figure 3.1 displays
1545 the general arrangement that has been adopted for the far-field blast trials undertaken at
1546 Buxton, which makes use of two rigid reflective surfaces in the form of a reinforced con-
1547 crete bunker 4m in height and a blockwork wall 2.20m in height, 4.46m in width, separated
1548 exactly 10.0 m apart. For each of the trials, hemispherical explosive charges, were formed

1549 using a 3D-printed mould (see Figure 3.2), and detonated at varying stand-off distances
1550 between the two walls.

1551



Figure 3.1: General arrangement of the test pad at the University of Sheffield Blast and Impact Laboratory: Site photograph taken approximately from the location of the high speed video camera

1552 Kulite HKM-375 piezo-resistive pressure gauges were used to record the reflected pressure-
1553 time histories in each test at both locations. The gauges were threaded through, and made
1554 flush to the surface of a small steel plate (approximately $110 \times 150 \times 10$ mm) which was
1555 fixed to these walls. The plates were fixed to the two reflective surfaces, achieving a
1556 10mm height from the centre of gauge to the ground surface level to ensure pressures
1557 recorded were normal to the charge. The charges were placed on a small steel plate
1558 ($150 \times 150 \times 25$ mm) prior to detonation, in order to avoid repeated damage to the con-
1559 crete testing pad. The pressure was recorded using a 16-bit digital oscilloscope and TiePie
1560 software, with a average sampling rate of 195 kHz at 16-bit resolution. The recording was
1561 triggered automatically using TiePie's 'out window' signal trigger on a bespoke break-wire
1562 signal, formed by a wire wrapped around the detonator. The 'out window' trigger initi-
1563 ated with a voltage drop outside the normal electrical noise experienced in the break-wire.
1564 This coincides with the detonation of the charge breaking the circuit.

1565



Figure 3.2: Photographs of the moulding stages of a 250g PE10 hemispherical charge in which the 3D printed mould is used to provide consistency in charge shape and density, which can be seen when removed from the casing.

1566 High speed video (HSV) recordings of the propagating shock front were taken in an attempt
1567 to quantify the variability of shock wave arrival time recordings. As seen in Figure 3.3,
1568 the explosive charges were situated between a Photron FASTCAM SA-Z high speed video
1569 camera, fitted with a Tamron SP AF70 70-200 mm zoom (F2.8) lens, and a zebra board ran
1570 perpendicular to the blockwork and bunker walls. The camera was positioned to record
1571 the shock wave as it propagated towards the blockwork wall and had the charge location
1572 out of the camera's field of view to avoid excessive light from the fireball corrupting the
1573 contrast of the images. The camera was placed inside an armoured protective housing to
1574 avoid any fragment damage during testing.

1575

1576 A long-span zebra board served to provide a high contrast background such that distor-
1577 tions in light, caused by changes in refractive index of the propagating shock front, would
1578 feature as sharp dark bands in the HSV recordings. This allows the position of the shock
1579 front to be identified in each frame, and therefore enables wider-field arrival time measure-
1580 ments from a given trial^[78]. The frame rates of the recordings and exposure time varied
1581 between 10,000–36,000 /s and 9.4–26.2 μ s due to varying weather/light conditions. Frame
1582 rates and exposure times were altered before to each test in order to maximise the contrast
1583 of the zebra board stripes, and the sharpness of the propagating shock front. Through
1584 trial and error it was found that a maximum of 20 μ s exposure time was allowed to achieve
1585 reliable data which exhibited minimal motion blur. If the lighting conditions meant that
1586 the video was not visible at this shutter speed or faster, then the data would be unreliable

1587 and therefore discarded from analysis. The resolution of the camera, when set to these
 1588 parameters, was 1024 x 512 which resulted in an average pixel size of around 50mm for the
 1589 far-field set up adopted. This resulted in a best possible accuracy of $Z=\pm 0.075\text{m}/\text{kg}^{1/3}$
 1590 per pixel.

1591

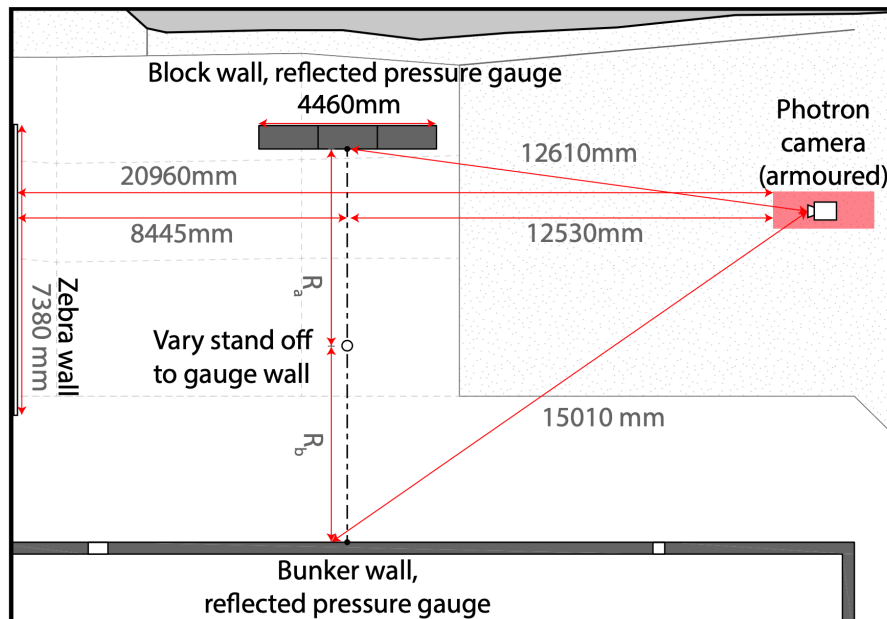


Figure 3.3: General arrangement of the test pad at the University of Sheffield Blast and Impact Laboratory: Schematic including HSV camera positioning

1592 The experimental work undertaken aims at achieving the highest precision during trial
 1593 setup which includes measurements of distance to a target, charge mass and shape consistency.
 1594 Despite being well regarded, the adopted methods still incorporated both systematic
 1595 experimental errors which had room for improvement alongside research dependant bias
 1596 into the assignment of blast parameters for a given trial. It was important to establish
 1597 methods for eliminating or minimising these errors as best as possible both analytically
 1598 and experimentally. The implementation of new experimental capabilities and methods of
 1599 analysis are discussed. Within the remainder of this chapter, analytical and experimental
 1600 improvements will be denoted by **Ai** and **Ei** respectively.

1601

1602 **3.2 Pressure Gauge Data**

1603 Detailed within this section is the discussion of experimental and analytical improve-
1604 ments made to testing regimes to ensure high fidelity data capture of free-air reflected
1605 blast parameters. The content outlines known systematic features within the adopted
1606 methodologies and aims to use fundamental knowledge of shock wave propagation, and
1607 its interaction with structures, to omit any errors and establish highly consistent data sets.

1608

1609 **3.2.1 Analysis of Positive Phase (Ai)**

1610 The raw pressure-time histories from far-field shots, recorded at the University of Sheffield,
1611 have tended to exhibit large uncertainties in the opening few microseconds after the arrival
1612 of a shock. This has been associated with sensor ringing and smaller variations in natural
1613 electrical noise from the recording apparatus, as seen in Figure 3.4^[135]. Precise determina-
1614 tion of key blast wave parameters is therefore challenging and alludes to an emphasis on the
1615 importance of functional signal analytics for improved blast parameter characterisation^[14].

1616

1617 The raw trend seen in Figure 3.4 shows reasonable levels of consistency to KB predic-
1618 tions, which initially provides justification to the quality of the semi-empirical tool for
1619 far-field shots. There are still some parameters which do not compare quite as accurately.
1620 Background electrical noise precludes accurate and automatic determination of the posi-
1621 tive phase duration, t_d . This is the time it takes for the overpressure to return back to
1622 atmospheric conditions. This feature is due to a recorded signal oscillating around the
1623 origin for a non-zero period of time. For the data presented in Figure 3.4, the end of the
1624 positive phase would be prescribed as between $12.6 - 12.7ms$ and the positive phase time
1625 duration would be prescribed as $t_d = 2.7 - 2.9ms$ depending on where the analyst defines
1626 the parameter.

1627

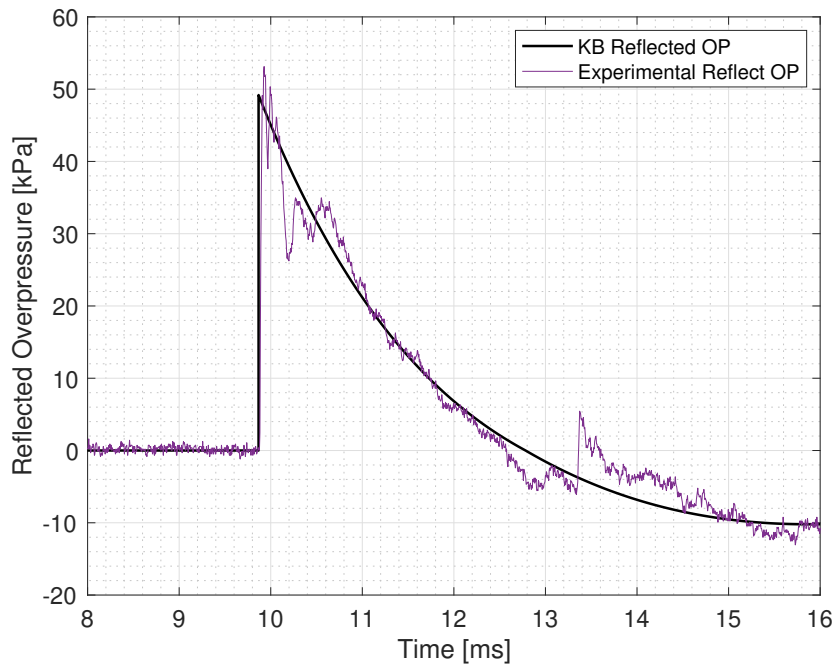


Figure 3.4: Example historic test results using a 250g PE4 hemispherical charge reflected pressure recording at 5 m stand-off compared to KB predictions for a 300g TNT hemisphere

1628 Prior to any analysis or scaling being applied to the data, it is important to establish
 1629 how nominally identical raw recordings compare. Figure 3.5 displays a compilation of
 1630 as-recorded positive phase pressure-time history profiles for 250g hemispherical PE4 det-
 1631 onations at various stand-off distances. Since pressure was recorded at two stand-off
 1632 distances for each tests (2m to the bunker wall and 8m to the blockwork wall in test 1, 3m
 1633 and 7m respectively in test 2, etc.), the results from different tests but identical stand-off
 1634 distances can be compared. Qualitatively, each pair of results are in excellent agreement,
 1635 with minimal variations in the blast pressure histories. The raw peak pressures at 2m
 1636 stand-off ($Z \sim 3 \text{ m/kg}^{1/3}$) exhibits a higher level of variability when compared to the other
 1637 pairings. This is in agreement with the working hypothesis of enhanced variability in the
 1638 regions described by Tyas^[173], hypothesised to be due to Rayleigh-Taylor (1882, 1950d)
 1639 and Richmyer-Meshkov (1969, 1960) surface instabilities.

1640

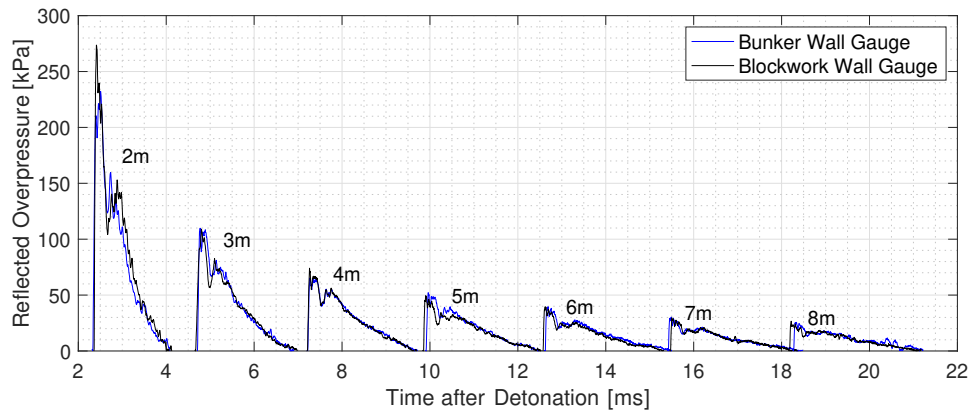


Figure 3.5: Compilation of the entire raw data set of 250 g PE4 hemispherical ground burst comprising of two recordings at each scaled distance. Positive phase only

1641 It has previously been shown that the specific impulse of a pressure signal can vary significantly
 1642 depending on which value of positive phase duration is taken^[96]. This alongside the
 1643 significant drop in the opening few milliseconds of the positive phase, can cause specific
 1644 impulse values to be significantly lower than of a perfect reflection recording (as repre-
 1645 sented by the KB prediction in Figure 3.4). A solution to overcome these errors, discussed
 1646 within literature^[14,49,135] is to fit a modified Friedlander curve, defined by Equation (2.1),
 1647 to the experimentally recorded data. No signal will ever be perfectly noiseless, and there-
 1648 fore the positive phase duration will always carry an enhanced level of uncertainty. Errors
 1649 or uncertainties in this parameter can contribute to the overall loading. However, a curve
 1650 fit can always be tailored with a different decay parameter in order to match a prescribed
 1651 specific impulse value. Hence, studying sources of uncertainty in positive phase duration
 1652 are of lesser importance than other positive phase parameters but it is still a reasonable
 1653 indicator of explosive yield.

1654 The aforementioned technique was examined through automatically fitting a modified
 1655 Friedlander curve to individual pressure signals. This is done to determine how best to
 1656 quantify variability of the pressure signal without introducing user error or losing physical
 1657 meaning of the signal. The aim of adopting an automated approach was to provide a
 1658 more reliable benchmark of explosive characterisation in far-field regions. The aim of this
 1659 was to achieve variabilities associated with the positive phase blast parameters that were
 1660 statistically unbiased. The automated method was initially validated using 65 of the 144
 1661 historic PE4 pressure-time histories recorded at the University of Sheffield^[128], and was
 1662 then used to process the remaining PE4 shots to assess its flexibility.

1663

1664 3.2.1.1 Determination of Best-Fitting Procedures

1665 Considering equation (2.1), the parameters required for the curve fitting approach are arrival time t_a , and time duration of the positive phase t_d . Through the use of MatLab's
1666 in-built curve fitting toolbox the max overpressure, P_{max} , and decay parameter, b , can be
1667 established using these known parameters. The process was coded to run through multiple
1668 variations of curves fitted to a given percentage of the recorded data after the arrival
1669 time was ignored. A curve was fitted to remaining data and extrapolated back to the
1670 defined arrival time. This provided a range of different blast parameters for a single given
1671 trial which could then be statistically investigated whilst trying to omit the effects of the
1672 'sensor' ringing within recorded data. The general algorithm of the curve-fitting routine
1673 is defined as follows:

1675 1. **Determination of t_a :** This parameter typically appears and is recorded as a sharp,
1676 unambiguous rise to peak pressure, with a maximum rise time of only a few (< 10)
1677 recorded samples ($\approx 50 \mu s$) depending on recording frequency of the chosen oscilloscope.
1678 This can be seen in Figure 3.7a and is easily discernible in the raw data, i.e. the main peak
1679 in the gradient of the pressure signal^[129].

1680

1681 2. **Assignment of t_d :** The electrical noise in each trial was recorded with a 10% pre-trigger
1682 to the detonation so an estimation of the effect of the noise could be made for more accurate
1683 t_d determination. As seen in Figure 3.6, the electrical noise is shown to be normally
1684 distributed around the true mean value meaning that when integrating pressure, with respect to
1685 time, the effect of electrical noise cancels out and has negligible effect on peak specific impulse.
1686 Therefore the difference between the assigned t_a and the time at which the maximum specific impulse
1687 occurred would be assigned as the positive phase duration, t_d , as seen in Figure 3.7b. This finding
1688 of normally distributed noise was consistent across all historical recordings.

1690

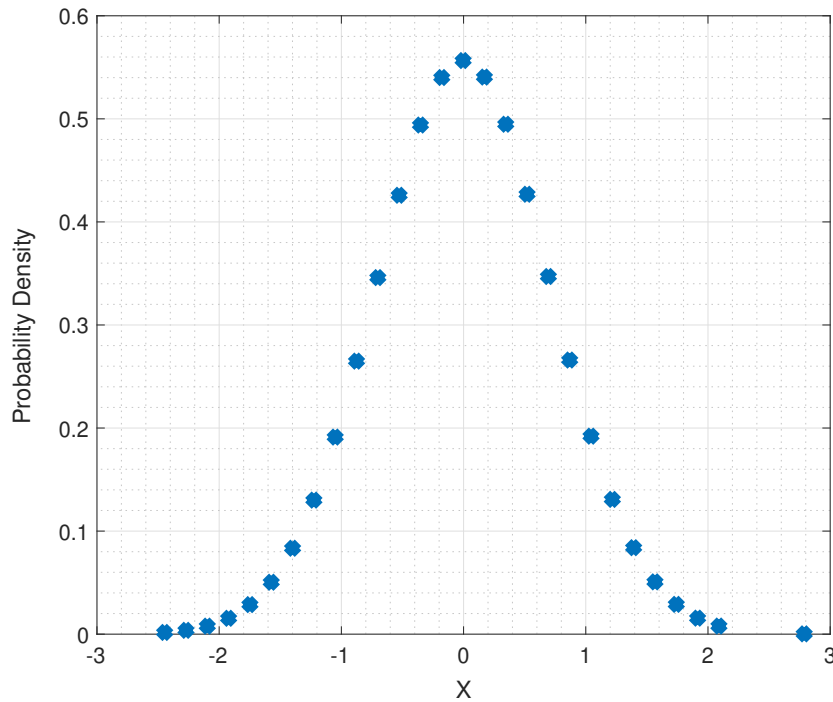


Figure 3.6: Example historic test recording of electrical noise prior to arrival of the shock wave with its distribution presented, where X is associated to the voltage recorded in the system

1691 **3. Incremental removal of start of positive phase:** Inputting the time-based pa-
 1692 rameters and the raw pressure-time data into equation (2.1) provides a generalised
 1693 curve with corresponding values for peak pressure and the decay coefficient, P_{max}
 1694 and b respectively. Using Matlab's curve fitting toolbox, these parameters were fit
 1695 to successively more curtailed pressure signals (in increments of a single percent,
 1696 starting with the full signal) and extrapolating back to the arrival time of the shock,
 1697 obtaining a spectrum of different peak pressures, specific impulses and decay coeffi-
 1698 cients from the results of a single test.

1699

1700 **4. Determination of optimal removal ratio:** Peak pressure and peak specific im-
 1701 pulse values were seen to form a 'fan' around a central representative value (see
 1702 Figure 3.8a). This was caused by fitting curves to the real physical features in the
 1703 data and the additional variation induced by sensor ringing and inherent electrical
 1704 noise. The mean values of the collated results for both peak pressure and specific
 1705 impulse were calculated and the curve fits which satisfied both mean values within
 1706 a $\pm 5\%$ region were defined as the 'optimal model fits' to the raw data. The mean

1707 of these parameters were then evaluated, giving the automatically-generated peak
 1708 pressure and peak specific impulse from that test. The decay coefficient was deter-
 1709 mined from iterations of the integral Equation 2.1, which enables the full pressure
 1710 signal to be reconstructed.

1711

1712 3.2.1.2 Example of Curve Fitting Procedure

1713 Initially, a simple verification of the automated procedure of blast parameter assignment
 1714 was undertaken. Figures 3.7a and 3.7b show the pressure and specific impulse signals
 1715 (through cumulative trapezoidal integration with respect to time) from a historic test
 1716 with a 250g PE4 at 4m stand-off ($Z = 6.3496 \text{ m/kg}^{1/3}$). Overlaid on both figures are
 1717 the automatically assigned time based blast parameters, t_a and t_d , by implementing steps
 1718 1 and 2 of the curve fitting algorithm. These automated values are in good agreement
 1719 with those determined from visual inspection of the raw data, and provide some initial
 1720 confidence in the accuracy of the self-defining parameter routines.

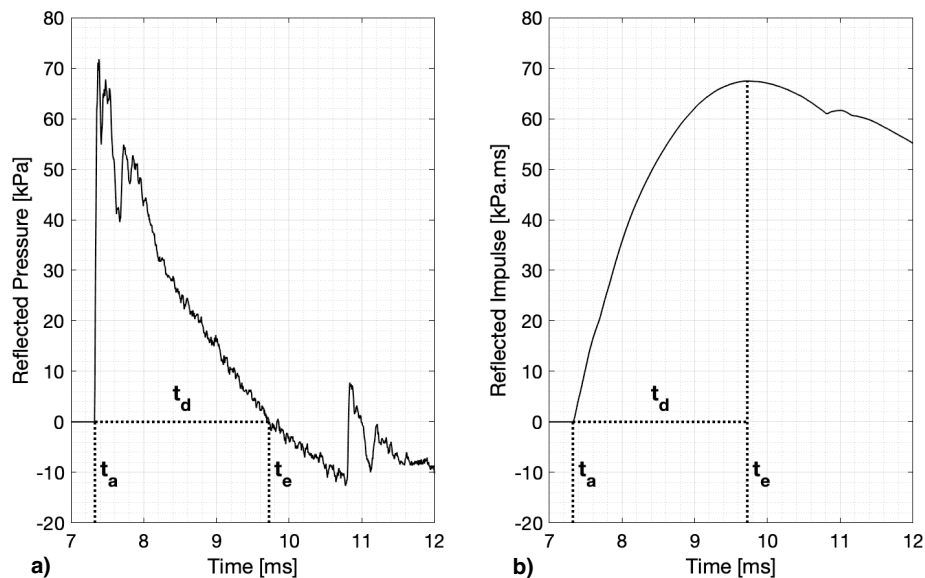


Figure 3.7: Example historic test results using a 250g PE4 hemispherical charge at 4 m stand-off: a) Raw pressure, and; b) Raw specific impulse with respect to time for a single test, highlighting the assignment of shock wave arrival time, t_a , and positive phase duration, t_d

1721 The results from step 3 of the algorithm are shown for the same test in Figures 3.8a
 1722 and 3.8b. With successive removal of between 0–50% of the positive phase data and back
 1723 extrapolating to the defined arrival time parameter, a variety of different fitted Friedlander
 1724 curves are seen, represented by the coloured ‘fanning’ envelope at $t = 0ms$ in Figure 3.8a.
 1725 Peak pressures between 50–72 kPa are determined at this stage of the algorithm which is
 1726 $\pm 12kPa$, or $\pm 10\%$, of the KB prediction for the same given scenario. For a non-statistical
 1727 based analysis, these variations are an improvement on those seen in articles expressing
 1728 the hypothesis of inherent randomness between nominally identical explosive trials. This
 1729 provides a potential explanation for sources of error introduced by the analyst.

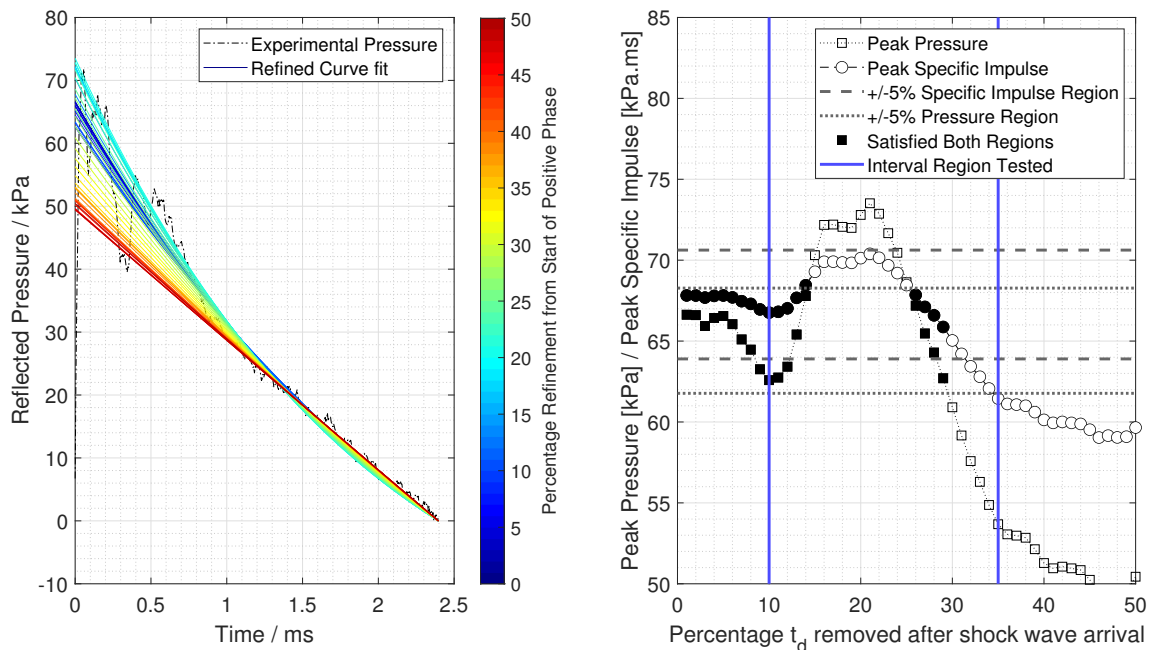


Figure 3.8: Example curve-fitting for historic test results using a 250g PE₄ hemispherical charge at 4 m stand-off: a) all fitted curves to raw data with iterations based on percentage t_d removed immediately after t_a , and; b) Variations in fitted peak pressure and specific impulse as a result of the percentage of t_d removed, and definition of final range (solid markers) for which mean parameters are evaluated

1730 Figure 3.8b visually represents the peak pressures and specific impulse envelopes with re-
 1731 spect to the percentage of t_d removed (from the start of the positive phase). The cyclic
 1732 pattern in the opening 0–35%, and the fanning envelope introduced previously, is a result
 1733 of fitting to different proportions of peaks and troughs in the electrical signal. The re-
 1734 ductions in both parameters after 35% represents the progressive loss of credible data and
 1735 thus would result in the inaccuracy of representing features of the given blast wave. The
 1736 dip in peak pressure and specific impulse around 10% is as a result of fitting to the first

1737 major sensor ringing loss, seen for this test at $t \approx 0.3ms$ after arrival in Figure 3.8a. A
1738 region of cyclic parameters was seen consistently at approximately 10–35% positive phase
1739 removal for all validation tests. Accordingly, this interval was set as the representative
1740 region over which the mean values were evaluated in step 4 of the algorithm, visually
1741 displayed by the vertical lines in Figure 3.8b.

1742

1743 Step 4 assigns the most statistically representative parameters within this reduced region.
1744 First, the mean peak pressure and mean peak specific impulse between the entire region of
1745 10–35% positive phase removal is calculated. Then, any percentage positive phase removal
1746 that results in a peak pressure or peak specific impulse deviating from this initial mean
1747 by more than 5% is discounted, and the mean values from each new, reduced dataset are
1748 calculated. This is shown visually in Figure 3.8b, where the vertical lines denote the initial
1749 region where the first mean is taken (10–35%), the hollow markers indicate those that are
1750 excluded from the final mean calculation, and the solid markers indicate those data points
1751 that are included in the final mean calculation that satisfy the requirements stated above
1752 but only the ones falling inside the 10–35% region were considered.

1753

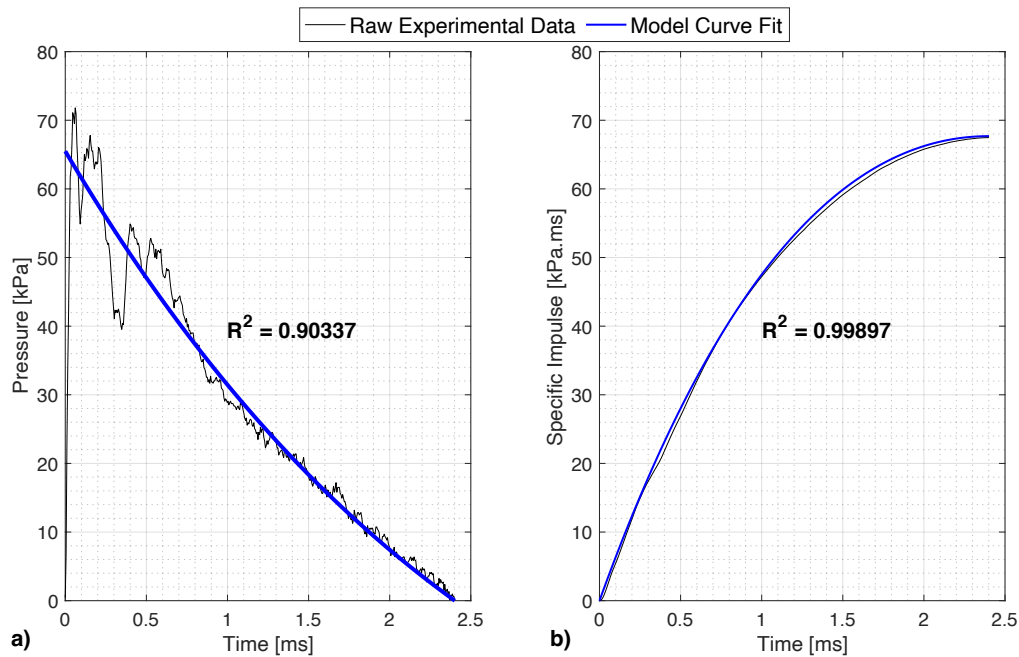


Figure 3.9: Example historic test results using a 250g PE4 hemispherical charge at 4 m stand-off with optimal fits overlaid: a) Overpressure, and; b) Specific impulse

1754 Finally, Figures 3.9a and 3.9b display the raw data from the example historic test (250 g
 1755 PE4 hemisphere at 4 m stand-off), and the resulting modified Friedlander curve fits de-
 1756 termined from the automated process. The high R^2 values noted on each figure show that
 1757 the generalised model fits compare well to the raw data. Due to the nature of pressure
 1758 signals being easily influenced by ringing and other electrical noise, the R^2 of the model
 1759 pressure fit will always be lower than that of a specific impulse fit. This justifies the
 1760 findings presented by Bogosian et al.^[16] which discuss the impulse measurements were in-
 1761 herently more reliable than those for peak pressure. It is worth reiterating that although
 1762 Figures 3.9a and 3.9b present optimal fits from a given test, they are evaluated from a
 1763 large number of different potential fits.

1764

1765 A secondary aim of this analysis was to confirm whether a consistent value of $0.25 t_d$ re-
 1766 moved after the arrival time was suitable as a general rule in line with conclusions made by
 1767 Rigby et al.^[135]. Only 10 of the 65 validation pressure-time histories resulted in optimum
 1768 curves whose parameters were determined from trial fits outside of $0.25 \pm 0.03 t_d$, and

1769 this was mainly due to the level of sensor ringing and electrical noise that was present.
1770 With low levels of sensor ringing, the optimal amount of data removal was found to range
1771 between $0-0.2t_d$, meaning that a Friedlander curve could effectively be fit to the entire
1772 data set without any loss of accuracy. For tests with intermediate to excessive sensor
1773 ringing, or for poor signal-to-noise ratios, a value of $0.25t_d$ was deemed suitable and can
1774 be used as a general rule in the absence of more sophisticated techniques such as the one
1775 presented in this article.

1776

1777 3.2.1.3 Compiled Results

1778 The optimal blast parameters determined from the automated process for the 65 PE4
1779 pressure-time histories used to validate the method^[135], the remaining 79 historical PE4
1780 recordings^[133,175] and the additional 10 PE4 trials conducted as part of this research are
1781 presented in Figures 3.10a–d as a function of scaled distance. Positive phase duration
1782 (Figure 3.10b) was not determined in the historic tests and so are omitted here.

1783

1784 The data presented in Figures 3.10a–d have been expressed using the Hopkinson-Cranz
1785 scaling^[33,74] assuming a constant TNT equivalence of PE4 of 1.20^[132]. For comparison,
1786 KB predictions have also been included in these figures. The recorded blast parameters
1787 demonstrate excellent agreement with the KB predictions, particularly in the region of
1788 $Z > 3 \text{ m/kg}^{1/3}$. The agreement between experiment and prediction reduces when ap-
1789 proaching smaller scaled distances, $Z < 3 \text{ m/kg}^{1/3}$, although the results are still typically
1790 within 10-20%. The fundamental cause in this deviation has not been fully determined,
1791 yet the working hypothesis is that it is likely due to irregular features of the fireball/air
1792 interface giving rise to non-consistent shock wave properties between tests in the mid-to-
1793 near field.

1794

1795 An additional consideration is a singular value of TNT equivalence may not be appropriate
1796 for scaled distance regions in which the fireball and shock wave propagate in unison as
1797 the chemical processes ongoing from a TNT detonation may not be directly scalable to
1798 those from another given explosive. Upon the shock wave detaching from the fireball, the
1799 recorded parameters are scalable as the comparison is made between shock waves travel-
1800 ling through the same medium, unaffected by any external factor. This is the subject of
1801 ongoing work considering the mechanisms of explosives.

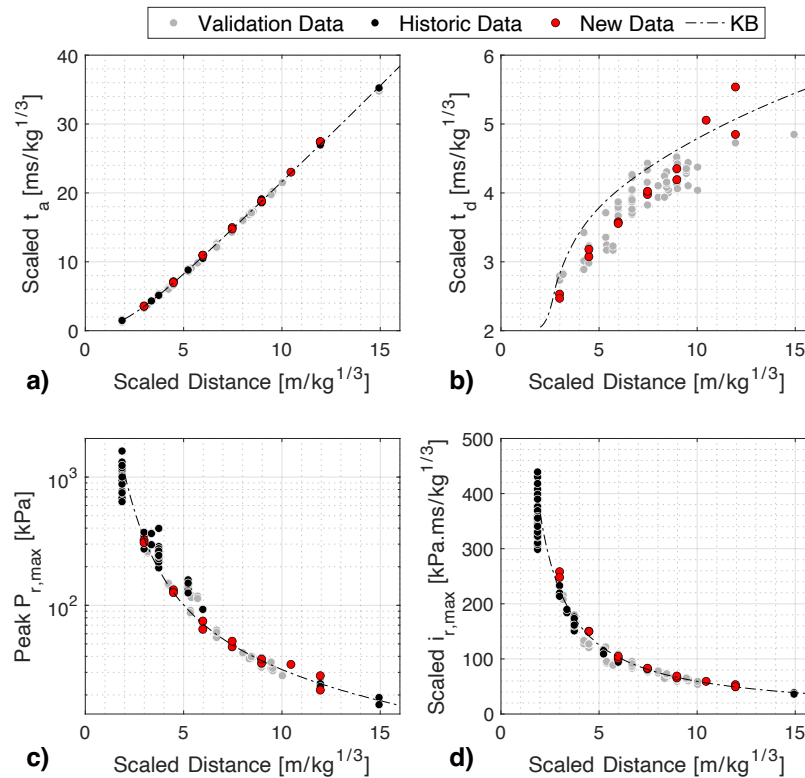


Figure 3.10: Compiled blast parameters from hemispherical PE4 trials as a function scaled distance, compared with KB predictions: a) Scaled arrival time, b) Scaled positive phase duration, c) Peak reflected pressure, d) Scaled reflected peak specific impulse

1803 When considering test-to-test repeatability, there are two main conclusions to be drawn,
 1804 both in agreement with findings published in the literature. Firstly, uncertainty in ex-
 1805 perimentally record blast parameters generally reduces with an increase in scaled dis-
 1806 tance^[16,115] seen predominantly in Figures 3.10c and 3.10d. Secondly, despite best efforts
 1807 to remove unconscious researcher-dependent bias, peak pressure, specific impulse and pos-
 1808 itive phase duration, in nominally identical scenarios, demonstrate much higher levels of
 1809 test-to-test variability than shock front arrival time. This can be seen when comparing
 1810 Figures 3.10b–d with 3.10a respectively. It is suggested that test-to-test variations are
 1811 non-negligible for $Z < 3 \text{ m/kg}^{1/3}$ and the source of these variations are to be investigated
 1812 in future works.

1813

1814 3.2.2 Modifications for Positive Phase Improvement (Ei)

1815 The results from the far-field trials presented in this thesis begin to build a coherent fun-
1816 damental understanding of the nature of experimentally recorded far-field loading from
1817 a given charge composition, shape and mass. The importance of these findings provides
1818 a starting point to establish characteristics of more challenging blast loading conditions
1819 both in the near-field and those in complex environments. Knowledge of which can be
1820 supplemented with high-fidelity, validated numerical modelling.

1821

1822 Considering Figure 3.5, the standard ‘sensor ringing’ feature can be seen across each of
1823 the positive phases recorded at each increment of standoff. The results of which have
1824 historically been assigned to being technical incapacibilities of the pressure gauges used in
1825 the trials. Although the aforementioned analytical techniques have shown to effectively
1826 omit the effects of this feature, it was important to establish its root cause and attempt
1827 to rectify it for future testing regimes.

1828

1829 Reisler et al.^[119] reviewed extensively the development of pressure gauges throughout the
1830 high explosive era. Examples of recorded pressure-time histories for different gauges across
1831 numerous researchers and trials were provided within. One interesting finding in this arti-
1832 cle was that represented by Figure 3.11, which showed similar oscillations in the opening
1833 25% of the positive phase duration recorded as the tests undertaken at the University of
1834 Sheffield. The fact that a similar finding was recorded using a pressure probe directed the
1835 thinking towards a real physical shock feature rather than a individual technical failure of
1836 the gauge causing early time oscillations.

1837

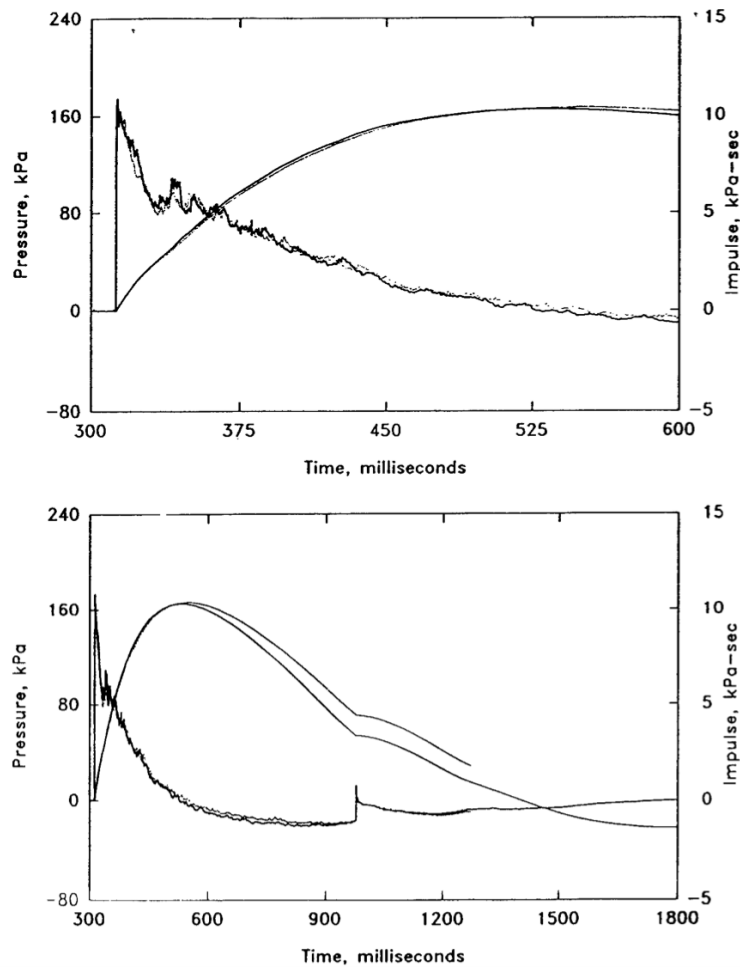


Figure 3.11: Example historic test results from a 1000lb high explosive height of burst shot recovered from Reisler et al.^[119], displaying a comparison between a pressure probe (dotted) and a Kulite pressure gauge within a baffle plate (solid) across two different time bases, similar to the methods adopted in this thesis.

1838 Understanding the mechanisms and magnitudes of blast loading on targets is of key im-
 1839 portance for the analysis and design of the response of protective structures. Vast amounts
 1840 of numerical work has been undertaken throughout literature with experimental work to
 1841 validate its accuracy^[179]. However, there has yet to be a synergistic approach to anomaly
 1842 or systematic feature finding in experimental trials using validated models. The experi-
 1843 mental results and analysis discussed within this thesis have been used to validate far-field
 1844 numerical models for various high explosives. These can be implemented into much more
 1845 complex numerical simulations to produce validated and accurate prediction for near-field
 1846 and complex conditions. It was important to use these models to try and reverse engineer

1847 a physical justification to any systematic errors which could contribute to the variabilities
1848 seen in experimental data.

1849

1850 3.2.2.1 Synergy of Modelling and Experimental Studies

1851 Whittaker et al.^[179] validated APOLLO blast simulator, developed by EMI, using an in-
1852 house explicit afterburn method (leveraging Cheetah and EXPLO5), against experimental
1853 data for PE4 and PE10 explosives and then for PE8 in works presented by Farrimond
1854 et al.^[50]. The agreement attained gives confidence in the standard equation of states
1855 (EOSs) and subsequent parameters developed for these particular explosives, which can
1856 be utilised in numerical modelling for more complex scenarios. Numerical modelling gen-
1857 erally is taken at its most simplified form to try to establish key explosive characteristics
1858 from basic scenarios which can then be mapped into more complex settings. These el-
1859 ementary numerical models usually do not account for minor environmental variations,
1860 such as terrain levels, reflective surface blemishes and energy losses, therefore omitting any
1861 shock wave mechanisms which could occur because of the aforementioned.

1862

1863 The oscillations identified in the raw pressure-time history traces from Figure 3.12a, and
1864 others within this thesis, have been previously attributed to a consistent feature of ‘sensor
1865 ringing’^[49,135], which was an assumed systematic technical limitation of the gauges used.
1866 The numerical models developed in the aforementioned articles with validated EOS’s for
1867 a given explosive previously utilise simplistic, and fairly unrealistic baseline conditions.
1868 This is denoted by the idealistic shock wave behaviour simulated against experimentally
1869 recorded reflected pressure and specific impulse for 250g PE10 hemisphere at 2m stand-off
1870 as seen in Figures 3.12a and 3.12b.

1871

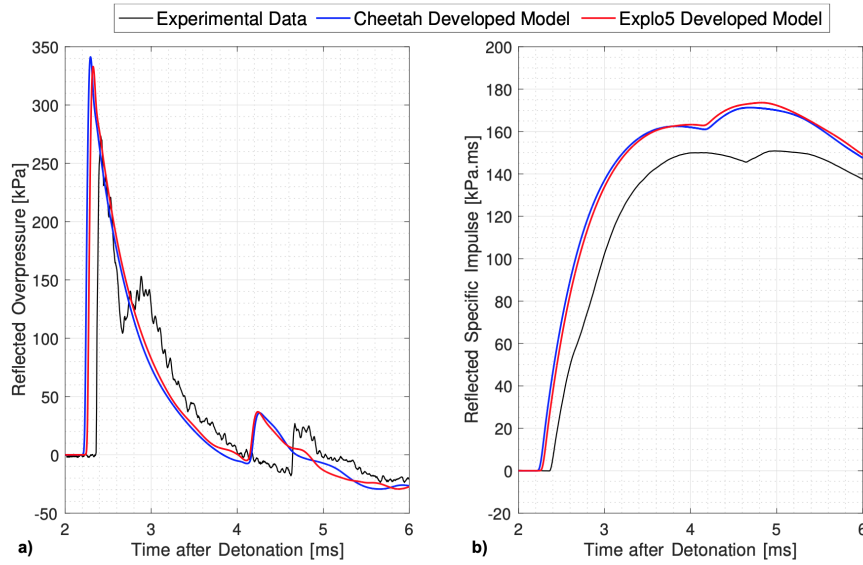


Figure 3.12: Comparison between experimentally recorded data for a 250 g hemispherical PE10 charge detonated 2 m away from a reflected gauge and Apollo numerical modelling (with EOS parameters developed using Cheetah and Explo5): a) Reflected overpressure, b) Reflected specific impulse, both with respect to time after detonation occurs

1872 The numerical models utilised quarter-symmetry, with both the ground, vertical symmetry
 1873 plane, and a boundary wall at the required stand-off distance defined as perfectly reflecting
 1874 surfaces with afterburn features included using an explicit method. The Explosive EOS
 1875 parameters used in this study were generated using the thermochemical code Cheetah
 1876 v7, which, due to export control reasons, are not available for publication. Therefore, an
 1877 alternative thermochemical code, EXPLO5, was also used to determine the EOS for the
 1878 explosives. The surrounding medium of the numerical simulation was assumed to be air
 1879 defined by the parameters in Table 3.1.

1880

	Parameter	Units	Value
Perfect Gas EOS	R	$Pa/(K.kg/m^3)$	288
	$E0$	J/kg	-2.375e5
Caloric EOS	c_{v1}	$J/kg.K$	723.3
	c_{v2}	$J/kg.K^2$	0.0749

Table 3.1: Table of EXPLO5 Equation of State Parameters for Air.

1881 These parameters have only been used for a complementary comparison to the Cheetah
 1882 study and have not been fully assessed or validated, but are provided as representative

1883 values in Table 3.2. It is important to note the developed parameters were produced purely
 1884 based on the explosive composition, correct density and a best guess of plasticiser material
 1885 (finding the exact composition may improve the results), all done using the same default
 1886 methodology. No calibrating to experimental data or tweaking of technique or parameters
 1887 to improve results was performed.

1888

	Parameter	Units	Explosive Type		
			PE4	PE8	PE10
JWL	Density	g/cm ³	1.59	1.57	1.55
	A	kPa	3.62E+08	3.35E+08	3.21E+08
	B	kPa	7.89E+06	7.24E+6	9.40E+06
	R1	N/A	3.99	3.94	4.40
	R2	N/A	1.15	1.13	1.23
	w	N/A	0.27	0.27	0.271
	Detonation Velocity	m/s	7700	7608	7735
Detonation Products	E_0	kJ/kg	-5.36E+06	-5.32E+06	-5.18E+06
	Gas Constant	J/kg.K	282	282	269
	CV1	J/kg.K ²	738.8	672.9	738.2
	CV2	J/kg.K ³	0.4321	0.4502	0.3847
Combustion Products	E_0	kJ/kg	-3.99E+06	-3.57E+06	-4.51E+06
	Gas Constant	J/kg.K	286	287	283
	CV1	J/kg.K ²	488	309	453
	CV2	J/kg.K ³	0.295	0.409	0.320
Other	Stoichiometric Ratio (Air/Exp)	N/A	0.716:0.284	0.726:0.274	0.709:0.291

Table 3.2: Table of EXPLO5 Equation of State Parameters for Explosives.

1889 Figure 3.12 shows both numerical models to be almost identical in this scenario, but with
 1890 both somewhat over-predicting the peak reflected pressure and impulse of the event whilst
 1891 not capturing the ringing seen in the opening $\sim 0.5ms$ of the positive phase. Hereafter,
 1892 APOLLO blastsimulator using the Cheetah-determined equation of state parameters was
 1893 the chosen method of numerical prediction due to the similarity in modelling results. The
 1894 arrival time of the primary shock at this scaled distance is predicted sooner than the ex-
 1895 perimentally recorded which in conjunction with larger pressure predictions too begins the
 1896 narrative that the models are experiencing a bigger energy release than what was recorded.

1897

1898 The arrival of the secondary shock in the numerical models is much earlier than that
 1899 seen in the experimental data, which can lead to artificially high specific impulse pre-
 1900 dictions as the secondary shock arrives during the positive phase of the event. Other
 1901 published literature describes the numerical secondary shock as arriving much *later* than
 1902 the experimentally-recorded value when using numerical codes which do not explicitly
 1903 account for afterburn^[129], and it has been shown that including a calibrated secondary

1904 energy release can bring the secondary shock in line with the experimental recordings^[141].
 1905 Whilst this method holds empirically based variable prescription, it is clear that the arrival
 1906 of the secondary shock is intimately linked to the post-detonation pressure–volume–energy
 1907 relations of the fireball, and is a known limitation of current modelling capabilities. This
 1908 feature is believed to be due to the over prediction of the sound speed within the fireball.
 1909

1910 The results from the simplest form of the numerical simulation led developments to con-
 1911 sider energy losses to the ground surface. In the experimental methodology, the explosive
 1912 charge was detailed as being placed on top of a steel anvil to reduce progressive damage
 1913 to the ground; this alone is enough evidence to suggest some energy is being lost to the
 1914 ground surface and therefore the first modification to the simplistic model was to take
 1915 these into account. A 10% energy loss was applied to the simulation, in line with TM5-
 1916 8858^[77], by using 225 g rather than 250 g of PE10, the results of which are shown in
 1917 Figure 3.13. This is conceptually similar to the 1.8 spherical equivalence factor adopted
 1918 in the KB predictions, and brings the modelling results much more in-line with the exper-
 1919 imental data.

1920

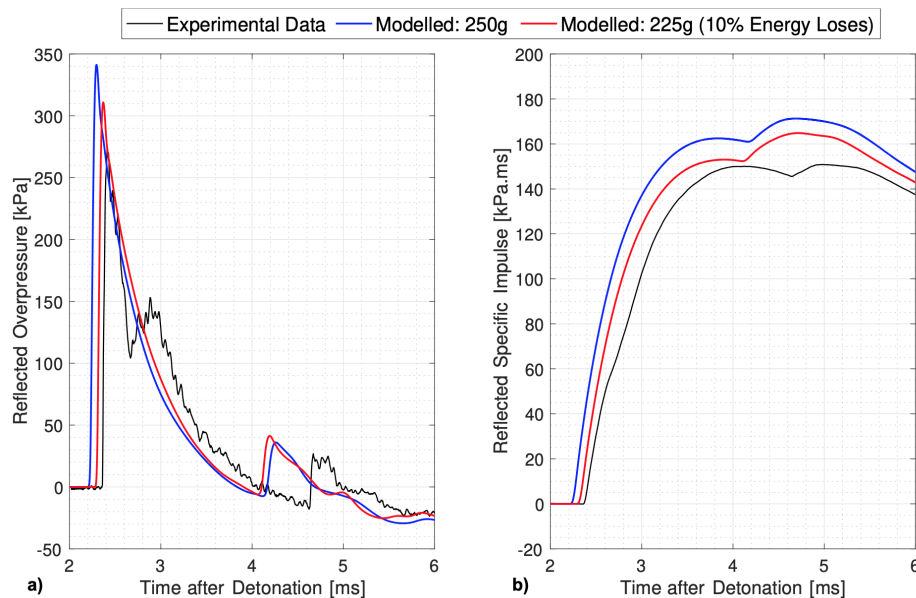


Figure 3.13: Comparison between experimentally recorded data for a 250g hemispherical PE10 charge detonated 2m away from a reflected gauge and numerical model evaluating using APOLLO when accounting for energy losses into the ground a) Reflected overpressure, b) Reflected specific impulse, both with respect to time after detonation occurs

1921 The ‘sensor ringing’ seen in the experimental data was still not captured in the model
 1922 detailed in Figures 3.13a and 3.13b. In higher fidelity numerical analysis, the pressure
 1923 gauges were detailed as being made flush to a steel plate which protruded 25 mm from
 1924 the reflecting surface. The significant drop in the pressure seen in Figure 3.14a over the
 1925 opening 25% of the positive phase is enough evidence to support the necessity of accurate
 1926 models to verify experimental procedures. By capturing some of the more complex early-
 1927 time behaviour initially attributed to ‘sensor ringing’, numerical analysis and experimental
 1928 procedures can work collaboratively to achieve higher levels of accuracy overall. This is
 1929 a key insight for future experimental trials as it means if a truly flush surface can be
 1930 achieved without any opportunity for clearing and reflected wave interactions, the statis-
 1931 tical accuracy of the recorded data quoted in this article may be improved on further and
 1932 the ‘sensor ringing’ maybe omitted. Shin et al.^[148] ran numerical analysis which however
 1933 suggests once in the extreme near-field region, $Z < 1.6 \text{ m/kg}^{1/3}$, the effects of clearing is
 1934 negligible and should be ignored for design purposes.

1935

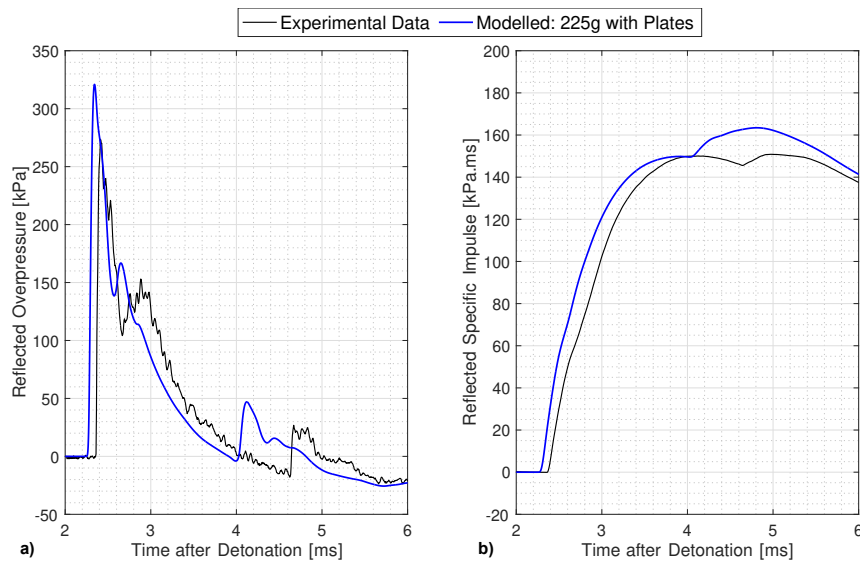


Figure 3.14: Comparison between experimentally recorded data for a 250g hemispherical PE10 charge detonated 2 m away from a reflected gauge and numerical model evaluating using APOLLO when accounting for energy losses into the ground and substantial terrain features a) Reflected overpressure, b) Reflected specific impulse, both with respect to time after detonation occurs

1936 3.2.2.2 Blast Arena Modification

1937 The fact that the numerical modelling has allowed the identification of systematic features
 1938 within the experimental arrangement is testament to the synergistic nature of modelling
 1939 and experimentation, especially when high levels of control are attained as they are in this
 1940 study. A larger steel plate, with a minimum distance of 1.5m from the gauge to the edges
 1941 of the plate, was affixed to the bunker wall to ensure that the aforementioned localised
 1942 clearing effects would not interfere with the positive phase of the shock.

1943

1944 A repeat trial was conducted using a 250g PE10 hemisphere detonated 5m away from this
 1945 surface and compared to the original trials, the data of which is presented in Figures 3.15a
 1946 and 3.15b. The new results indicate that whilst these 'sensor ringing' oscillations alter
 1947 the qualitative form of the pressure trace, their influence on the quantitative blast pa-
 1948 rameters, determined through curve fitting, or integration of the pressure signal to find
 1949 specific impulse, are negligible when comparing the raw data to the curve fit itself as seen
 1950 in Figures 3.9a and 3.9b.

1951

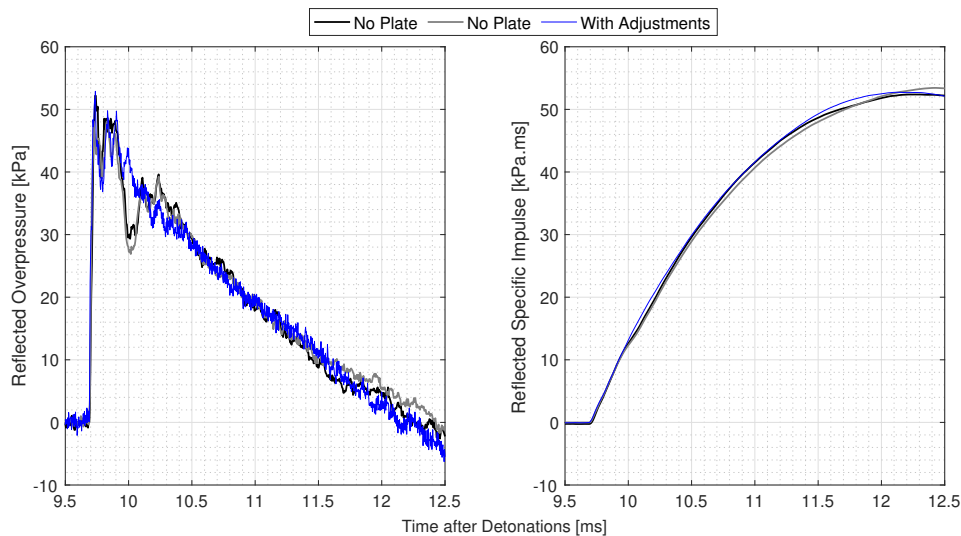


Figure 3.15: Comparison between experimentally recorded data, time-shifted so the arrival of the positive phases align exactly, for a 250 g hemispherical PE10 charge detonated 5 m away from a reflected gauge with adjustments to the blast arena and without: a) Reflected overpressure, b) Reflected specific impulse, both with respect to time after detonation occurs

1952 3.2.2.3 Top vs Bottom Detonation in Ideal Explosives

1953 Another important notion to take from the trial results presented in Figures 3.10 is the
1954 agreement of findings presented by Bogosian et al.^[16]. Both suggest that specific im-
1955 pulse values developed from curve fitting techniques are much more reliable than peak
1956 overpressure. However, the physical reasoning for the clear spread in the data seen at
1957 $Z < 3 \text{ m/kg}^{1/3}$ for the two parameters is still yet to be fully determined. The hypoth-
1958 esis detailed by Tyas^[173] made definitive justification to the experienced spread through
1959 links to fluid dynamic instabilities which is a physical and logical explanation. What was
1960 not considered in full detail however was how the experimental procedure of top detonat-
1961 ing the hemispherical charge could effect the close-in shock wave/fireball blast parameters.

1962

1963 Sherkar et al.^[146] undertook an analysis on the influence of a charge shape and detonation
1964 point on explosive parameters and found that for scaled distances of $Z > 3.15 \text{ m/kg}^{1/3}$ the
1965 independent variable made no difference to shock wave parameters. This finding is in line
1966 with the working hypothesis defined by Tyas^[173]. The predictions at smaller scaled dis-
1967 tances are however significantly variable as charge shape and detonation point is altered.
1968 As the majority of the data presented within this thesis, unless specifically specified, was
1969 the result of a top detonated hemisphere it was important to investigate, using both new
1970 experimental data and validated numerical modelling, whether or not detonator position
1971 could be an influencer of parameter variability in the $Z < 3 \text{ m/kg}^{1/3}$ region.

1972

1973 Using a validated EOS for PE4, both top and bottom detonated scenarios were con-
1974 sidered for the standoff range of 1.25-2m assuming a TNTe=1.2, corresponding to $Z =$
1975 $1.87 - 2.99 \text{ m/kg}^{1/3}$. Experimental data used for a direct comparison to both the CFD
1976 simulations and KB predictions presented. According to numerical simulations presented
1977 in Shin et al.^[148], the near-field modelling can ignore the effect of clearing as it is negligi-
1978 ble. The gauge plate features discussed in Section 3.2.2 was therefore removed from the
1979 simulation, resulting in minor changes to the overall pressure-time history. It is however
1980 important to note that the aforementioned article makes use of AUTODYN for modelling
1981 capabilities where as the study presented in Figure 3.16 uses APOLLO.

1982

1983 The nearest standoff of 1.25m, a scaled distance of $Z = 1.87 \text{ m/kg}^{1/3}$, considered was
1984 chosen as it shows to exhibit the most variability in the resulting output blast parame-

1985 ters from historic trials. The results from experimental tests, numerical simulations and
1986 KB predictions are presented in Figures 3.16a-d for this region to both consider the ef-
1987 fects of detonator position but also test the validity of KB predictions at these scaled
1988 distances. What is evident from these plots is the distinct differences between top and
1989 bottom detonated results when considering the general form of the experimental pressure-
1990 time histories.

1991

1992 The bottom detonated trials, represented by Figures 3.16b and 3.16d, show that both
1993 KB predictions and numerical simulation over-predict the peak reflected pressure, but the
1994 specific impulse values agree reasonably well. This begins to reiterate shock wave develop-
1995 ment and formation theory discussed by Kinney and Graham^[90] in Chapter 2. The theory
1996 shows a clear transition of the pressure-time history from a rounded pressure pulse, which
1997 then begins to expand radially at its own speed for each portion of the pulse, therefore
1998 the higher pressure and temperature regions expanding faster, resulting in a progressively
1999 sharper rise to peak pressure, and thus the shock discontinuity.

2000

2001 Figures 3.16b and 3.16d are believed to display a shock in this transition period. The
2002 overall energy released from the detonation is representative of KB and numerical model
2003 predictions for a 250g PE4 charge but the peak pressure is lower and time duration is longer
2004 in the experimental trial. It is clear that neither numerical modelling or KB predictions
2005 are able to accurately capture the transition period of the shock front as both over-predict
2006 the peak pressure considerably. At these scaled distances KB predictions were developed
2007 using numerical simulation which explains why the two compare reasonably well. It is
2008 clear that these methods assume a fully formed shock discontinuity occurs sooner than in
2009 reality and therefore the real physical behaviour of the early development of the shock is
2010 not captured correctly.

2011

2012 Figures 3.16a and 3.16c represent the top detonation of 250g PE4 hemisphere. These
2013 result in a much lower peak pressure when compared to KB predictions but peak specific
2014 impulse values presents much lower too. This is a direct result of the varying detonation
2015 dynamics, resulting in differences in rates of energy release. This induces a more incon-
2016 sistent behaviour when compared to a standard free-air assumption. Reassuring is the
2017 strength of numerical modelling at this scaled distance which although has not captured
2018 either the pressure-time or specific impulse-time histories perfectly, the same general trend

2019 can be qualitatively seen. Quantitatively, the numerical simulation is different to the ex-
 2020 perimental data which is speculated to be related to the exact position and orientation
 2021 of the detonator. In the numerical simulation, a perfect, infinitesimally small detonation
 2022 point is prescribed at the very top of the charge. In reality despite best efforts, the det-
 2023 onators used in testing are approximately 5mm into the top of a charge, do not expand
 2024 radially and may not be perfectly orientated. This therefore will result in different det-
 2025 onation dynamics. The fact, that qualitatively, the trends show similarities gives rise to
 2026 the testing methodology but validates the sensitivity of the results at this scale.

2027

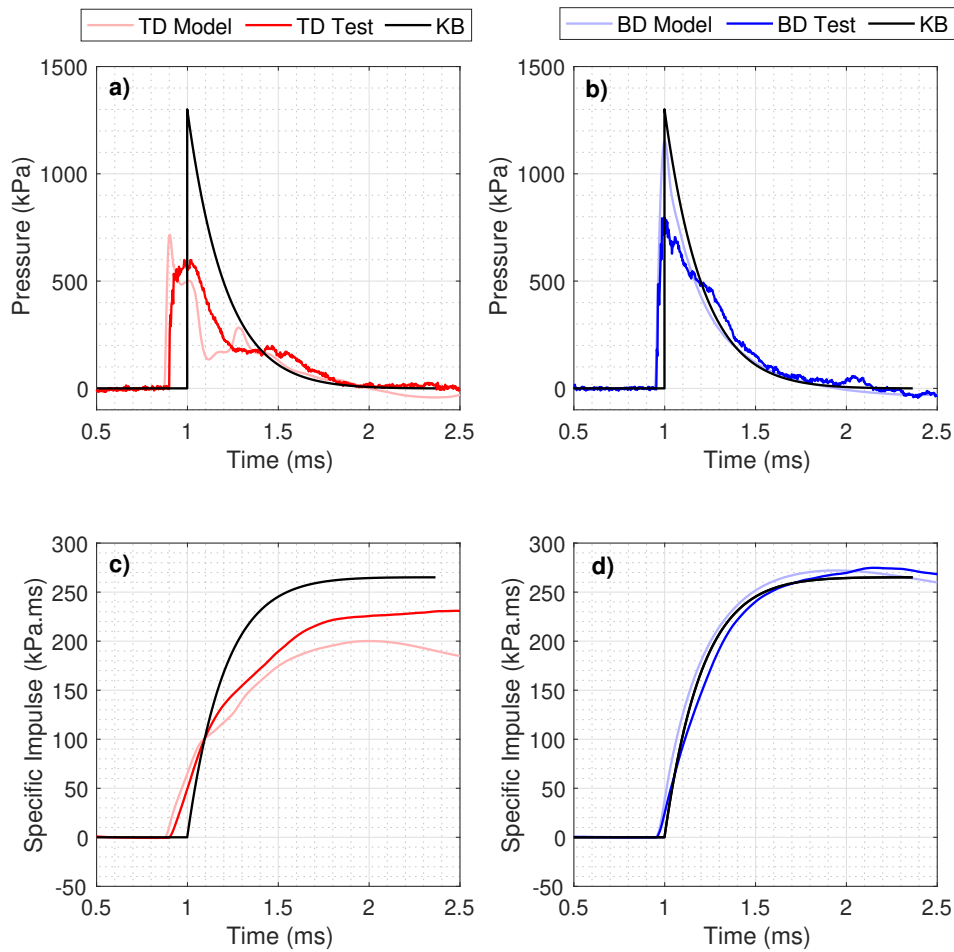


Figure 3.16: A comparison of top and bottom detonated 250g hemispherical PE4 charges recordings of reflected pressure and specific impulse against time at 1.25m standoff (corresponding to plots a and c, and b and d respectively) which make reference to KB predictions and numerical simulations.

2028 For the top detonated trials, there is believed to be a slower expanding detonation cloud
 2029 and resulting shock wave associated with the top portion of the charge. This expands
 2030 prior to the detonation wave reaching the bottom of the charge. When the detonation

2031 wave reaches the bottom of the charge, and the reflective surface, the resulting detonation
2032 products would begin to expand laterally along the reflected surface in the form of jets,
2033 and radially back outwards. The jetting-like behaviour has been attributed to the first rise
2034 in the pressure-time histories seen in Figure 3.16a, with the secondary radial expansion
2035 being attributed to the rise approximately 0.3ms later experienced in both the numerical
2036 simulation and experimental data.

2037

2038 The additional near-field experimental results, considering the standoff range of between
2039 1.25-2m, are presented in Figures 3.17a-h. These trials aimed at deducing the distance
2040 the effect of detonator position becomes irrelevant to the overall blast parameter data.
2041 The raw data presented in the aforementioned plots is considered in a qualitative sense
2042 against KB predictions. The results suggest that for standoff distances greater than 1.5m,
2043 $Z > 2.24 \text{ m/kg}^{1/3}$, both top and bottom detonated trials behave consistently in terms of
2044 the overall energy released, and thus the resulting peak specific impulse trends which are
2045 captured well by KB predictions as shown in Figures 3.17f and 3.17h. As the standoff
2046 distance is reduced, the KB predictions generally capture the energy released in bottom
2047 detonated trials, but top detonated results show a lower energy release overall. This is
2048 directly related to the non-spherical expansion of the detonation wave and its products,
2049 resulting in inconsistent energy releases, seen in Figures 3.17b and 3.17d. KB predictions
2050 generally over-predict the peak overpressure at these scaled distances too, which is related
2051 to the lack of comprehension in the physical shock formation in the empirical tool at these
2052 scaled distances.

2053

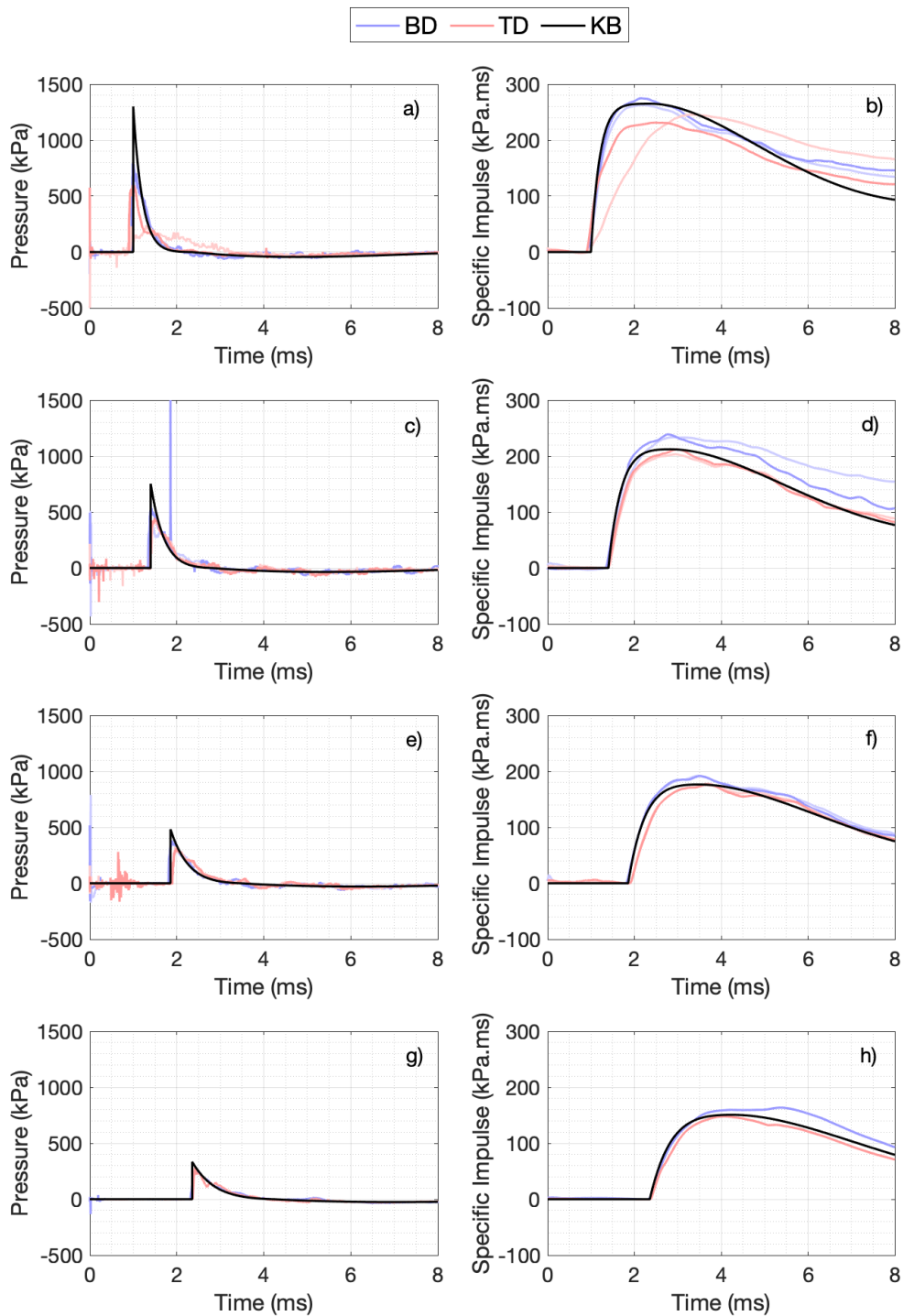


Figure 3.17: A comparison of two top and bottom detonated 250g hemispherical PE4 charges recordings of reflected pressure and specific impulse against time at 1.25m, 1.5m, 1.75m and 2m standoffs (corresponding to plots a, c, e and g, and b, d, f and h respectively) which make reference to KB predictions.

2054 Interestingly, the blast parameters from the new top detonated data at 1.25m standoff,
2055 shows to be in line with the lower bound of those presented in Figures 3.10c and 3.10d.
2056 The historic data presented in both Figures 2.12a-d and Figures 2.13a-d was conducted by
2057 different research teams, which reinforces the narrative of ensuring the systematic differ-
2058 ences between testing regimes are considered when comparing results. The historic data
2059 at $Z > 3 \text{ m/kg}^{1/3}$ shows agreement with the newly conducted trials, which validates the
2060 hypothesis that at this scaled distance blast parameters are less sensitive to the imperfec-
2061 tions, but are for near-field scenarios, $Z < 3 \text{ m/kg}^{1/3}$. The team conducting the historic
2062 trials may not have been as strict in their control measures, with variations in the deto-
2063 nator placement resulting in greater variability. The fact the repeat shots in this study
2064 show low levels of variability suggests that these are a more appropriate characterisation
2065 of the detonated explosives. Future tests will look at studying the near-field region with
2066 much more rigour to establish the consistency in the blast parameters similarly to that
2067 discussed within this thesis for far-field scenarios.

2068

2069 These trials show that the top detonated experimental methodology utilised for the ma-
2070 jority of the data referred to in this thesis, will give rise to spreads of data in near-field
2071 regions, $Z < 3 \text{ m/kg}^{1/3}$. Data recorded in this region will be influenced by detonator
2072 position, orientation, charge geometry and placement imperfections alongside any other
2073 systematic experimental and analytical errors induced during the trials. This finding pro-
2074 vides evidence towards the overlooked systematic errors incorporated into blast related
2075 experimental work which result in the justifications of ‘inherent’ high levels of explosive
2076 yield variability.

2077

2078 3.2.3 Analysis of Negative Phase (Ai)

2079 Rigby et al.^[133] undertook an intensive assessment of the negative phase of the ideal
2080 explosive, PE4, and compared various methods of approximating the associated trends.
2081 The secondary, and subsequent, shock phenomenon are not accounted for in any empir-
2082 ical predictions presented within the aforementioned article. This is justified by the low
2083 magnitudes they exhibit compared to the negative phase presented in ideal explosives.
2084 Further noted is the affect clearing has on a pressure trace, a given time after the arrival
2085 of the primary wave. This is associated with the speed of a clearing wave being 340m/s
2086 and then taking the distance from the nearest edge of a reflective surface. Clearing and

2087 reflections from other surfaces are wave phenomena which will be picked up on experimen-
2088 tal recordings. Without a fundamental understanding of the systematic errors associated
2089 with a testing methodology, data could be misinterpreted, with these features considered
2090 to be real physical parameters of a free-air shock interacting with a single surface.

2091

2092 Whilst this thesis neglects both these experimental features, the ‘Cubic Negative’ curve fit
2093 approximation (detailed in the second case of Equation 2.6), developed by Granstrom^[66],
2094 presents as the most effective method of approximating the negative phase both qualita-
2095 tively and quantitatively for the ideal explosive in question. This provides further justifi-
2096 cation as to the source of negative phase semi-empirical curves. Shown in Figures 3.18a-d
2097 is the considerable effects experimental features can have on the overall loading from the
2098 same explosive detonation. It is important to note that the data presented in the aforemen-
2099 tioned figure was recorded during the same trial using two different recording locations.
2100 Interestingly, the general form of the raw data exhibits rather obvious differences when
2101 $t=17.5-22\text{ms}$ which is approximately 7-12ms after the arrival of the primary shock.

2102

2103 Using a 340m/s clearing wave speed and the distances to the nearest sides of the walls,
2104 the time at which the clearing waves would arrive at the respective pressure gauges would
2105 be approximately 12ms and 7ms after the arrival of the primary shock. With the bunker
2106 wall being effectively infinitely long but 4m tall, only one clearing wave would interact
2107 with the negative phase data. The blockwork wall on the other hand has three finite
2108 edges, two of which were 2.25m resulting in two clearing waves interacting with negative
2109 phase simultaneously around 17ms after detonation. This is clearly denoted by the lower
2110 pressures in Figure 3.18a associated with the time periods mentioned.

2111

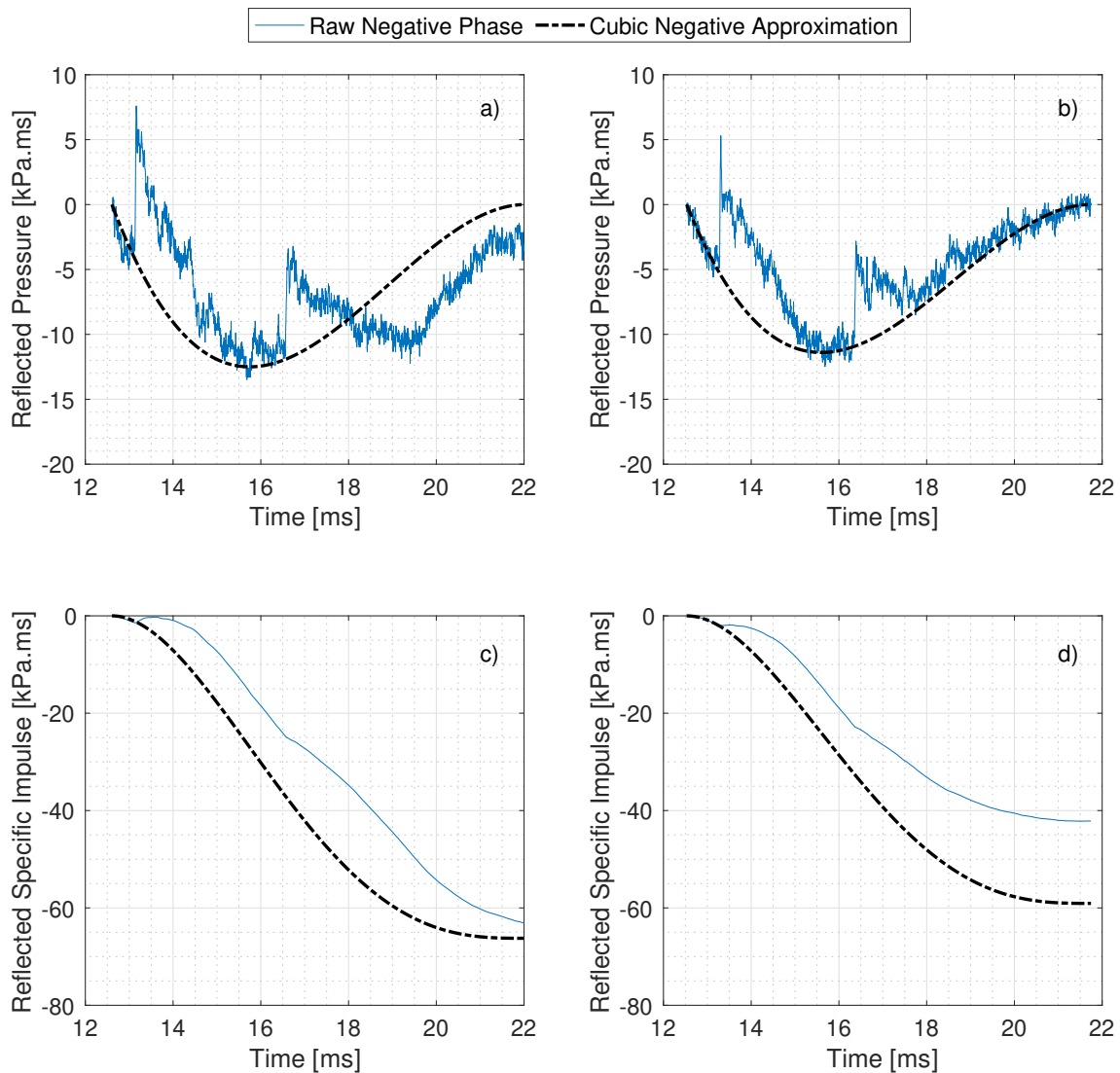


Figure 3.18: Comparison of negative phase results from a 250g PE4 trial with two reflective recording points 5m either side of the explosive, where plots a) and b) represent the raw negative pressure-time plots and plot c) and d) are the corresponding impulse-time plots for the blockwork and bunker walls respectively.

2112 3.2.3.1 Compiled Negative Phase Parameters

2113 The ‘Cubic Negative’ approximation investigated by Rigby et al.^[133] was implemented on
 2114 this data set to validate the findings further as presented in Figure 3.18. It is clear to
 2115 see a reasonable prediction for the negative phase loading is established across all three
 2116 parameters regardless of secondary shocks and any other experimental features.

2117

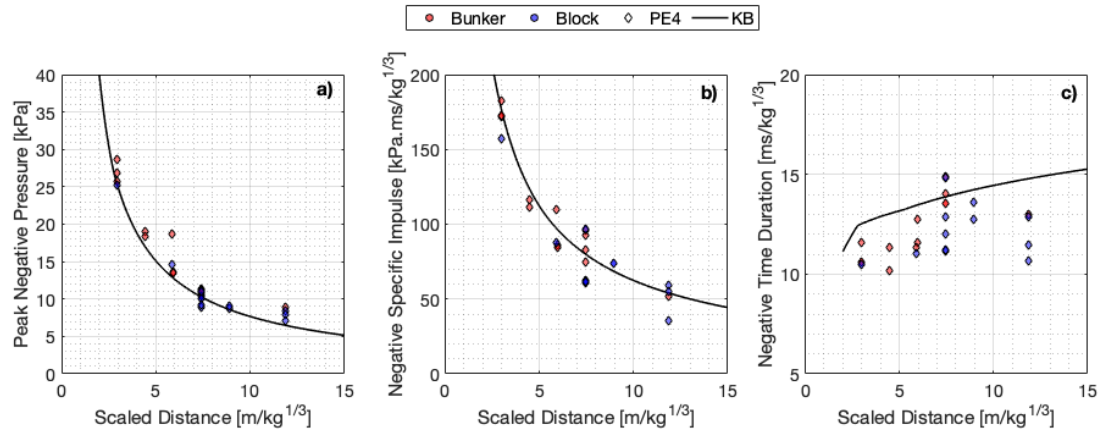


Figure 3.19: Compilation of scaled negative phase parameters resulting from visual analysis of 250g PE4 hemispherical charge using a $TNTe=1.22$: a) Negative Pressure, b) Scaled Negative Specific Impulse and c) Negative Phase Time Duration.

2118 What is yet to be established is a rigorous validation of quoted negative phase parameters.
 2119 The difficulty in characterising these parameters is associated with the lower magnitudes
 2120 of the negative phase parameters and therefore it is much easier to misinterpret the ac-
 2121 tual values within the noise of the recorded data. Reducing the noise for the secondary
 2122 shock is not possible as its magnitudes is inherent to the rating of the gauge. To avoid
 2123 gauge failure, the rating needs to be high enough to withstand the primary blast wave
 2124 prior to secondary waves arriving. That alongside experimental features, related mainly
 2125 to clearing, means that developing automated analytical tools, similar to that presented
 2126 in Section 3.2.1, hold more difficulty. Taking this into account, the negative phase param-
 2127 eters are established visually within the contents of this thesis to validate both the ‘Cubic
 2128 Negative’ and semi-empirical approximations across a number of different explosives.

2129

2130 Figures 3.19a-c displays visually interpreted negative phase parameters from 250g hemi-
 2131 spherical PE4 trials across the far-field testing range. This data was scaled using a
 2132 $TNTe=1.22$, in line with findings from Farrimond et al.^[50], and compared directly to
 2133 the empirical curves established by Granstrom^[66]. The agreement with predictions of
 2134 peak negative pressure, $P_{r,min}$, suggest that whilst the experimental methodology induced
 2135 variations during the negative phase, the general maximum pressure experienced is in
 2136 line with approximations. It is important to note, if the reflecting surfaces were smaller,
 2137 clearing waves would arrive at the recording positions sooner, resulting in variations to
 2138 the established $P_{r,min}$ values. Figures 3.19b and 3.19c present as comparing well to the

2139 approximations due to including the behaviours of secondary shock and clearing features
2140 respectively discussed by Rigby et al.^[133]. Section 3.2.3.2 discusses the strength in the
2141 ‘Cubic Negative’ approximations when empirically accounting for secondary shock loading.

2142

2143 3.2.3.2 Secondary Shock Phenomena

2144 The secondary, and successive shocks, of an explosive are noticeable features in a pressure-
2145 time history. Despite some preliminary efforts of establishing yield prediction parameters
2146 through experimental work^[62,129,133], there is yet to be a rigorous investigation of all shock
2147 wave parameters. Noted in Rigby and Gitterman^[129] is the inability of numerical mod-
2148 elling to capture the negative phase and secondary shock behaviours accurately. The root
2149 cause is currently unknown but is speculated to be related to erroneous detonation me-
2150 chanics integrated within simulations, proving it necessary to establish the behaviour of
2151 secondary shocks empirically. The secondary shock, unlike the initial shock, has to travel
2152 through the detonation product fireball and therefore, the properties of the surrounding
2153 medium (such as temperature, sound speed etc.) are believed to strongly influence the
2154 behaviour and timing of the secondary shock.

2155

2156 In Rigby et al.^[133] there is no mention of the inclusion of afterburn within the numerical
2157 simulations and therefore an under prediction of the fireball properties results in addi-
2158 tional reactions being omitted and the secondary shock arriving later. The APOLLO
2159 simulations documented in Farrimond et al.^[50], use explicit afterburn which results in an
2160 early arrival of the secondary shock. This was related to inaccuracies in defined caloric
2161 EOS detonation and combustion products over predicting the speed of sound within the
2162 fireball^[179]. Whilst the arrival time of secondary shock is a known flaw in numerical sim-
2163 ulation, the loading form of the shock is captured accurately and parameters compared to
2164 experimental work for validation.

2165

2166 With the successive shock waves occurring in the wake of the primary shock, alongside
2167 any systematic experimental features within the recordings, producing a standardised and
2168 automated analytical tool for all parameters currently holds difficulty. Figures 3.20a-
2169 d present scaled secondary shock parameters, using a $TNTe = 1.22$, which have been
2170 established and compared to KB predictions for primary shock waves.

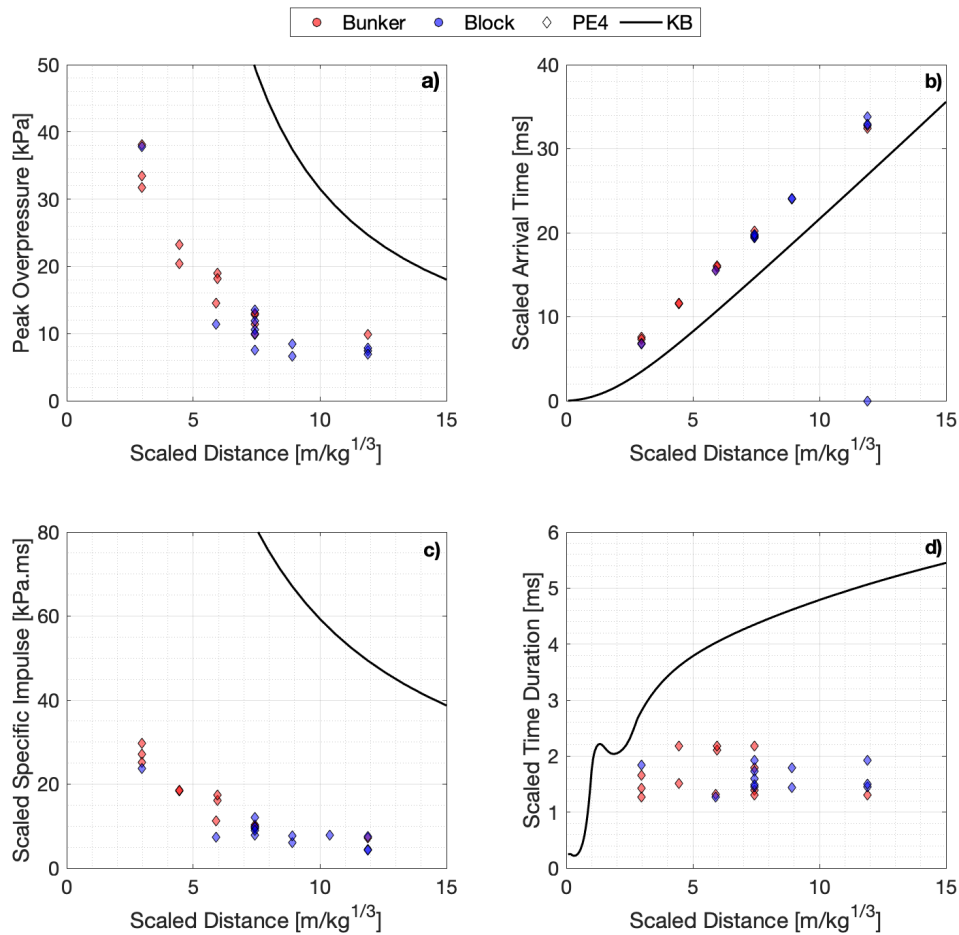


Figure 3.20: Compilation of scaled secondary shock parameters resulting from visual analysis of 250g PE4 hemispherical charge using a $TN_{Te}=1.22$: a) Negative Pressure, b) Scaled Negative Specific Impulse and c) Negative Phase Time Duration.

2171 KB predictions across the four blast parameters visually shows a lack of agreement when
 2172 scaling the PE4 secondary shock data based on the corresponding $TN_{Te}=1.22$. General
 2173 consistency within the data itself is experienced at each increment of scaled distance. This
 2174 finding was surprising based on both the difficulty in assigning these parameters within
 2175 the negative and the unpredictability of the temporal characteristics of a secondary shock
 2176 wave as it propagates through chaotic fireball and detonation products. When comparing
 2177 the result directly with those presented in Figures 3.10a-d, similar behaviour are exhibited
 2178 with respect to scaled distance suggesting that secondary shock waves present similarities
 2179 to a free-air shock wave for far-field scenarios. Future work should consider methods of
 2180 predicting secondary shock parameters through the development of similar semi-empirical
 2181 prediction curves to that presented by [87].

2182

2183 The delayed offset of $\sim 5ms$ on average in scaled arrival time of the secondary shock, when
2184 compared to the KB predictions, is displayed in Figure 3.20b. The initial conclusions to
2185 this was that it was directly linked to the time delay between a detonation wave travelling
2186 to the charge extents, a rarefaction wave collapsing towards the centre of detonation,
2187 coalescing and then reflecting back out as the secondary shock. However, this mechanism
2188 in 250g hemispherical charges would be over in around 0.01-0.02ms, when considering a
2189 constant detonation velocity of 7800m/s in PE4. The question then posed was what makes
2190 up the remaining 99% of the offset time in secondary shock arrival time. This phenomena
2191 is discussed in more detail in Section 4.2.3.1 when more data is available across a number
2192 of different explosives.

2193 3.2.4 Future Considerations for Pressure Gauge Work

2194 The main concern with current practices of pressure gauge related blast trials is the lack
2195 of systematic feature consideration when designing a testing facility and analysing any
2196 resulting data. The shock wave mechanisms of clearing and reflection, discussed in Sec-
2197 tion 2, can have detrimental effects to a pressure-time history depending on the features
2198 the analyst wants to consider. Despite best efforts, the blast arena trials conducted at the
2199 University of Sheffield are not absent of systematic features but make best attempts at
2200 reducing them, as discussed in Section 3.2.

2201

2202 Future pressure gauge work and blast trials should make more use of numerical models and
2203 create synergetic solutions to how data is captured. Through a fundamental understanding
2204 of the parameter, variations to experimental methodologies can be made to improve the
2205 accuracy and consistency in the desired blast parameter. This will push the community in
2206 the direction of a clear and coherent understanding of blast wave mechanics and achieve
2207 general consistency across the full range of scaled distances.

2208 3.2.5 Pressure Gauge Summary

2209 This section focusses on the analytical and experimental techniques which can be adopted
2210 for pressure gauge type recordings across the far-field range. Within this thesis, the ap-
2211 plication of the aforementioned methods on PE4 hemispherical detonations, demonstrate
2212 high levels of accuracy of output blast parameters from both the positive and negative

2213 phase. These have been used to validate the KB predicting tools across the far-field range.
2214 These tools will be referenced throughout the duration of this thesis to assess the validity
2215 of their application across both ideal and non-ideal explosives in the far-field range.

2216

2217 **3.3 High Speed Video (HSV) Recordings**

2218 Detailed within this section of the thesis is the discussion of how high speed video record-
2219 ings can be used to develop a more robust understanding of free-air blast parameters.
2220 Ultra high speed recordings are shown to provide both qualitative and quantitative infor-
2221 mation on the detonation wave physics, alongside the early expansion of the fireball and
2222 subsequent shock wave.

2223

2224 **3.3.1 Far-Field HSV Analysis (Ai)**

2225 As mentioned in Section 3.1.1, high speed video cameras were utilised for recording the
2226 shock wave propagation through the surrounding air medium as the result of an explosive
2227 detonation. Visual representation of an explosive trial offers a full spatial and temporal
2228 understanding of a detonation event, within the confines of camera resolution. If utilised
2229 correctly, HSV provides a much more robust understanding of shock wave development in
2230 far-field scenarios whilst reducing the number of trials required to get a similar represen-
2231 tation using pressure gauges alone. The following sections discuss the various procedures
2232 which have been iteratively developed to track shock waves within the literature, and
2233 present newly automated tools to analyse high speed video recordings of far-field explo-
2234 sive detonations.

2235 **3.3.1.1 Shock Wave Edge Detection Process**

2236 Rigby et al.^[130] presented an edge-detection process for explosive video recordings using
2237 high speed video cameras. The inbuilt MatLab function, *edge*^[21], was suitable for optical
2238 tracking of the fireball/air interface in near-field explosive analysis due to the presence of
2239 a clear contrast between the luminous fireball and the ambient air. However, when testing
2240 the modified ‘*edge*’ function for far-field scenarios the absence of a fireball in the images
2241 and low variations in light intensity across the shock front resulted in the most noticeable

2242 edges being those of the zebra board behind the shock front. This caused any analysis
2243 using this method to output alone the outlines of the background image rather than any
2244 shock wave propagation.

2245

2246 Hargather and Settles^[71] presented an image subtraction method which identifies the
2247 regions of greatest variation in pixel brightness between two consecutive frames. This
2248 method is particularly suited for situations with imagery of low contrast but with a clear
2249 schlieren disturbance^[59]. A simplified approach was developed based on this idea is pre-
2250 sented in Equation 3.1.

2251

$$N_{i,j,k} = F_{i,j,k+1} - F_{i,j,k} \quad (3.1)$$

2252

2253

2254 where N is the resulting image subtracted video file, F the recorded video file, i and j are
2255 the pixel locations in terms of columns and rows respectively, and k is the frame number.
2256 A comparison exercise was undertaken between the Hargather and Settles^[71] method and
2257 the simplified approach developed during this research in Equation 3.1. The results were
2258 nominally identical for the videos tested but the simplified approach completed much faster
2259 due to the equation being simplistic. Despite the image subtraction method being suited
2260 for low levels of lighting, weather conditions at the University of Sheffield testing site vary
2261 massively throughout the year and therefore had justification for some trial recordings
2262 yielding impractical footage for analytical purposes.

2263

2264 Whilst the method presented in Hargather and Settles^[71] has been shown to work well
2265 against a structured, repeatable background, other work using similar methods^[103] has
2266 been shown to yield accurate results with a highly variable “noisy” background. The
2267 accuracy of this method for images with low contrast backgrounds has yet to be proven,
2268 however it is envisaged that more sophisticated techniques (e.g. PIV) would be needed
2269 instead of the simple edge detection and image subtraction technique described herein.

2270

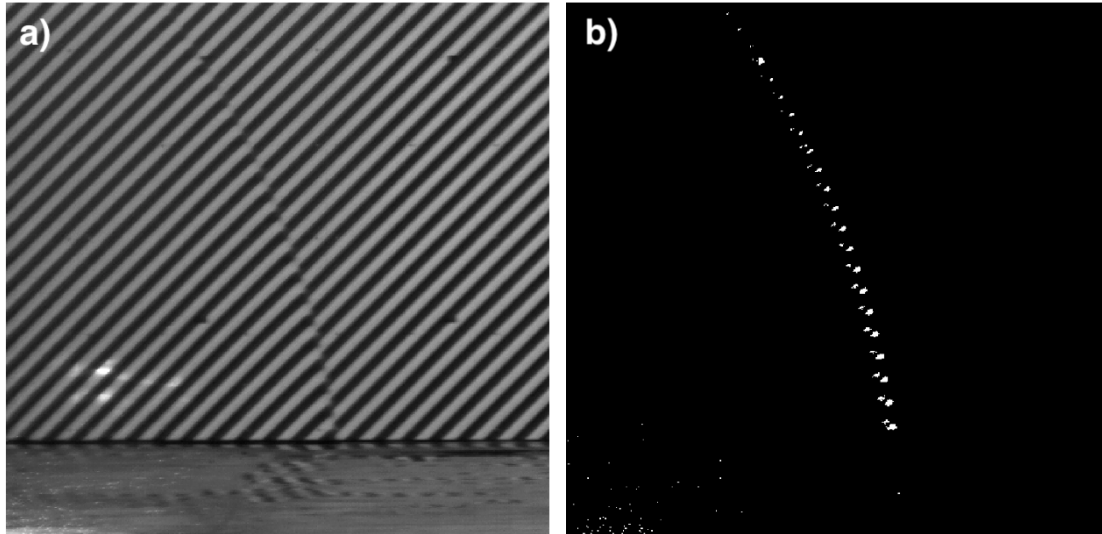


Figure 3.21: HSV stills from a test using 250 g PE4 hemisphere at 4 m stand-off: a) Raw video footage, and; b) Corresponding image subtraction

2271 Utilising the simplistic image subtraction method results in frame-by-frame outlines of
2272 the disturbances caused by the propagating shock wave as seen in Figure 3.21b. Whilst
2273 it is clear from the presented images where the spherical expansion of the shock front
2274 is generally, prescribing an actual position of the front held some ambiguity. A bespoke
2275 algorithm was written to accurately locate the propagating shock wave amongst random
2276 light variations between frames from the subtracted images which aimed at identifying
2277 large disturbance regions and assigning a specific singular pixel within said region as the
2278 position of the shock front. The process consists of three stages: *Pixel cluster identifica-*
2279 *tion*, *Intensity changes along radial spokes* and *Radial distance confirmation*, the concept
2280 of which is shown Figure 3.22.

2281

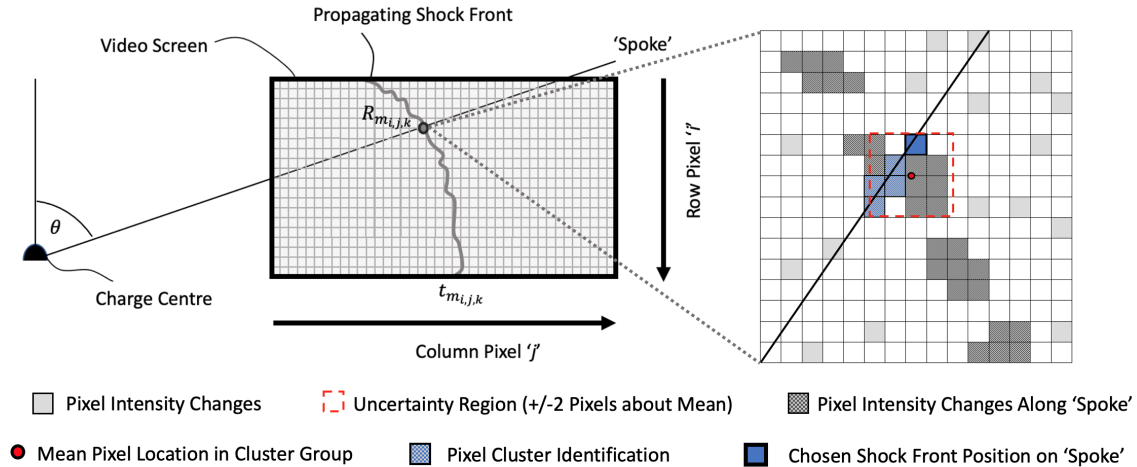


Figure 3.22: Method for detecting location of Shock Front through intensity changes on a virtual 'spoke' overlaid with pixel cluster identification

- 2282 1. **Pixel cluster identification:** After eliminating any random light variation between
 2283 subsequent frames through light intensity filtration, a clearly defined propagating
 2284 shock wave can be identified by clusters of pixels, as seen in Figure 3.21b. The
 2285 clustering size is directly related to the camera's shutter speed, i.e. the lower frame
 2286 rates resulted in an increased blurring effect and large clusters. A MatLab script was
 2287 written to locate the areas of each frame which had five or more connected pixels
 2288 with large levels of light change in the image subtracted video, N . It is anticipated
 2289 that with a higher frame rate, the number of connected pixels may reduce along
 2290 with a more defined shock front, but 5 pixels was sufficient for all far-field videos
 2291 discussed within this thesis.
- 2292 2. **Intensity changes along radial spokes:** In a similar manner to previous stud-
 2293 ies^[130], the second stage of the algorithm discretised the camera resolutions domain
 2294 into 1° increments along virtual 'spokes' which span from the charge centre to the
 2295 extents of the recording boundary. The location of any pixels which have light in-
 2296 tensity changes along each 'spoke' are stored for each increment of angle and each
 2297 frame. Due to the explosive charge being out of shot in each recording, the location
 2298 was approximated through scaling-based calculations using the size of a pixel in the
 2299 plane of the charge and known geometric relationships between the charge location,
 2300 zebra board, camera and the pressure gauge in field of view, following procedures
 2301 outlined by Gerasimov and Trepalov^[60] and Kucera et al.^[93]. Fiducial markers on
 2302 the zebra board, spaced 1.23 m horizontally and 1.33 m vertically, were used to cali-

2303 brate the corresponding pixel size of ~ 3 mm for video resolution of 1024x560, which
 2304 was consistent across all far-field videos discussed.

2305 **3. Radial Distance Confirmation:** The pixel locations saved as a result of stage one
 2306 and two were then compared directly to automate the shock front locating algorithm
 2307 along each ‘*spoke*’ and remove any random light variation between frames. A MatLab
 2308 script was written to compare stages one and two with an uncertainty region of ± 2
 2309 pixels about the mean pixel location of the stage one results. This presented all
 2310 the pixels which satisfied both stages of the analysis. The difficulty then came with
 2311 assigning a distance to the regions of pixels which have been assigned in 3D space
 2312 from a 2D image. Due to the spherical expansion of the shock wave and a fixed
 2313 camera location, considerations had to be made to account for the camera recording
 2314 the front of the shock wave in a different position to where the front would be along
 2315 the gauge line used as the reference for calibration, as seen in Figure 3.23. To find
 2316 the radial distance a given pixel is away from the charge, the locations of the camera,
 2317 explosive and shock wave pixel location on the zebra board, denoted by subscript
 2318 i , j and k respectively, were set up with coordinates in 3D space, x , y and z . The
 2319 distances between each of the three locations were calculated using Equations 3.2,
 2320 3.3 and 3.4, where a is the distance between the explosive and the camera, b is the
 2321 distance between the explosive and pixel’s relative location on the zebra board, and
 2322 finally c is the distance between the camera and the pixel’s relative location on the
 2323 zebra board (again see Figure 3.23 for reference).

$$2324 \quad a = \sqrt{(x_i - x_j)^2 + (y_i - y_j)^2 + (z_i - z_j)^2} \quad (3.2)$$

$$2325 \quad b = \sqrt{(x_j - x_k)^2 + (y_j - y_k)^2 + (z_j - z_k)^2} \quad (3.3)$$

$$2326 \quad c = \sqrt{(x_i - x_k)^2 + (y_i - y_k)^2 + (z_i - z_k)^2} \quad (3.4)$$

2327
 2328 The distances evaluated from the above outline algorithm are then used to calculate the
 2329 area of the triangle, A , enclosed between the three points in 3D space using Heron’s for-
 2330 mula (Equations 3.5 and 3.6). Assuming that the shock wave is expanding spherically,
 2331 and the line between the camera and the shock wave pixel location is a tangent to the
 2332 sphere, and therefore the radius, r , of the shock front would be equal to the height of the

2333 triangle, abc .

2334

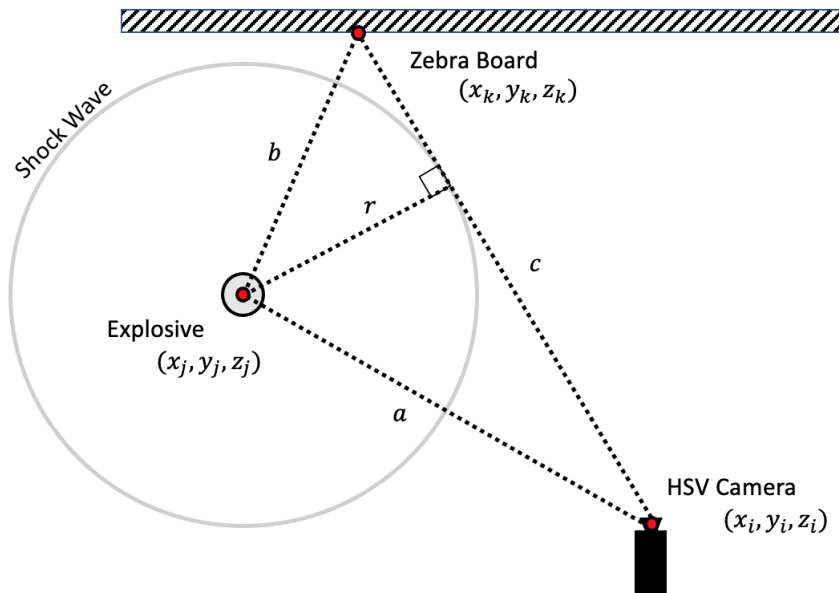


Figure 3.23: Schematic of shock wave radius calculation in 3D space to be used in conjunction with Equations 3.2–3.6

2335 After calibrating, the pixel furthest away from the charge centre was identified as the
 2336 position of the shock front along a particular ‘spoke’, at a given time. This was done
 2337 to achieve test-to-test consistency regardless of the frame rate and associated blurring.
 2338 It is important to note that although strict experimental procedures were in place, focal
 2339 lengths were difficult to record as a manual zoom was used. Despite the levels of zoom
 2340 being consistent, lens distortion effects subsequently were not accounted for during each
 2341 test and thus would cause minor variations between videos recording different stand-off
 2342 distances. Previous research has suggested that lens distortion results in minimal levels
 2343 of uncertainty in pixel size calibration^[136] but across a large testing pad could begin to
 2344 induce some variation between different test setups. The presented within this thesis for
 2345 far-field trial minimised the severity of lens distortion by comparing only like-for-like tests
 2346 which would therefore remove the uncertainty associated with recordings made at the ex-
 2347 tents of a camera resolution for one standoff which corresponds to a more central region
 2348 for a different standoff tested.

2349

$$A = \sqrt{s(s-a)(s-b)(s-c)} \quad (3.5)$$

$$s = \frac{a+b+c}{2} \quad (3.6)$$

$$r = \frac{2A}{c} \quad (3.7)$$

2350

2351

2352 3.3.1.2 Example of Edge Detection Algorithm

2353 The results from the edge detection algorithm for a single test (250g PE4 hemisphere at
2354 5m stand-off) are presented in Figure 3.24. KB predictions are also presented on the afore-
2355 mentioned figure, again using Hopkinson-Cranz scaling and a TNT equivalence of 1.22 to
2356 compare the raw data directly to an equivalent prediction. The high levels of agreement
2357 highlights the repeatability and accuracy in the results from the presented automated im-
2358 age tracking and arrival time interpolation algorithms discussed throughout the section
2359 and therefore validates them for use across all other far-field high speed video recordings
2360 of shock front propagation. When comparing Figures 3.10a and 3.24, the full-field data
2361 extracted from a single video gives a clear justification for the use of HSV over pressure
2362 gauges when studying shock wave arrival time.

2363

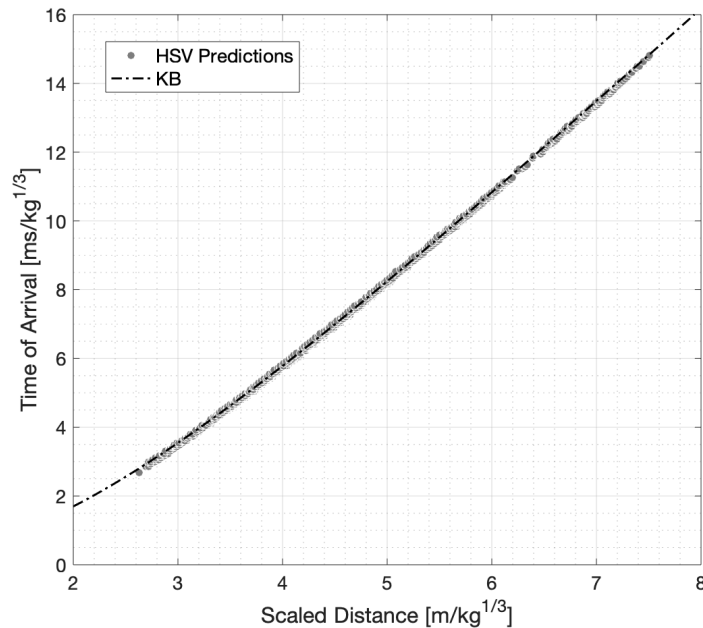


Figure 3.24: Shock front arrival time from a 250g PE4 hemispherical charge placed at a 5m stand-off, recorded using HSV and scaled using 1.22 TNT equivalence factor to compare to KB predictions

2364 The results from the edge detection analysis of the primary shock, presented in Figure 3.24,
2365 provides confidence in the methodology introduced which could be utilised for all explo-
2366 sives tested in the far-field testing regime. Unfortunately for the PE4 trials tested, the
2367 technique was in development and therefore the critical parameters required to achieve
2368 accurate credible data were not yet known. Despite the method being suited for low levels
2369 of lighting, when weather conditions were unfavourable, videos yielded data which exhibit
2370 unclear shock discontinuities and therefore were omitted from detailed analysis. Provided
2371 the lighting conditions during testing are ideal, the analysis presents clearly defined edges
2372 of the shock with less noise from over-exposing the video. In certain instances when the
2373 conditions were preferable, the analysis was able to extract secondary shock wave data
2374 which was visible to the naked eye providing another benefit for this analytical method;
2375 this will be discussed in Chapter 4.3.3.

2376

2377 3.3.1.3 Shock Wave Arrival Time Interpolation

2378 Blast parameters such as peak incident and reflected pressure can be linked to shock
 2379 velocity (Mach number) through the well-known Rankine-Hugoniot jump conditions^[90].
 2380 There is motivation, therefore, to understand how blast wave arrival time (and thus Mach
 2381 number) varies with changing distance from the centre of the explosive, as this will pro-
 2382 vide insights into the variability of pressure and impulse parameters and the implications
 2383 therefore on designs.

2384

2385 A method, presented visually in Figure 3.25, was proposed to interpolate arrival time data
 2386 at increments of $0.1 \text{ m/kg}^{1/3}$ scaled distance along each virtual ‘spoke’ for a given far-field
 2387 shock wave tracking HSV recording. In doing this, a direct comparison could be made
 2388 between pressure gauge recordings of arrival time, with discrete scaled distances, and high
 2389 speed video recordings with discrete increments of time between each frame but a variety
 2390 of processed scaled distances using the methods detailed in Section 3.3.1.1.

2391

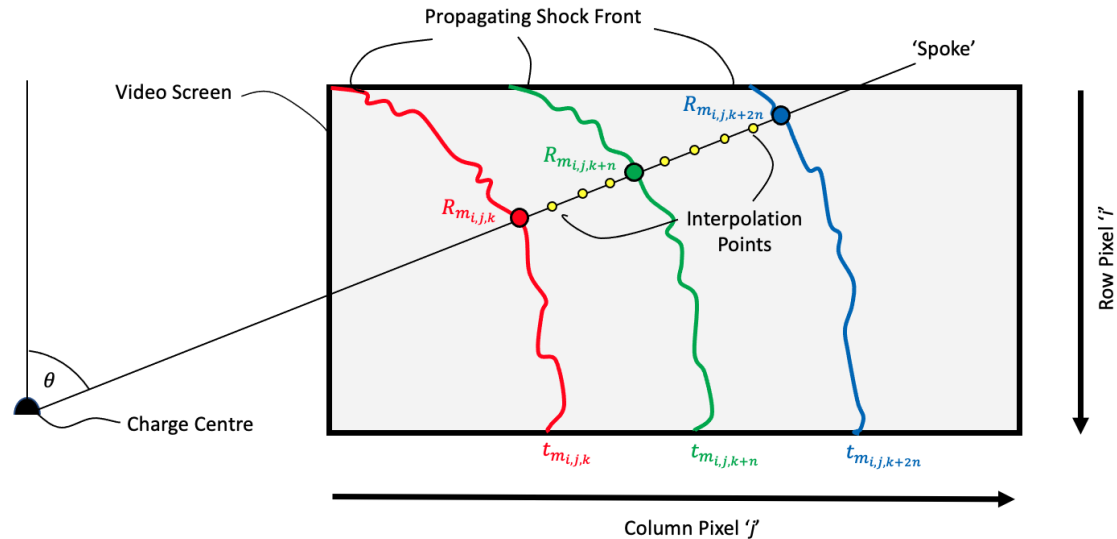


Figure 3.25: Method of interpolation of shock front arrival times along a virtual ‘spoke’ for prescribed scaled distances

2392 Through collecting the interpolated arrival times with respect to scaled distance along
 2393 each radial spoke, different types of analysis become available for comparing the results
 2394 from HSV trials directly to those corresponding the pressure gauge trials. The synergy

2395 between two testing methodology provides validation of the each data acquisition and
 2396 analytical method adopted alongside providing both precise measurements from discrete
 2397 individual gauges and more generalised full-field representation using camera footage.

2398

2399 3.3.1.4 Extrapolating Blast Wave Features

2400 An important thing to note, is that the techniques discussed in Section 3.3.1 can be used
 2401 to evaluate a more robust understanding of other features of free-air blast waves, provided
 2402 that the experimental conditions are favourable. The tools aforementioned could be im-
 2403 plemented to track not just the primary wave propagation but all subsequent waves and
 2404 reflections. The ability of tracking secondary shocks is purely down to the sharpness of the
 2405 visible propagation on the video recording, improved providing the lighting is adequate.
 2406 This data extraction can be used to develop the understanding of the secondary shock
 2407 phenomena discussed in Section 3.2.3.2 when acquired through pressure gauge recordings.

2408

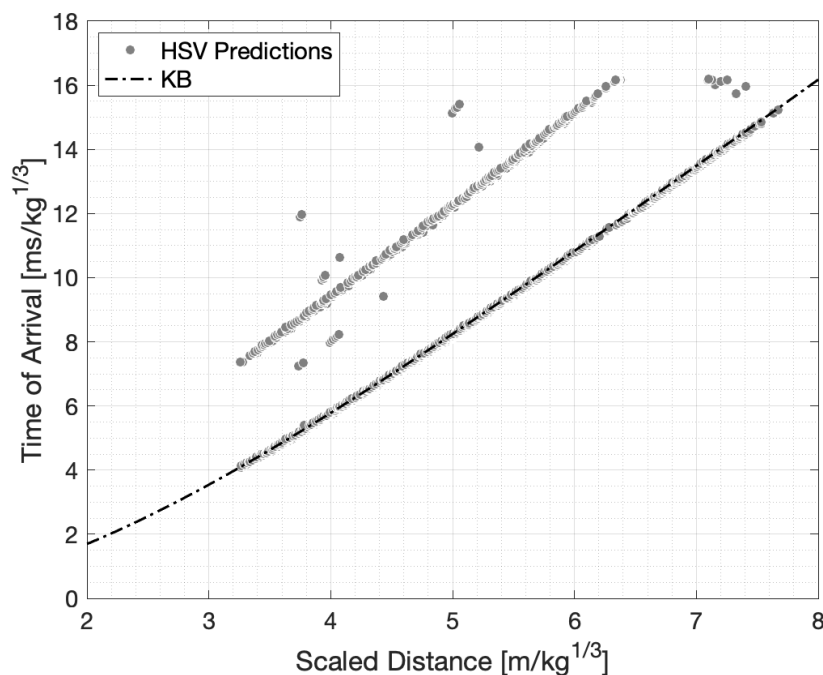


Figure 3.26: Shock front arrival time from a 250g PE₄ hemispherical charge placed at a 5m stand-off, recorded using HSV and scaled using 1.22 TNT equivalence factor to compare to KB predictions, detailing the capture of the secondary shock propagation.

2409 Figure 3.26 is an example of extracting subsequent shock data from a 250g hemispherical

2410 PE4 charge placed at a 5m standoff distance. It is important to note that despite the
2411 clear trends of both primary and secondary shock propagation there are other data points
2412 which disagree with the data. The cluster of data points at the extents of the camera
2413 view, $Z > 7\text{m}/\text{kg}^{1/3}$ in this test, is a result of the primary wave reflecting back towards the
2414 charge centre detailed by a reduction in distance travelled with time. The reflected wave
2415 propagation was not of interest for the contents of this thesis and so improvements to the
2416 tracking algorithm was not made to account for this change in direction of the wave. The
2417 other anomalous results which appear around the secondary shock data points are analyt-
2418 ical errors associated with the light intensity of pixels during the image subtraction being
2419 more variable for weaker shocks. As mentioned above, with preferable lighting conditions,
2420 the HSV extracted data for both primary and subsequent shocks will exhibit less scatter.
2421

2422 **3.4 Future Considerations for HSV Work**

2423 **3.4.1 Far-Field Methodology Errors (Ei)**

2424 Despite the clear capability of the presented HSV data capture techniques and analysis,
2425 there are still concerns with the accuracy due to systematic errors which have yet to
2426 be fully considered and mitigated from the current experimental practice. Figure 3.27
2427 displays the results of implementing the shock wave arrival time tracking algorithm, dis-
2428 cussed in Section 3.3.1.1, for a nominally identical trial to that presented in Figure 3.24
2429 - a 250g PE4 hemispherical detonation at 5m standoff from a reflected surface. It is im-
2430 portant to note during this trial, an arbitrary intentional but non-obvious skew of the
2431 camera's field of view was made during the experimental setup away from the charge to
2432 assess the sensitivity of the analytical techniques to systematic errors in camera placement.
2433

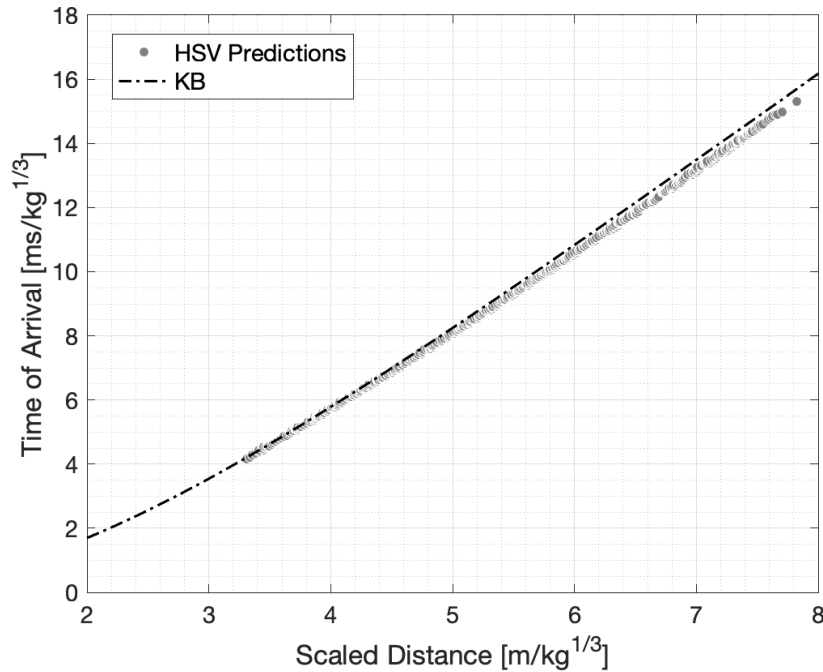


Figure 3.27: Shock front arrival time from a 250g PE4 hemispherical charge placed at a 5m stand-off, recorded using HSV and scaled using 1.22 TNT equivalence factor to compare to KB predictions, showing the effect of having testing procedure not perfectly square.

2434 The systematic rotation in any direction would skew the number of pixels contained be-
 2435 tween the reference markers used to calibrate each frame of the video. This causes it to
 2436 appear a smaller distance than what it would be if all equipment was perpendicular as
 2437 per the analytical techniques assumptions. If a lower number of pixels is measured for the
 2438 same known distance, each individual pixel size would increase in value. As the lengths
 2439 calculated at each stage of the analytical process are directly proportional to these pixel
 2440 sizes, a reduction in size would result in everything appearing to travel faster than a real
 2441 shock wave. This is visible from the behaviour presented between the experimental data
 2442 and KB predictions within Figure 3.27. With this knowledge, a reduction is required to
 2443 the calibrated pixel size to account for the skew in the experimental setup.

2444

2445 In doing so, a skewed angle between the camera and the zebra board of approximately 2°
 2446 was evaluated. This value of angular rotation was used to investigate the sensitivity of
 2447 evaluated shock wave positions when only minor tweaks to assumed camera positions are
 2448 made. The magnitude of this rotation is minor but across the distances covered during this
 2449 experiential setup accounts for almost 750mm additional distance theoretically assigned

2450 hence the shock wave tending to arrive sooner than KB predictions in the experimental
 2451 data.

2452

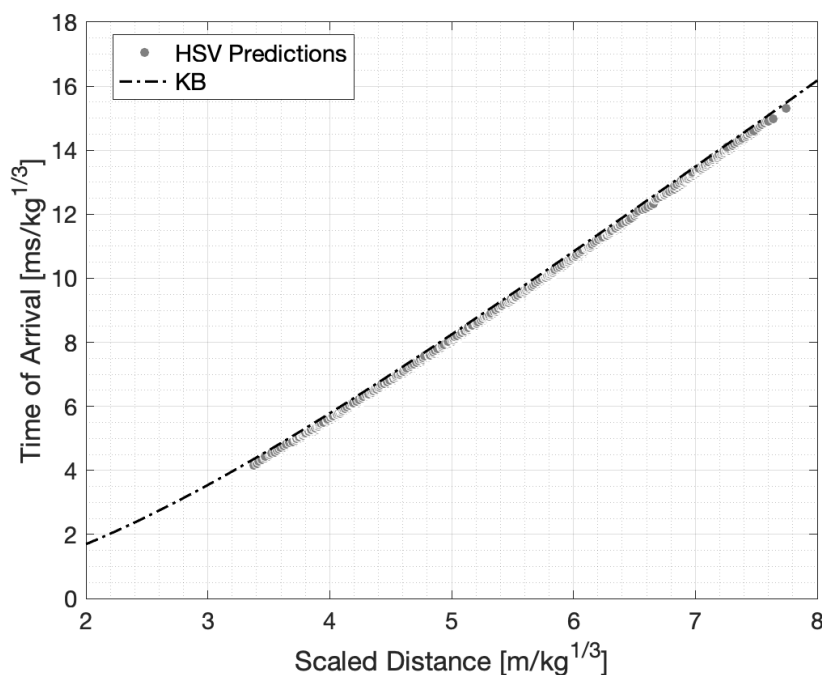


Figure 3.28: Re-analysed data from Figure 3.27 with an adjustment to the calibrated pixel sized to account for the skewing of the FOV during camera rotation.

2453 The HSV data when adjusted for this skew presented a more reasonable comparison to
 2454 KB predictions with an offset in the data. This generally suggests the shock wave has
 2455 travelled further than predictions in the same amount of time, as seen in Figure 3.28.
 2456 This has been deduced to be an effect of the prescribed initial boundary conditions set
 2457 for the theoretical charge location, not visible within the video but is known through
 2458 real-world measurement. The distance to which the camera's FOV extends is calculated
 2459 using the visible pressure gauge position and the pixel size established during calibration.
 2460 This distance was then subtracted from the known standoff to assign the location of the
 2461 theoretical charge with respect to the pressure gauge in the respective video. Knowing
 2462 the pixel size is larger during the skewed test means that the location of the explosive off
 2463 screen has been re-defined as being closer to the gauge wall than what it is in reality. This
 2464 is why generally a faster arriving shock wave is established.

2465

2466 It is essential to ensure the correct positioning of all initial reference markers and bound-

2467 ary conditions prior to conducting any future far-field HSV recordings. Despite the large
2468 quantity of data which can be extracted from HSV recordings, if the positions are not
2469 in-line with assumption made in the analytical procedures then the results will be inaccur-
2470 ate. The time committed to setting up cameras with high levels of precision is significant
2471 but the gain in data quantity in comparison to that collected during pressure gauge trials
2472 outweighs this.

2473

2474 This is just one of many errors which could result in strange behaviours of the analyt-
2475 ically established arrival time data which should be consistent with KB predictions in
2476 far-field regimes. Ensuring the assumptions made in the analytical procedures align with
2477 the actual experimental setup clearly holds importance for data accuracy. This finding is
2478 linked with that discussed in Section 2.9 where the clear disregard for systematic errors
2479 in methodologies and analysis have resulted in published articles suggesting far-field blast
2480 parameters exhibit inherent variability. In actual fact, if these errors are accounted for,
2481 nominally identical data is consistent and is highly comparable to KB predictions.

2482

2483 3.4.2 Extracting More Parameters (Ai)

2484 The strength in using high speed video techniques for experimental trial data extraction
2485 is the vast quantity obtained from a single trial. Despite work being undertaken relating
2486 arrival time data to pressure within the available published literature using Rankine-
2487 Hugoniot principles, there is still a gap in extracting the specific impulse of a shock wave
2488 using optical methods.

2489

2490 Murphy and Adrian^[104] looked at developing a particle image velocimetry (PIV) system
2491 which measures velocity fields as a result of detonating explosives. The method was effec-
2492 tive in far-field scaled distance ranges when comparing to assumed triangular blast wave
2493 profiles, but struggled in near-field instability and turbulence regions. Hargather and Set-
2494 tles^[70] made efforts at using optical techniques to measure and attempt to scale explosive
2495 parameters in the gram-range. Whilst the authors made efforts at characterising arrival
2496 time and pressure through Rankine-Hugoniot conditions and time duration through the-
2497 oretical assumptions proposed by Kinney and Graham^[90], no link was made to specific
2498 impulse of the shock wave.

2499

2500 The current background orientated schlieren methodology used for far-field HSV data cap-
2501 ture at the University of Sheffield could be developed to incorporate PIV analysis. To do
2502 so, it is essential for the far-field recordings to be calibrated to account for lens distortion
2503 and any further experimental errors associated with camera orientation which can be done
2504 so effectively using MATLAB's in-built camera calibration software. Once calibrated, an
2505 interesting development of this method would be to assess the velocity field measured
2506 using PIVlab, another in-built MATLAB GUI, in the attempt assign the time at which
2507 the positive phase ends for a propagating shock wave. Adopting a triangular blast wave
2508 profile approximation, the pressure at a given position, evaluated through arrival time,
2509 and the time at which positive phase ends, an estimation of specific impulse of the shock
2510 wave can be achieved at that given point.

2511

2512 **3.5 Summary**

2513 The main body of established blast literature covers yield results of idealised high ex-
2514 plosives when detonated in various scenarios. As mentioned throughout Chapter 3, the
2515 variety in conclusive findings does pose difficulty for truly understanding the fundamen-
2516 tals of blast wave phenomena alongside the associated inherent variability of recorded data.

2517

2518 The real-world implications of this lack of clarity is the uncertainty of both the effects
2519 an explosive threat yields alongside the confidence in protective measures to withstand
2520 the given threat. Engineers face a constant battle of providing improved efficiency and
2521 aesthetics in design, both of which may result in a reduction in the intrinsic robustness
2522 of a structure, thereby increasing the need for considered and holistic blast protection
2523 measures. Whilst ever there is a major disagreement between the consistency of yields
2524 resulting from a given explosive, there will always be a lack of confidence in protective
2525 structures implemented and therefore massive safety factors applied to ensure the upper
2526 bound of quoted variability is accounted for.

2527

2528 The need for a precise intrinsic quantification and characterisation of prescribed blast
2529 scenarios will enable better probabilistic assessments of threats which can be taken into
2530 consideration with more confidence at design stages of any protection regime. The idea
2531 that extrinsic sources of variability are included in experimental work used for numeri-

2532 cal model validation is concerning when considering that some article quote spreads in
2533 pressure and impulse $\pm 50\%$. This uncertainty should be prescribed at a later stage when
2534 considering the likely threats rather than including extrinsic error through poorly executed
2535 research.

2536

2537 This chapter has focussed on developing validated methods of experimental and analytical
2538 capabilities which result in consistent data from nominally identical trials. A detailed
2539 review of historical blast data has been presented which opens up the discussion, in line
2540 with Objective 1 of this thesis, of whether recorded blast parameters are in fact deter-
2541 ministic across a range of scaled distances. The historic data provides confidence in the
2542 consistency in explosive yields across two well documented compositions used during the
2543 high explosive era, but more recent research has highlighted variability regions as high as
2544 $\pm 130\%$.

2545

2546 This chapter has scrutinised experimental methodologies adopted during free-air blast
2547 loading trials and the subsequent data analysis with the aim of providing validated tech-
2548 niques and procedures. This can provide data which facilitates development in blast wave
2549 theory as opposed to causing further confusion within the industry as to whether explo-
2550 sions are inherently variable or not. The presented techniques in this chapter have shown
2551 that for far-field scenarios, $Z > 3\text{m}/\text{kg}^{1/3}$, blast parameter variability can be minimised
2552 and is therefore considered deterministic regardless of early-stage detonation mechanics.
2553 The techniques have however highlighted the need for strict experimental procedures, espe-
2554 cially when considering near-field loading conditions as these are highly sensitive to spatial
2555 variation of the energy release upon detonation. The remainder of this thesis will focus
2556 primarily on far-field studies to characterise the effects of a variety of different explosive
2557 compositions across a range of scaled distances.

2558 Chapter 4

2559 Ideal Explosive Blast

2560 Characterisation

2561 4.1 Introduction

2562 Considering the established hypothesis in Chapter 3, in that explosive yields are in fact
2563 deterministic when experimental methodologies adopt strict experimental control and au-
2564 tomated analytical tools, a variety of different ideal high explosives have been examined
2565 to simultaneously further validate the tools presented in the aforementioned chapter. The
2566 aims of these analyses are to improve data consistency across a range of different ideal
2567 explosives and to quantify the levels of variability each blast parameter exhibits across a
2568 range of scaled distances.

2569

2570 Through conducting well-controlled trials across a number of ideal explosives and scaled
2571 distances a comprehensive understanding of definitive variability regions, in line with the
2572 hypothesis proposed by Tyas^[173], can be established. When comparing the extracted data
2573 set independently and to semi-empirical predictions, developments to the fundamental un-
2574 derstanding of explosive air shock and confidence in KB predictions and TNT equivalence
2575 can be established for different ranges of scaled distance.

2576

2577 4.2 Pressure Gauge Data

2578 In similar fashion to Chapter 3, this section will consider the data captured using pres-
2579 sure gauges during experimental trials. This will be used in the attempt to both validate
2580 the on-going hypothesis of far-field blast parameters being deterministic and to assess the
2581 capability of the methods developed for analysis across a number of ideal explosives.

2582

2583 4.2.1 Positive Phase Blast Parameter Variability

2584 This section will evaluate the explosive output of three ideal high explosives; PE4 (nomi-
2585 nally 88% RDX with 12% Plasticiser/Taggant), PE8 (nominally 86.5% RDX with 13.5%
2586 Plasticiser/Taggant) and PE10 (nominally 86% PETN with 14% Plasticiser/Taggant),
2587 using tools established in Chapter 3. The choice of explosives used for the assessment
2588 techniques was based on their usage within a military setting, ease of acquisition through
2589 UK energetics vendors and predominantly due to the vast amount of historical testing
2590 undertaken at the University of Sheffield.

2591

2592 PE4 and PE8 are effectively the same explosive, with similar RDX and binder percentages,
2593 despite the variation in their binder compositions. The former explosive, PE4, was de-
2594 signed to match the American standard military grade explosive of C4 which is confirmed
2595 in the findings presented by Bogosian et al.^[15], Rae and McAfee^[115]. PE8 was a later
2596 developed RDX based explosive which was designed to supersede PE4, therefore providing
2597 no change for the end users of the explosive. PE10 again was designed to provide a similar
2598 explosive yield to PE4 but was PETN based rather than RDX offering a different explosive
2599 composition. The reasoning behind the development of this explosive was two fold: firstly
2600 to work better in colder environments; and secondly due to RDX being recognised as a
2601 problematic and bioavailable explosive which has tendencies to infiltrate watercourses^[171].

2602

2603 With all the above mentioned, the results of any trials adopting the same methodology
2604 and data analysis for these three explosives is hypothesised to yield the same blast param-
2605 eters. This was assumed true for far-field free-air scenarios, where any afterburn related
2606 to secondary chemical reactions ongoing in the fireball minimally affects the overall pos-
2607 itive phase pressure-time history. The results of the far-field tests will be assessed and

2608 compared against both semi-empirical predictions and high-fidelity numerical modelling.
2609 This comparison had the overall aim of demonstrating that far-field blast parameters,
2610 $Z > 3\text{m}/\text{kg}^{1/3}$, can essentially be considered deterministic, whilst providing an example
2611 of the synergistic nature between the two.

2612

2613 4.2.1.1 Intrinsic Data Set Variability

2614 To assess the intrinsic variability in a given data set, each positive phase blast parameter
2615 has been normalised against the mean value of that particular parameter for nominally
2616 identical recordings. By adopting this analysis approach, the consistency in the data set
2617 can be deduced and can help determine as to whether the working hypothesis of deter-
2618 ministic blast parameters holds credibility across a range of scaled distances and explosive
2619 types. The analysis considered the three explosives mentioned above to determine the
2620 inherent variability in the explosive yield of three ideal high explosives when adopting val-
2621 idated methods of conducting experimental trials and data analysis discussed in Chapter 3.

2622

2623 Figures 4.1a-d represent the findings from the initial intrinsic data set variability assess-
2624 ment of each positive phase blast parameter after adopting experimental procedures and
2625 analysis as detailed in Chapter 3. Notably what can be seen is a high level of repeatability
2626 in the processed positive phase blast parameters across all three explosives and scaled
2627 distances considered.

2628

2629 Reflected peak pressure, peak specific impulse and positive phase duration exhibit similar
2630 levels of spread across the entire range of scale distances tested; typically around $\pm 6\text{--}8\%$,
2631 with a slight reduction in consistency as scaled distance reduces seen in Figures 4.1a, 4.1b
2632 and 4.1d.

2633

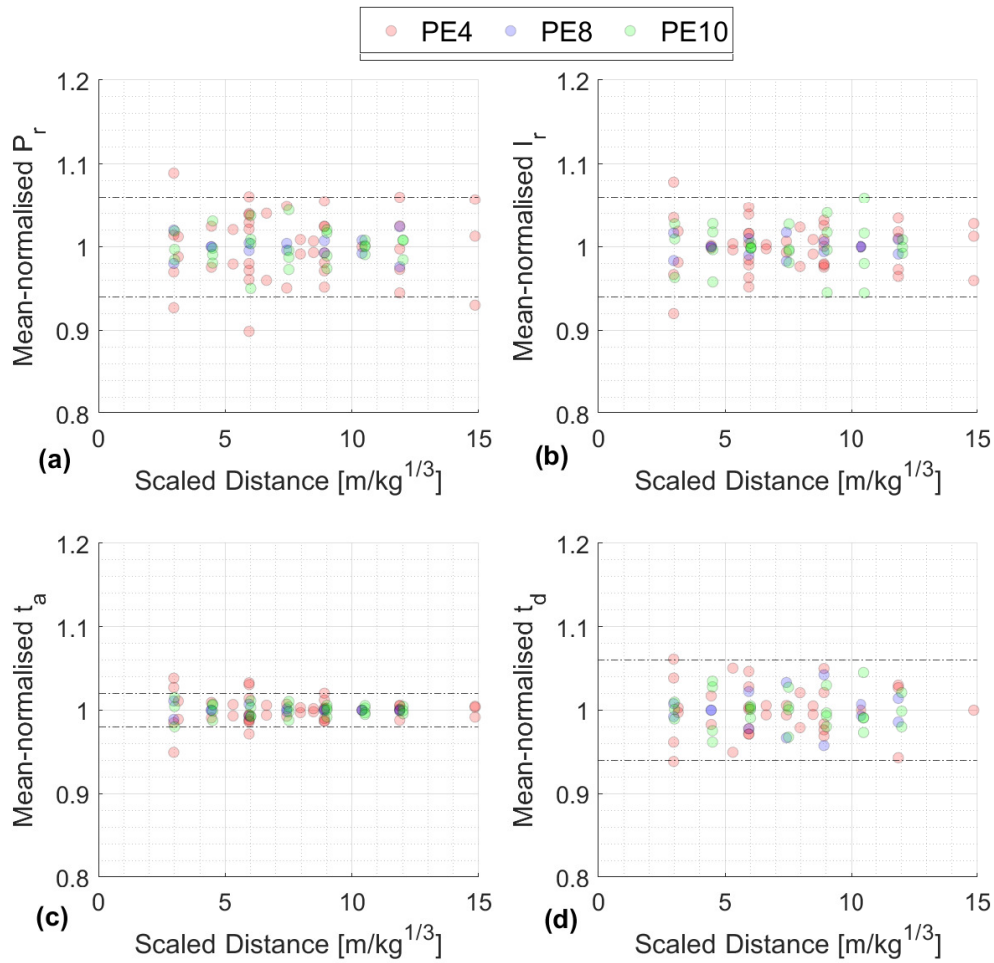


Figure 4.1: Compiled blast parameters from RDX and PETN based explosive trials as a function scaled distance, normalised against the mean of nominally identical trials: a) Peak reflected overpressure (Mean-Normalised P_r), b) Scaled reflected peak specific impulse (Mean-Normalised I_r), c) Scaled arrival time (Mean-Normalised t_a), d) Scaled positive phase duration (Mean-Normalised t_d)

2634 The arrival time of the shock wave, in Figure 4.1c, exhibits a considerably smaller spread,
 2635 within $\pm 2\%$ when $Z > 3 \text{ m/kg}^{1/3}$, which is to be expected for a parameter of generally
 2636 much lower magnitudes. Again, a noticeable increase in variability as scaled distances
 2637 reduces in line with findings presented in Rigby^[127].

2638

2639 The consistency of the experimental results presents a clear indication that for small-scale,
 2640 far-field, geometrically simple scenarios, the blast positive phase parameters are essentially
 2641 deterministic with quantifiable but limited levels of variability. Despite only a few data
 2642 points at each scaled distance, there is a clear reduction in arrival time variability as
 2643 scaled distance increases which holds true for all three explosive types. Although not as

2644 evident across the other blast parameters, a reduction is present, which agrees with the
2645 idea proposed by Stoner and Bleakney^[160] and Bogosian et al.^[16] that variability levels
2646 differ with scaled distance.

2647

2648 The results of this analysis provides enough evidence to suggest that as the near-field is
2649 approached, $Z < 3\text{m/kg}^{1/3}$, a much greater spread in the arrival time data, and thus
2650 other blast parameters, is observed which agreed with findings presented by Simoens and
2651 Lefebvre^[153]. This not only provides evidence to the hypothesis of scaled distance regions
2652 over which fireball surface instabilities are prominent, as discussed by Tyas et al.^[174], Rae
2653 and McAfee^[115] and Rigby et al.^[130], but when compared with other published works
2654 on blast variability starts to build a more robust understanding of the development of
2655 explosive shock fronts.

2656

2657 4.2.1.2 Comparison Against Semi-Empirical Predictions

2658 To assess the robustness of the KB prediction method and the accuracy of the data pre-
2659 sented within this thesis, a Mean Absolute Error (MAE) analysis was undertaken between
2660 the results from each explosive and those predicted for varying quantities of TNT. This
2661 analysis was done to establish the best possible TNT equivalence value to represent each
2662 explosive. It is expected that due to the three explosives in question being designed to
2663 result in similar explosive yields on detonation, that the TNT equivalence values for far-
2664 field free air scenarios should be similar across all explosives tested.

2665

2666 The results presented in Figure 4.2d shows that generally speaking the positive phase du-
2667 ration tends to hold higher levels of variability due to the difficulty in assigning a specific
2668 value to the parameter. This is due to noise in the signal making the true point at which
2669 conditions return briefly back to atmospheric difficult to determine. It is further deduced
2670 that experimentally recorded results are around 90% of KB predictions which would cause
2671 any MAE analysis for a TNT equivalence value to be skewed. Thus, positive phase dura-
2672 tion has been omitted from the considered MAE analysis.

2673

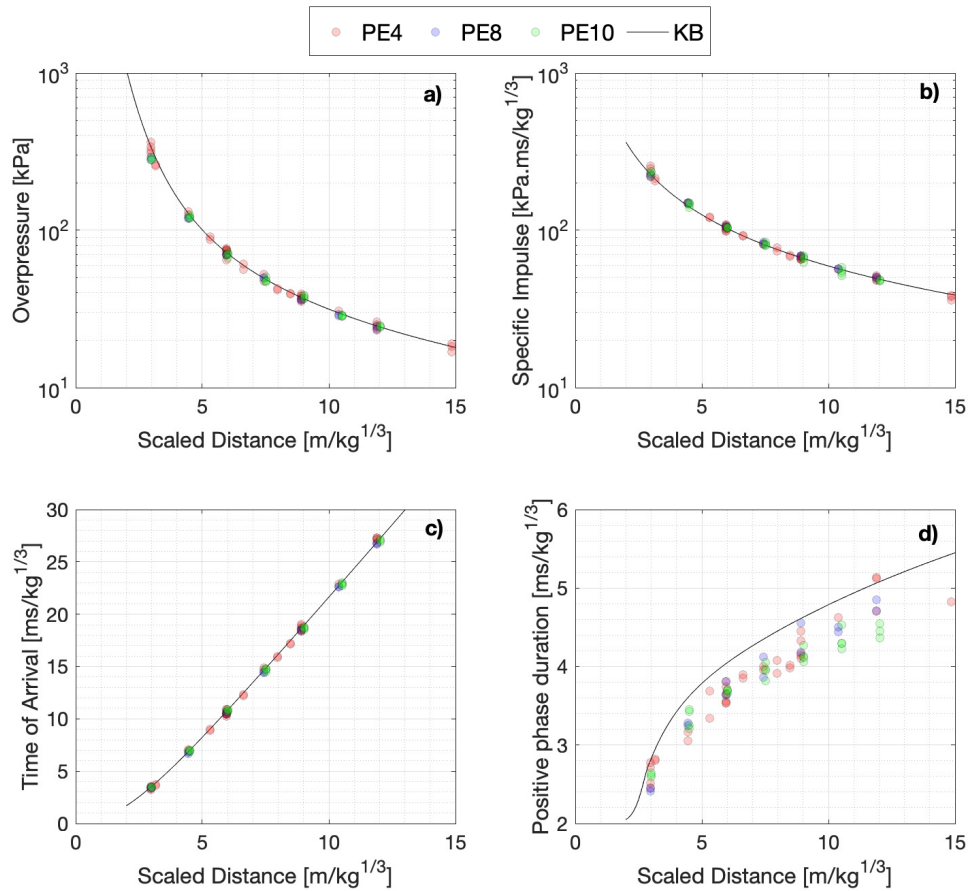


Figure 4.2: Compiled blast parameters from RDX and PETN based explosive trials as a function scaled distance, compared with KB predictions: a) Peak reflected overpressure, b) Scaled reflected peak specific impulse, c) Scaled arrival time, d) Scaled positive phase duration

2674 The analysis was performed for definitive far-field scenarios, $Z > 3\text{m}/\text{kg}^{1/3}$, in light of
 2675 the fact that for smaller scaled distances the variability levels rise. This believed to be
 2676 a result of not only systematic experimental errors but also ongoing chemical reactions
 2677 in the fireball resulting in fluid dynamic instabilities which differ from composition to
 2678 composition. Therefore the decision was made that evaluating a TNT equivalence value
 2679 for only the free-air shock propagation in the far-field was the philosophy of this analysis.
 2680 This was undertaken through comparing the scaled mass equivalent shock produced from
 2681 the explosives in question to KB predictions for 1kg of TNT.

2682

2683 It was found through averaging of the MAE values for pressure, specific impulse and ar-
 2684 rival time at all far-field scaled distances, the three explosives all resulted in very similar
 2685 TNT equivalency factors: PE4 and PE10 resulting in an equivalence of 1.22; and PE8

2686 resulting in an equivalence of 1.24. This was a further reassurance and validation in both
 2687 the experimental and data analysis techniques adopted for these trials as the three explo-
 2688 sives were designed to have the same explosive yield. The TNT equivalence factors were
 2689 applied to each of the corresponding compositions and subsequent recorded blast param-
 2690 eters presented in Figures 4.2a-d. The scaled results show a striking agreement between
 2691 each blast parameter (expressed as a 1kg hemispherical TNT charge), and KB predictions.
 2692

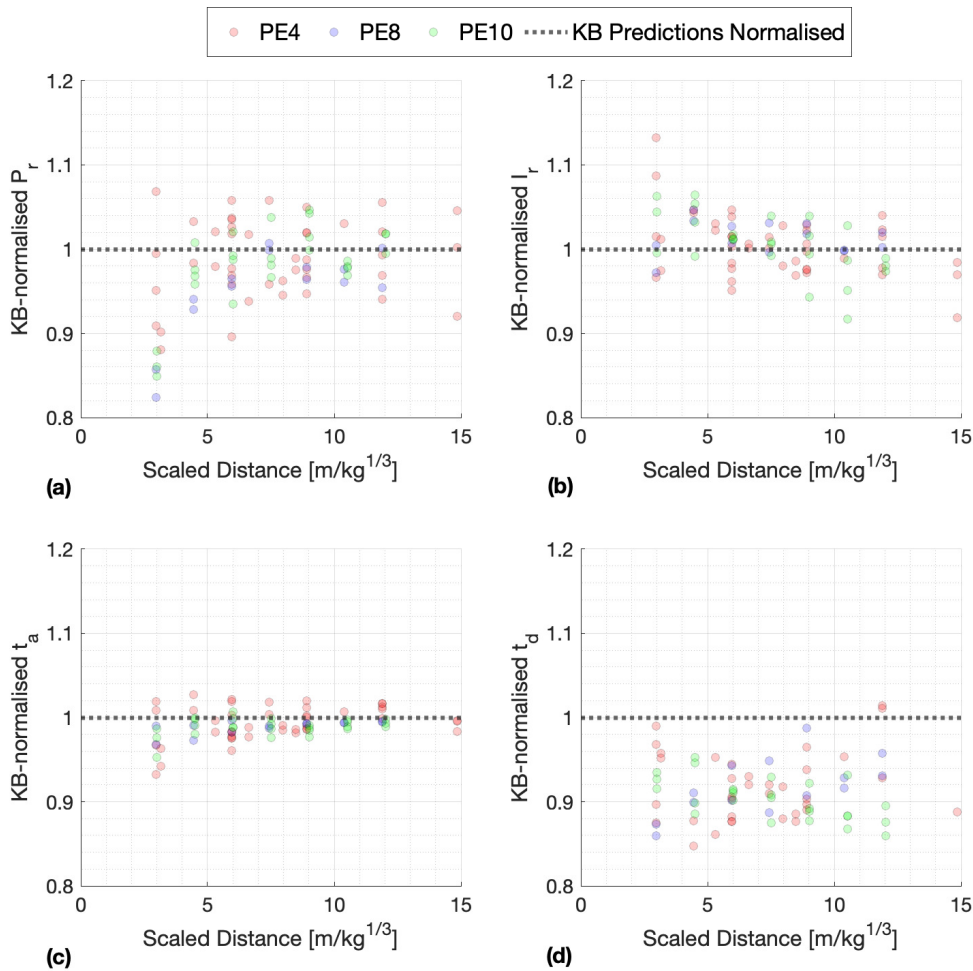


Figure 4.3: Compiled blast parameters from RDX and PETN based explosive trials as a function scaled distance, normalised against KB predictions for nominally identical trials: a) Peak reflected overpressure (KB-Normalised P_r), b) Scaled reflected peak specific impulse (KB-Normalised I_r), c) Scaled arrival time (KB-Normalised t_a), d) Scaled positive phase duration (KB-Normalised t_d)

2693 Despite Figure 4.2d again showing experimentally recorded values generally lower than
 2694 KB predictions, this finding follows the same trends presented in Figures, 2.12d, 2.13d
 2695 and 3.10b. This, along with the agreement of each blast parameter with KB predictions

2696 in all of the remaining subplots in the aforementioned figures, validates the hypothesis
2697 that for far-field scenarios, ideal high explosive blast parameters are deterministic despite
2698 the variety in researcher, methodology, composition, analysis techniques and finally the
2699 year the trials were undertaken.

2700

2701 In order to establish specific levels of variability that the scaled experimental recordings
2702 had in comparison to KB predictions, a similar approach to that discussed in Section 4.2.1.1
2703 was adopted in which the experimental recordings were normalised but by the KB predic-
2704 tion of the parameter at each scaled distance for which data was available.

2705

2706 Figures 4.3a-d present the results of the KB normalisation, which shows that for far-field
2707 loading in simple geometrical scenarios, existing semi-empirical predictions are in fact
2708 remarkably accurate and compare well to experimental data. This suggests that semi-
2709 empirical methods can be used with confidence as a first-order approach for quantifying
2710 the blast load conditions from a small-scale high explosive in simplified far-field situations.
2711 The slight increase in the variability as the scaled distance reduces to the extent of those
2712 presented in these trials. This gives rise to the ongoing hypothesis that as recordings
2713 approach the near-field there is less certainty in the data captured.

2714

2715 Interestingly to note from Figures 4.3a and 4.3b is the opposing correlation of recorded
2716 peak pressure and specific impulse to KB predictions. Taking into account the fact that
2717 the data presented was the result of a purely top detonated testing regime, the findings
2718 presented in Figure 3.16, found in Section 3.2.2.3, show comparable results to facilitate
2719 this discussion.

2720

2721 The peak pressure recorded in a top detonated blast is much lower than that of a bottom
2722 detonated simulation as result of inconsistent detonation product break out. The faster
2723 arrival time associated with top detonation is a direct result of the early expansion and
2724 the un-spherical propagation of products and shocks associated with the decomposition
2725 at the top of the charge. This then re-adjusts when the detonation wave reaches the
2726 bottom of the charge, effectively releasing all of the potential energy and the remaining
2727 detonation products begin to radially expand. This results in mach stems formation and
2728 thus a superposition of waves resulting in standard friedlander pressure-time histories at
2729 greater scaled distances. This finding provides clear explanation as to why experimen-

2730 tally the peak pressure values are lower than KB predictions for top detonated charges
2731 presented in Figures 4.3a. This does however reveal a systematic error in the majority of
2732 the presented data within this thesis, in that for near-field scenarios the complex interac-
2733 tion of the shock wave and detonation products results in increased uncertainty of the data.
2734

2735 Taylor^[167, 168, 169] discussed in great detail the theory behind how blast waves form and
2736 hypothesised an instantaneous release of energy from an infinitely small point in space
2737 resulting in radial expansion of the products and shock. Whilst the data presented for
2738 the far-field scaled distances agree with equivalency scaling of spherical shock wave pa-
2739 rameters, the smaller scale distances highlight a greater sensitivity of resulting blast pa-
2740 rameters. This has been related directly to the variable detonation mechanics in shaped
2741 charges alongside detonation position and orientation.

2742

2743 The presence of the gauge plate clearing effect discussed in Section 3.2.2.3, shows that
2744 with near-field conditions, the reduction in the peak specific impulse value is significant
2745 when comparing numerical simulations directly to KB predictions. Whittaker et al.^[179]
2746 ran numerical simulations to quantify the effects clearing has in this scaled distance region
2747 and showed that when $Z < 1.6\text{m/kg}^{1/3}$ clearing can be effectively omitted. The 1.25m
2748 simulation outlined in Figure 3.16 falls very close to this region and therefore it important
2749 to only consider the trends where clearing has been effectively ignored.

2750

2751 When removing the clearing effect from the 1.25m numerical model, the peak pressure for
2752 a bottom detonated trial over predicts that of a top detonated shot whilst the impulse
2753 values resort to comparing with the KB predictions remarkably well. The danger in com-
2754 paring purely numerical analysis to the KB predictions is that the numerical models, and
2755 their corresponding EOS's were developed using far-field data, $Z > 3\text{m/kg}^{1/3}$. In this
2756 region the expanding fireball is no longer effecting the detached shock front. The model
2757 replicates an energy released instantaneously which results in a friedlander-like shock wave
2758 rather than one affected by the ongoing chemistry of the fireball. It would therefore be
2759 incorrect to suggest at this scale, the real-world physical mechanisms from the detonation
2760 mechanics and fireball is accurately captured.

2761

2762 The results presented in Figures 4.3b are processed using the curve-fitting method pre-
2763 sented in Section 3.2.1.1, which does its best attempt at omitting the clearing effects from

2764 purely experimental data. This raises the question as to why an increase in the specific
2765 impulse is seen in comparison to KB predictions. This can be used to further explain the
2766 narrative of why the semi-empirical method becomes inaccurate for the near-field scenarios
2767 tests. Considering Figure 4.3b, the under-prediction of impulse suggests that the energy
2768 released from these high explosives is higher than TNT within the fireball region. This
2769 difference suggest one single TNT equivalency factor cannot be definitively assigned across
2770 all scaled distances.

2771

2772 The experimentally established positive phase duration, shown in Figure 4.3d, presents
2773 lower values to those evaluated by KB predictions by up to 15%. The large spread in the
2774 experimental recordings is directly related to the difficulty in determining this parameter,
2775 as discussed previously. This generally becomes more difficult with increased scaled dis-
2776 tance as: (a) signal:noise is typically lower, and (b) the gradient of the latter stages of the
2777 positive phase is shallower, as can be seen in Figure 3.5.

2778

2779 4.2.1.3 Time of Arrival Variability

2780 Despite the visibly low levels of variability shown in both Figure 3.10a and 4.3c, shock
2781 front time of arrival, t_a , is a somewhat overlooked parameter when compared to peak pres-
2782 sure and peak specific impulse despite its qualitative comparison to KB predictions. The
2783 arrival time of a blast wave can be used to evaluate the resulting pressures and impulses
2784 using Rankine-Hugoniot principles which presents a utility of arrival time as a parameter
2785 for blast design purposes. The main aim of this section is to investigate and quantify the
2786 variability of shock wave arrival times, and investigate how it varies with scaled distance.

2787

2788 Due to the consistency of the results presented across all three high explosives tested,
2789 it was decided that arrival time of PE4 would be assessed due to having a much larger
2790 data set, with multiple repeat shots at each scaled distance. To evaluate the precision
2791 and repeatability of the data set, the relative standard deviation (RSTD) was calculated
2792 from the pressure-gauge arrival time data (from the historic and current tests presented
2793 in this thesis: 157 in total). The results of which can be used to assess variations in the
2794 measurements from this particular experimental method with respect to scaled distance,
2795 presented in Figure 4.4.

2796

2797 To provide a more generalised relationship of arrival time variability with respect to scaled
2798 distances it was important to assess results which had enough repeat trials to represent
2799 real physical behaviours. Of all the scaled distances tested, there was a variety in which
2800 at least 5 nominally identical tests were conducted and so should provide a reasonable
2801 approximation to the general variability behaviour. These groups are shown in Figure 4.4
2802 as solid markers, whilst the hollow markers presented show the remaining scaled distances
2803 tested which had less than 5 nominally identical results. Overlaid on the plot is an ex-
2804ponential trend connecting the groupings with 5 or more trial, which when considering
2805the same qualitative behaviour is seen in the hollow markers, decreasing variability with
2806increasing scaled distance, is believed to be a physical representation of how variability in
2807arrival time changes with scaled distances specifically for top detonated charges. It has
2808been assumed however that the same behaviour would be experienced by bottom deto-
2809nated charges as a result of fluid dynamic instability formation, however to magnitude of
2810the RSTD would be lower for equivalence scaled distances.

2811

2812 It is important to note the curve has been extrapolated for larger scaled distance values,
2813 $Z > 12\text{m}/\text{kg}^{1/3}$, but not for near-field conditions, $Z < 3\text{m}/\text{kg}^{1/3}$. The assumption made
2814 here is that for far-field scenarios, the propagating shock wave is unaffected by anything
2815 other than the medium it travels through, and therefore a similar relationship of variability
2816 is expected. For the near-field recordings it is however assumed the data would be affected
2817 by the sensitive detonation mechanics and to the idea of instabilities in the shock front
2818 forming and diminishing, in the intermediate range of scaled distances. The exponential
2819 relationship has been defined for the region of scaled distances tested and extrapolated as
2820 a guideline prediction outside these ranges. Future works will undertake more testing in
2821 near-field conditions, $Z < 3\text{m}/\text{kg}^{1/3}$, in the hope to show a reduction in arrival time vari-
2822 ability associated with a smooth shock front before instabilities start to form at extreme
2823 near-field in the support of the hypothesis of shock wave instability regions^[173]. It could
2824 however be that the magnitude of arrival time variability presented in Figure 4.4 would
2825 reduce if systematic detonation errors are omitted within the methodology.

2826

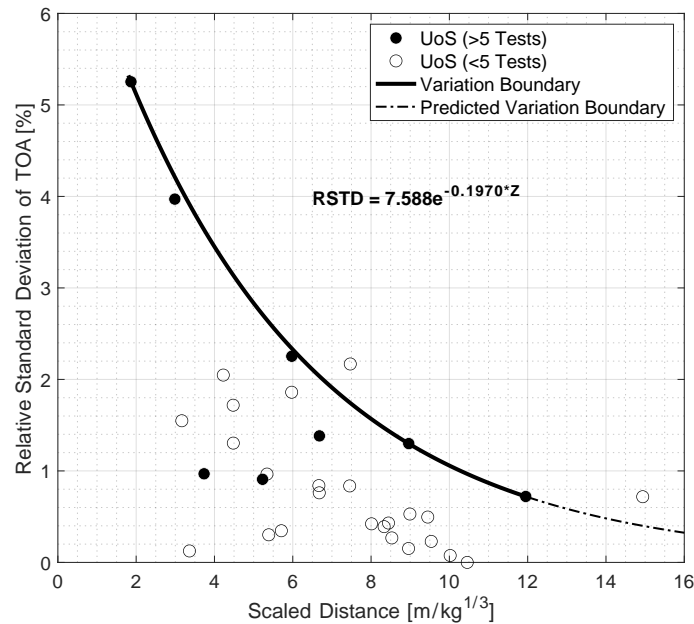


Figure 4.4: *RSTD* in shock wave arrival time with respect to scaled distance, from historic and current pressure gauge data

2827 The main point to address from this analysis is that quantifying arrival time, t_a , variability
 2828 with respect to scaled distance, which the parameter itself can be chosen with little ambi-
 2829 guity, provides a further insight into the development of a shock front and the formation of
 2830 any un-spherical nature, especially in top detonated scenarios. Current predictive methods
 2831 provide discrete blast parameters for a given scaled distance an explosive is away from a
 2832 target. Through understanding the spread of experimentally measured t_a , improvements
 2833 can be made to fast running predictive methods to provide probabilistic blast parameters
 2834 within the uncertainty range for a given distance.

2835

2836 4.2.2 Numerical Simulation

2837 The pressure gauge results presented within this Chapter, begin to build a coherent fun-
 2838 damental understanding of the nature of experimentally recorded far-field loading from
 2839 a given charge composition, shape and mass. The importance of these findings provides
 2840 a starting point to establish characteristics of more challenging blast loading conditions

2841 both in the near-field and those in complex environments, knowledge of which can be
2842 supplemented with high-fidelity numerical modelling.

2843

2844 Whilst experimental data is beneficial for understanding the real world yield and sub-
2845 sequent effects of a particular explosive detonation, both the costs and safety concerns
2846 related to blast trials create a need for high fidelity numerical simulations which are able
2847 to capture the behaviours within any scenario.

2848

2849 Understanding the mechanisms and magnitudes of blast loading on targets from both
2850 near-field detonations of high explosive, and those in complex environments, is of key
2851 importance for the analysis and design of the response of protective structures. How-
2852 ever, there is little definitive experimental data on the measurement of these loads and
2853 consequently the predictions of numerical models of near-field blast loading are largely
2854 unvalidated.

2855

2856 The experimental results presented with Chapter 4 have been used to validate far-field
2857 numerical models for PE4, PE8 and PE10, using the methodology outlined in Whittaker
2858 et al.^[179]. The intention was to implement the developed equations of state (EOS) for
2859 each explosive, into much more complex numerical simulations to produce validated and
2860 accurate predictions.

2861

2862 It is important to note that the numerical modelling discussed within this thesis is collab-
2863 orative work between the University of Sheffield, Blastech LTD and DSTL, which together
2864 are attempting characterise explosives with high levels of accuracy in order to map out
2865 the effects a given charge composition, mass and shape may have in a prescribed envi-
2866 ronment to better plan and protect civilian infrastructure. Whilst DSTL have conducted
2867 the numerical simulation, the author has undertaken all experimental data capture and
2868 comparative studies.

2869

2870 **4.2.2.1 Model Description**

2871 The baseline numerical model consisted of 250g hemispherical charges placed on the floor
2872 which were centrally detonated (of an equivalent sphere) to match the experimental pro-

2873 cedures adopted with the testing regime. A reflective boundary was set at the edge of
2874 the model used to represent the reflective wall (be that the bunker or the blockwork wall)
2875 with a size of 2.2m in height and 4.4m in width to prevent any clearing effects within the
2876 positive phase of the reflected shock. An additional reflective boundary was set up along
2877 the ground surface to represent the concrete pad used during the explosive trials.

2878

2879 Afterburn is included in these free-air models to show that thermochemical analysis can
2880 suitably parametrise numerical models rather than relying on experimental methods which
2881 are very expensive and time consuming processes. The EOS was generated in a few hours
2882 using a thermochemical code. Afterburn should be included in free-field models to achieve
2883 an accurate EOS. This is to show that the model behaves in a representative way when it
2884 is included for all scenarios, whether that is relatively small amounts in free-field or large
2885 amounts in confined scenarios.

2886

2887 The numerical model was solved using APOLLO blastsimulator, which makes use of Adap-
2888 tive Mesh Refinement (AMR), and zoom levels (distance-dependant AMR) to allow finer
2889 mesh resolution to be used within the complex regions of numerical analysis^[36,109]. AMR
2890 is based on a mesh which exhibits an inconsistent resolution globally. In regions of the
2891 domain which requires a more refined resolution (i.e the detonation point, the shock front
2892 etc), the mesh refines to make more precise predictions. Other area of the mesh will
2893 reduce in resolution to allow for simulations to be undertaken faster. As the simulation
2894 progresses, as do the refined areas, and therefore providing high resolutions in the area
2895 which require them. This has the benefit of saving the computational power for the ar-
2896 eas that require it through reducing the computational demand from areas where it is
2897 not needed. The AMR process requires a user-defined zone length, corresponding to the
2898 coarsest cell size, and a maximum resolution level size which corresponds to the smallest
2899 allowable cell size. The software then refines and un-refines different zones within the
2900 model (based on differentials of pressure, material etc) to accurately simulate the event
2901 while maximising efficiency. The model also uses “zoom levels” which allows a higher res-
2902 olution level to be used for a fixed radius from the charge centre (e.g. a zoom level of 1 for
2903 200mm would increase the maximum resolution level by 1 until a disturbance is registered
2904 at 200mm, the model would then only allow the initial maximum resolution level to be
2905 achieved for the remainder of the model). As an example, the 2m models reviewed within
2906 the remainder of this document had the following zone lengths and resolution levels:

- 2907 • Zone length = 200mm, maximum resolution level = 5.
- 2908 • Zoom level 3 for 200mm from charge centre.
- 2909 • Zoom level 2 for 600mm from charge centre.
- 2910 • Zoom level 1 for 1400m from charge centre.
- 2911 • Then maximum resolution level of 5 (6.25mm) for the remainder of the model (6ms).

2912 The Explosive EOS parameters used in this study were generated using the thermochemi-
2913 cal code Cheetah v7, which, due to export control reasons, are not available for publication.
2914 Therefore, an alternative thermochemical code, EXPLO5, was also used to determine the
2915 EOS for the explosives. These parameters have only been used for a complementary
2916 comparison to the Cheetah study and have not been fully assessed or validated, but are
2917 provided as representative values in Table 3.2. It is important to note the developed pa-
2918 rameters were produced purely based on the explosive composition, correct density and
2919 a best guess of plasticiser material (finding the exact composition may improve the re-
2920 sults), all done using the same default methodology. No calibrating to experimental data
2921 or tweaking of technique or parameters to improve results was performed. The numerical
2922 model construction is also discussed in more detail within Section 3.2.2.1 which provides
2923 additional narrative to the discussion for reference.

2924

2925 4.2.2.2 Numerical Model Validation

2926 Figure 4.5 presents a comparison between numerical and experimental pressure histories
2927 across a range of standoff distances for each ideal explosive tested. A high degree of
2928 similarity in the qualitative form of the pressure-time histories, for all three explosives,
2929 across the entire range of far-field stand off distances, provides confidence in the modelling
2930 approach and parameters. This finding alone gives rise to the hypothesis presented by
2931 Tyas^[173] that for at least far-field scenarios, $Z > 3 \text{ m/kg}^{1/3}$, explosive characterisation is
2932 deterministic with low levels of variability. The comparable results, when adopting a more
2933 realistic representation of the test conditions, highlights the synergistic nature of these
2934 two techniques. Numerical models have become powerful enough to not rely solely on
2935 experimental data for validation and can now be used to fault-find systemic flaws within
2936 experimental procedures.

2937

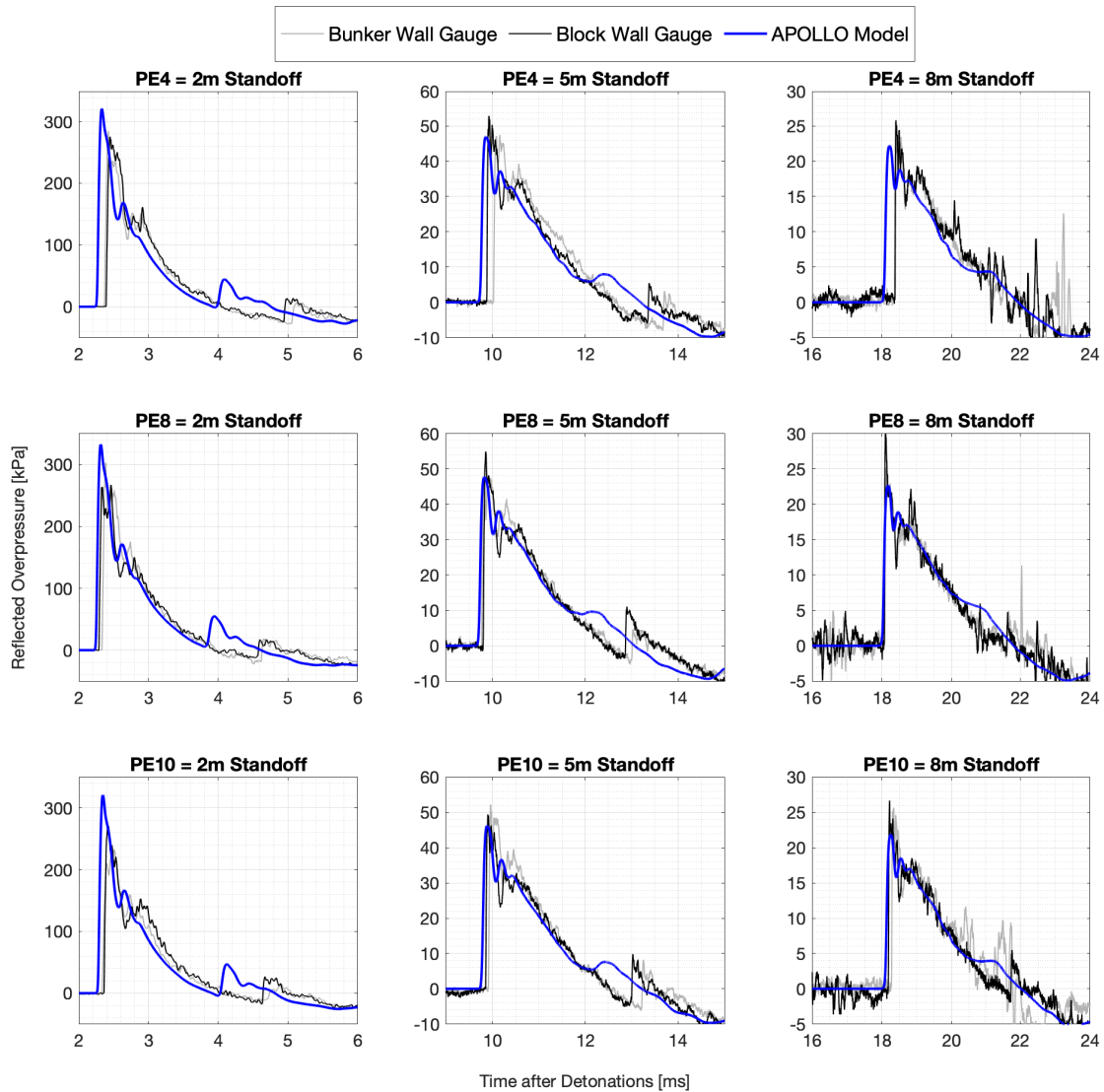


Figure 4.5: A compilation of reflected overpressure-time history plots for the three explosives tested across three different stand-off distances compared to the corresponding validated and APOLLO model predictions

2938 Interestingly what is exhibited across each numerical simulation displayed in Figure 4.5 is
 2939 a much faster arrival of the secondary shock wave when compared to experimental trials.
 2940 This feature can lead to artificially high specific impulse predictions with the secondary
 2941 shock arrival coinciding with the positive phase of the primary shock event. As mentioned
 2942 in Section 3.2.2.1, the secondary shock arrival time varies depending on how afterburn is
 2943 modelled. The current understanding of the secondary shock is that it is intimately linked
 2944 to the post-detonation pressure–volume–energy relations of the fireball, which currently is
 2945 not captured accurately without manual tweaks to modelling parameters^[141].

2946

2947 **4.2.3 The Negative Phase**

2948 To evaluate the findings in Rigby et al. [133], which displayed comparable predictions of neg-
 2949 ative phase parameters to well-controlled experimental testing of PE4, the same method-
 2950 ology and analytical techniques were adopted again for the remaining ideal explosives of
 2951 PE8 and PE10. The result of which have been scaled accordingly and are presented in
 2952 Figures 4.6a-c.

2953

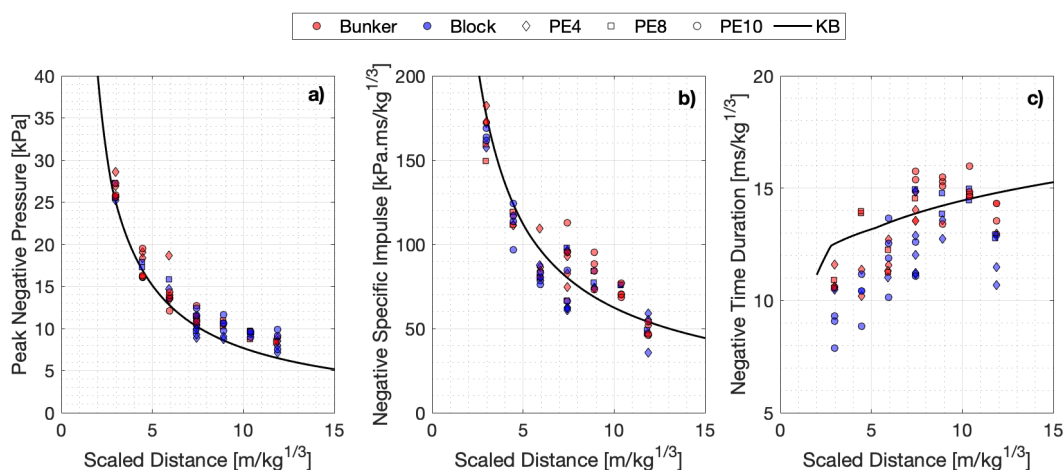


Figure 4.6: Compilation of scaled negative phase parameters results of 250g PE4, PE8 and PE10 hemispherical charges using the representative TNTe: a) Negative Pressure, b) Scaled Negative Specific Impulse and c) Negative Phase Time Duration.

2954 The variability in each negative phase parameter is visibly greater than the positive phase
 2955 parameters presented in Figures 4.2a-d for the same explosive trials. This has been linked
 2956 to the increased difficulty in prescribing parameters with lower magnitudes being sensitive
 2957 to electrical noise and systematic experimental features. Despite this, the main conclu-
 2958 sions to be drawn from the data presented in the aforementioned plots is that the plastic
 2959 explosives tested behave almost exactly in line with KB predictions for far-field scenarios
 2960 when conducted under strict conditions. The same cannot yet be deduced across all scaled
 2961 distances as the behaviour of near field loading conditions are yet to be investigated and
 2962 will be part of future planned works.

2963

2964 4.2.3.1 Secondary Shock Data

2965 Section 3.2.3.2 began the investigation into the secondary shock and its associated blast
2966 parameters for PE4 detonations. The study presented a consistent general behaviour
2967 with KB predictions across the far-field range of scaled distance. There was however
2968 significant discrepancies in terms of the magnitudes due to being compared directly to
2969 a primary shock. Whilst the KB predictive tools in their current form do not capture
2970 secondary shock waves accurately, the presented behaviours of the associated parameters
2971 could provide insights into the mechanisms involved in their development. This behaviour
2972 is believed to be essential for fundamentally predicting them. With this idea in mind,
2973 the secondary shock data from both PE8 and PE10 was analysed in the same way, using
2974 the corresponding TNTe values evaluated in Section 4.2.1.2 are presented in Figures 4.7a-d.

2975

2976 Across all three plastic explosives tested in the far-field range, the scaled secondary shock
2977 positive phase parameters compare remarkably well when compared to one another at
2978 each increment of scaled distance. The general increase in variability associated with each
2979 of the parameters, seen in Figure 4.2, has been associated with the chaotic detonation
2980 cloud medium the secondary shock has to propagate through. This, similarly to primary
2981 wave data, tends to reduce with an increase in scaled distance seen in Figures 4.7a and
2982 4.7c. This suggests that variations related to this changing medium are insignificant once
2983 measurements are far enough away from charge centre.

2984

2985 A feature which is exhibited within the pressure data recorded for the secondary shock
2986 (Figure 4.7a), is an apparent plateau at $Z \simeq 10m/kg^{1/3}$ and an increase in pressure values
2987 experienced some distance afterwards. The plateauing behaviour is justified through the
2988 curtailment of pressure as distance increases until the wave effectively is obsolete, but the
2989 apparent rise in pressure holds no physical explanation. This visible increase is a system-
2990 atic error within the data capture at these large scaled distances. Pressure magnitudes
2991 are so low for the secondary shocks that the assignment of parameters is extremely sen-
2992 sitive. Using gauges that are rated to 70 times greater than these magnitudes is flawed.
2993 The inherent electrical noise in the system accounts for $\pm 1kPa$ and so contributes to the
2994 spreads seen. It is difficult to improve this using this methodology as gauges are required
2995 to be higher rated to survive the pressure from the primary wave.

2996

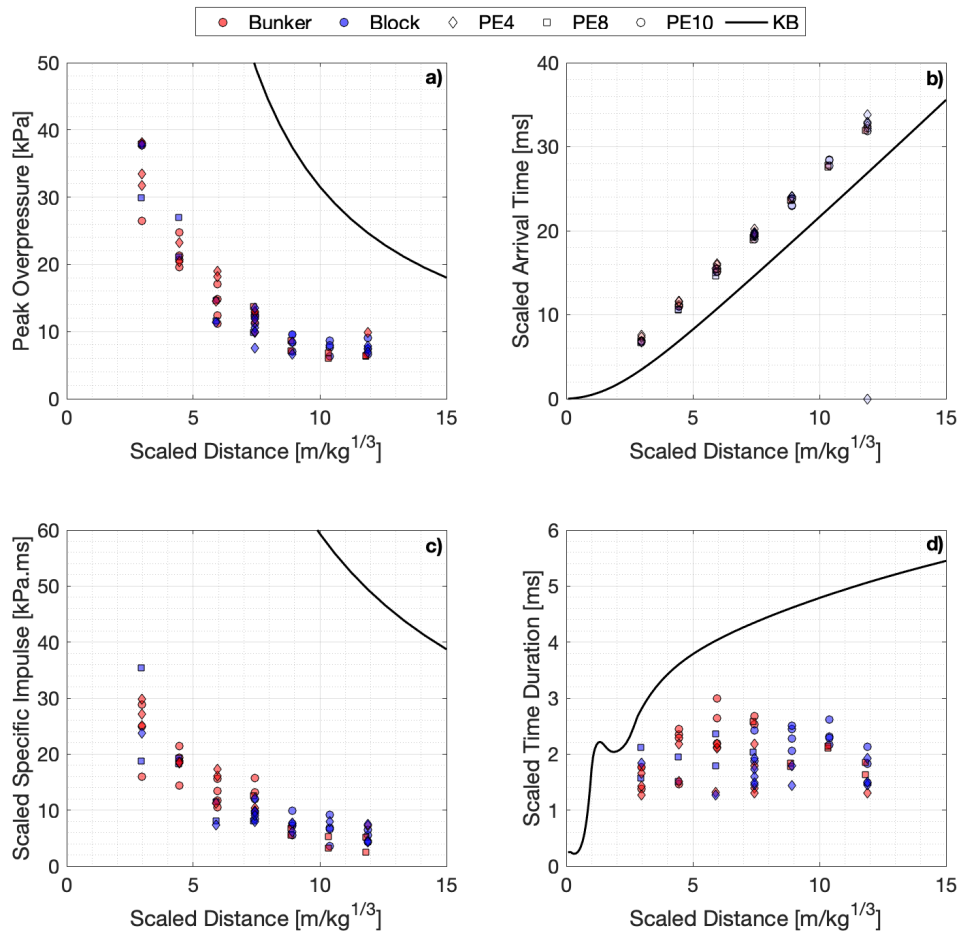


Figure 4.7: Compilation of scaled secondary shock parameters resulting from visual analysis of 250g hemispherical charges of PE4, PE8 and PE10 using the representative TNTe values: a) Negative Pressure, b) Scaled Negative Specific Impulse and c) Negative Phase Time Duration.

2997 Clearly the ideal explosives investigated within this thesis exhibit similar behaviours to a
 2998 primary wave but are not captured by KB predictions when scaling the results based on
 2999 the original charge mass and established TNT equivalency values.

3000

3001 4.2.3.2 Secondary Shock Delay Predictor

3002 The delay in time between the arrival of both the primary and secondary shocks is a fea-
 3003 ture which has been previously investigated success to quantify equivalent TNT yields [129].

3004 Figure 4.8 presents the scaled secondary shock delay data across the three ideal explosives
 3005 considered in this thesis. Although exhibiting a consistent behaviour with scaled distance,

3006 this could be related to the fact PE4, PE8 and PE10 have been almost identical across
 3007 each parameter investigated. Rather than assuming this behaviour is consistent for all
 3008 ideal explosives, it was therefore important to compare these results to historical data,
 3009 providing confidence in the presented data within this thesis alongside the predictive tool
 3010 developed by Rigby and Gitterman^[129].

3011

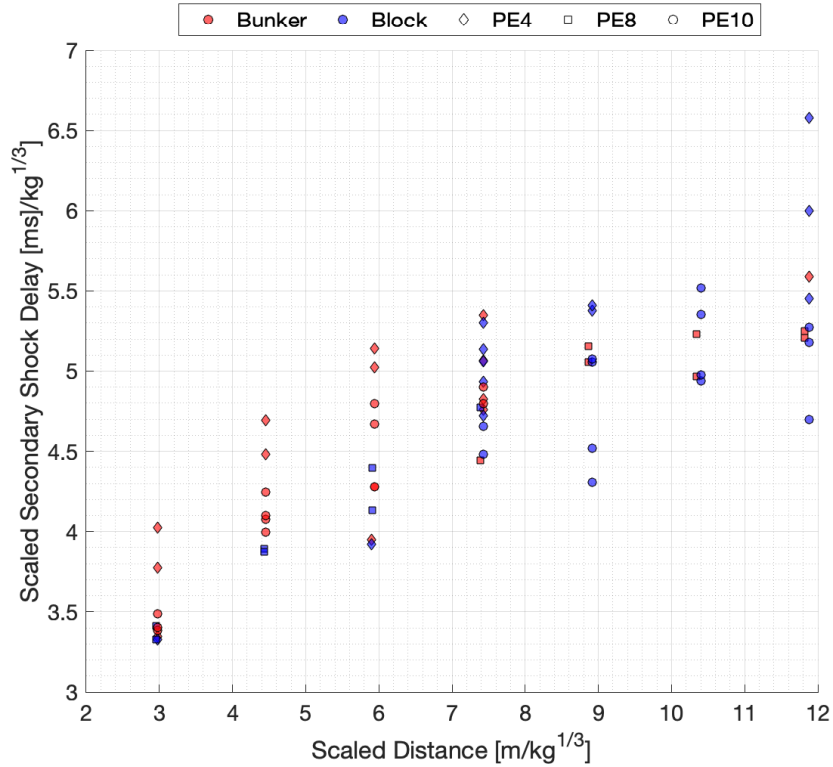


Figure 4.8: Scaled secondary shock delay with respect to scaled distance for all three ideal explosives using representative $TNTe$ values

3012 Through manipulation of the data, presented in Figure 2.8, by dividing the secondary
 3013 shock parameter, τ_{ss} (derived by Equation 4.1), by scaled distance squared, the paramete-
 3014 ter becomes dimensionless and therefore required no physical justification. The result of
 3015 this data manipulation is displayed in Figure 4.9 with a log-polynomial fit represented by
 3016 Equation 4.2.

3017

$$\tau_{ss} = \delta t_{ss} V_{od} / Z^2 (\rho W)^{1/3} \quad (4.1)$$

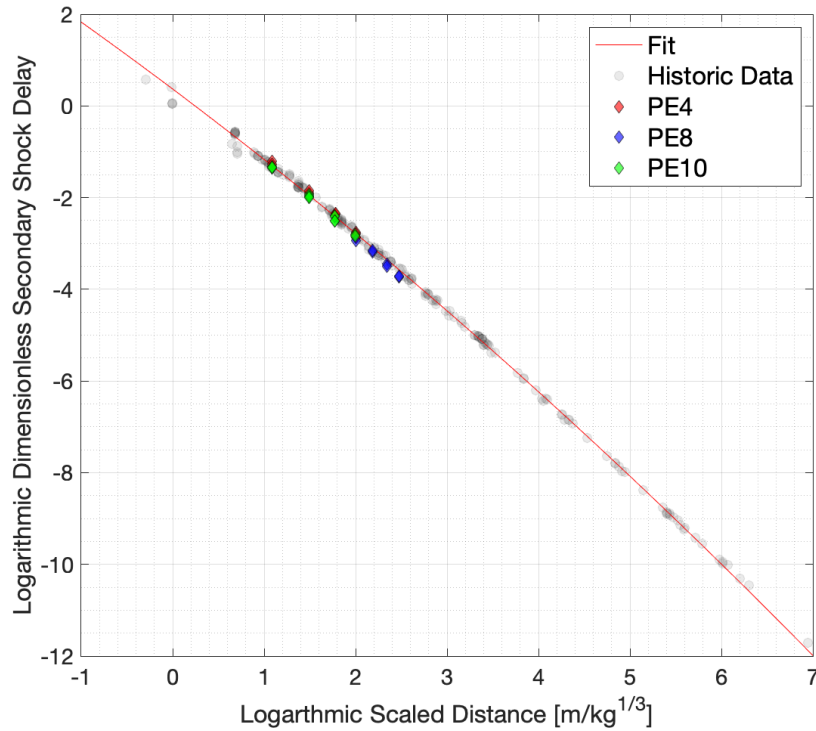


Figure 4.9: Dimensionless secondary shock delay parameter evaluated for all ideal explosives scaled accordingly compared with polynomial fit to historical data

3018 Plotted alongside the historic data are the three ideal explosives investigated in this thesis,
 3019 showing high levels of consistency when compared to all the experimental data and the
 3020 newly presented fit. It is important to note that whilst the scaling of these secondary
 3021 shock parameters is based on some physical mechanism, the fundamental processes under-
 3022 pinning the secondary shock behaviour is not yet fully understood. The fact there is such a
 3023 close relationship between all the explosives considered, despite differences in velocities of
 3024 detonation, packing densities and TNT equivalences, suggests that there is a real physical
 3025 process relating the secondary shock to these parameters.

3026

$$\log(\tau_{ss}) = 0.3682 - 0.0351\log_{10}(Z)^2 - 1.498\log_{10}(Z) \quad (4.2)$$

3027 4.2.4 Pressure Gauge Summary

3028 This section clearly validates the analytical and experimental techniques discussed in
 3029 Chapter 3 for pressure gauge type recordings across the far-field range for three plas-

3030 tic explosives with ideal behaviours. Alongside this, there is definitive evidence that with
3031 well-controlled experimental trials, data only exhibit high levels of consistency but both
3032 positive and negative phase blast parameters are captured remarkably well by KB pre-
3033 dictions. With this finding, it is hoped that the industry will recognise that for far-field
3034 regimes, the yield of explosives is not inherently variable and steps can be made to improve
3035 the fundamental understanding of blast wave mechanics.

3036

3037 Detailed analysis into the secondary shock phenomena was undertaken in the attempt
3038 to understand the physical mechanisms behind the subsequent wave propagation and at-
3039 tempt to both quantify and characterise its behaviour. The work developed in this chapter
3040 provides a reasonable and logical assessment of explosive blast wave features and presents
3041 a validated prediction tool which could potentially be integrated into numerical simula-
3042 tions to better capture its behaviour. Whilst the fundamental mechanisms are yet to be
3043 captured accurately for the secondary shock features, these developments can hopefully
3044 drive improvements to the knowledge.

3045

3046 Future experimental data is required from explosives which exhibit widely different TNT
3047 equivalency values to distinguish whether the trends seen through Figures 4.7a-d are con-
3048 sistent across explosive composition, mass and distance. If the resulting data was con-
3049 sistent, considerations into a more appropriate way of scaling the results extracted from
3050 secondary shock data would be required to either agree with KB predictions of a primary
3051 wave or to develop a set of new empirical curve to supplement the established well vali-
3052 dated predictive tool.

3053

3054 4.3 High Speed Video Data

3055 This section will begin to consider the data captured using high speed cameras during
3056 far-field testing, tracking specifically the propagating shock wave in real time away from
3057 the explosive centre. With this technique of analysis being relatively new for the blast
3058 industry, it has not been fully explored. The tools developed in Sections 3.3.1.1 and 3.3.1.3
3059 have been tested against the three ideal explosives discussed in this chapter. Comparisons
3060 have been made directly to corresponding pressure gauge data to both further validate
3061 the methods developed and to provide greater insights into the shock wave phenomena.

3062

3063 **4.3.1 Time of Arrival Variability**

3064 With arrival time being the main feature captured during HSV recordings, it was im-
3065 portant to assess how the parameter varied both spatially and temporally during the
3066 spherical expansion of the shock wave. Implementing the shock wave edge detection and
3067 arrival time interpolation tools for far-field high speed video recordings, as described in
3068 Sections 3.3.1.1 and 3.3.1.3 respectfully, arrival time, t_a , values were established at every
3069 $0.1 \text{ m/kg}^{1/3}$ scaled distance along each virtual ‘*spokes*’ which start at the charge centre
3070 and plot radially outwards, in increments of 1° . The data captured through this analysis
3071 allows for a direct comparison to pressure gauge recordings which records pressure-time
3072 histories at discrete locations. The 3,163 data points as a result of implementing these
3073 methods on 10 HSV recordings of PE4 far-field detonations, 4,313 data points for PE8
3074 from 6 recordings, and 3,639 PE10 data points from 4 recordings, were collated with re-
3075 spect to each increment of scaled distance. The RSTD was calculated similarly to that
3076 done for the gauge recordings seen in Figure 4.11.

3077

3078 The standardised range of interpolated HSV arrival time results are compared directly
3079 to pressure gauge variation of PE4 trials, with respect to scaled distance, presented in
3080 Figure 4.10. As lens distortion was not accounted for during these trials, the videos were
3081 not used as a group of collated data, and only those with the same test set up and explo-
3082 sive composition were compared. Regardless of this, the relative variability of arrival time
3083 recordings at specific scaled distances held fairly consistent between groupings of tests
3084 when compared to the relationship pressure gauge data had with scaled distance.

3085

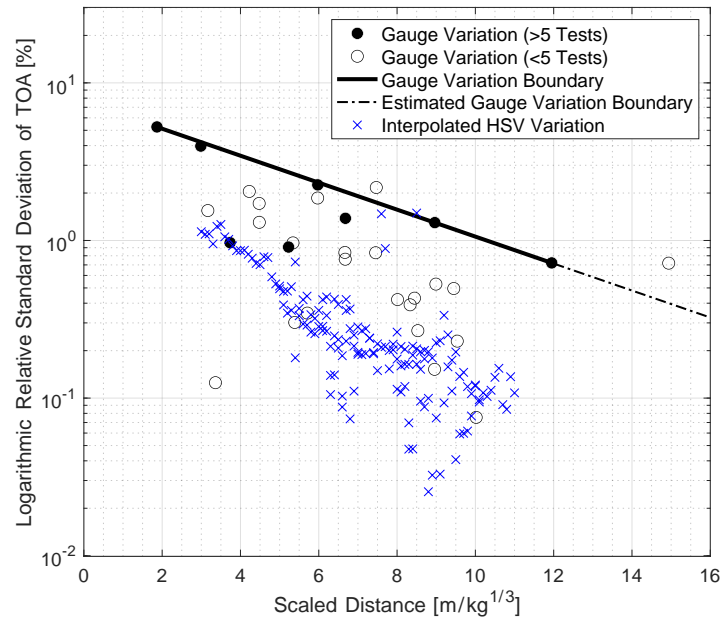


Figure 4.10: RSTD in shock wave arrival time with respect to scaled distance, when comparing pressure gauge and HSV data

3086 A significant reduction in the variability of processed arrival times across the entire range
 3087 of tested distances is experienced when using high speed video for PE4 detonations. In
 3088 part, this reduction in variability is associated with the volume of data points created
 3089 through the interpolation at each scaled distance. However, some of the scaled distances
 3090 in question had a similar benchmark number of data points as the pressure gauge results.
 3091 That being said, for the entire range tests, the two extremes had a low number of record-
 3092 ings established and yet there is still a noticeable reduction in processed t_a variability
 3093 as scaled distance increases. This alone provides clear justification for the use of video
 3094 recordings during explosive trials to enhance the precision and repeatability of blast pa-
 3095 rameters between tests.

3096

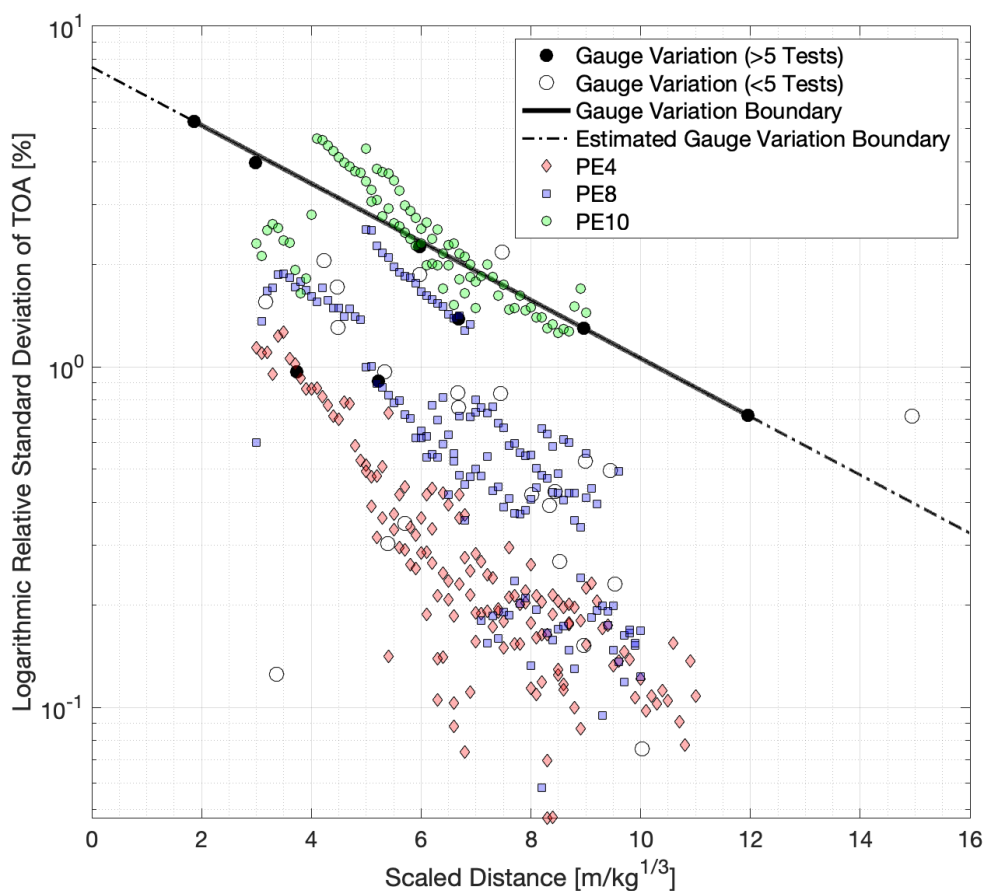


Figure 4.11: Logarithmic RSTD in shock wave arrival time as a function of scaled distance for both pressure gauge and high speed video recordings of 250g hemispheres of PE4

3097 One thing to note is the substantial volume of interpolated data from just 10 processed
 3098 PE4 videos when compared to 157 different pressure gauge recordings. The larger data
 3099 set (3,163 recordings), alongside the clear improvement in repeatability and accuracy from
 3100 just 10 explosive trials, presents a more cost-effective solution to experimental testing of
 3101 high explosives.

3102

3103 The interpolated values from HSV recordings of PE8 and PE10 detonations also experi-
 3104 ence the same behaviour of reduced variability as scaled distance increases. The reason
 3105 gauge data boundaries have not been presented in Figure 4.11 is due a lack of data to
 3106 extract a reasonable estimate of arrival time variation with distance. Comparisons to that
 3107 for PE4 are made instead to facilitate the discussion. The videos for PE10 were by far
 3108 the hardest to analyse, with the lighting conditions when conducting these trials making

3109 extracting shock wave data difficult. That being said the 4 videos which data could be
3110 used provided similar levels of variation from 157 PE4 pressure gauge trials, highlighting
3111 the efficiency of this testing method. With an increase in the number of videos available,
3112 a reduction in the variability is experienced when looking at the results from the 6 PE8,
3113 and then finally the 10 PE4 videos.

3114

3115 The significance of Figure 4.11 is the comparable relationship between t_a variation with
3116 respect to scaled distance for both HSV and pressure gauge recordings. This is shown
3117 on a logarithmic scale to enable a clear comparison between the processed results from
3118 both methods of recording. A clear exponential reduction in t_a variability is seen with
3119 an increase in scaled distance across both methods of experimental analysis. The typical
3120 variations in the HSV data are approximately an order of magnitude lower than those in
3121 the pressure gauge data. The speculation that blast parameters exhibit high levels of in-
3122 consistency across the far-field range is disproved with these findings, explicitly suggesting
3123 far-field regimes and the corresponding blast parameters are deterministic.

3124

3125 4.3.2 Derivation of Explosive Yield from Arrival Time Data

3126 To consider arrival time data as a metric for forensic explosive analysis, it is important to
3127 validate it against empirical data alongside quantifying the variability in the parameter.
3128 Rigby et al.^[131], Stennett et al.^[158] and Dewey^[39] all present methods of quantifying the
3129 explosive yield of a real-world accidental explosion through arrival time data analysis with
3130 respect to best estimations of stand-off from the explosive centre. Their findings presented
3131 clear agreement of explosive yield and thus the TNT equivalence factor of the explosive
3132 detonated.

3133

3134 To verify the shock wave tracking and interpolation algorithm as an accurate means of
3135 gaining an understanding of how arrival time varies spatially across a full range of scale
3136 distances, the method was tested to estimate the apparent yield of the explosive used
3137 during the tests, which is known to be 250g PE4.

3138

3139 Figure 4.12 shows how the HSV-measured values of arrival time compare to a variety of
3140 different sized TNT charges and their corresponding KB predictions. It is important to

3141 note that the axes have been scaled down to visualise a clear distinction between the pre-
 3142 dictive TNT curves due to modest changes in charge mass. The HSV data shows excellent
 3143 agreement with the curves for TNT between 290-320g, equating to a TNT equivalence of
 3144 1.16-1.28. The minimum mean absolute error is associated with a TNT mass of 305g, i.e.
 3145 an equivalence of 1.22, which is very close to the value of 1.20 found in related studies [132].
 3146 This suggests that the optical technique presented is a reliable method for determining
 3147 the TNT equivalence of explosives.

3148

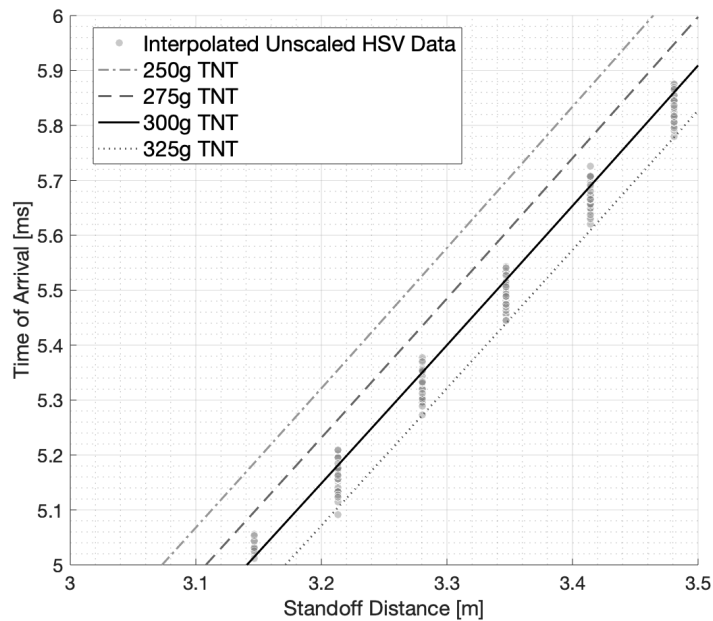


Figure 4.12: Interpolated and unscaled HSV shock wave arrival time from a 250 g PE4 hemispherical charge at 5 m stand-off compared with KB predictions for TNT hemispherical charges with different charge mass

3149 Whilst the spread in TNT equivalence extracted from the HSV data is not significant,
 3150 this could be vastly improved when considering what position on the shock wave you take
 3151 to be the definitive front. The algorithm discussed in Section 3.3.1.1 makes use of im-
 3152 age subtraction which results in clusters of pixels exhibiting change. Despite best efforts
 3153 of the edge detection method, variations are therefore recorded in the shock wave position.

3154

3155 The Abel transformation relates a spherically-symmetric 3D field to a two-dimensional
3156 projection^[34,91] which has been adapted for use in blast wave tracking through inversion
3157 of the process for tiny charges in well-controlled lab conditions^[172]. The inversion how-
3158 ever requires refraction angles obtained through effective lens calibration which was not
3159 conducted during this testing regime. Future tests should consider the use of the Abel
3160 transformation inversion to improve the accuracy of shock wave position extrapolated from
3161 a 2D video of a 3D spherical expansion.

3162

3163 4.3.3 Capturing Secondary Shock Data

3164 The secondary shock parameter has been shown to be as deterministic as the primary
3165 shock in Section 3.2.3.2, provided the right assumptions are made when attempting to
3166 characterise it. However, what has yet to be established is whether secondary shock data
3167 can be captured using HSV techniques which could help provide a more fundamental un-
3168 derstanding of the phenomena.

3169

3170 Adopting the same methods discussed in Sections 3.3.1.1 and 3.3.1.3 proved successful
3171 for capturing the propagating subsequent shock provided that lighting conditions were
3172 favourable and the explosive was not located too far out of the camera field of view. As
3173 the secondary shock is much weaker, the shock discontinuity is a lot more challenging to
3174 extract and therefore a lot of the videos provided unextractable data.

3175

3176 Despite the difficulty in extracting secondary shock data across each explosive, Figure 4.13
3177 presents the overall secondary shock delay parameter, as discussed in Section 3.2.3.2,
3178 against the unscaled compiled pressure gauge data presented in Figure 4.8. The first
3179 thing to notice is the comparison in behaviours of the HSV extracted data to the pressure
3180 gauge data. This relationship provides enough evidence to support the validity of the
3181 ext extraction methodology for secondary shock data. The HSV data however seems to
3182 exhibit a smaller delay parameter across the explosives analysed. This is believed to be
3183 directly related to the HSV data tracking incident wave data throughout its duration for
3184 both primary and secondary shocks, where as the pressure gauges record at a discrete
3185 instance of distance. This means the secondary shock would be forced to expand against
3186 a reflected wave, decelerating its propagation and thus increasing its time of arrival when
3187 compared to the HSV data.

3188

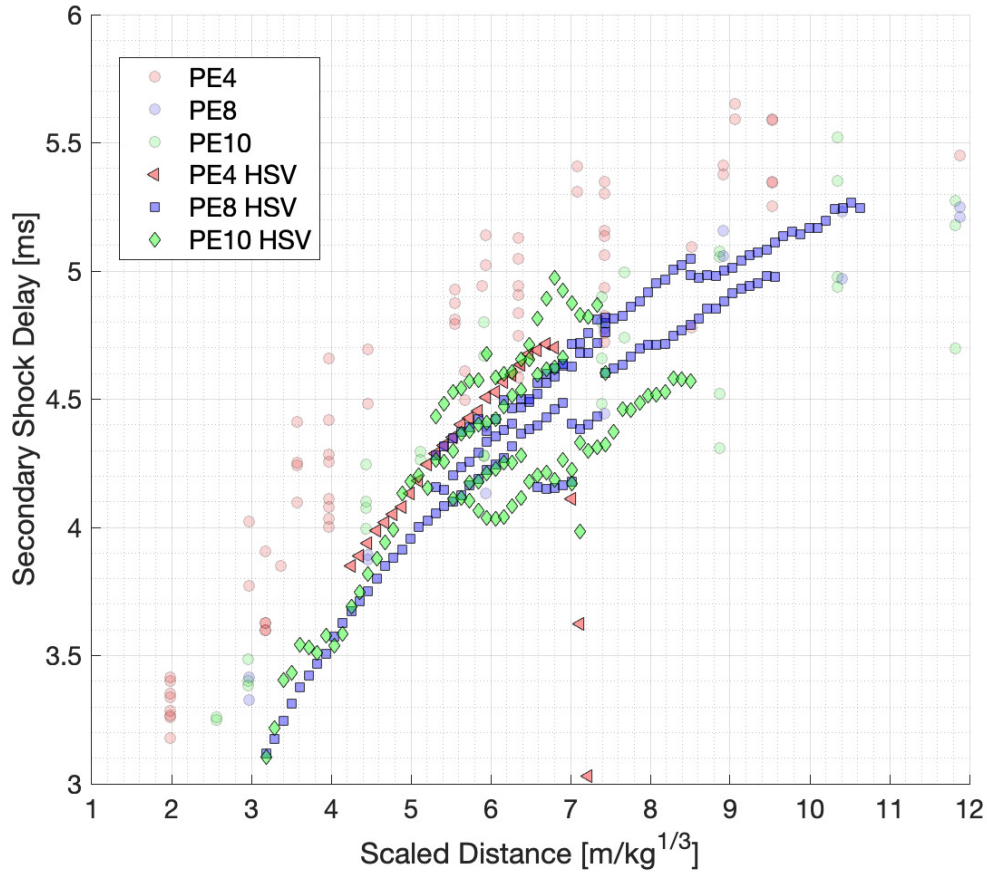


Figure 4.13: Interpolated and unscaled HSV secondary shock delay time from 250 g hemispherical ideal charges compared with pressure gauge recordings of PE4, PE8 and PE10

3189 A learning point from early testing analysis of PE4 and PE10 was highlighted when the
 3190 extracted secondary shock propagation data was either unusable or varied considerably,
 3191 as seen in Figure 4.13. This knowledge was considered for the PE8 trials which were only
 3192 conducted within favourable conditions resulting in all 8 shots providing credible data as
 3193 opposed to only 5 of 20 trials in total for PE4 and PE10 combined.

3194

3195 4.3.4 High Speed Video Summary

3196 This section makes efforts at implementing the tools developed in Chapter 3 to explore
 3197 the other data which can be extracted on the three ideal explosives tested. Highlighted is
 3198 the requirement for recording conditions to be considered prior to a test if only recording

3199 data using cameras. If lighting conditions are unfavourable most of the videos become
3200 difficult to extract data from without artificially editing image brightness which induces a
3201 margin of doubt in what data is being considered.

3202 Importantly, the HSV methods have shown to produce highly accurate results for explosive
3203 yield and secondary shock delay when compared with known experimental features and
3204 the corresponding pressure gauge recordings respectively. With the secondary shock is
3205 not fully understood, work of this kind which shows clear differences between incident and
3206 reflected delay data provides further insights which can be developed to fully characterise
3207 the physical process.

3208

3209 **4.4 Future Research of Explosive Characterisation**

3210 **4.4.1 Near-Field Loading**

3211 Whilst already conducted is an in-depth and rigorous analysis of ideal explosives within
3212 far-field regimes, there is limited published work considering near-field loading conditions.
3213 Even fewer of these contain experimental-based evidence for the definitive mechanisms re-
3214 sulting in variability of near field blast parameters. The fireball of detonation products in
3215 this region is still expanding and driving the shock wave and so interacts with a structure
3216 in unison. KB predictions in this region are poorly defined, with limited experimental data
3217 of which exhibits significant scatter and divergence to physics-based numerical models^[150].

3218

3219 To establish the root cause of the discrepancies between numerical simulation and the
3220 KB predictions, it is essential to record experimental data of both spatial and temporal
3221 blast parameter distribution in near-field regions. In doing so, insights into how energy
3222 is released during the initial stages of detonation can be made helping to characterise the
3223 near field effects on a composition basis.

3224

3225 With developments in near-field loading measurements discussed in Barr et al.^[10], a rigor-
3226 ous test plan should be undertaken in the future to characterise the temporal and spatial
3227 behaviour of detonation within close-proximity of a reflective surface to measure localised
3228 loading conditions. The experimental trials will require high levels of precision with blast
3229 parameters showing considerable sensitivity inaccuracies in detonator placement within

3230 close proximity of the charge. This data will be used to expand the bank of data used
3231 to establish the KB predictions and hopefully provided insights into the discrepancies be-
3232 tween that and numerical models. Future work should also build on that presented by
3233 Rigby et al.^[130], which looks at implementing similar HSV techniques but for near-field
3234 fireball expansion tracking.

3235

3236 4.4.2 TNT Equivalence

3237 Despite the vast quantities of published literature on TNT equivalence across a range
3238 of explosive compositions, there still holds discrepancies between researchers as to what
3239 realistic values are. This can only be a result of a general lack of consistency in exper-
3240 imental processes and analytics. What has never been fully justified in literature is the
3241 idea that TNT equivalence varies with scaled distance. Within near-field regimes where
3242 the propagating shock wave is sensitive to fireball thermodynamics, the statement that
3243 TNT equivalence varying may be valid. Compositions will decompose and combust at
3244 different rates which should result in variations to energy release when compared to that
3245 of TNT. However, in far-field regimes, the explosive detonation can be considered as a
3246 point source energy release^[167] and therefore once the shock wave detaches from the fire-
3247 ball, the localised differences in detonation thermodynamics can be ignored. Effectively
3248 at these greater scaled distances, shock waves are comparable based on the speed in which
3249 they propagate through free-air and thus is believed to exhibit a constant TNT equivalence.

3250

3251 The techniques developed in Chapter 3 when used on a variety of ideal explosives demon-
3252 strates highly consistent blast parameter data in the far-field range which compare well
3253 to KB predictions for TNT. This provides justification and validation of developing tech-
3254 nical procedures for data capture and analysis which considers the fundamental physical
3255 principles of shock wave propagation, relating to Objectives 2 and 3. With this in mind,
3256 the benefit of understanding how to improve our recorded data can be adopted to re-
3257 characterising other ideal explosives, attempting to quantify their effects and compare
3258 with KB predictions to establish realistic and consistent TNT equivalence values across
3259 the far-field range.^[164] undertook well-controlled blast trials using PE4 across a range of
3260 mass scales, and showed general consistency of shock wave parameters regardless the mass
3261 tested. Although these masses tested are beyond what is possible for the University of
3262 Sheffield testing site, the fact that ideal explosives behave consistently both in mass and

3263 distance in the far-field range provides justification for this work. In doing so, more opti-
3264 mised numerical simulations can be developed to better quantify the effects these explosive
3265 compositions could have in more complex environments.

3266

3267 4.5 Summary

3268 With experimental blast parameters providing the primary source of evidence to supple-
3269 ment the fundamental understanding of shock loading conditions on a structure, it is of
3270 paramount importance to characterise and quantify these effects resulting from a variety
3271 of compositions across a range of distances.

3272

3273 Stress-testing the tools developed in Chapter 3 using a large data set of small-scale hemi-
3274 spherical PE4 detonations against other explosives of similar idealistic behaviour was the
3275 main aim of this Chapter. Presented is a rigorous investigation into the effects various
3276 ideal explosives have on structures across the far-field range. The data was collected using
3277 both pressure gauges and high speed video recordings to further validate the methods and
3278 results through comparison of nominally identical shots.

3279

3280 The findings within Chapter 4 highlight the strength of KB predictions for both positive
3281 and negative phases of free-air shock wave parameters, specifically for far field scenarios,
3282 alongside the high consistency in output data from consistent well-controlled experimen-
3283 tal trials. Despite only testing 250g hemispheres, the findings from^[164] provide enough
3284 evidence that the results of consistency will transfer across mass and distance with the
3285 far-field range when considering ideal explosives of spherical shape. The methods of data
3286 capture and analysis have the potential to be tested further when considering other non-
3287 ideal explosives, charge shape effects and near-field loading conditions.

3288

3289 Chapter 5

3290 Non-Ideal Explosive Blast 3291 Characterisation

3292 5.1 Introduction

3293 Chapters 3 and 4 present validated experimental methodologies and analytical tools for
3294 accessing the far-field blast parameters from small scale ideal explosives in free-air, which
3295 have shown to improve the repeatability of recorded data and provide confidence in deter-
3296 ministic predictions of far-field detonation yields. Although having a fundamental under-
3297 standing of ideal explosives is required, in reality, common explosive threats are home-made
3298 due to the relative ease and low costs associated with acquiring their constituents. This
3299 has led to a critical need to understand the effects of detonating non-ideal explosives to
3300 improve civilian safety against potential terror attacks in more realistic situations.

3301

3302 This Chapter will exercise the validated methods of conducting experimental work, and
3303 the corresponding data analysis, discussed in Chapter 3, to explore the yields of small
3304 scale ANFO (Ammonium Nitrate 94% and Fuel Oil 6% mixtures). The overall aim is to
3305 characterise the composition across a range of scaled distances at a smaller mass range
3306 to quantify any differences in the behaviour between that and large scale charges. The
3307 processed data will be compared to numerical simulations to assess the validity of the
3308 underlying physics assumed within standard JWL equations of state. An assumption that
3309 non-ideal explosives vary on mass as well as scaled distance has been made, meaning a
3310 single value for TNTe can not be assumed. If proven correct within the findings of this

chapter, it would result in discrepancies between simulated non-ideal detonations and the results from experimental trials, motioning a requirement for characterisation revisions of non-ideal explosives to capture realistic physical mechanisms ongoing in small scaled charges.

5.2 Experimental Setup

In the attempt to quantify the effects of small scale ANFO detonations a variety of experimental techniques were conducted to investigate the non-ideal behaviour which may be exhibited. This section outlines the basis of these trials making use of previously validated methodologies in Chapter 3 for far-field studies but introducing a new technique of analysis for rate-stick trials.

5.2.1 Far-Field Trials

A total of 37 far-field experiments were performed at the University of Sheffield (UoS) Blast and Impact Laboratory in Buxton, UK. These trials were conducted using surface detonated hemispherical charges with mass varying between 250-1000g comprising of two different chemical compositions (Ammonium Nitrate Fuel Oil (ANFO): 35 tests and Ammonium Nitrate (AN): 2 tests). Both explosives were commercially purchased from EPC-UK to provide comparable studies both of which presented material properties as seen in Table 5.1 quoted by the supplier.

Parameter	Ammonium Nitrate + Fuel Oil (ANFO)	Ammonium Nitrate (AN)
Composition	92.5 - 95% AN, 5.0 - 7.5% FO	99% AN, ~1% Moisture
Density	0.8 g/cm^3	0.73 - 0.89 g/cm^3

Table 5.1: EPC-UK material properties of ANFO and AN which have proven useful for this body of research and article

The charges were boosted with a 3g sphere of PE10 to ensure reliable shot-to-shot detonation^[112], and were done so at standoff distances of 1–9m from gauge instrumentation. It is important to note that a secondary objective from these trials considered the position of the booster/detonator and their effects on non-ideal explosive yield. With this in mind, 16 of the trials were top detonated and 21 bottom detonated. Figure 5.1 is a schematic

3336 displaying the different booster and detonator positions adopted for this testing regime.
3337 For each of the far-field trials, clear plastic sheets were heated and vacuum-formed around
3338 a 3D printed hemispherical mould, to provide a lightweight casing to facilitate negligible
3339 confinement of the ANFO prill whilst keeping a consistent shape. This was implemented
3340 to compare the results with the large bank of ideal explosive data available within the
3341 literature^[49]. A small circular cut was made in the top of the vacuum formed plastic
3342 casing to enable a consistent method of filling the charge with prill.

3343

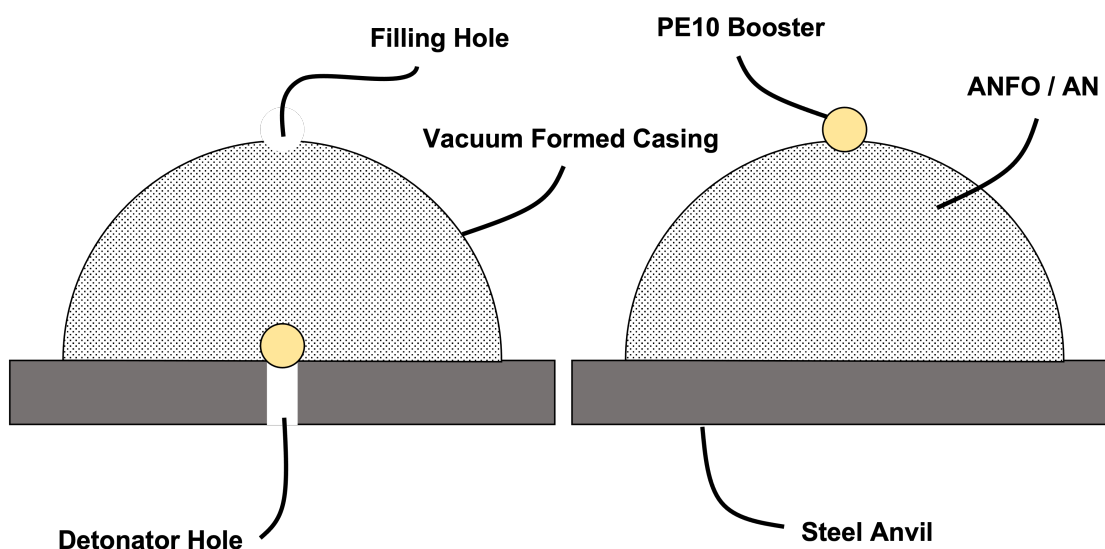


Figure 5.1: Schematic of the charge and booster locations used throughout this testing regime

3344 The far-field experimental methodology for the non-ideal tests mirrors that discussed in
3345 Chapter 3. The explosive charges were detonated at varying stand-off distances, R_a and
3346 R_b in Figure 3.3, perpendicular to two rigid reflective surfaces in the form of a reinforced
3347 concrete bunker and a blockwork wall, separated exactly 10.0 m apart. Data was recorded
3348 using the same collection of Kulite HKM-375 piezo-resistive pressure gauges as were used
3349 for the recording of the ideal explosive detonations. The gauges were threaded through,
3350 and made flush to the surface of a small steel plate (approximately $110 \times 150 \times 10$ mm) which
3351 was fixed to these walls. The charges were placed on a small steel plate ($150 \times 150 \times 25$ mm)
3352 prior to detonation, in order to avoid repeated damage to the concrete testing pad.

3353

3354 For bottom detonated trials, the steel anvil had a machined hole through the centre to
3355 allow for the detonator to be in contact with the PE10 booster. The pressure was recorded

3356 using a 16-bit digital oscilloscope and TiePie software, with a average sampling rate of
3357 195 kHz at 16-bit resolution. The recording was triggered automatically using TiePie's
3358 'out window' signal trigger on a bespoke break-wire signal, formed by a wire wrapped
3359 around the detonator. The 'out window' trigger initiated with a voltage drop outside the
3360 normal electrical noise experienced in the break-wire. This coincides with the detonation
3361 of the charge breaking the circuit.

3362

3363 Data was recorded using the same collection of piezo-resistive pressure gauges as were
3364 used for the recording of the ideal explosive detonations. It is important to note that 30 of
3365 the ANFO trials were undertaken as part of a separate research project funded by CPNI
3366 (Centre for the Protection of National Infrastructure) and DSTL (Defence Science and
3367 Technology Laboratory) that has been granted permission of use in this thesis and hence
3368 denoted throughout.

3369



Figure 5.2: Top detonated arrangement for hemispherical ANFO charges boosted with 3g PE10 sphere; the yellow tubes are installed as guides for the detonator to be positioned vertically into the booster

3370 The Photron FASTCAM SA-Z high speed video camera, fitted with a Tamron SP AF70
3371 70-200mm zoom (F2.8) lens, and a zebra board which ran perpendicular to the blockwork
3372 and bunker walls, was again utilised to capture far-field shock propagation. In conjunction

3373 to this, a Shimadzu ultra high speed video camera was used to record detonation wave
3374 propagation and fireball breakout of the explosive to provide insights in the early stages
3375 of the event.

3376

3377 5.2.2 Rate Stick Trials

3378 As an attempt to characterise ANFO small scale shots across a whole range of parameter
3379 metrics, there was a need to establish the velocity of detonation (VOD) of ANFO and
3380 how it develops within a charge. To evaluate this, 6 rate stick trials were undertaken
3381 making use of two Shimadzu ultra high speed cameras. ANFO prill was contained within
3382 300mm long, 3mm walled clear acrylic/PMMA (Polymethyl-Methacrylate) pipes which
3383 had varying diameters in the attempt to establish the critical diameter of ANFO and the
3384 explosives corresponding VOD. The three internal diameters of pipe tested were 50mm,
3385 75mm and 102mm which correspond to ranges quoted within published literature for the
3386 critical diameter of ANFO [20,23,24,47,97].

3387

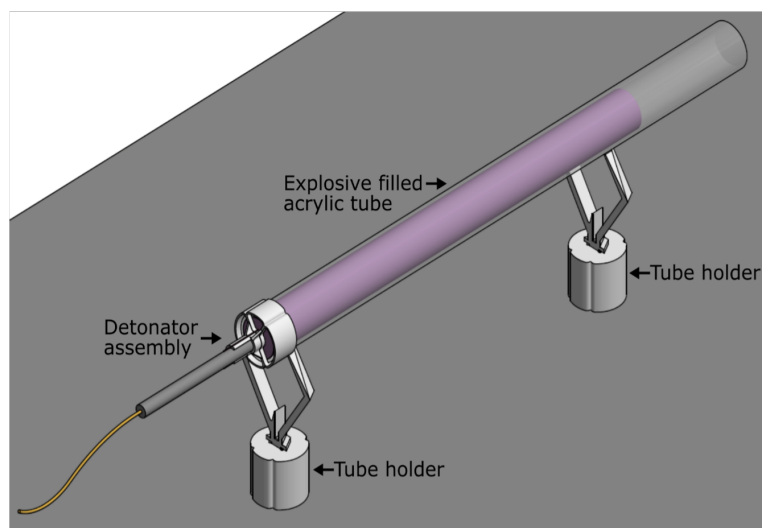


Figure 5.3: General arrangement of the rate stick trials using a 3mm wall thickness PVC clear tube, a 3D printed end cap which allows for detonator and booster placement and a reference measurement guide to enable distance tracking in the high speed videos

3388 The pipes were sealed using a 3D printed detonator holder at one end and a lightweight
3389 fibreglass material at the other to provide minimal confinement at the pipes extents as
3390 seen in Figure 5.3. To trigger the camera, a breakwire system was wrapped around the
3391 detonator so that upon detonation a voltage drop is experienced, triggering the camera

3392 to record. The general arrangement of the detonator assembly can be seen in Figure 5.4.
3393 Five out of the six trials utilised a 3g PE10 booster to initiate the detonation process to
3394 be comparable with the data collected within the far-field studies. The final test looked
3395 at a non-boosted 50mm diameter rate stick trial.

3396

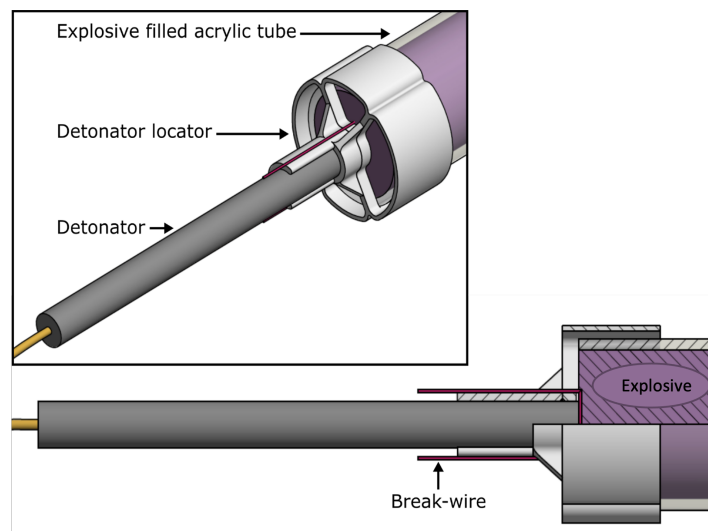


Figure 5.4: General arrangement of the rate stick trials using a 3mm wall thickness PVC clear tube, a 3D printed end cap which allows for detonator and booster placement and a reference measurement guide to enable distance tracking in the high speed videos

3397 5.3 Pressure Gauge Results

3398 5.3.1 Positive Phase

3399 Having little knowledge on the yield and characteristics of small scale ANFO charges, it
3400 was essential to establish an understanding on the general qualitative behaviour recorded
3401 during testing. The aim of this is to assess if non-ideal explosives behave as consistently
3402 as ideal explosives, as seen in Figure 3.5.

3403

3404 Presented in Figure 5.5 are 18 individual pressure gauge recordings as a result from 250g
3405 hemispherical ANFO detonations. These were conducted across a variety of standoff
3406 ranges tested alongside varying detonator position between top and bottom detonated.
3407 For data recorded between 2–9m, it is evident that although a discrepancy in the arrival
3408 times is consistently recorded (discussed further in Section 5.3.2), the overall trends in the

3409 positive phase of the pressure-time histories show considerable agreement. This ultimately
3410 suggests that within far-field regions, the detonator position is irrelevant. A similar level
3411 of consistency is also to be experienced between non-ideal and ideal explosives. The traces
3412 at 1m standoff do present variability in the overall peak overpressure recorded. This has
3413 been directly linked to the spherical uniformity and irregularity of the detonation cloud
3414 breakout related to bottom and top detonated tests respectively.

3415

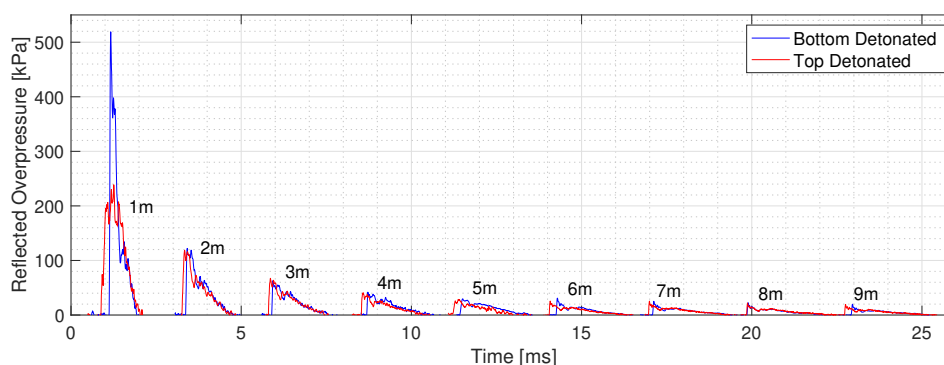


Figure 5.5: Compilation of raw data set of 250g ANFO hemispherical ground bursts, comprising of both top and bottom detonated charges, across the entire range of standoff distances tested in this regime. Positive phase is presented only.

3416 To establish the behaviour small scale ANFO explosives exhibit when compared to TNT, it
3417 was important to adopt the same experimental analysis tools, validated for ideal explosives.
3418 This is undertaken to assess how the tools perform for non-ideal explosive detonations.
3419 Since Friedlander-like behaviours are presented in Figure 5.5, the curve fitted method was
3420 used to process the small scale ANFO data. The caveat is that this accounted for the gauge
3421 plate ringing (the phenomena discussed in Section 3.2.2) as the blast arena amendments
3422 had not been made prior to these trials being conducted. The output blast parameters
3423 resulting from the curve fitting method were collated. A TNT equivalency assessment
3424 made using an MAE analysis, comparing the ANFO data set to semi-empirical predic-
3425 tions based on varying masses of TNT. As per the findings of Section 3.2.1.1, the positive
3426 phase duration exhibited higher levels of variability when assigning a specific value to the
3427 parameter. This was directly linked to noise in the signal, making the true point at which
3428 conditions return briefly back to atmospheric difficult to precisely determine. In common
3429 practice, an inaccurate time duration can be remedied by tailoring the decay coefficient
3430 in the Friedlander equation such that the specific impulse is preserved. With this noted,
3431 positive phase duration was omitted from the MAE analysis in line with the previous
3432 assessment on ideal explosives.

3433

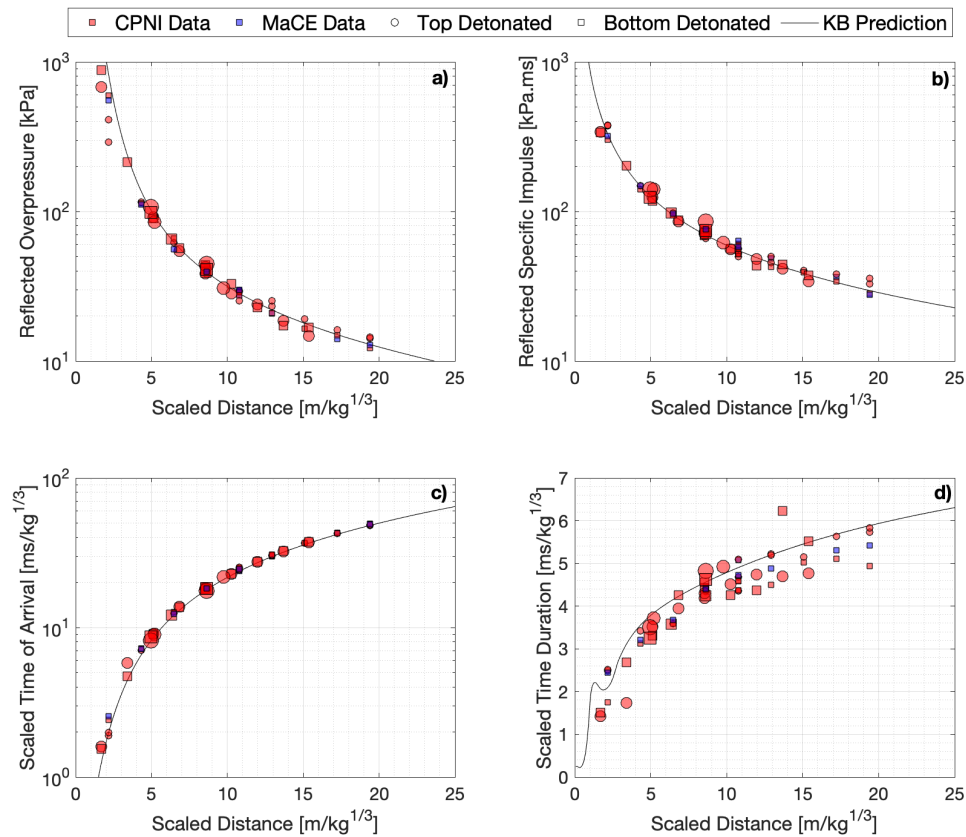


Figure 5.6: Compiled scaled blast parameters from ANFO explosive trials as a function scaled distance, compared with KB predictions, using a $TNTe=0.395$, with increasing marker sizes to indicate the mass of charge: a) Peak reflected overpressure, b) Scaled reflected peak specific impulse, c) Scaled arrival time, d) Scaled positive phase duration

3434 To be in line with variability regions hypothesised Tyas^[173], the MAE analysis was per-
 3435 formed on all AN-based tests within the definitive far-field range. The aim was to attribute
 3436 a $TNTe$ factor at which the shock wave is no longer affected by ongoing fireball chemistry,
 3437 instabilities and detonation cloud propagation. To establish the effects of detonator po-
 3438 sition and charge mass, the MAE analysis was undertaken for the data when separated
 3439 into both nominally identical tests alongside the full data set. The results from the MAE
 3440 analysis of peak reflected pressure and specific impulse converged on a $TNTe = 0.395$
 3441 across the entire range of scaled distances, for both charge mass and detonator position
 3442 groupings. Arrival time however, presented a $TNTe = 0.28$ and 0.295 , for bottom and top
 3443 detonated trials respectively. This suggests that the shock waves resulting from small scale
 3444 ANFO detonations propagate slower than an equivalent TNT mass with approximately

3445 40% of the mass of the detonated ANFO charge. This offset in arrival time, is visually
3446 represented in Figure 5.5, and will be discussed in detail in Section 5.3.2. The arrival time
3447 of the shock waves will however be treated with caution for discussions relating directly
3448 to explosive yield.

3449

3450 To assess the explosive yield alone, the $TNTe = 0.395$ was applied to each of the recorded
3451 blast parameters and presented in Figures 5.6a-d which show a striking agreement between
3452 each experimentally recorded blast parameter (expressed as a TNT equivalent mass), and
3453 the KB predictions. It is important to note that there were 2 preliminary trials which
3454 resulted in much lower explosives yields related to not using a PE10 booster, which have
3455 been removed from the presented data to avoid speculative comparison of different testing
3456 regimes. This finding does however provide enough evidence to suggest that small scale
3457 ANFO charges cannot be reliably detonated without the presence of a booster charge.

3458

3459 Interestingly to note is the behaviour of the peak pressure and specific impulse values in
3460 near field scenarios, $Z < 3m/kg^{1/3}$, when comparing top and bottom detonate shots in
3461 Figures 5.6a and 5.6b. The top detonated peak pressure values recorded in the exper-
3462 imental trials seem When comparing peak pressure and specific impulse values, the top
3463 detonated trials exhibit different evaluated $TNTe$ factors when compared directly to KB
3464 predictions. The bottom detonated trials present agreement in $TNTe$ for both parame-
3465 ters with KB predictions. This suggests that ANFO small scale charges are sensitive to
3466 detonator position in the near-field due to differences in the detonation product mechan-
3467 ics occurring on breakout in line with the findings presented in Figure 5.5. It can be
3468 concluded that the energy release of an explosive, and therefore the specific impulse, is
3469 independent of detonator position. The peak pressure however is directly related to the
3470 fireball breakout and propagating shock wave shape.

3471

3472 What can be deduced from this finding however is that specific impulse provides a more
3473 representative $TNTe$ across all explosives and scaled distances regardless of whether a
3474 ‘Friedlander’ shock wave has fully formed. Specific impulse loading is directly related to
3475 the total energy released from detonation rather than pressure which is time-dependant
3476 based on shock wave forming. For the near-field it is important which parameter is con-
3477 sidered but far-field can make use of both to establish equivalency in line with findings
3478 presented by Bogosian et al.^[16].

3479

3480 The results presented in Figures 5.6a-d contradict quoted values of TNTe for ANFO be-
 3481 ing around 0.82 across all scaled distances and masses, therefore posing implications for
 3482 numerical modelling which have incorporated these assumptions within simulations. To
 3483 assess the variability of the data set, it was scaled by mass alone and compared to vary-
 3484 ing KB prediction curves using a TNTe = 0.3-0.5, presented in Figure 5.7. A TNTe =
 3485 0.395 ± 0.1 does agree with the results presented in Figure 2.14 by Figuli et al. [52], sug-
 3486 gesting that a $\pm 10\%$ spread about the MAE evaluated TNTe could be related to inherent
 3487 variability in ANFO trials as result of a combination of the factors mentioned in Section ??.

3488

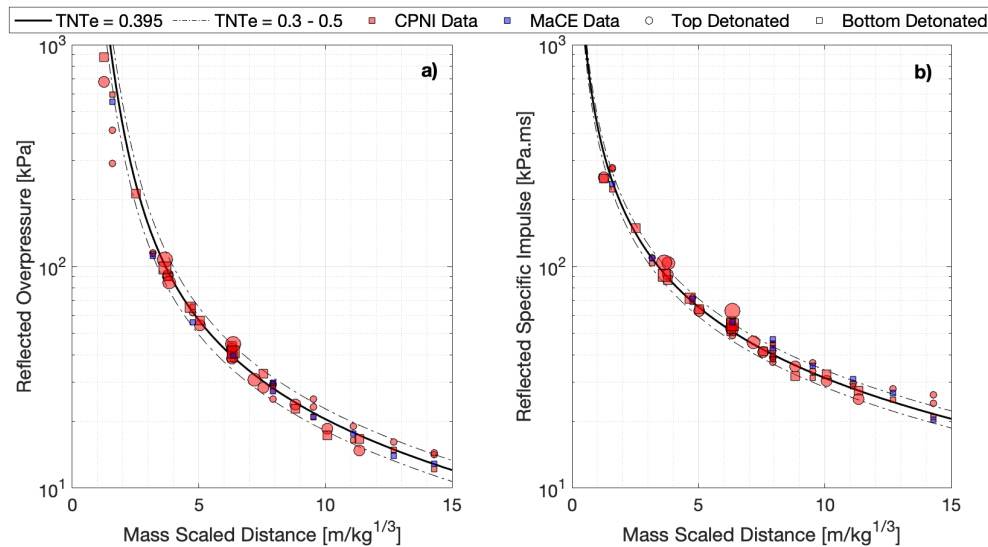


Figure 5.7: Compiled mass scaled blast parameters from ANFO explosive trials as a function mass scaled distance, compared with KB predictions, using a $TNTe=0.395 \pm 0.1$ to present variability bounds, with increasing marker sizes to indicate the mass of charge: a) Peak reflected overpressure, b) Scaled reflected peak specific impulse

3489 Across a variety of experimental trials, it would become evident if the gauge data was
 3490 invalid, especially where gauges were swapped out depending on their pressure rating and
 3491 where the charge was situated, providing confidence in the recorded and processed blast
 3492 parameters. The confidence in the data provides us with enough evidence to suggest that
 3493 non-ideal explosives are not scalable across all masses and standoff distances and therefore
 3494 KB predictions may not hold as much credibility when assessing these types of explosives.

3495

3496 For ANFO in particular, larger charges masses (i.e a greater self-confinement and longer
 3497 distance for detonation wave to travel without energy losses) foster the full detonation and

3498 combustion of both AN and FO components of the composition, but in smaller charges
3499 a percentage of the energy is not fully released prior to the propagation of the air shock.
3500 The question which begins to be highlighted is why ANFO is described to exhibit a TNTe
3501 of around 0.82 across all scale distances and little further experimental work undertaken
3502 to assess the robustness of this factor across a range of masses and scaled distances other
3503 than that presented in Figure 2.14. The implication this has on the fundamental under-
3504 standing of non-ideal explosives is significant and will result in any numerically modelled
3505 simulation of ANFO of low charge mass will result in over predictions in the behaviour.
3506 The contents of this chapter will make efforts at assessing the robustness of KB predictions
3507 against small mass-scale ANFO charges whilst making comparisons to the historical data
3508 captured from large scale charges.

3509

3510 5.3.2 Arrival Time Offset

3511 Using the findings from the MAE analysis on peak reflected pressure and specific impulse,
3512 KB predictions of the positive phase, using a $TNTe = 0.395$, were overlaid on the raw
3513 data to evaluate whether the equivalency factor provided qualitatively similar trends in
3514 Figure 5.8. Whilst both the pressure and impulse trends are captured reasonably well, the
3515 arrival of the recorded shock waves is much later than an equivalent TNT blast.

3516

3517 With further inspection, the MAE analysis for arrival time highlighted TNTe being 0.28
3518 and 0.295 as providing more comparable scaled arrival time values to TNT predictions for
3519 both bottom and top detonated trials respectively which equates to on average ~ 0.3 ms
3520 delay in the arrival time in comparison to the TNTe establish for pressure and impulse.
3521 This suggests that despite exhibiting TNT equivalences of around 0.395 for energy related
3522 parameters, time-based shock wave parameters from small scale ANFO charges are equiv-
3523 alent to even weaker detonations and therefore arrival later. It is has been hypothesised
3524 that the delay in arrival time to be a result of a combination of different factors, of which
3525 will be discussed further.

3526

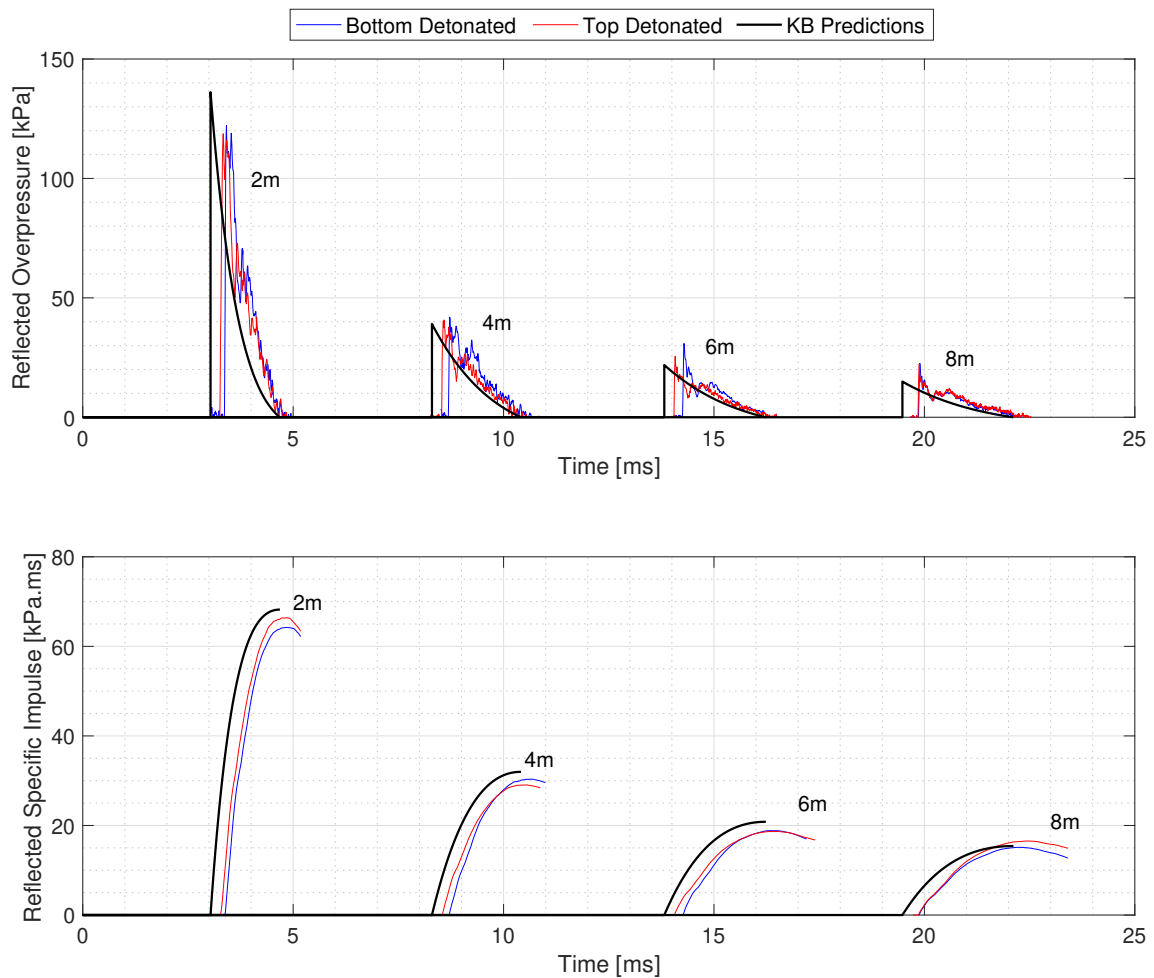


Figure 5.8: Compiled raw data from both top and bottom detonated trials at 2m, 4m, 6m and 8m standoff distance compared to KB predictions assuming a $TNTe=0.395$: Pressure-time history (Top) and Impulse-time history (Bottom), showing only positive phase

3527 5.3.2.1 Mesostructure of ANFO prill

3528 The heterogeneity of small scale ANFO charges, when compared to homogenous cast TNT
 3529 explosives used to develop the KB prediction tools, is a factor which could affect the ve-
 3530 locity of the detonation wave and therefore resulting blast parameters. Zygmunt and
 3531 Buczkowski^[183] investigated the influence of ANFO prill characteristics on detonation ve-
 3532 locities and energy release and found that the prill structure and the size in these types
 3533 of charges are massive influencers. Salyer et al.^[140] compared the detonation velocities
 3534 of two commercial ANFO types which have vastly different prill sizes and mesostructures
 3535 when prepared. The results suggested that the small prill resulted in much more ideal
 3536 detonation behaviours relating directly to reduction in air voids allowing for a more stable

3537 expansion of the detonation wave.

3538

3539 Petes et al.^[112] discusses this inhomogeneity characteristic as being the potential reason-
3540 ing for non-spherical propagation anomalies in the shock wave and fireball. This results
3541 in quoted blast parameter variability spreads of $\sim 7\%$ when comparing nominal trials
3542 detonating cast TNT, which is endorsed by^[9] who discusses instability formation relating
3543 to imperfections in the charge surface and homogeneity.

3544

3545 Considering large scale ANFO charges, it is highly likely the bulk density increases due
3546 to more centrally located prill layers exhibiting void collapse through self-weight compres-
3547 sion. Petes et al.^[112] detailed that an increased bulk density will result in faster detonation
3548 velocities and thus the resulting blast parameters exhibiting larger yields for large scale
3549 charges. McKay et al.^[99] and Dobrilovic et al.^[41] both experimented with reducing the
3550 density of ANFO by removing prill and exchanging it with polystyrene balls to assess the
3551 changes to the velocity of detonation in rate stick trials. Zygmunt and Buczkowski^[183]
3552 showed that increasing the density of ANFO charges using ground prill much higher deto-
3553 nation velocities were experienced, which with alongside the aforementioned articles show
3554 agreement with the theory stated by Petes et al.^[112].

3555

3556 Conversely, the works presented by Fabin and Jarosz^[47] undertook analytical studies in
3557 trying to improve the explosive yield of ANFO charges and discovered the morphology
3558 and prill size distribution did not necessarily follow the same theory. A dependence on
3559 FO absorption features of AN has shown to result in lower detonation velocities for higher
3560 density ANFO, therefore inducing variability in detonation mechanics. Sadwin and Swis-
3561 dak^[139] reinforced these ideas when it was reported that layer-to-layer density differences
3562 were measured throughout their large scale ANFO charge through varying detonation
3563 velocities and pressures. Consequentially to these density regions, hydrodynamic insta-
3564 bilities would develop leading to a considerable amount of internal turbulence and thus
3565 a degree of detonation wave propagation impedance when compared to cast TNT, all of
3566 which would result in spreads in explosive yield.

3567

3568 For these small scale charges, where the mass itself is not enough to provide reasonable
3569 compaction to induce full/ideal detonation of the ANFO, the air voids between each prill
3570 are believe to have a impedance influence on velocity of detonation. It is assumed that

3571 cellular disturbances of the detonation wave are generated when propagating through the
3572 prill structure, fostering multiple interactions within the complex mesostructure and dis-
3573 rupting the smooth isentropic expansion of the detonation wave and products.

3574

3575 Mi et al.^[102] numerically modelled the shock-to-detonation transition in nitromethane, a
3576 non-ideal heterogeneous explosive, when incorporating different arrangements of air voids
3577 and applying varying energy shocks to the medium to induce initiation. This model pre-
3578 sented delays during the initiation process when cavities were present in comparison to
3579 a homogeneous mixture. This provides further understanding to the complexity of the
3580 detonation process and some validation to the delays seen in Figure 5.8a-b.

3581

3582 Numerical simulations undertaken by Baer et al.^[6] demonstrated that heterogeneous ma-
3583 terials, like ANFO, which undergo shock loading, do not exhibit one jump state as de-
3584 picted in ideal explosives but consist of a distribution of states. In the early stages of
3585 being shocked, ANFO behaves elastically, with temperature and stress corresponding to
3586 those defined by the shock hugoniot of ammonium nitrate. As the the shock wave travels
3587 through the complex mesostructure, the distribution of states show elastic and plastic de-
3588 formation effects, corresponding to induced ANFO prill pore and air void collapse behind
3589 the shock front. Thereafter, when all available void space is filled, compression waves
3590 traverse through the materials that eventually equilibrate to a spatially-uniform stress
3591 state which was also hypothesised by McKay et al.^[99]. When directly comparing this to
3592 a ideal homogeneous explosive, which jumps states effectively instantaneously, the time
3593 difference between those reactions could also have an effect on the offset seen in the small
3594 scale charges.

3595

3596 The ANFO used in these trials had a bulk density of $0.824\text{g}/\text{cm}^3$ measured using a known
3597 volume and mass container, which is significantly lower than an individual prills density,
3598 quoted as between $1.2 - 1.4\text{g}/\text{cm}^3$ ^[138]. Density, or air voids percentage, is documented as
3599 having a direct relationship with the detonation velocity and thus the resulting shock wave
3600 parameters^[154]. Adopting rudimentary calculations, the ratio between the bulk density of
3601 a given amount of ANFO and an individual prill results in around 60% – 70%, suggesting
3602 30 – 40% is attributed to the air voids which is in line with quoted amounts by Cetner
3603 and Maranda^[24]. If the ratio of measured charged density to potential density is applied
3604 to the TNTe factor established for the best fit of reflected pressure and specific impulse,

3605 a new $TNTe \simeq 0.246 - 0.328$ is established, which is in line with the MAE analysis for
3606 arrival time when compared to KB predictions.

3607

3608 The arrival time delay could be attributed to the complex mesostructural interaction be-
3609 tween detonation wave, ANFO prill and the air voids. This phenomena will be much
3610 more prominent in small scale charges, with a lack of self-compaction, when compared to
3611 homogeneous TNT charges.

3612

3613 5.3.2.2 Detonator Position

3614 Comparisons between top and bottom detonated charges were undertaken to establish the
3615 delay in the shock wave arrival time related to detonator position. Using the $TNTe$ value
3616 of 0.395, the scaled distances and arrival times for the small scale ANFO trials were cal-
3617 culated, and through the use of interpolation, the corresponding KB prediction of arrival
3618 time could be established. The difference between the recorded scaled arrival times and
3619 the KB predictions was the offset recorded and presented in Figure 5.9.

3620

3621 Interestingly the findings, when comparing top and bottom detonated arrival time offset,
3622 show a consistently larger offset value attributed to bottom rather than top detonated
3623 charges, whilst also exhibiting a smaller degree of variability. This finding suggests that
3624 top detonated charges result in shock waves which travel faster than a bottom detonated
3625 charge. The velocity of detonation in ANFO has been shown to increase with charge ra-
3626 dius^[12,43,80,98] up to a maximum value and thus in top detonated charges, the detonation
3627 wave has more explosive to pass through prior to breakout. The correct orientation of the
3628 detonator when used in the top detonated scenario was subjected to human error due to
3629 being free-standing in the testing methodology. Conversely, the bottom detonated charges
3630 passed through a hole in a steel anvil, only large enough for the detonator to fit, therefore
3631 improving the directionality and thus showing a higher degree of consistency in the offset
3632 parameters at each given scaled distance seen in Figure 5.9.

3633

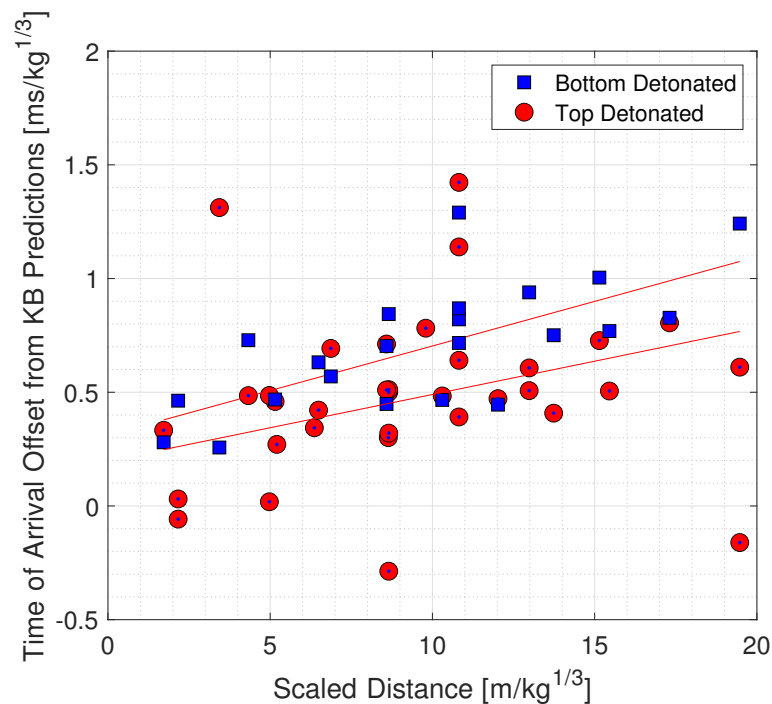


Figure 5.9: Scaled time of arrival from experimental work offset from KB predictions based on a $TNTe=0.395$, against scaled distance for top and bottom detonated tests with reference best fit lines

3634 Figure 5.9 does highlight that the offset in arrival time compared to KB increased with
 3635 scaled distance for both top and bottom detonated charges. It is hypothesised that this is
 3636 a feature of the ANFO shock wave de-accelerating faster than an equivalent TNT explosive
 3637 based on the 39.5% equivalency. Despite detonator position making slight differences to
 3638 the arrival time offset, this is second order mechanism and a more fundamental process
 3639 causes the discrepancy in arrival time data.

3640

3641 5.3.2.3 Proportionality in the Velocity of Detonation with Charge Radius

3642 The offsets in arrival time tends to increase with scaled distance in both scenarios of det-
 3643 onator position. This suggests for small scale ANFO charges the shock wave velocity is
 3644 decaying faster than what a TNT charge scaled accordingly would. This finding is indica-
 3645 tive of the relationship presented in Figure 5.6c which suggests that the decay of shock
 3646 wave velocity is on a different gradient to what is experienced in TNT.

3647

3648 The factors mentioned above all contribute to the main overarching reason for the arrival

3649 time offset but these are believed to be second order to the real mechanism resulting in the
3650 magnitudes of offset exhibited. TNT has a detonation velocity of around 6940m/s when
3651 compared to the published figure of approximately between 4250-4800m/s for ANFO, again
3652 related to large scale trials. However, TNT reaches this quoted maximum speed effectively
3653 instantaneously due to its ideal-like detonation characteristics. ANFO detonations on the
3654 other hand requires a specific amount of time in which the charge is sustaining, and not
3655 losing energy through propagation of products (i.e., similarly to being confined), allowing
3656 for a steady-state detonation to occur exhibiting the quoted velocity. It is therefore a rea-
3657 sonable assumption that comparing the internal conditions of small scale ANFO charges
3658 to TNT will result in slower detonation velocities at the instant of initiation, with the gap
3659 closing as charge diameter increases.

3660

3661 Figure 5.10 displays a schematic of what is hypothesised to be occurring in the detonation
3662 procedure of ANFO when compared to TNT. Upon the detonation of TNT, the velocity
3663 of the shock wave travelling through the explosive is fairly constant until breakout. At
3664 this point the propagation of the detonation products and air shock results in progressive
3665 energy loss and reductions in speed. Conversely for ANFO, the maximum velocity is not
3666 achieved instantaneously with detonation, but instead ramps up based on the secondary,
3667 non-ideal combustion of the FO component of the charge which occurs over a longer time
3668 scale. Depending on charge size, the breakout of detonation products could happen any-
3669 where along the curve ramping up to the maximum velocity of detonation of ANFO and
3670 then will behave similarly to TNT as a free air shock, just at a lower velocity.

3671

3672 In the experimental trials presented within this thesis, the ANFO charges would fall into
3673 smaller scale behaviours meaning that the when comparing the trend to TNT, it is clear
3674 that the velocity of the shock reduces faster in the ANFO detonation. This would there-
3675 fore create an artificial offset to arrival time data as the TNT shock wave travels faster
3676 for longer. Eventually the offset would level off when the changes in velocity with respect
3677 to time are almost the same (i.e when then curves begin to plateau and converge). The
3678 larger the ANFO charge, the closer to the maximum detonation velocity and therefore
3679 less of an offset would be induced. The initial acceleration of velocity in ANFO would
3680 result in some error arrival time within a close proximity to the charge which begins to
3681 even out at greater scaled distances. This exact feature is seen in Figure 5.6b, from the
3682 large scale ANFO trials that exhibits recordings in arrival time which are greater than KB

3683 predictions in near-field scaled distances.

3684

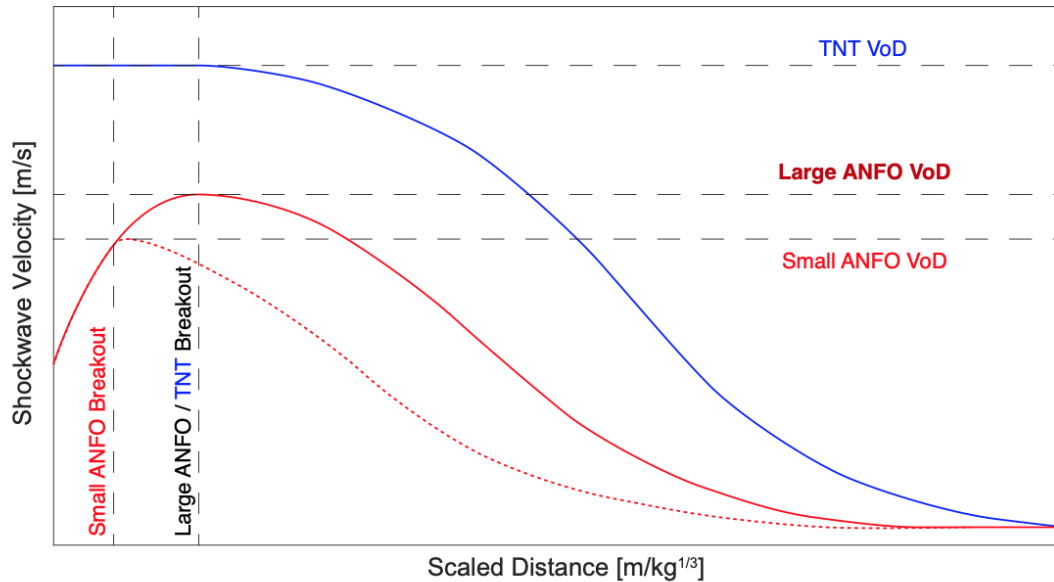


Figure 5.10: Theoretical schematic of different detonation and shock wave velocities with respect to scaled distance for TNT (Blue) and ANFO charges of various sizes (Red), with reference to the times at which detonation breakout occurs respectively to one another.

3685 The initial offset is something which can be quantified using evaluated shock wave data for
 3686 a TNT charge from KB prediction. If a velocity at detonation breakout can be recovered
 3687 from these small scale ANFO charges then so can a time in which TNT would take to
 3688 reach that same velocity. This would provide an initial offset at small scale distances and
 3689 will be revisited in Section 5.4.1. Quantification of the increasing offset seen in Figure 5.9
 3690 is only possibly through understanding the behaviours of the shock wave from the moment
 3691 it detonates up to far-field ranges.

3692

3693 5.3.2.4 Summary of Arrival Time Offset

3694 Whilst all of the above offer a reasonable hypothesis as to why the captured arrival time
 3695 data holds discrepancies to KB predictions, it is important to test these through other
 3696 methods of analysis to provide further validation. Whilst varying detonator positions in a
 3697 robust testing plan would be of interest to assess the variability of explosive yields within
 3698 the near field, the results would not produce a significant progression in the characterisa-

3699 tion of small scale ANFO charges.

3700

3701 Ensuring the exact formation and prill-to-prill interactions between tests is difficult for
3702 ANFO charge and therefore variations related to this are accepted as inherent to the
3703 heterogeneity of composition. With strict control measures the differences in prill size,
3704 charge density and oil absorption could be considered but it is believed to be a second
3705 order mechanism when establishing why the offset in arrival time for ANFO charges is so
3706 large.

3707

3708 The proportionality in the velocities of detonation between TNT and ANFO, and their
3709 general trend prior to detonation product breakout are essential for understanding the
3710 arrival time offset. HSV techniques can be used to track the velocity of both the fireball
3711 expansion and the resulting shock wave propagation from the ANFO charges which is
3712 discussed further in Section 5.4.1.

3713

3714 5.3.3 Disagreement to Published TNT Equivalence

3715 The fact this small scale testing regime resulted in a generally low TNT equivalence (TNTE
3716 $\simeq 0.4$) for ANFO is a contradiction to the widely accepted figure of approximately 0.80
3717 within the available published literature. Giglio-Tos and Reisler^[61] undertook a experi-
3718 mental research consisting of the detonation of three large scale, 20-100 Ton, hemispherical
3719 ANFO charges which recorded blast parameters using pressure transducers and magnetic
3720 tape recording systems, resulting in a TNTE = 0.83. Petes et al.^[112] expressed the desir-
3721 ability to use ANFO for explosive testing and provided a detailed compilation of historic
3722 testing results. These concluded on the idea that ANFO had approximately 80% of the
3723 energetic output of TNT. This thesis begins the discussion around why small scaled ANFO
3724 charges present a much lower energetic output than historically reported and what the
3725 real world implications of this are. The list below were the preliminary hypotheses based
3726 on the experimental results:

3727

3728 **5.3.4 Was the composition of the explosive actually ANFO?**

3729 Findings from published literature details that purely ammonium nitrate detonation re-
3730 sults in around $TNTe = 0.32-0.4$ [3,28,110,131,158]. This has led to considering the possibility
3731 the composition of ANFO being incorrect and fuel oil not being present as requested on
3732 purchase. With this in mind, the composition was tested using a gravimetric analysis, sep-
3733 arating the substance using petane, resulting in a yield of $5.55\pm 0.09\%$ and $94.45\pm 0.09\%$
3734 of the tested mass found to be fuel oil and ammonium nitrate respectively.

3735

3736 During the detonation of ANFO consisting of 6% FO, theoretically an ideal and stoichio-
3737 metrically balanced reaction occurs, meaning the oxygen molecules released as a result of
3738 the AN decomposition are enough to cause the full deflagration of FO. In our trials, with
3739 a lower percentage of FO at 5.5%, the detonation would result in a reduction in energy
3740 released $\simeq 5\%$ [92] through the development of nitrous oxides rather than full combustion
3741 occurring. This however does not account for the 50% effective energy loss exhibited in
3742 the recorded data. This finding led to the hypothesis being removed from consideration.

3743

3744 **5.3.4.1 Does the scale of the charge affect the chemical reactions occurring** 3745 **during detonation?**

3746 Due to the fact the FO constituent of the composition not being chemically bonded to
3747 the AN, there is a lag in the reaction time, related to reaction zone size. If the detonation
3748 wave is to reach the extent of the charge prior to this chemical reaction fully occurring,
3749 the potential energy release is considered to not be fully achieved.

3750

3751 In small scale charges, it was discussed whether the aforementioned reaction actually oc-
3752 curs at all, and therefore the yield of these tests is the result of only AN detonating. To
3753 test this hypothesis, Ammonium Nitrate prill was purchased from the same provider and
3754 tested under an identical methodology to the ANFO shots (250g hemisphere and boosted
3755 with a 1g sphere of PE10). In these trials no detonation occurred in any of the pure
3756 AN shots despite being boosted. This result was evidence to suggest the fuel oil oxygen
3757 reaction had to be occurring to sustain the detonation during the ANFO trials.

3758

3759 It was proposed that a percentage of the potential energy from the fuel oil was released
3760 from the ANFO prill directly surrounding the booster with the PE10 providing enough
3761 energy to near-instantaneously combust the FO in its immediate vicinity. However it was
3762 speculated, within the small sized charges, that the reaction of FO combustion could not
3763 sustain effectively when the reaction zone reached the edge of the charge where a lack of
3764 confinement results in pressure and temperature losses. This behaviour would result in
3765 the detonation of ammonium nitrate, with a tiny percentage of fuel oil, similar to the first
3766 hypothesis.

3767

3768 Numerical modelling of small scale AN charges was undertaken, using similar methods
3769 discussed in Section 4.2.2, to establish whether or not the reduced energy release from small
3770 scale ANFO charges was captured by simulating the detonation of AN. Both Figures 5.11a-
3771 d and 5.12a-b represented experimental results for small scale ANFO trials when top and
3772 bottom detonated were compared with numerical simulations of 211.5g of AN detonation,
3773 established through removing the 5.5% FO composition and then an additional 10% mass
3774 for energy loses to the ground, in line with findings in Section 3.2.2.3.

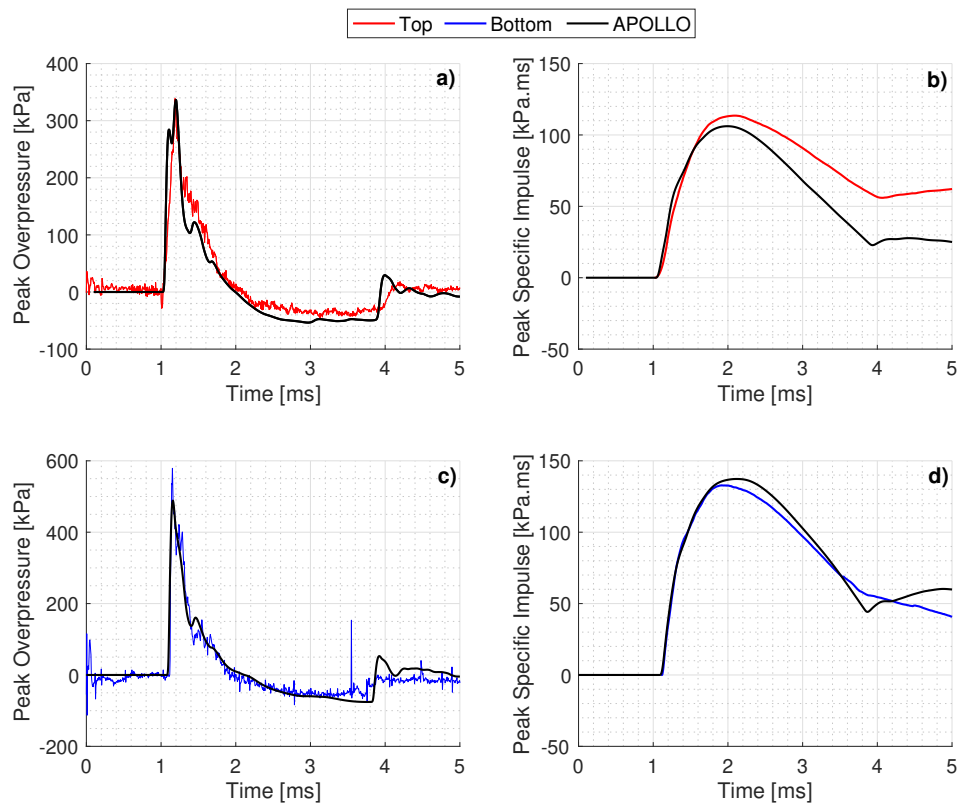


Figure 5.11: Compilation of pressure-time and impulse-time histories from 1m standoff distances which related to top (red) and bottom (blue) detonated trials compared to APOLLO simulations of AN: a) Top P-T, b) Top I-T, c) Bottom P-T, and d) Bottom I-T.

3775 The numerical simulations provide remarkable agreement with the experimental data de-
 3776 spite the fact a completely different explosive composition and corresponding JWL pa-
 3777 rameters have used to what is known to have been tested. The fact that the detonator
 3778 position effects are captured effectively by the models provides confidence that assuming
 3779 the ANFO composition behaves like pure AN at these mass scales is valid. The question
 3780 posed here however is at what point does modelling the explosive as AN become invalid.
 3781 Future research would consider testing a wide range of ANFO masses in free-field blast en-
 3782 vironments to quantify at what mass specifically results in a doubling of the yield through
 3783 the fuel oil reaction occurring effectively.

3784

3785 Johansson^[82] presented the numerical modelling schemes of small scale ANFO charges
 3786 which were validated using experimental work extracted from the literatures which de-
 3787 tailed a lack of accuracy when using standard JWL parameters for capturing the behaviour
 3788 on ANFO. The reason for this is because in a JWL EoS, there is an assumption of in-
 3789 stantaneous energy release which for non-ideal explosives it is known not to be the case.

3790 Presented is an Ignition and Growth (I&G) EoS which when adopted for the same simula-
 3791 tions, captures the behaviour of experimental ANFO detonation rather accurately. Future
 3792 numerical simulation will consider the I&G EoS for non-ideal explosives in the attempt
 3793 to capture their behaviours across a whole range of mass scales, alongside identifying at
 3794 which regions are JWL parameters valid.

3795

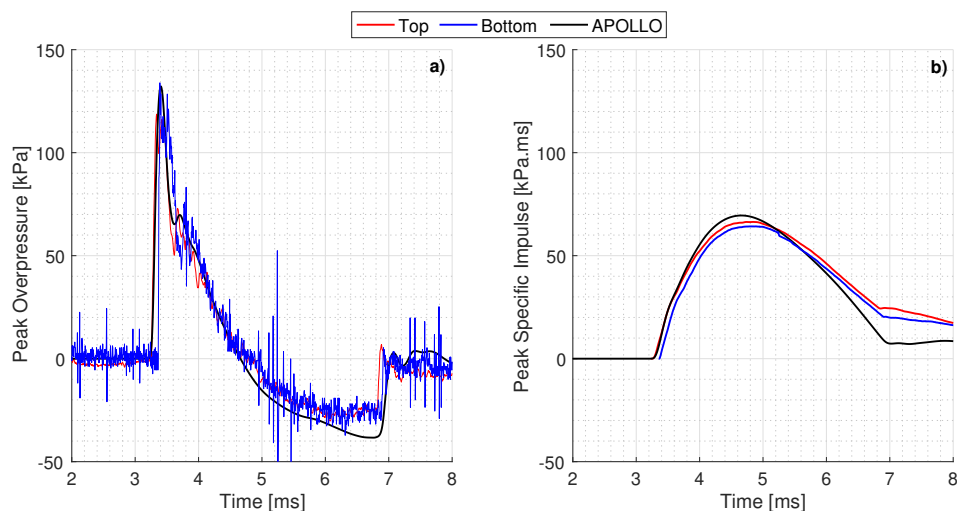


Figure 5.12: Compilation of pressure-time and impulse-time histories from 1m standoff distances which related to top (red) and bottom (blue) detonated trials: a) Top P-T, b) Top I-T, c) Bottom P-T, and d) Bottom I-T.

3796 5.3.4.2 Is the energy equivalence reduction related to unreacted ANFO prill 3797 projectiles?

3798 The findings from the previous two hypothesis tests lead to a further theory that the
 3799 overall explosive yield reduction, compared to quoted values in literature, could be di-
 3800 rectly linked to the quoted reaction zones of non-ideal explosives being anywhere from
 3801 in 10-100mm^[29]. A rudimental analysis assessed the effect this radius of ANFO being
 3802 un-reacted would have on the TNT equivalence for a variety of hemispherical charge sizes
 3803 tested as seen in Figure 5.13.

3804

3805 The analysis assumed a constant reaction zone size of 13mm and a $TNTe = 0.82$ to
 3806 compute a mass ratio of ANFO for the reduced mass based on the reaction zone being
 3807 un-reacted compared to the actual mass tested. The ratio was multiplied by the TNTe
 3808 factor to assume an actually reacted TNTe factor in these small scale trials which is dis-

3809 played by the plotted curve in Figure 5.13. This plot outlines both the charge masses
 3810 tested in this series, which would result in a range of $TNTe = 0.3 - 0.5$, closely matching
 3811 the spread seen for the experimental data conducted within this series (denoted by the
 3812 red and blue markers), and charge masses up to those tested by Petes et al. [112]. Based on
 3813 these assumptions, to achieve the $TNTe = 0.82$ based on this mechanism alone, a charge
 3814 mass of 10,000kg would be required.

3815

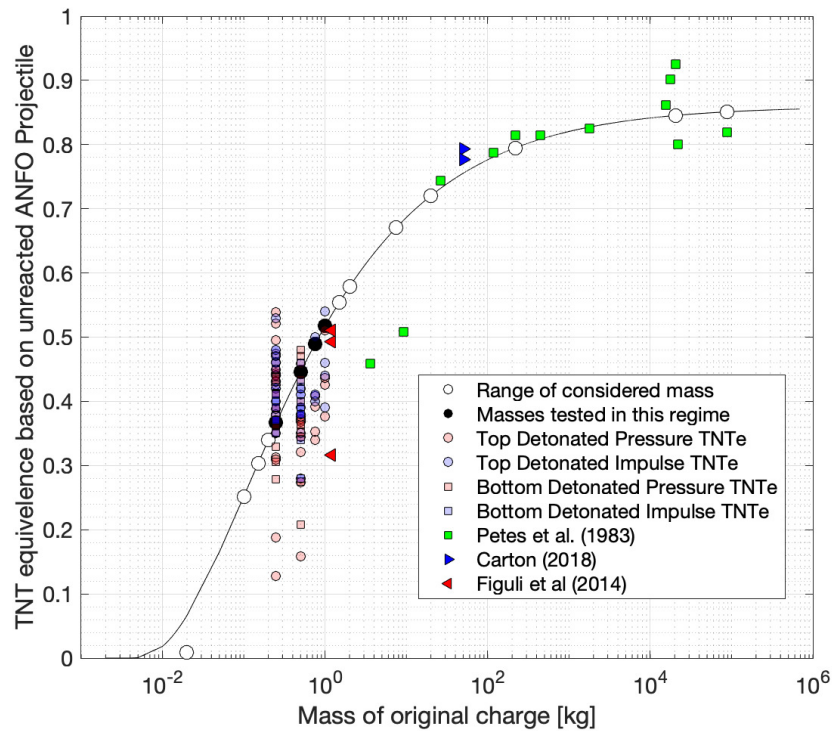


Figure 5.13: Equivalent $TNTe$ values resulting from applying a nominal 13mm reaction zone to a given ANFO charge and assuming that the region has not fully reacted and instead is fired off as a projectile compared with extract $TNTe$ from experimental pressure and impulse data

3816 It is important to note that the value of $TNTe$ and size of reaction zone have been chosen
 3817 based on published values and appropriate fits the available experimental data. The ex-
 3818 perimental recordings which are much lower than the quoted range of $TNTe = 0.3 - 0.5$
 3819 are the result of being within a scaled distance of $Z < 1m / kg^{1/3}$, to which the top deto-
 3820 nated position effects (see Figure 5.5) and fluid dynamic instabilities have a much greater
 3821 influence on the recorded parameters and thus the inferred $TNTe$ values, as discussed in
 3822 Section 5.3.1.

3823

3824 5.3.4.3 Additional Testing

3825 Whilst a mechanism which captures the behaviour of the small scale charges, it is not
3826 necessarily the definitive physical process. The irony with the potential mechanisms con-
3827 sidered in Section 5.3.4 is that both result in an effective yield reduction of around 50% for
3828 charges of this size, which agrees with the ratio between the recorded data and published
3829 values of ANFO yield, however, neither can be definitively confirmed.

3830

3831 It is actually hypothesised to be a combination of the above contributors which makes
3832 quantifying and characterising the fundamental process difficult. Further test were con-
3833 ducted using small scale charges but with a variety of ANFO/AN composition mixes. The
3834 idea behind these tests were to investigate and quantify the influence of FO has on the
3835 primary shock wave parameters and to establish which of the aforementioned mechanisms
3836 contributed to the over energetic output.

3837

3838 The first lot of testing which was of interest was to vary the general composition of the
3839 AN/FO percentages to assess the effects this has for small scale ANFO charges. Petes
3840 et al. [112] presented findings in large scales charge which suggested a drop in energy release
3841 when the FO percentage varied from being stoichiometric mix at 6%. To test this feature
3842 in small scale charges to establish whether the FO was making a difference in small scale
3843 charges was to change the composition by adding in a proportion of pure AN to reduce
3844 the global percentage of FO in each charge. A total of 10 tests (5 different FO% with a
3845 repeat test) were conducted recording the pressure-time history at two standoff distances
3846 resulting in 20 individual recordings. The data was analysed using standard methods dis-
3847 cussed throughout this thesis, and an average TNTe factor was established for the tests
3848 at each FO percentage investigated, as seen in Figure 5.14.

3849

3850 What this investigation did prove was that even in these small scale charges, the percent-
3851 age of FO does still play an effect on the overall explosive yield output from the detonation.
3852 This was enough evidence to suggest that during the detonation of ANFO in small quan-
3853 tities the mechanisms as to why a lower TNTe value is evaluated is not related to just AN
3854 detonating, and is a slightly more complicated phenomena. When no FO was present, the
3855 AN composition was unable to detonate, concluding that some FO is required and is part
3856 of the energy release process for small scale charges.

3857

3858 It was therefore important to consider what would happen if the FO percentage was lower
3859 within the composition, but rather than a well mixed composition of ANFO with AN,
3860 having two distinct regions of AN and ANFO. This would investigate whether the reduced
3861 TNTe value seen for small scale ANFO is related to an unreacted shell of prill projecting
3862 away from the charge proportional to reaction zone sizes.

3863

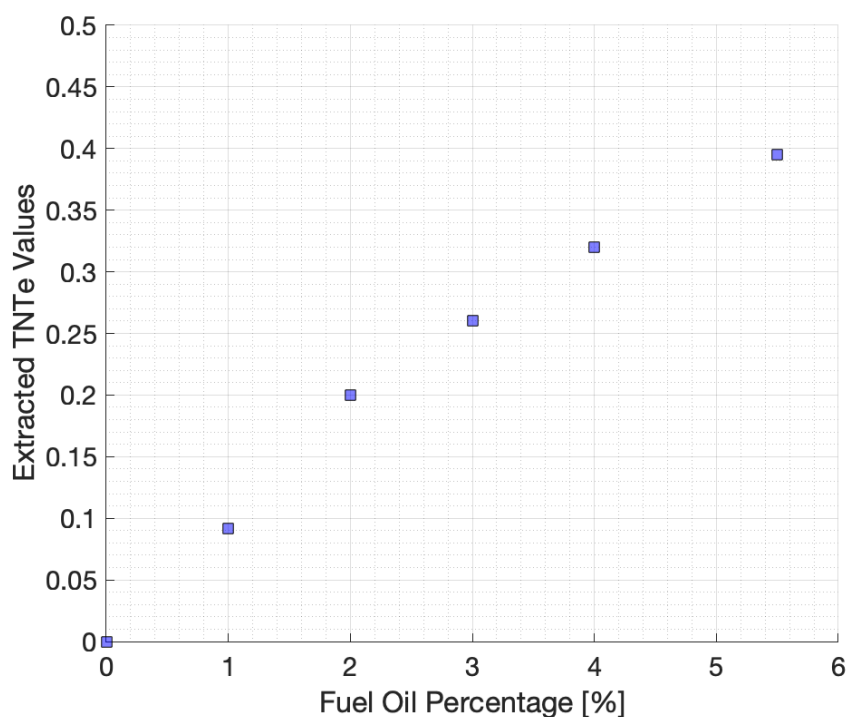


Figure 5.14: Extracted TNTe values from 250g hemispherical charge trials of ANFO varying the FO percentage by addition of pure AN

3864 To test theories proposed in which only the FO in the direct vicinity of the booster charge
3865 is fully combusting and the remaining energy release is from the detonation of AN. To test
3866 this a double shelled hemisphere was used with the the inner shell containing ANFO and
3867 the outer shell consisting of pure AN both with 125g of the material in theory lowering
3868 the overall FO percentage down to 2.75%. With low levels of FO, it has been reported a
3869 incomplete reaction occurs and therefore lower energy releases and subsequently slower
3870 detonation velocities^[183]. Obviously with a inner vacuum formed shell to keep the two
3871 compositions separate, an extra level of confinement, albeit minimal, is provided which
3872 could have resulted in additional combustion of the FO^[12].

3873

3874 Figure 5.15a-d shows the results of a shelled hemisphere test of AN/ANFO mix when
3875 compared to a full 250g ANFO hemisphere measured at two different standoff distances.
3876 Considering the pressure-time history plots in Figures 5.15a and 5.15b, there is very little
3877 difference between the positive phases of the shelled trials and the full ANFO tests which
3878 alone suggests that the hypothesis that the outer layer of the charge provides very little
3879 to the initial primary shocks blast parameters. However, taking into account the specific
3880 impulse plots seen in Figures 5.15c and 5.15d, there is a clear reduction between the
3881 bench mark 250g ANFO and the hypothesis testing AN/ANFO shell. Whilst the 250g
3882 ANFO trial results in a $TNTe \simeq 0.36$, the shell trial exhibited a $TNTe \simeq 0.3$ assuming a full
3883 250g mass for both, suggesting there is a reduction in energy release but not what would
3884 be expected when removing 2.75% of the fuel oil which when looking at the trend seen in
3885 Figure 5.14, this composition would be expected to result in a $TNTe \simeq 0.23$. The question
3886 posed here is what the reason for an explosive with half the mass of ANFO and an outer
3887 shell of AN to provide a different behaviour to the same quantities of the material but
3888 evenly mixed.

3889

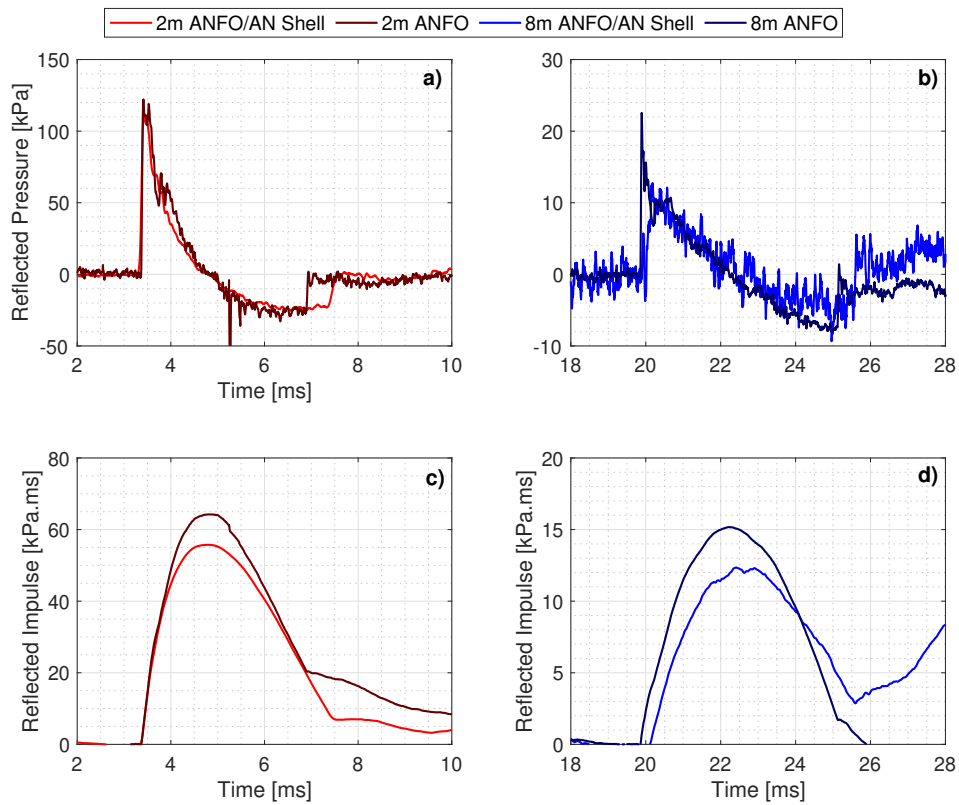


Figure 5.15: Compilation of pressure-time and impulse-time histories at 2m (red) and 8m (blue) standoff of 50/50 shells of ANFO/AN compared to 250g ANFO trials (black): a) 2m P-T, b) 8m P-T, c) 2m I-T, and d) 8m I-T.

3890 It was considered whether the outer shell of the explosive exhibits detonation and/or com-
 3891 bustion in any of the tests conducted. The proof of some energy release in this region is
 3892 exhibited through the differences in specific impulses when comparing the shelled tested
 3893 to full ANFO in Figure 5.15, therefore it can be induced that the reactions are ongoing
 3894 in the outer shell and cannot be fully assumed to be unreacted material projectiles. This
 3895 leads on to the idea that close to the charge centre there is enough pressure, confinement
 3896 and time for the ANFO to fully react prior to any energy losses, but the outer shell does not
 3897 and therefore exhibits state changes but never fully reacted by the time breakout occurs.
 3898

3899 A realistic conclusion to make from the results of the small scale ANFO tests, and the
 3900 corresponding positive phase data, is that friedlander pressure-time histories are achieved
 3901 similar to those resulting from ideal explosive detonation but for non-ideal explosives the
 3902 reaction zone is the main mechanism which causes the reduction in TNTe and thus is
 3903 not scalable with charge mass. The shelled tests exhibit specific impulse reductions when
 3904 comparing to 250g ANFO shots which are directly related to the lack of FO in the final

3905 section of the charge, and therefore no combustion could occur. It is assumed however that
3906 the outer shell of the charges only partially reacts, which is why only a small differences in
3907 the TNTe values extracted from the data is experience between the two trial types. With
3908 large charges, the reaction zone size becomes irrelevant when looking at its percentage of
3909 the whole charge mass and therefore the detonation fosters confinement, higher pressures,
3910 temperatures and more time for the reactions to occur, resulting in TNTe values close
3911 to the theoretical thermochemically derived value of $TNTe=0.82$. HSV work, discussed
3912 in Section 5.4, will look at quantifying a more realistic reaction zone size to improve the
3913 validity of the empirical curve derived in Figure 5.13.

3914

3915 5.3.5 Negative Phase

3916 The negative phase of the small scale ANFO trials was of particular interest as again there
3917 is no previous literature which outlines its behaviour, and with the findings of discussed in
3918 Section 5.3.1 showing discrepancies between assumed knowledge and experimental data,
3919 further developments to understanding non-ideal detonation could be made. Figures 5.16a-
3920 c represent the extracted and scaled negative phase parameters for the small scale ANFO
3921 trials which do not compare to KB predictions as well as the ideal explosives discussed in
3922 Figures 4.6a-c, which was somewhat expected.

3923

3924 Whilst Figure 4.6c presents a general consistency in the negative phase duration parame-
3925 ters extracted, the spreads observed are a consequence of the difficulty in prescribing the
3926 parameter generally, and within this low pressure region even more so. It is important
3927 to note that going forward the time duration of the negative phase will be omitted from
3928 further analysis in line with the notion presented within this thesis of it not providing
3929 truly valid data despite KB predictions capturing the parameter reasonably well.

3930

3931 Considered the negative pressure in Figure 5.16a, when comparing to KB predictions there
3932 seems to be a much greater negative phase as a result of small scale ANFO detonations,
3933 resulting in a $TNTe \approx 0.82$ providing a more appropriate fit. Needham^[107] discussed the
3934 phenomena of the negative phase and detailed the mechanism of the rising fireball creating
3935 lower pressure regions near the ground surface, which pulls the surround air towards the
3936 detonation point. This process further lowers the air density of the medium which has
3937 over-expanded during the shock waves propagation and therefore results in large negative

3938 phases within an energetic fireball. What this suggests for the small-scale ANFO trials
 3939 is that the energy contained within the fireball, and subsequent shocks, is much greater
 3940 than what is to be expected for standard ideal explosives, which has been related directly
 3941 to the combustion of FO that was unable to occur during the initial detonation process.
 3942

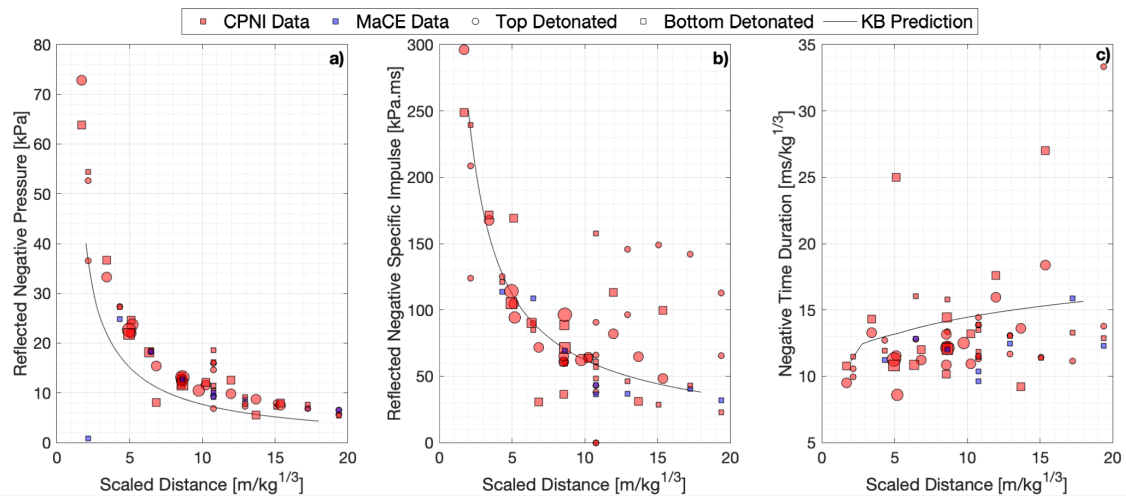


Figure 5.16: Compiled scaled negative phase blast parameters from ANFO explosive trials as a function scaled distance, compared with KB predictions, using a $TNTe=0.395$, with increasing marker sizes to indicate the mass of charge: a) Reflected negative pressure, b) Reflected negative specific impulse, c) Scaled negative phase duration

3943 Scaling all the negative phase data by a larger $TNTe$ value based on the pressure data
 3944 alone would result in a shift away from the KB predictions for the negative impulse in Fig-
 3945 ure 5.16b, which based on a $TNTe=0.395$ seems to be captured reasonably well. This led
 3946 to question whether KB predictions were reasonable representations of ANFO, and other
 3947 non-ideal explosives, for the negative phase as different $TNTe$ values could be deduced
 3948 for each parameter, which is uncondusive when compared to how well ideal explosives be-
 3949 have against prediction tool across every considered parameter in far-field regimes. With
 3950 ANFO still exhibiting ideal-like behaviours, but with a lack of scalability with mass, for
 3951 each increment of mass scale there should be a valid $TNTe$ value which works across the far-
 3952 field range and therefore a systematic error in the method of analysis has been highlighted.
 3953

3954 The method of establishing the specific impulse values in Figures 5.16b and 4.6b, dis-
 3955 regards the effect of the secondary shock during the calculation due it being so small in
 3956 ideal explosives it could be omitted making minimal differences to the output results^[133].
 3957 Interestingly, small scale ANFO presents as if the secondary shock is more influential on

3958 the negative phase impulse as there is disagreement between the TNTe values required for
 3959 KB predictions to capture the data. An investigation into the inclusion of capturing the
 3960 secondary shock parameters when calculating the overall negative phase specific impulse.
 3961 In doing so, a more appropriate value of a TNTe could be established which fits across
 3962 each of the blast parameters in line with findings for ideal explosions in far-field scenarios
 3963 seen in Chapter 4.

3964

3965 With a more through examination of the data to account for the negative phase a stand
 3966 alone feature of the primary blast wave by effectively omitting the effects of secondary
 3967 shock was undertaken. As mentioned previously in Figure 5.16a, the negative pressure as-
 3968 sociated with small-scale ANFO seems to be better captured with a TNTe \approx 0.82 which had
 3969 been related to the FO combustion occurring within the confines of the fireball detonation
 3970 cloud. It has been therefore assumed that all parameters which are extracted after the
 3971 positive phase can be scaled accordingly with the idea that there is enough temperature,
 3972 pressure and time for the full energy release from the partially reacted prill projectiles due
 3973 to the confinement of the detonation cloud itself. Figures 5.17a-b present the negative
 3974 phase parameters, scaled using a TNTe=0.82 when considering the behaviour as a perfect
 3975 blast wave and removing effects of secondary shocks.

3976

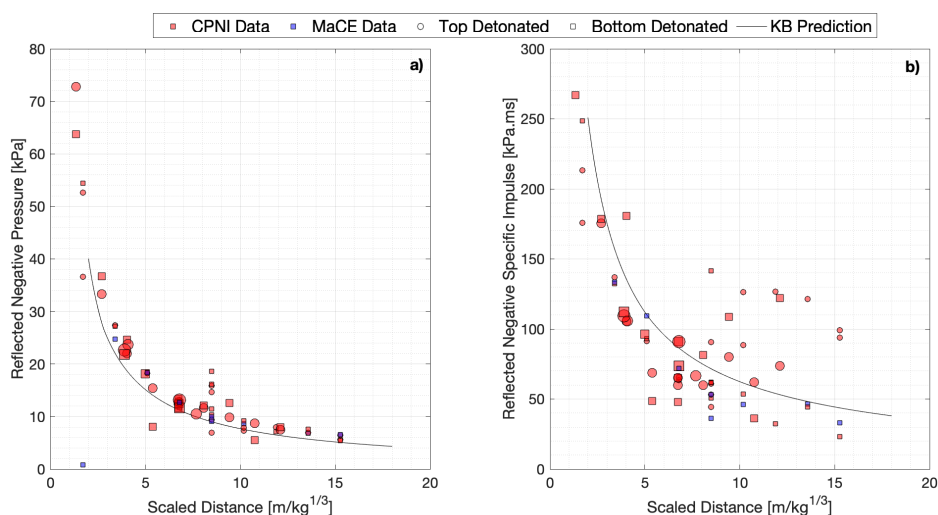


Figure 5.17: Compiled negative phase blast parameters re-worked to include secondary shock capture from ANFO explosive trials as a function scaled distance, compared with KB predictions, using a TNTe=0.82, with increasing marker sizes to indicate the mass of charge: a) Peak reflected overpressure, b) Scaled reflected peak specific impulse

3977 Clearly seen in both Figures 5.17a and 5.17b is a general improvement on how well KB

3978 predictions capture the negative phase experimental data. This finding provides justifi-
3979 cation the KB predictions capture blast parameters extremely well in the far-field regime
3980 when considering a perfectly ideal shock wave and thus the basis of future work which
3981 could look at the superposition of ideal primary and subsequent waves to establish accu-
3982 rate predictions for a real-world explosive event.

3983

3984 **5.3.5.1 Secondary Shock Data**

3985 The scaled secondary shock data, when presented against KB predictions in Figures 5.18a-
3986 d, again shows similar behaviour to the ideal explosives, with a large difference between
3987 predicted values of a primary wave but exhibiting considerable consistency across the en-
3988 tire data set with respect to scale distance. The general behaviours of the data and the
3989 predictive curves are comparable which does give rise to the idea that secondary shock
3990 data is as deterministic as the positive phase and future work should consider the devel-
3991 opment of similar empirical curves from a large quantity of secondary shock data resented
3992 within this thesis and published literature.

3993

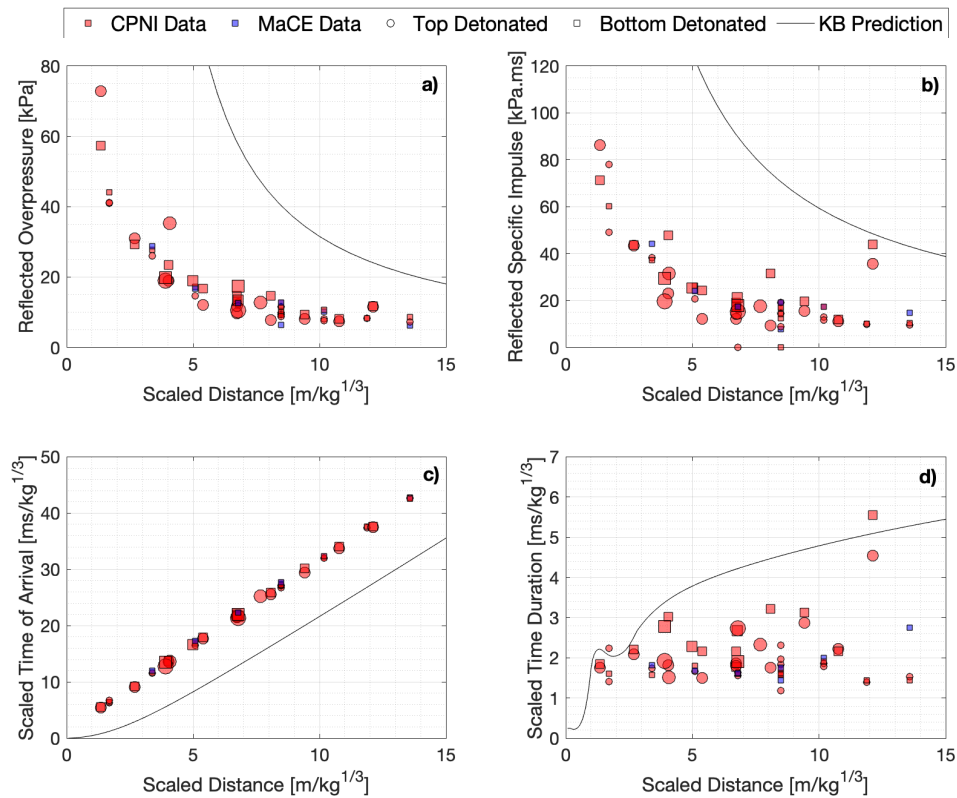


Figure 5.18: Compiled scaled secondary shock blast parameters from ANFO explosive trials as a function scaled distance, compared with KB predictions, using a $TNTe=0.82$, with increasing marker sizes to indicate the mass of charge: a) Peak reflected overpressure, b) Scaled reflected peak specific impulse, c) Scaled arrival time, d) Scaled positive phase duration

3994 The issue currently faced is the idea that the secondary shock is highly dependent on the
 3995 spatial and temporal characteristics of the fireball and therefore variations in the internal
 3996 chemistry and energy release will increase the variation in extracted experimental data.
 3997 Figures 5.19a-d present a compilation of all the secondary shock data collected within this
 3998 thesis, scaled accordingly with the corresponding $TNTe$ values.

3999

4000 Although the scaled data generally holds a level of comparison between the ideal and non-
 4001 ideal explosives there are some clear differences exhibited which could be key to further
 4002 the understanding of small scale ANFO detonations and thus other non-ideal explosives.
 4003 With reference to Figures 5.19b and 5.19c, the ideal explosives seem to exhibit less en-
 4004 ergy released in the form of specific impulse but have a time of arrival which seems to be
 4005 smaller than that of the ANFO trials, suggesting that the secondary shock travels faster
 4006 but is a much smaller quantity of energy associated with it. In reality this finding goes

4007 back to the idea of the arrival time offset discussed in Section 5.3.2, which discussed the
 4008 biggest influencer of arrival time offset to be the proportionality between the velocity of
 4009 a ANFO shock wave, in comparison to one from an equivalent TNT mass. Interestingly,
 4010 when approaching the near-field, $Z < 3m/kg^{1/3}$, the data for secondary shock arrival time
 4011 tends towards the same point suggesting the offset between primary and secondary shock
 4012 is consistent across all explosive types, related directly to the time in which it takes the
 4013 secondary shock to begin to propagate. The data quickly trends away as the secondary
 4014 shock in ANFO is travelling at a much lower velocity than that of the ideal explosives.
 4015 The velocity proportionality theory could be explored in future research to acquire a re-
 4016 lationship between primary and secondary shock velocities for both ideal and non-ideal
 4017 explosives to then quantify the offsets presented in Figure 5.19c in a robust manner.

4018

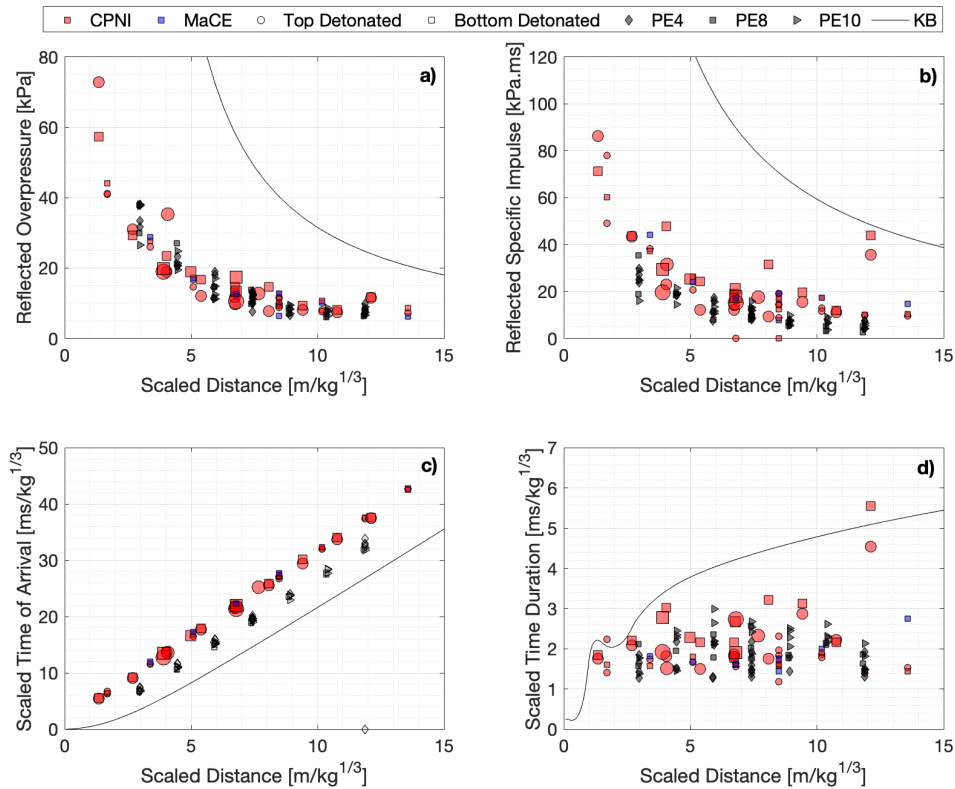


Figure 5.19: Compiled scaled secondary shock blast parameters from all considered explosive trials as a function scaled distance, compared with KB predictions, using each corresponding TNTe value, with increasing marker sizes to indicate the mass of charge: a) Peak reflected overpressure, b) Scaled reflected peak specific impulse, c) Scaled arrival time, d) Scaled positive phase duration

4019 It was important to test the dimensionless secondary shock delay prediction tool against
 4020 small scale ANFO secondary shock data provided the knowledge that all parameters after

4021 the primary wave is to be scaled with a $TNT_e=0.82$. In order to express the secondary
 4022 shock delay extracted from the data as dimensionless, there is a requirement to prescribe
 4023 a velocity of detonation and packing density for the small scale ANFO. Whilst the packing
 4024 density was measured pre-test to be approximately $820\text{kg}/\text{m}^3$, the velocity of detonation
 4025 was uncertain from pressure gauge analysis alone. It was questioned whether the velocity
 4026 of detonation used in this analysis associated to the primary shock, the secondary shock
 4027 or the maximum velocity the detonation wave can achieve in the given composition. On
 4028 closer inspection of the data presented within Rigby and Gitterman [129] assumed the ve-
 4029 locity of detonation to be that associated with the maximum velocity of detonation the
 4030 composition could achieve, meaning that regardless of mass scaling and the composition,
 4031 the explosive secondary shock delay is normalised based on an ideal detonation. With
 4032 that being said, the small scale ANFO data collected within this thesis was normalised
 4033 using a velocity of detonation of $4,200\text{m}/\text{s}$ in line with values quoted in Petes et al. [112],
 4034 to which is presented in Figure 5.20.

4035

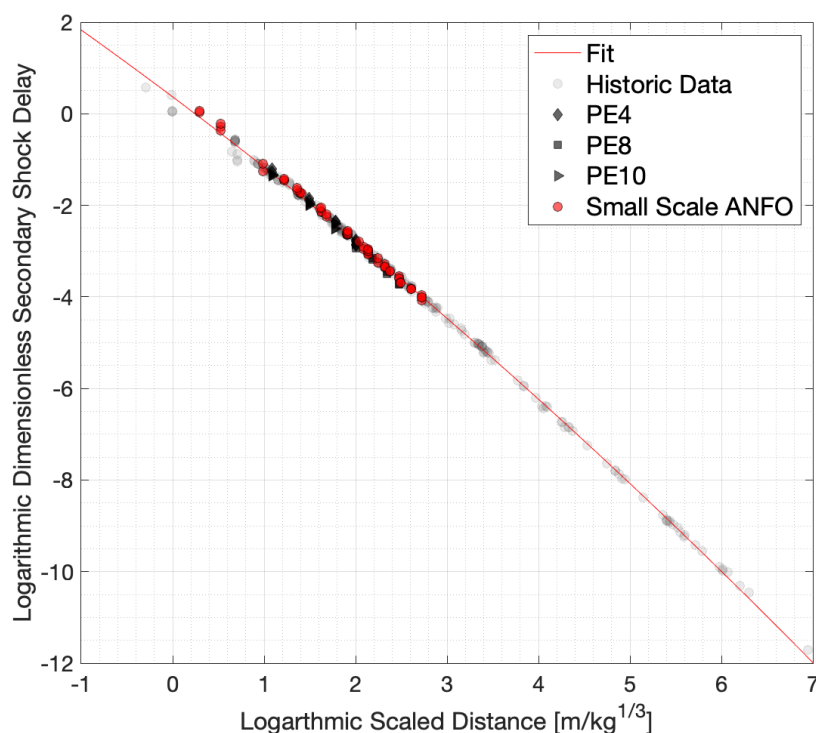


Figure 5.20: Dimensionless secondary shock delay parameter evaluated for all explosive tested in this thesis, scaled accordingly, and compared with polynomial fit to historic data

4036 The clear comparable relationship between the normalised data and the fitted prediction

4037 provides confidence in the tool across explosive composition mass and scaled distance.
4038 The concern however in considering small scale ANFO with a velocity of detonation of
4039 4,200m/s, similarly to large scale detonations, seems fragmented because the total energy
4040 released during detonation are different and thus the secondary shock behaviour should be
4041 too. It proposed that the energy which is not released during the development of the pri-
4042 mary shock is gained during the propagation of the secondary shock, resulting in a halving
4043 of the blast parameters associated with the primary shock and doubling of the secondary
4044 shock parameters. This would explain why the delay time between primary and secondary
4045 shock arrival time is unaffected by this process as one the primary wave propagates slower
4046 and the secondary faster than they should when compared to large scale charges. This
4047 knowledge justifies the reason for assuming the maximum detonation velocity regardless
4048 of composition and mass scaling because even during non-ideal detonation the same quan-
4049 tity of energy is released just over much lower time scales, and with understanding the
4050 two aforementioned parameters proportionality even if the energy is not captured by the
4051 primary wave, subsequent waves will.

4052

4053 **5.3.6 Pressure Gauge Summary**

4054 The section has considered the pressure gauge data recorded from the detonation of small
4055 scale ANFO charges with a thorough investigation into the mechanisms resulting in a
4056 $TNTe=0.395$ for primary shock wave positive phase parameters, and a $TNTe=0.82$ for all
4057 parameters subsequent to the breakout of the primary shock (i.e negative phase of the
4058 primary wave and the secondary shock parameters). Discrepancies were exhibited in the
4059 $TNTe$ of arrival time for both primary and secondary shocks which was related directly to
4060 the proportionality of velocity between small scale ANFO and TNT, which was not visible
4061 for ideal explosives discussed in Chapter 4, but in reality is probably a feature to consider
4062 during future analysis.

4063

4064 The pressure gauge analysis of small scale ANFO has provided insights into the behaviours
4065 of non-ideal explosives provided the reaction zone size is order of magnitudes larger than
4066 the explosive charges diameter. At each increment of mass, ANFO detonations result in
4067 friedlander shock wave behaviours which could be considered to be ideal, with a constant
4068 $TNTe$, across an entire range of standoff when compared to KB predictions. However,
4069 there is proportionality between mass and the associated $TNTe$, until a limit is reached

4070 where no amount of additional charge mass will effect the overall explosive yield. It is
4071 therefore paramount for a more robust mechanism of predicting non-ideal explosive be-
4072 haviour.

4073

4074 Interestingly, the findings for the secondary shock analysis suggest that of the total po-
4075 tential energy of a given charge, there is a ratio of energy split between the primary and
4076 subsequent shocks. The amount of energy released during the primary shock development
4077 in small scale ANFO is approximately half than what was expected, however the devel-
4078 opment of subsequent shocks encompassed around double the amount of energy. This
4079 idea has been speculated when considering the fact all the measured parameters which are
4080 confined within, and affected by, the detonation cloud for longer durations seem to exhibit
4081 far more energy in small scale ANFO than what is experienced in ideal explosives.

4082

4083 5.4 High Speed Video Analysis

4084 The methods discussed in Chapter 4 have been adopted for the far-field trials using hemi-
4085 spherical ANFO charges to test their applicability to non-ideal explosives. The results from
4086 an individual trial at 5m standoff from a reflective wall was processed using the aforemen-
4087 tioned methods and is presented in Figure 5.21. The data required a $TNTe = 0.3$ in
4088 order for the scaled arrival time data to replicate KB predictions. This is parallel to
4089 the findings from the MAE analysis undertaken on the pressure gauge data evaluating a
4090 lower TNTe factor (TNTe=0.28) for arrival time when compared to other blast param-
4091 eters (TNTe=0.395). This finding establishes confidence in the pressure gauge recordings
4092 triggering at the correct time and the time of arrival offset being a real physical feature
4093 of small scale ANFO charges. The reason for the offset in arrival time has yet to be fully
4094 established and therefore it is hoped that with the use of high speed video in conjunction
4095 with the theory of shock wave velocities presented in Figure 5.10, and their relationship
4096 to TNTe, an approximation for the offset could be deduced.

4097

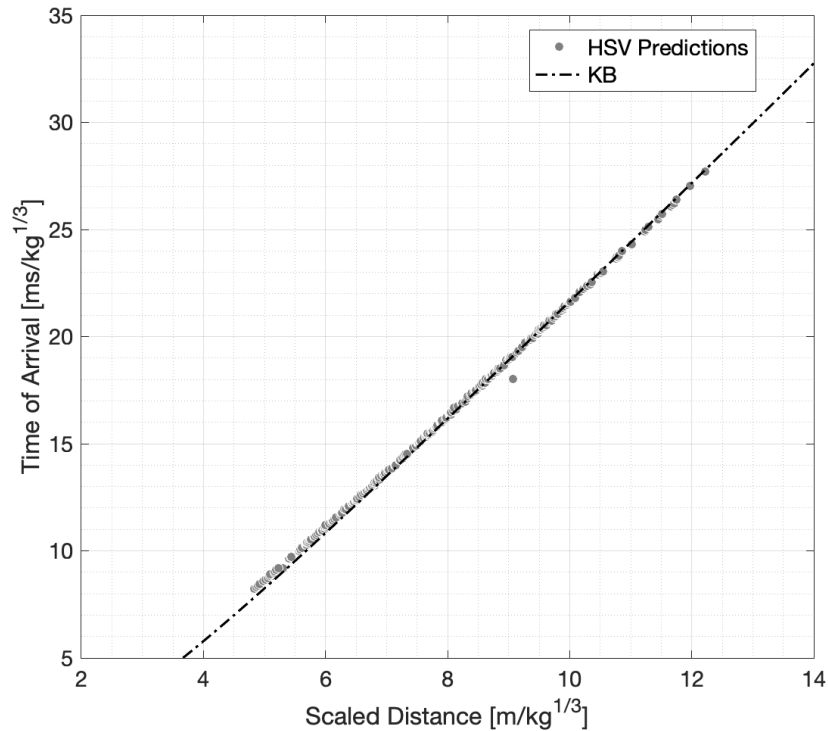


Figure 5.21: Shock front arrival time from a bottom detonated 250g ANFO hemispherical charge placed at a 5 m stand-off, recorded using HSV and scaled using TNT equivalence factor of 0.3 to compare to KB predictions

4098 5.4.1 Fireball Breakout

4099 Figure 5.22 displays two snapshot images taken from a bottom detonated hemispherical
 4100 250g charge recorded at 1,000,000fps. The average velocity of the detonation wave can
 4101 be approximated using the known distance the wave has travelled in these tests, 53mm,
 4102 divided through by the time taken to reach this position which was 28,000ns, resulting
 4103 in a average velocity of $\sim 1900\text{m/s}$. Considering Figure 5.10, it is a safe assumption that
 4104 the velocity at breakout would be higher than this value due to the acceleration in ANFO
 4105 detonation velocity from point source to charge extents and therefore is probably closer
 4106 to a velocity of $\simeq 2700\text{m/s}$, calculated from using the $\text{TNTe}=0.395$ established from pres-
 4107 sure gauges data analysis and multiplying it by the detonation velocity of TNT which is
 4108 6950m/s . This finding is a reasonable approximation for supporting the notions regarding
 4109 the lower TNTe value established from the far-field trials.

4110



Figure 5.22: Ultra high speed video footage of a bottom detonated 250g hemispherical trial displaying the comparison between charge diameter at the time of detonation and at the point in which the detonation wave reached the charge's extents and products began to expand; this is denoted by the charge size extending passed the green lines spaced 80pixels apart in both plots.

4111 However, as discussed in Section 5.3.2, the lower TNTe value established from shock wave
4112 arrival time is when the data is compared to free-air shock propagation for TNT which
4113 during the early stage detonation process behaves very differently to ideal ANFO. It can
4114 be assumed that TNT from the moment it detonates, it does so with a velocity of around
4115 6950m/s across the entirety of its charge radius and then begin to decay when breakout
4116 occurs. As ANFO behaves differently, with an increasing velocity with charge radius,
4117 the best approximation which can be made for ANFO a average velocity of the detona-
4118 tion wave across the radius to make it comparable to an almost perfect average velocity
4119 of 6950m/s in TNT. By making this assumption of average velocities, the difference in
4120 time it takes for a 250g TNT hemisphere to reduce down from 6950m/s to the average
4121 $\sim 1900\text{m/s}$ would provide a reasonable theoretical approach to why an offset is recorded.
4122

4123 Figure 5.23 outlines the aforementioned theoretical approach and highlights an offset in
4124 arrival time of around 0.32ms between the detonation a 250g hemisphere of TNT and the
4125 time in which it begins to behave like a 250g hemisphere of ANFO based on HSV analysis
4126 of the fireball breakout. This offset value is similar to that presented in Section 5.3.2 when
4127 looking at the arrival time offset attributed to the pressure gauge data when comparing
4128 between the pressure and impulse evaluated TNTe=0.395 and the KB predictive curve,
4129 which gives justification towards this theory. Whilst ideal explosives exhibit similar veloc-
4130 ity profiles to TNT, non-ideal explosive like ANFO need to be characterised differently,
4131 with shock wave arrival time being investigated based on the propagation dynamics of the
4132 detonation wave and subsequent shock wave.

4133

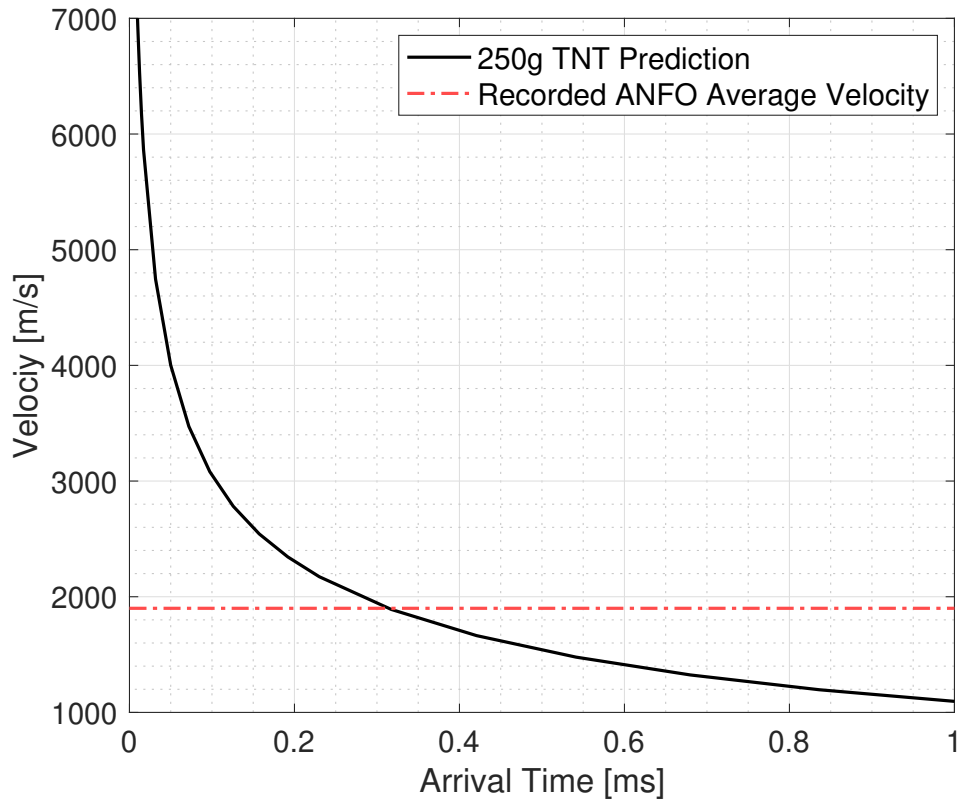


Figure 5.23: Shock wave velocity with respect to arrival time for a 250g TNT explosives evaluated through KB predictions with reference to the average velocity extracted from the HSV of hemispherical ANFO detonation

4134 5.4.2 Variability Analysis

4135 The same variability analysis undertaken in Section 4.3.1 to establish whether ANFO also
 4136 behaves in a deterministic manner when relating the consistency of arrival time data to
 4137 the scaled distance at which the recording was taken. Figure 5.24 shows arrival time vari-
 4138 ability across the top and bottom detonated (denoted by TD and BD respectively) ANFO
 4139 hemispherical detonations with respect to scaled distance. The estimated gauge variation
 4140 boundary (GVB) presented in Figure 4.11 for PE4 trials was used to test the confidence
 4141 in the fit as a deterministic predictor for far-field free-air shock waves developed from
 4142 different explosives.

4143

4144 The estimated GVB can be seen to generally hold credibility for the ANFO hemispheres
 4145 tested and this provides confidence as this an upper bound for arrival time variability
 4146 across the far-field range for all explosives. It is important to note that at the two ex-

4147 tremities of the recording scaled distances, the variability in the data is higher based on
 4148 the sheer lack of data points recorded. These positions correspond to the furthest parts
 4149 of the cameras field of view and therefore are subject to lens distortion and can be omitted.

4150

4151 One finding that can be deduced which is in line with that of the pressure gauge data is
 4152 that the variability of both bottom and top detonated shots are similar which gives rise
 4153 to the consistency of each individual trial and the increase in variability as scaled distance
 4154 enter the instability regions between $0.1 < Z < 3m/kg^{1/3}$ regardless of the differences in
 4155 arrival time associated with detonator position discussed in Figure 5.9.

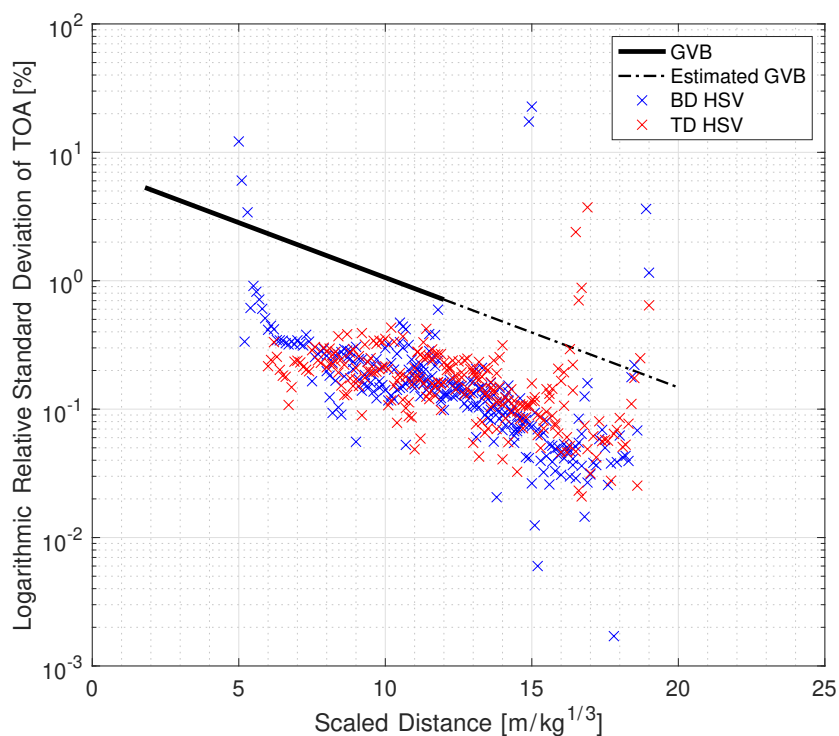


Figure 5.24: Logarithmic RSTD in shock wave arrival time of 250g hemispherical ANFO trials as a function of scaled distance for high speed video recordings, compared to empirical model detailed within Farrimond et al.^[49]

4156 5.4.3 Rate Stick Trials

4157 The rate stick tests documented within this thesis were recorded using ultra high speed
 4158 video cameras recording at between 250,000-500,000fps in order to capture the detona-
 4159 tion wave propagation throughout each trial. The main reason for the variability of the

4160 frame rate is due to the filming capability of the camera used limited to a set number of
4161 frames, therefore in order to capture the fireball/shock wave propagation from the end of
4162 the cylinders a slower frame rate was required.

4163

4164 The video recordings displayed in Figure 5.25 shows the general behaviour of the ANFO
4165 rate stick trials, which was consistent across all pipe sizes tested and therefore is not a
4166 diameter dependent feature within the bounds of this experimental regime. The first snap-
4167 shot shows the 3g PE10 booster detonating and causing a preliminary flash to be seen
4168 percolate through the voids in the prill. As the snapshots progress through time intervals
4169 0.05ms, a clearly defined detonation wave can be seen to form, with a seemingly expanding
4170 bright region to which has been attribute to the ongoing chemical reactions attempting to
4171 find an equilibrium. The final snapshot shows the front of the reaction zone reaching the
4172 end of the pipe, at which point the bright zone succeeding it begins to compress until it
4173 reaches the end of the pipe too. The fireball/shock wave only begins to propagate away
4174 from the end of the tube when the bright zone is fully compressed which provides the
4175 understanding that the propagating pressure wave would be related to the position at
4176 which the cylinder begins to expand (the back of the bright zone) and the front face of the
4177 bright zone is a reaction wave which has effectively little-to-no pressure associated with it
4178 but induces decomposition of the explosive. The brighter region size of particular interest
4179 as it is speculated that this could be the reaction zone of ANFO which if quantified could
4180 be included in Figure 5.13 to both verify the projectile theory and the assumption of a
4181 13mm reaction zone for this sized charge.

4182

4183 Using similar methods discussed by Rigby et al.^[130] to track near-field blast loading,
4184 the displacement of the front and back faces of this bright region were tracked using the
4185 ‘Canny’ edge detection in-built MATLAB function, as it was found to perform optimally
4186 for high contrast tracking. The back face of the bright region was defined by the position
4187 at which the cylinders began to laterally expand which provided to be more challenging to
4188 capture due to chaotic nature of the expanding detonation products. This was overcome
4189 by utilizing a threshold procedure to capture the brightest pixels, between a value of 0-255
4190 within each frame, and then assigning a range between the max value and a percentage
4191 (determined through trial and error) of the assigned pixel intensity, to which the edge
4192 detection function would consider. Anything outside of this threshold was omitted from
4193 the analysis, and assumed to be too dark to be the reacting region of the charge. To verify

4194 the assigned positions of the reaction zone, the tracked positions were plotted at regular
4195 intervals over the whole frame itself as seen in Figure 5.26.

4196

4197 Using a pre-test photo taken from the high speed camera, displaying known dimensions
4198 of the rate stick, the video was calibrated through assigning define boundaries of the pipe
4199 external walls to achieve a real-world pixel size. The tracked positions from the source of
4200 detonation were calibrated to be a real-world distance travelled along the rate stick which
4201 was also differentiated, with respect to time, resulting in the velocity of the propagation
4202 of each front.

4203

4204 Rather than viewing the whole pipe in its entirety with one analysis and the attempt to
4205 nullify any irregularities in the fronts bright regions behaviours, three randomly assigned
4206 box strips across the width of the rate stick were considered for the regions of analysis
4207 in the attempt minimise any curvature effects and irregularities across the entire front.
4208 These three independent processed data sets are to be used as self-validation of the results
4209 extracted from recordings.

4210

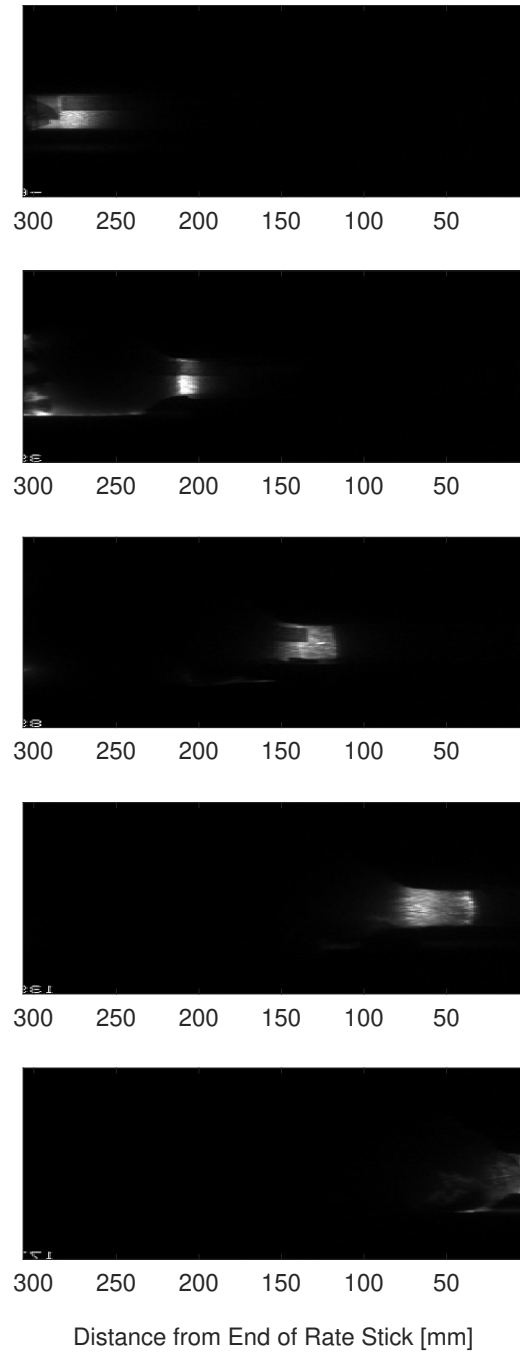


Figure 5.25: High speed video snapshots with 0.05ms intervals between each plot displaying the propagation of the detonation wave along a 300mm length 50mm diameter PMMA pipe

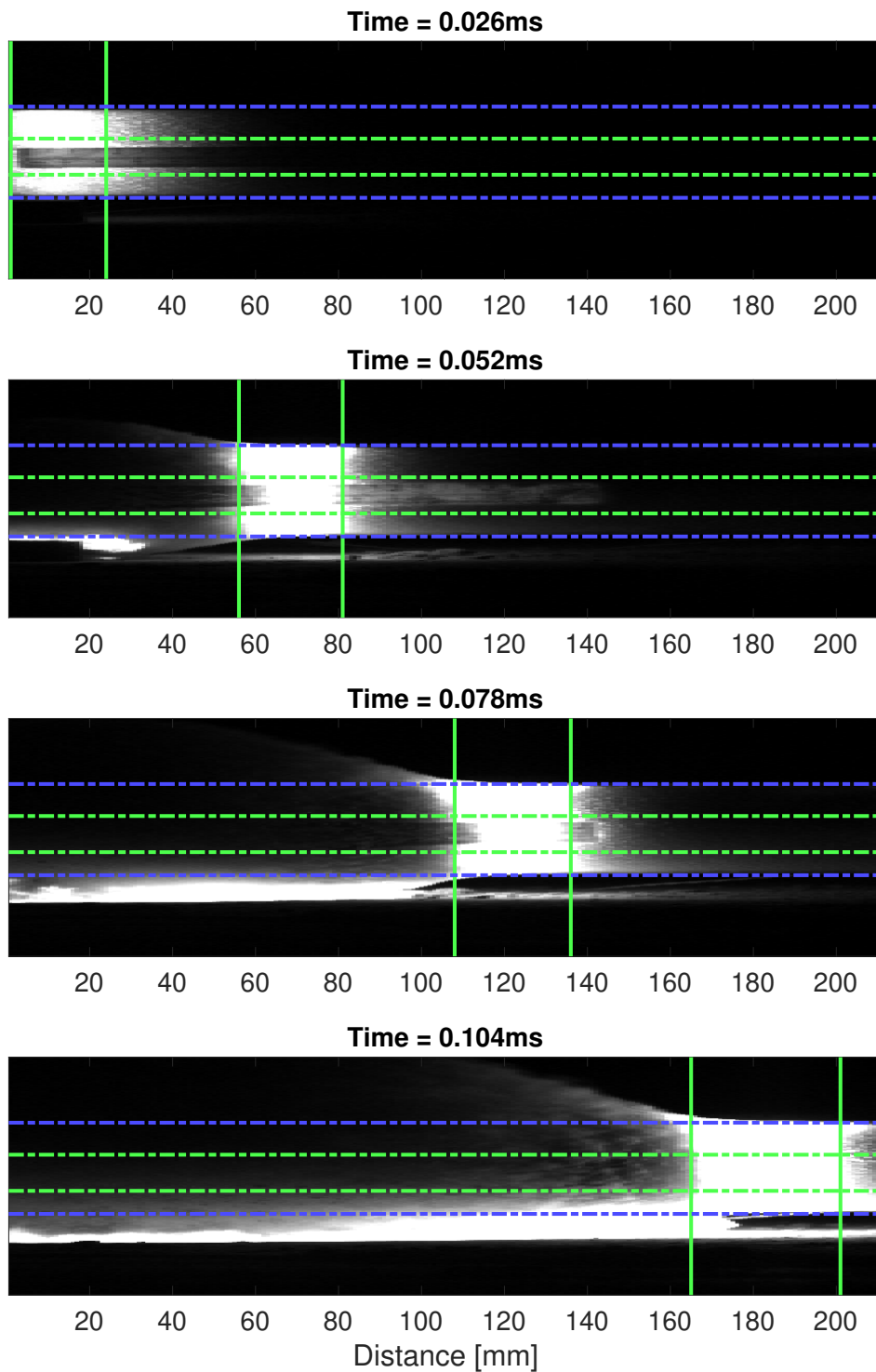


Figure 5.26: High speed video snapshots of 104mm internal diameter ANFO filled PMMA tube at $t = 0.026, 0.052, 0.078$ and 0.104 ms after detonation, highlighting the reaction zone positions assigned (green lines) for the given region considered (green dashed lines) alongside the widths of the pipe (blue dashed lines).

4211 Figure 5.27 displays the same video recording as Figure 5.26 but only considering the

4212 region defined by the green dashed lines. The reduced region of analysis results in a con-
 4213 sistent reaction zone and therefore a definitive position can be established for each trial.
 4214 The three regions of analysis were chosen at random between the confines of the rate sticks
 4215 width, to establish a probabilistic catheterisation of reaction front position and velocity.
 4216 These parameters were evaluated from both horizontal and vertical calibration pixel sizes
 4217 across all three analysis regions and were compared to assess the accuracy and validity of
 4218 the analytical data processing tool which can be seen in Figure 5.28.

4219

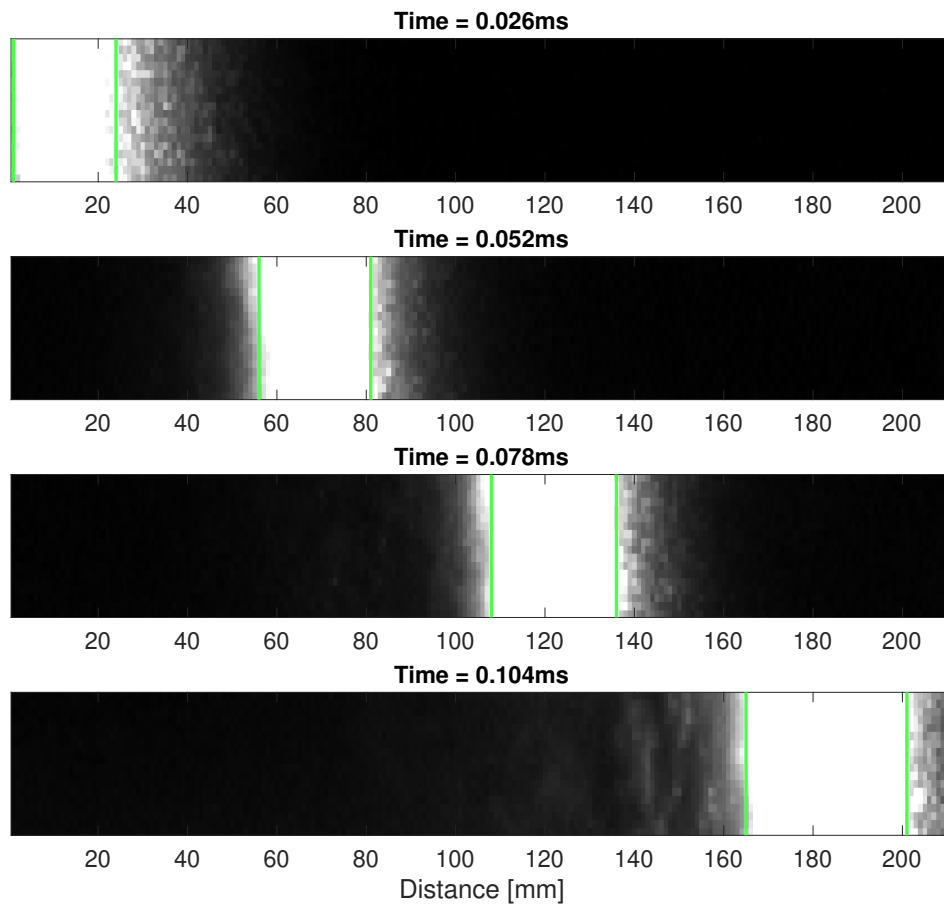


Figure 5.27: High speed video snapshots of 104mm internal diameter ANFO filled PMMA tube at $t = 0.026, 0.052, 0.078$ and 0.104ms after detonation, zoomed into the area box region considered.

4220 Figures 5.28a and 5.28b display the validation of the methodology process which consid-
 4221 ered the three different regions of analysis, all of which that are calibrated using both
 4222 vertical and horizontal pixels sizes. The results from these different processes suggest
 4223 not only is the calibration correct, resulting in accurate data, but the regional choices
 4224 on the pipe result in a general consistency of both velocity and reaction zone size. It

4225 is believed the initial flash duration discussed in Figure 5.25, corresponds to the time it
 4226 takes before the measured reaction zone of the ANFO charges to visually appear without
 4227 doubt through the illuminated prill. The initial flash from the detonator/booster reduces
 4228 the precision of the visual interpretation of the ANFO detonation wave alone hence why,
 4229 during the opening $\sim 80\text{mm}$ along the rate stick across both figures, spikes in the data is
 4230 experienced which was a finding verified by Bohanek et al. [17]. After this, the data begins
 4231 to behave consistently and therefore can be considered to be credible and representative
 4232 of the decomposition of ANFO, similar to the discussion made by Araos and Onederra [4],
 4233 until $\sim 200\text{-}250\text{mm}$ at which point, the data become corrupted by either detonation prod-
 4234 uct breakout or systematic experimental oversights of securing tape blocking the view of
 4235 the reaction zone. With this in mind, all subsequent plots discussing the rate stick trials
 4236 will be confined to these limits of distance along the pipe to avoid discussing the inefficacy
 4237 of unverified data.

4238

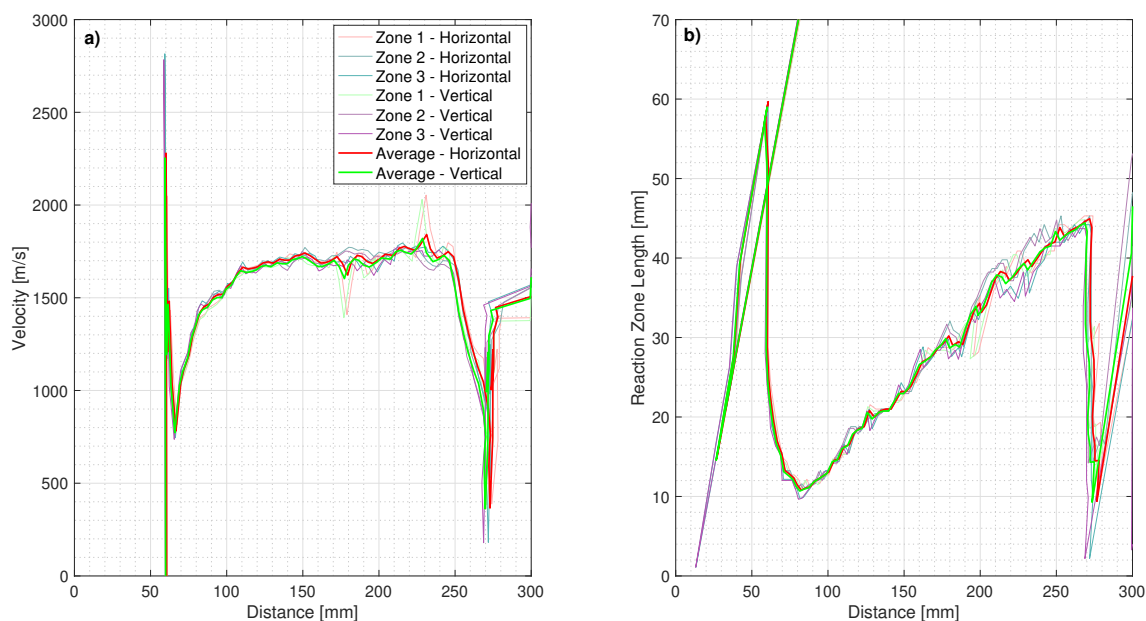


Figure 5.28: HSV tracking results from a 50mm internal diameter rate stick trial, calibrated both vertically and horizontally, smoothed using a 15 point moving average to present: a) Detonation front velocity and b) Reaction zone length with respect to distance travelled along the pipe.

4239 5.4.3.1 Compiled Results

4240 The results from the six different conducted rate stick trials subjected to the aforemen-
 4241 tioned analytical procedure are presented in Figures 5.29a and 5.29b for the lengths along
 4242 the rate sticks which provide valid data which is representative of the decomposition an

4243 detonation of ANFO. When considering the smallest diameter pipes (50mm internal) which
 4244 made use of a PE10 booster charge, denoted by the grey coloured lines, all exhibit consis-
 4245 tency in velocity and reaction zone size across three nominally identical shots, providing
 4246 confidence in the level of repeatability in the analytical methods developed between test,
 4247 therefore verifying its application for the remaining tests to be qualitatively compared.

4248

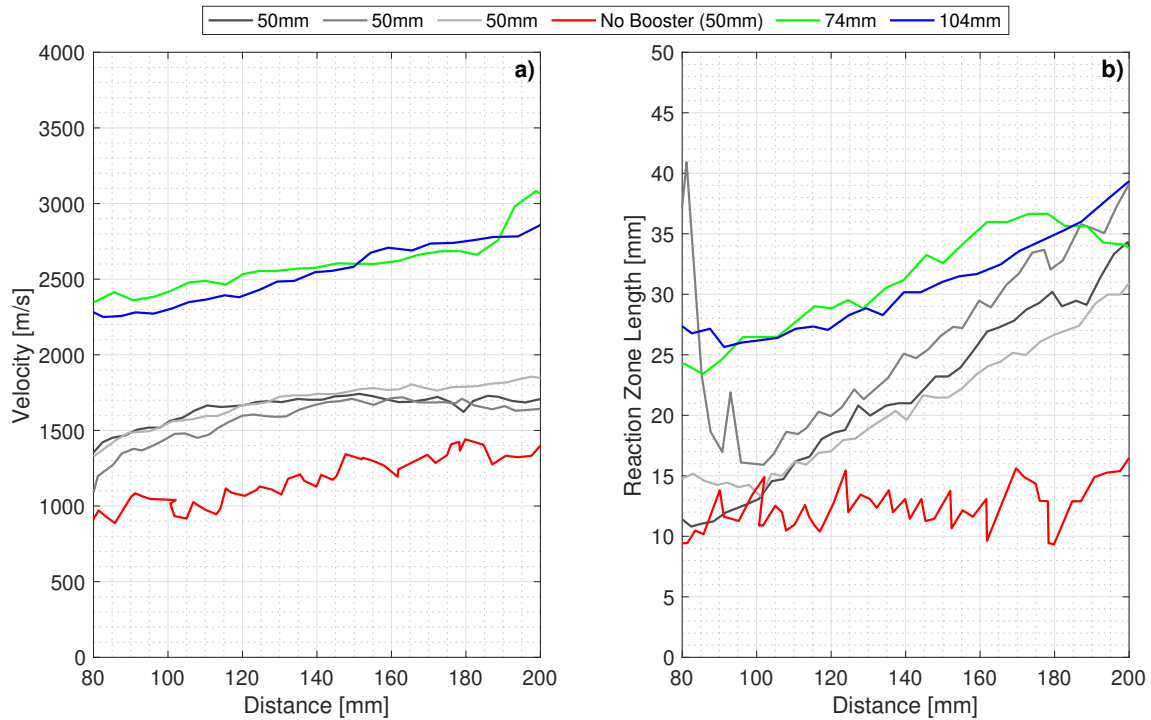


Figure 5.29: HSV tracking results from all ANFO rate stick trials conducted within this thesis, smoothed using a 15 point moving average to present: a) Detonation front velocity and b) Reaction zone length with respect to distance travelled along the pipe.

4249 The non boosted 50mm diameter pipe shot presented much lower overall velocities and
 4250 reaction zone than the boosted trials, with only a slight general increase in both with
 4251 respect to distance along the rate stick. This finding is thought to infer the critical di-
 4252 ameter of ANFO to be around that of this pipe used in these trials due to the fact the
 4253 energy loses due to the expansion of detonation products almost equals the energy released
 4254 through detonation. What this does begin suggest is that the minimum bright zone size
 4255 required to detonate ANFO successfully is represented by around 12mm which begins to
 4256 point towards the theory proposed within Figure 5.13.

4257

4258 The interesting finding when considering the 50mm internal diameter pipes is that upon

4259 exhibiting behavior representative of ANFO detonation, a velocity of detonation measure-
4260 ment of 1000m/s and 1200-1400m/s has been extracted for the non-boosted and boosted
4261 trials respectively. The latter of which reaches steady state detonation recording a velocity
4262 of ~ 1700 - 1800 m/s which is much lower than the quoted value for ANFO (~ 4200 m/s). The
4263 non-boosted 50mm rate stick test results in a consistent velocity of ~ 1000 m/s, due to the
4264 lack of additional initial energy provided by the booster to induce the chemical reaction
4265 process within the confinement of the charge prior to any detonation product expansion.
4266 The 50mm diameter rate sticks clearly are large enough to sustain the detonation reaction
4267 but the energy losses through the lateral fireball and product expansion is enough to freeze
4268 out any further energy gained from the decomposition of the ANFO. With these values
4269 in mind, a direct comparison can be made to the quoted value which results in a ratio of
4270 between $\simeq 25$ - 43% depending on the values chosen which is believed to be further evidence
4271 to the reduced yield from small scale ANFO charges when to large charge trials conducted.

4272

4273 The larger diameter rate stick trials (74mm and 104mm internal diameters) exhibited a
4274 much higher initial velocity which has been initially tracked ~ 2300 m/s, both increasing at
4275 similar rates to peak values of around 2800m/s after a 200mm run along the rate stick but
4276 no levelling off like what is exhibited in the 50mm diameter pipes. This is related to the
4277 fact the large diameter pipes induce a longer duration of confinement and therefore the
4278 energy loss through lateral expansion of these rate stick was no longer enough to restrain
4279 a minimum steady-state velocity but instead result in an increase in energy released and
4280 front velocity with respect to distance along the rate stick.

4281

4282 Considering the trajectory of these velocities, it is approximated that a rate stick of around
4283 500mm long is required to exhibit the quote velocities of large scale ANFO, assuming no
4284 plateauing occurs at these charge diameters. The 74mm and 104mm internal diameter
4285 pipes show similar patterns of behaviour for both reaction zone length and velocity devel-
4286 opment which highlights the fact that the energy loss through lateral expansion of these
4287 rate sticks was no longer enough to restrain steady-velocities of the recorded fronts but
4288 instead resulted in an increase in energy release and velocity. This begins the debate as
4289 to whether the definition of the critical diameter of an explosive is the minimum size at
4290 which a cylindrical charge has to be to sustain a steady-state detonation is correct for
4291 non-ideal explosives and whether this could be the reason for a variety of different critical
4292 diameters quoted within published literature.

4293

4294 Qualitatively, the 104mm pipe seems to result in an increasing velocity at a greater rate
4295 than those recorded in the 74mm pipe. This is logical if less energy is being lost later-
4296 ally whilst simultaneously gaining energy from the decomposition of ANFO but there is
4297 a larger surface area associated with the 104mm diameter pipe which will result in larger
4298 energy losses. This finding could be an artefact of the 15 point moving average applied
4299 to the raw data and should be treated with caution, assuming that the general physical
4300 mechanisms are similar. It is expected that with an increased minimum radius until det-
4301 onation product breakout, the quicker the rise time in velocity to maximum theoretical
4302 value for ANFO. Using the data presented, this finding is not something which can be
4303 concluded on and would require a much more extensive rate-stick test plan across a range
4304 of pipe sizes and lengths with multiple repeats to establish its validity.

4305

4306 The measured reaction zone sized across all trials conducted with a booster exhibits an
4307 increase with distance travelled along the rate stick. As discussed previously, it is posited
4308 that as the ANFO detonates, under optimal conditions of confinement, the reaction zone
4309 length will increase in size. With the detonation cloud expansion only starting to occur
4310 when the back face of the zone reaches the charge extents, it is a reasonable suggestion
4311 that the larger the measured reaction zone, the more decomposition of ANFO occurs. This
4312 would result in a greater efficiency in the energy release of the composition as there would
4313 be more time and confinement for the deflagration of FO to occur. With this in mind,
4314 there key measurement of the reaction zone is at the beginning of the rate stick prior to
4315 any increase in length as this will identify the the amount of ANFO which could in theory
4316 will not react in any detonation.

4317

4318 It was hypothesised that the reaction zone measured in the rate stick trials could infer the
4319 range in size the zone could be depending on the test-to-test variability in the charges,
4320 directly relating to prill form and their macrostructural arrangement. Taking the non-
4321 boosted 50mm trials presented in Figures 5.29a and 5.29b, a near steady-state detonation
4322 is displayed and therefore the trial was close to the critical diameter of non-boosted ANFO.
4323 With this in mind, it would suggest the constant behaviour in the measured reaction zone
4324 length of between 9-17mm, taking the range recorded, is the minimum size required to
4325 detonate ANFO and therefore the size in which quantifies the amount of ANFO which
4326 may be unreacted on breakout of the detonation products and the primary shock wave.

4327

4328 Undertaking a similar analysis presented in Figure 5.13, the TNTe values were calculated
4329 for effective masses of ANFO resulting from the reaction zone sizes of 9mm and 17mm in
4330 line with the aforementioned findings and is presented in Figure 5.30. What can instantly
4331 be deduced from this plot is the fact that almost all the collected data, both from this
4332 research and historical is captured by the theoretical mechanism proposed and therefore
4333 provides justification to the theory of unreacted prill projectiles. Whilst there is agree-
4334 ment, the definitive mechanism is still to be established and future experimental testing
4335 should consider both the attempt of capturing this mechanism using high speed video and
4336 investigate the mass range which minimal data is presented for between 1-50kg, to verify
4337 if a general increase in TNTe is recorded in line with theoretical prediction.

4338

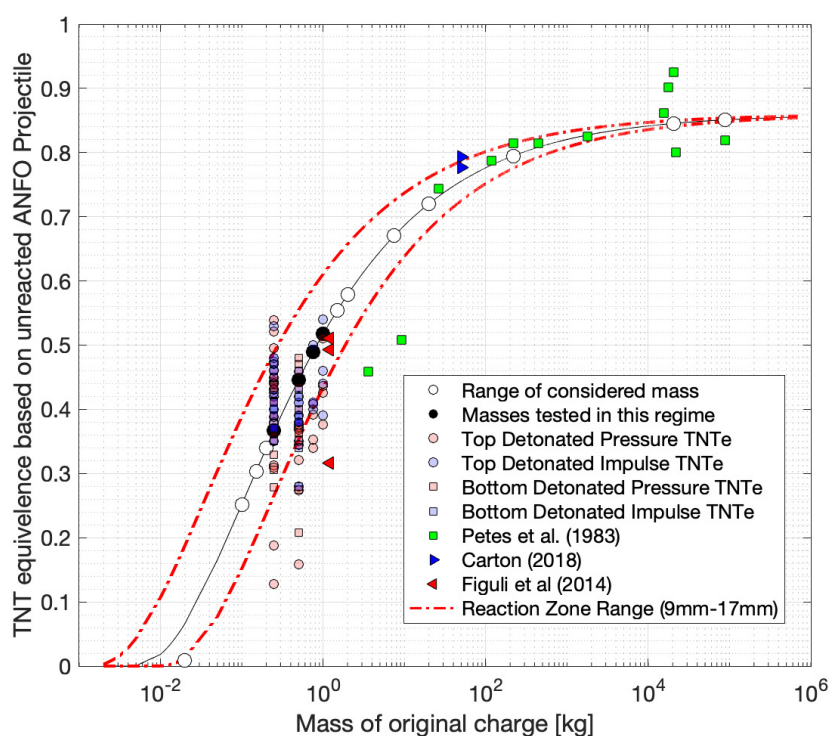


Figure 5.30: Equivalent TNTe values resulting from applying a nominal 13mm reaction zone to a given ANFO charge and assuming that the region has not fully reacted and instead is fired off as a projectile compared with extracted TNTe values from experimental pressure and impulse data, making reference to the reaction zone range recorded Figure 5.29b.

4339 5.4.4 High Speed Video Summary

4340 The validated HSV data processing tools developed within this thesis have been com-
4341 pared to the non-ideal explosive ANFO, in the attempt to characterise the mechanisms
4342 underlying the lower explosive yields associated with small scale charges and provide both
4343 theoretical and empirical quantifications to support the hypotheses made.

4344

4345 The results from using the originally developed tools provided confidence than small scale
4346 ANFO charges at each increment of mass behave with ideal-like characteristics, exhibiting
4347 the ability to be scaled by a single given TNTe value in far-field ranges.

4348

4349 The use of KB velocity relationships provided further insights into why the arrival time of
4350 small scale ANFO charges do not have a directly comparable relationship to TNT when
4351 comparing to other blast parameters extracted. The development of detonation velocity
4352 within the ANFO composition was not measured as part of these tests, but an average
4353 velocity from detonation to breakout was inferred which allowed comparison to the be-
4354 haviour of an ideal explosive and therefore TNT. Implementing this theoretical approach
4355 resulted in arrival time offset which closely matched the pressure gauge data when com-
4356 pared to KB predictions.

4357

4358 Rate stick trials were undertaken in the attempt to further understand the internal det-
4359 onation mechanisms and processed within smaller scale ANFO charges. An automated
4360 tracking algorithm was developed and self-validated in terms of its accuracy of tracking
4361 and the consistency in the test data through comparing three nominally identical shots
4362 which resulted in highly consistent data. The remaining trials were the conducted using a
4363 variety of pipe diameters and non-boosted charges. Through the analysis there is a clear
4364 relationship presented between confinement and/or charge diameter with the development
4365 of velocity.

4366

4367 One rate stick trial which was at quoted values of the critical diameter of ANFO (50mm
4368 internal diameter) and was not boosted, resulted in near steady state conditions, which not
4369 only verified the critical diameter but also presented the minimum conditions to achieve
4370 detonation. The reaction zone measurement for this trial was utilised to improve the
4371 mechanism of unreacted ANFO projectile prediction presented in Figure 5.30, which jus-

4372 tified the behaviour of both small and large scale ANFO trials and presents ranges of
4373 potentially TNTe fluctuations based on the uncontrollable variables associated with het-
4374 erogeneous prilled ANFO detonation. Whilst reasonable justification for these mechanisms
4375 have been presented, a much larger and rigorous testing regime for ANFO, across mass
4376 scales is required to help further validate the hypothesis presented.

4377

4378 **5.5 Future Research into Non-Ideal Detonations**

4379 The investigation into small scale ANFO has provided a significant gap in the current
4380 understanding of non-ideal explosive characterisation and the mechanisms which result in
4381 there effective and overall potential explosive yields. Whilst there are studies which at-
4382 tempt to quantify a range of explosives which exhibit non-ideal behaviours, a distinct lack
4383 of thorough investigation is presented within published literature to quantify yields based
4384 on both mass and distance. The implication of assuming non-ideal explosives exhibit the
4385 same chemical processes during detonation as ideal explosives across an entire mass range
4386 is a general over prediction of the explosive output. If numerical models are not able
4387 to capture the yield of particular composition accurately across a full range of masses,
4388 a uncertainty within design regimes occur with the knock on effect of over-engineering
4389 each element of protection. Conducting well-controlled experiential trials for a number of
4390 commonly used home made explosives, varying both mass and standoff distance, will en-
4391 able a global understanding of explosive loading conditions which can be integrated within
4392 probabilistic load characterisation for optimised protective schemes of element design.

4393

4394 In addition, this compilations of experimental data can be used to develop more robust
4395 numerical modelling schemes through the validation of thermochemical code values devel-
4396 oped for EoS of each explosive. The method presented within this thesis makes use of the
4397 JWL EoS for small scale but adjusted to model AN alone which is known to be incorrect
4398 despite the accuracy of the simulation across the far-field range. Further investigation into
4399 the use of Ignition and Growth (I&G) EoS should be undertaken to accurately capture
4400 the detonation mechanics of non-ideal explosives across a whole range of masses.

4401

4402 Future work should look at undertaking a quantitative analysis of shock wave velocity
4403 across a full range of distances as a result from small scale ANFO charges using timer pins

4404 within the composition itself, alongside tracking the wave using incident gauges and/or
4405 HSV from detonation through to free-air propagation. In doing so, empirical comparisons
4406 can be made between the recorded data and the KB derived shock wave velocities for TNT,
4407 to further access the hypothesis suggested in Figure 5.10 and to attempt to quantify the
4408 offset associated with the disproportional shock wave velocity of TNT and ANFO.

4409

4410 Chapter 6

4411 Summary, Conclusions and Future 4412 Work

4413 6.1 Summary

4414 This thesis has aimed to characterise and quantify the effects of a shock wave in far-field
4415 regimes resulting from the detonation of both ideal, and non-ideal, explosives using vali-
4416 dated analytical data processing methods.

4417

4418 Explosive detonation and the subsequent parameters which follow have been investigated
4419 for many decades but despite best attempts to understand their effects, a clear divide has
4420 been presented for the fundamental nature of shock wave dynamics and the variability
4421 exhibits. Whilst designing infrastructure to withstand extreme dynamic loading poses dif-
4422 ficulty when considering where a charge may be in relation to the structure, alongside its
4423 composition and mass, making it harder is the uncertainty on how deterministic the yields
4424 of an explosive are. If explosives exhibit high levels of variability within simple scenarios,
4425 the engineer can have no confidence of ensuring designs are robust enough to withstand
4426 extreme loading.

4427

4428 If variability levels can be established with respect to the distance a given charge is away
4429 from a target, a more probabilistic approach can be undertaken by a designer which ex-
4430 hibits a higher level of safety confidence. It is therefore of high importance to assess
4431 current practices of experimental work to highlight reasons as to why explosive yields

4432 are viewed as chaotic by some in the industry and provided a robust approach to un-
4433 dertaken well-controlled trials to reduce the inconsistency in data. Chapter 2 provides a
4434 review of the current literature as well as background theory to blast wave phenomenol-
4435 ogy alongside state-of-the-art predictive methods for explosive yields in free-air scenarios.
4436 Chapter 3 proposed the development of numerous data processing tools which are exposed
4437 to PE4 experimental data and undergo synergetic validation with numerical simulations
4438 to highlight systematic features within methodologies, their effects and how they can be
4439 mitigated. Chapter 4 exercises the tools on a large data set of three ideal explosives in
4440 the attempt to both validate the tools across a wide range of ideal explosives and to iden-
4441 tify whether ideal explosives exhibit deterministic behaviours across the far-field range of
4442 scaled distances. The methods which were validated in the aforementioned chapters were
4443 the used in Chapter 5 to study their applicability to small-scale non-ideal explosives, such
4444 as ANFO. The results of which provided fundamental insights into the characterisation of
4445 non-ideal explosives.

4446

4447 Findings from the PhD thesis provide a robust and validated far-field data set across a
4448 number of small-scale explosive compositions which should be used as a benchmark, not
4449 only for the level of repeatability achievable from well-controlled far-field trials, but as a
4450 final highlight to the blast community that far-field parameters are deterministic and are
4451 captured remarkably well by semi-empirical KB predictions. The industry's more mod-
4452 ern construct of explosive detonation exhibiting significant and inherent variability across
4453 the entire range of scaled distances is the result of probabilistic based regimes which col-
4454 lated trial data which had systematic errors in their approach of conducting and analysing.

4455

4456 The investigation into the detonation of small scale non-ideal explosive, ANFO, has pro-
4457 vided fundamental developments to the understanding of how detonation mechanics effect
4458 the overall charge yield. Despite non-ideal behaviour, when a composition exhibits deto-
4459 nation, provided the minimum reaction zone size is smaller than the charge diameter, it
4460 will result in a Friedlander ideal-like pressure-time history but the equivalency to TNT
4461 will vary based on the minimum charge radius and therefore mass. Through a theoretical
4462 approach, validated with experimental data, a fast look up tool for ANFO detonations is
4463 provided to estimate the explosive parameters depending on mass scaling.

4464

6.2 Conclusions

The resulting work documented within this thesis can be summarised in the following conclusions which are listed in chronological order of appearance within the thesis:

- When considering the far-field range alone, well-controlled experimental trials result in highly deterministic blast parameters across both the positive and negative due to exhibiting low levels of variability from nominal identical testing regimes. This finding has been shown across both ideal and non-ideal explosives in Figures 3.10 and 5.6.
- The prediction curves developed by Kingery and Bulmash^[87] not only capture the results of ideal explosive detonation in far-field regimes remarkably well, but are a sign of incredible prowess in both theoretical and empirical understanding of blast wave phenomenology during the times the trials were conducted.
- Systematic features, such as detonator position (Figure 3.16), gauge mounting apparatus (Figure 3.15) and so on, albeit seemingly negligible in the grand scheme of explosive detonation, present issues in acquiring deterministic behaviours as scaled distances reduce into near-field regimes due to detonation mechanisms and shock wave interactions being highly sensitive to small changes respectively.
- The secondary shock phenomena presents similar levels of consistency with scaled distance in the far-field regime as the primary shock and therefore prediction curves, similar to those presented by Kingery and Bulmash^[87], has been presented in Figure 4.7.
- The use of high speed video recordings and subsequent validated data processing methods has shown to result in comparable data to that recorded with pressure gauges (Figure 4.11), but providing parameters across a much broader range of scaled distances from a small number of recordings.
- Non-ideal explosives, although exhibit friedlander-like pressure-time histories, are not scalable with mass like ideal explosives. The small-scale ANFO detonation results in a much lower explosive yield than quoted TNTe values within published literature suggest, established from large scale trials, depicted in Figure 5.7.
- Small scale ANFO detonations exhibit a seemingly delayed arrival time when compared to equivalent TNT detonations in Figure 5.7. This has been related to the

4496 detonation and shock wave velocity profile inconsistencies between an ideal explosive
4497 behaviour and that of ANFO. Upon detonation of an ideal explosive, the detonation
4498 wave is assumed to propagate as its maximum velocity until reaching the edge of the
4499 charge at which point a decay in the velocity occurs. ANFO however sees an increase
4500 in detonation wave speed with the distance it has to travel through the explosive
4501 charge. This results in a clear discrepancy between arrival time predictions across
4502 the smaller charge diameters tested, but becomes increasingly negligible in line with
4503 charge diameter, in line with Figure 5.10.

- 4504 • The negative phase of small scale ANFO detonations present reasonably higher mag-
4505 nitudes of parameters, comparing better with quoted figures of ANFO $TNTe=0.82$
4506 in Figure 5.17. This has been deduced to be the effect of any ANFO which did not
4507 have enough time to full detonated during the first pass of the primary detonation
4508 wave, releases its energy sometime after, creating a more vigorous and energetic
4509 fireball which in turns results in an increased suction effect at ground level and thus
4510 large negative phases pressures and more powerful secondary shock features.
- 4511 • The mechanism underlying the reason as to why small scale ANFO resulting in much
4512 lower yield in for positive phase parameters has been directly linked to the behaviour
4513 of the characteristic reaction zone size for ANFO, projecting away from the charge
4514 effectively unreactive. The mechanism has been comprehensively compared against
4515 test data from a range of scales and is found to encapsulate the data remarkably
4516 well in Figure 5.30.
- 4517 • Non-ideal explosives can be predicted by both empirical prediction tools and nu-
4518 merical simulations, therefore exhibiting ideal-like behaviours at each increment of
4519 mass. It is therefore important to make use of a third dimension to prediction tools
4520 for non-ideal explosions which consider the characteristic reaction zone projectile
4521 mechanism for a given composition which can be established using rate stick trials.

4522 6.3 Evaluation and Future Work

4523 Chapter 3 made a distinct effort to develop data processing methodologies for standard
4524 free-air explosive trials in the attempt to assess the validity of deterministic regions of ex-
4525 plosive yield presented by Tyas^[173]. An extensive data set, comprising of 144 recordings
4526 of small-scale hemispherical PE4 detonations over the far-field range, was utilised to verify

4527 the processing tools ability to capture blast wave parameters and evaluate them through
4528 comparisons to Kingery and Bulmash^[87] semi-empirical predictions. The background to
4529 characterising explosive yield, particularly in the far-field regime, is lack of agreement
4530 in the consistency of recorded parameters. Borenstein^[18] state the reasons believed for
4531 a wide range of variability assumption have been made for explosive yields is down to
4532 the way each individual decided to assess the results. Extrinsic and intrinsic sources of
4533 variability were not completely outlined in each published article, with conclusions drawn
4534 from the data which may not have been scientifically accurate. This has led parts of
4535 the industry to believe explosive detonation is inherently variable, which fosters anoma-
4536 lous, or widely varying, data to be accepted as given without a through critique being
4537 undertaken. Whilst a generalised understanding of explosive yield regardless of explosive
4538 size, shape, composition and separation distance from a target (extrinsic) is important for
4539 probabilistic-based analyses, as to is a more robust and scientific understanding of a defini-
4540 tive situation is required for high confidence in prescribed loading for specific scenarios.
4541 Through establish well-controlled experimental methodologies, validated and improved
4542 through numerical simulation, clear improvements in the understanding of explosive vari-
4543 ability has been established, with both positive and negative phase parameters not only
4544 exhibiting deterministic behaviours but are also captured remarkably well by the KB pre-
4545 diction tools, agreeing with the works of Esparza^[45] and Rickman and Murrell^[125] and
4546 verifying the hypothesis posed by Tyas^[173].

4547

4548 In near-field regimes, where the blast wave parameters are driven by the expanding det-
4549 onation products and fireball, higher levels of variability has been recorded. This ties in
4550 with the uncertainty in of KB predictions in this regime where physics-based numerical
4551 simulations are shown diverge rapidly as scaled distances are reduced Cormie et al.^[32].
4552 Whilst preliminary work at the University of Sheffield has made attempts at capturing
4553 spatial and temporal variations of near-field loading^[51,134,174], there is still a lot of work
4554 required to provide a robust understanding of these extreme conditions. Barr et al.^[10]
4555 has developed a new capability which vastly improves the resolution of the data capture
4556 presented in the aforementioned articles. Through a rigorous testing regime using this
4557 capability of near-field scaled distances and explosive charge compositions, the results
4558 would be used to provide better guidance for explosive quantification and characterisation
4559 through the full detonation process.

4560

4561 Non-ideal explosives are discussed within literatures as exhibiting varying rates of energy
4562 release, depending on homogeneity, with discussion into how detonation mechanics inter-
4563 nally within the charge, measuring velocity of detonation, changes with charge diameter
4564 and/or confinement [5,80]. There is little investigation into the characterisation of shock
4565 wave development from non-ideal explosives and with what is available consisting of pre-
4566 dominantly large scale trials which has been taken as a given across all mass scales [68].
4567 This is contradictory to the aforementioned articles which finds significant variations when
4568 considering the release of energy from non-ideal explosives being across longer timescales.
4569 To exhibit full potential release, non-ideal explosives need confinement from either a sur-
4570 rounding material or additional charge mass which results in enough time before breakout
4571 to exhibit the full chemical decomposition. Smaller scale charges by nature therefore result
4572 in partial energy releases and cannot be defined by a single value of TNTe in line with
4573 the theories presented for ideal explosives. Chapter 5 has proven this finding by adopting
4574 the same well-controlled methodologies and validated data processing tools whilst deto-
4575 nating small scale ANFO charges which resulted in much lower TNTe values than what
4576 has historically been published. Whilst a mechanism for the reduction in yield has been
4577 presented, it is important to test this theory with future larger scale trials, in the attempt
4578 to categorise at which mass scales do normal TNT equivalency rules apply, in which a
4579 non-ideal explosive exhibits ideal-like behaviours, and where the explosive yield becomes
4580 highly dependent on the mass.

4581

4582 Although ANFO was the only non-ideal explosive tested in this thesis, it is not the only
4583 one of interest or threat. Future works will consider the effects of other non-ideal explosive
4584 compositions to see if they too exhibit behaviours similar to ANFO with another dimen-
4585 sion of scaling with mass. A more comprehensive dataset of non-ideal detonations across
4586 both mass and distance scales is required to provide confidence in the ability to not only
4587 understand explosive events but to capture them accurately with numerical simulation.

4588

4589 The conclusions made throughout the entirety of this thesis should provide evidence to
4590 the idea of deterministic blast wave features across a number of explosives in particular
4591 for far-field scenarios, and therefore any other suggestion should be put down to either
4592 systematic errors in data acquisition or results reporting extrinsic variability. The findings
4593 should ultimately be taken forward into the analysis of near-field, where the fundamental
4594 quantification is lacking, to see if similar characterisation methods can be adopted.

4595

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